Integrated Advanced Microwave Sounding Unit-A (AMSU-A)

Performance Verification Reports
Initial Comprehensive Performance Test Report

P/N: 1356008-1-IT, S/N: 202/A1

Contract No. NAS 5-32314 CDRL 208

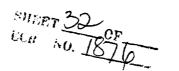
Submitted to:

National Aeronautics and Space Administration Goddard Space Flight Center Greenbelt, Maryland 20771

Submitted by:

Aerojet 1100 West Hollyvale Street Azusa, California 91702

Legens as appropriated TO STORES ENGINEERING CHANGE NOTICE A. Wieto Madge X.2103 AE 2615619 NC EOSIAMSU-AI 8 Rea. CONFIGURATION MOR. Tosts 4 Design Verif. INCORPORATE OF MATERIAL NSTALLED. ON ORDER IN STOCK DISAPPROVE ROMBACLE LUNGER DATE 712/198 ADVANCE REL. EQ ENDITEM SAN 305 see attached Master Mark-UP MAND LTBT CHANGE CODE 14. JUSTIFICATION/REASON FOR CHANGE 18 7-22-98 ONTE: 714998 MAND NAS 5-32314 PART NUMBER(S) 17. NASA COI CLASSIFICAD PO Box 296 Azusa, CA 91702 CAGE Code 70143 Hallist HAMDWARE SOFTWARE CECCER 30 x 298561 9081 12. DESCRIPTION OF CHANGE ITEM ZONE CAMSU-2. ECH NUMBER B. MULTIPLE DOCUMENTS YES LND AFFECTED 13. BIOMATURES Routine . CAC ORP Design Varilt, Specer. 1. PROGRAM COMBINED Design Verif., Dwg. PT. (Engl AMSU Mfg Eng CCRS 8661 14:58



Select MATH

Select Next

Select Intgrt: Note the display changes to present an integrated value of the current waveform.

Select X;

Move the X marker to the maximum right of the display. The Y value is indicative of the integrated current value over the entire 8 second period. Plot this waveform and attach a hard copy of the scan to TDS 5.

Multiply the maximum Y value by the current/ div as selected on the current amplifier, then divide by 8 seconds to acquire an average current/ second value. As an example: if the current amplifier is set up to display 200 ma/ 10 mv per division, and the maximum Y value = 32.4 mv:

 $[32.4 \text{mv} \times (200 \text{ma}/10 \text{mv})]/8 \text{ sec} = 81 \text{ma}/\text{ sec}$

Enter this value on TDS 5

- 18. Compute the operating peak and average power in watts from the measured values in steps 16 and 17 above. Record the computed values on TDS 5. Compute Noisy Bus current during the integrrate/hold, dumpt (I/H, D) time period (Refer to Figure 15-A). Record the data on TDS 5.
- 19. With the multimeter, adjust the external power supply PS1 to 29±0.10vdc as measured between J1-5 (high) and J1-7 (low). Record this voltage on TDS 5.
- 20. Repeat steps 13 through 18.
- With the multimeter, adjust the external power supply PS1 to 31±0.10vdc as measured between J1-5 (high) and J1-7 (low). Record this voltage on TDS 5.
- 22. Repeat steps 13 through 18.

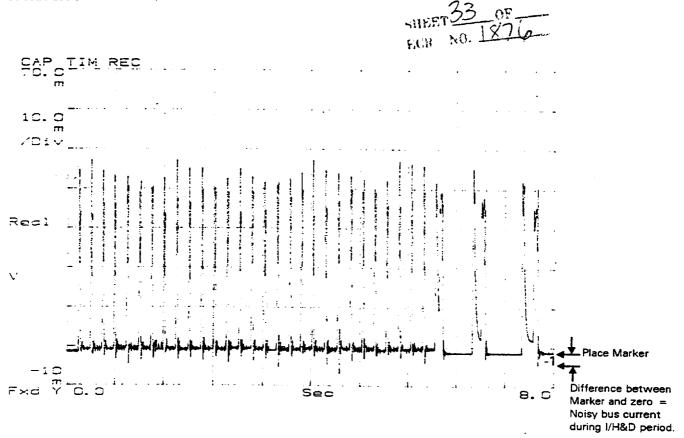


Figure 15-A. Typical Noisy Power Bus Current

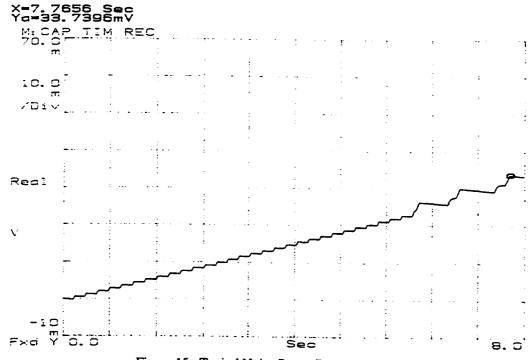
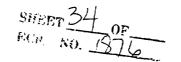


Figure 15. Typical Noisy Power Bus Integration



3.3.3.2.2 Noisy power bus turn on transient test. The Noisy Power Bus turn on transient shall be verified at +31 volts as follows:

- 1. The setup should be intact from paragraph 3.3.3.2.1 testing
- 2. Verify the external power supply (PS1) is adjusted to 31±.1vdc, make appropriate adjustments, and the unit is in WARM CAL position.
- 3. Configure the Dynamic Signal Analyzer (DSA) as follows:

Select MEAS MODE

Select Time Capture

Select Capture Select

Select Capture Length; Enter 80.0, Select msec

Select FREO

Select Freq Span; Enter 100.0; Select KHz

Select E SMPL Off

Select Time Length; Enter 8.0; Select msec

Select SELECT MEAS

Select Power Spec

Select CH1 Active

Select WINDOW

Select Hann

Select SOURCE

Select Source Off

Select AVG

Select Avg Off

Select Tim Av Off

Select RANGE

Select Chan 1 Range; Enter 1; Select V

Select INPUT COUPLE

Select CH1 DC

Select CH 1 Ground

Select INPUT TRIG

Select Trig Level; Enter 100; Select mv

Select Arm AU

Select Chan I Input

Select Slope +

Select TRIG DELAY

Enter 0.0; Select Sec

Select COORD

Select Real

Select VIEW INPUT

Select Time Buff

Select SCALE

Select X Fixd Scale; Enter 0.0, 80.0; Select msec

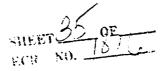
Select Y Fixd Scale; Enter 0, 640.0; Select mv

Select UNITS

Select Hz (sec)

-NOTE-

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.



- a) Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
- b) Depress "Start Capture" on the DSA.
- c) With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
- d) Position the current probe to its original location in accordance with Figure 8.
- 4. Adjust PS2 for +28vdc.
- 5. Start the DSA signal capture by depressing "Start Capture"; wait for the DSA message "waiting for trigger" before proceeding.
- 6. On the Relay Board, turn the switch ON and obtain a record of the Noisy Bus Turn on current waveform. On the Relay Board, turn the switch OFF. Adjust the display time base and voltage sensitivity to allow for adequate current and pulse duration measurements. Plot the obtained waveform and attach a hard copy of the scan to TDS 6. A representative Noisy Bus Turn-on is shown in Figures 16-A and 16-B.
- Measure the Turn On pulse width; record this value in TDS 6.
- 8. Compute the peak current as follows: Multiply the maximum Y value by the current/ div as selected on the current amplifier. As an example: if the current amplifier is set up to display 200 ma/ 10 mv per division, and the maximum Y value = 276mv:

$$276$$
mv x $(200$ ma/ 10 mv) = 5520 ma = 5.52 amps

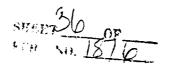
Record this value on TDS 6.

9. The 1st derivative of the current waveform must be calculated. Compute the dI/dT as follows:

The most probable location of the greatest current demand is during the first positive transition after voltage application. If this is the case, expand that segment of the display and measure the greatest voltage transition in the smallest time transition. The change in voltage times the current/ div as selected on the current amplifier produces the change in current. Next divide this change in current by the change in time (in microseconds). This value is dI/dT. Example:

 $144 \text{mv} \times (200 \text{ma}/ 10 \text{mv}) / 19.5 \text{ us} = 147.7 \text{ma per us}$

- 10. Record the computed value on TDS 6.
- With the multimeter, adjust the external power supply PS1 to 29±0.10vdc as measured between J1-5 (high) and J1-7 (low).
- 12. Repeat steps 3 through 10.
- 13. With the multimeter, adjust the external power supply PS1 to 27±0.10vdc as measured between J1-5 (high) and J1-7 (low).



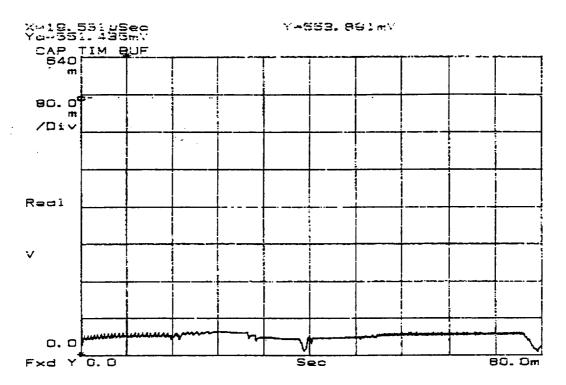


Figure 16-A. Typical Noisy Bus Turn On Transient

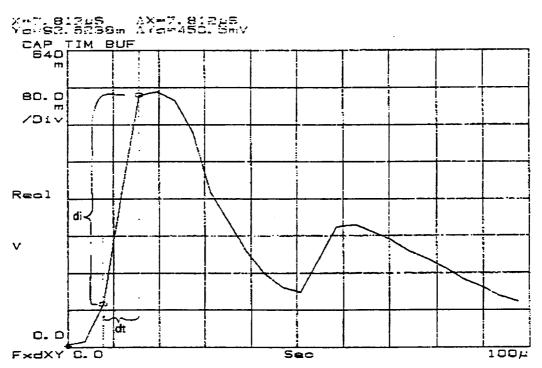
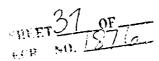


Figure 16-B. Typical Noisy Bus Turn On Expanded



- 14. Repeat steps 3 through 10.
- 15. Turn the STE power supply panel Q/ pulse switch OFF (refer to Figure 3).
- 16. Turn the STE power supply panel main power switch OFF (refer to Figure 3).
- 3.3.3.3 Survival heater power bus interface tests. The operational characteristics of the redundant survival buses A and B shall be verified during ambient thermal cycle testing using test procedure AE-26151/9. For final CPT attach data sheet from Survival Heater Test to this data package.
- 3.3.4 Passive analog interface test. This test provides the verification of the passive analog telemetry requirements found in the following documents:

UIID None

GIRD Sections 4.5.2, 4.5.3, and 6.3

POS Section 4.6.3.6 (8)

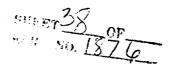
ICD Sections 4.5 and 6.3

Passive analog telemetry signals are output from the unit through the spacecraft interface connector J2. To verify these signals, perform the following procedures:

- 1. The unit should be configured as shown in Figure 12 if performing an LPT or Figure 14 if performing a CPT. Turn the STE main power switch on (computer should be on, STE power panel should be off). From the A1 directory and at the "\$" prompt, enter the command to the STE "RUN E1". The EOS/AMSU-A1 software program should be running as evidenced by the STE screen shown in Figure 9.
- 2. Enter the STE command "[2] MONITOR ONLY". The screen should now be as shown in Figure 10.
- 3. Enter the STE command "[12] UNPOWERED THERMISTORS". The screen should now be as shown in Figure 17.
- 4. The thermistor data should update every 8 seconds. Enter STE command "[2]" to print the screen. Enter the data on TDS 7 and attach the printout to TDS 7.

3.3.5 Command and data handling bus interface test

- 3.3.5.1 Formal qualification test of the EOS/AMSU-A1 firmware (protoflight model 1st CPT only. On 3/21/97, an initial Formal Qualification Test (FQT) of the EOS AMSU-A firmware was conducted using Test Procedure AE-26600 (CDRL 415). The results of that test were documented in Report 10974 (CDRL 217). As stated in that report, a final FQT would be performed as a part of the initial instrument CPT for the EOS protoflight models A1 and A2 to validate the firmware requirements (Report 10458, CDRL 306-2b) which could not be validated during the initial FQT. The purpose of this test is to perform that validation by repeating Test Procedure AE-26600 and conducting additional system level testing with the unit connected to the Special Test Equipment (STE). At the conclusion of paragraph 3.3.5 testing, the firmware will be validated. Perform Test Procedure AE-26600 with the following clarifications:
 - Paragraph 4.1, Load bonded Software the last half of the paragraph beginning with "The tape labeled N7 ..." to the end of the paragraph should be ignored because the unit configuration uses flight CCAs.
 - Paragraph 4.2, Configure the test environment replace this paragraph with the instructions provided in paragraph 3.3.5.2 steps 1 through 9 of this procedure.
 - Paragraph 4.4.4, C through L. These are replaced by section 3.3.5.3 of this procedure.



EOS A1 - XX OB.A1] [5] SCIENCE DATA		0000	29-SEP-97	14:44:25 SCAN I	NUMBER
[6] CONTROL/STATU	S ELEMENT	00		•	
[7] ENGINEERING	ELEMENT	00			
	UNPOWERE	D THED	MICTODS		
NO	DATA		MISTORS	TEMP C	
1 A1-1	SCAN MOTOR	TEMPE	ם א דין ווס ב	23.50	
	SCAN MOTOR			23.40	
	RF SHELF TEM			20.00	
	RF SHELF TEM			19.90	
	WARM LOAD			20.30	
	WARM LOAD			20.50	
· · · · · · · · · · · · · · · · · · ·	RF SHELF TEM	PERAT	URE #2	20.05	
	RF SHELF TEM			19.95	
	armavar (.	
POWER ON			CALC		SA29
SCREEN SELECT BUTTON	ONLY [2] PR	UNT [3	J FULL	[1] RETUR	UN .

Figure 17. EOS/AMSU-A1 STE Unpowered Thermistors Screen

- 3.3.5.2 Instrument commanding test. This test provides the verification of the instrument commanding capability. Each of the commands shown in Table III with the exception of [21] GSE Modes will be sent to the unit and verified that it was received and carried out by the unit. GSE Modes will be verified during test point interface testing (paragraph 3.3.6). Perform the following procedures.
 - 1. Configure the unit as shown in Figure 12. If the unit is already configured, skip to step 7.
 - 2. Connect a 25 pin breakout box to J1 of the instrument. Connect a 37 pin breakout box to J4 of the instrument.
 - 3. Connect the STE to the instrument using the following STE interface cables:
 - a. STE interface cable J1 (1356648-1)
 - b. STE interface cable J2 (1356648-2)
 - c. STE interface cable J3 (1356648-3)
 - 4. Connect STE interface cable J1 from EOS J1 found on the STE power panel shown in Figure 4 to the remaining end of the 25 pin breakout box connected to J1 on the unit.

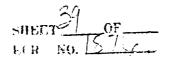
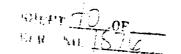


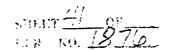
Table III. EOS/AMSU-A1 Instrument Commands

STE Command Screen Number	STE Command	Instrument Status	
· [9]	Scanner A1-1 Power	ON / OFF	
[10]	Scanner A1-2 Power	ON/OFF	
[11]	Antenna Full Scan Mode	YES / NO	
[12]	Antenna Warm Cal Mode	YES / NO	
[13]	Antenna Cold Cal Mode	YES / NO	
[14]	Antenna Nadir Mode	YES / NO	
[15]	PLO Power	PLO #1 / PLO #2	
[16]	Cold Cal Position 1	YES / NO	
[17]	Cold Cal Position 2	YES / NO	
[18]	Cold Cal Position 3	YES / NO	
[19]	Cold Cal Position 4	YES / NO	
[20]	Reset C&DH Processor	Resets 1553 firmware	
[21]	GSE Modes	YES / NO	

- 5. Connect STE interface cable J2 from EOS J2 found on the STE test panel shown in Figure 5 to J2 on the unit.
- 6. Connect STE interface cable J3 from EOS A&B J1 found on the STE interface panel shown in Figure 6 to J3 on the unit.
- 7. Turn the STE MAIN POWER switch on (refer to Figures 2 and 3 (computer should be on, STE power panel should be off)). From the A1 directory and at the "\$" prompt, enter the command to the STE "RUN E1". The EOS/AMSU-A1 software program should be running as evidenced by the STE screen shown in Figure 9.
- 8. Turn the STE power supply panel Q/MAIN switch on (refer to Figure 3). With a multimeter, adjust the Quiet Bus voltage at the breakout box to 29 ± 0.10 volts (between J1-1 and J1-3).
- 9. Turn the STE power supply panel N/PULSE switch on (refer to Figure 3). With a multimeter, adjust the Noisy Bus voltage at the breakout box to 29 ± 0.10 volts (between J1-5 and J1-7).
- 10. Go to the Commands screen on the STE. From the Main screen shown in Figure 9, enter the STE command "[2] MONITOR ONLY". The screen should now be as shown in Figure 10. Enter the STE command "[14] COMMANDS". The screen should now be as shown in Figure 11.
- 11. The instrument commands shown in Table III are now ready to be tested.
- 12. Enter the STE command "[11] ANTENNA FULL SCAN MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES). Record the status on TDS 8.

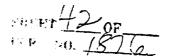


- 13. Enter the STE command "[9] SCANNER A1-1 POWER". Look at the commands screen to see that the command was received by the instrument {the state of that command should go from NO (OFF) to YES (ON)}. The A1-1 scan motor should now be scanning. Record the status on TDS 8.
- 14. Enter the STE command "[9] SCANNER A1-1 POWER". Look at the commands screen to see that the command was received by the instrument {the state of that command should go from YES (ON) to NO (OFF)}. The A1-1 scan motor should stop scanning. Record the status on TDS 8.
- 15. Enter the STE command "[10] SCANNER A1-2 POWER". Look at the commands screen to see that the command was received by the instrument {the state of that command should go from NO (OFF) to YES (ON)}. The A1-2 scan motor should now be scanning. Record the status on TDS 8.
- 16. Enter the STE command "[10] SCANNER A1-2 POWER". Look at the commands screen to see that the command was received by the instrument {the state of that command should go from YES (ON) to NO (OFF)}. The A1-2 scan motor should stop scanning. Record the status on TDS 8.
- 17. Enter the STE command "[9] SCANNER A1-1 POWER". Wait for at least 18 seconds and then enter the STE command "[10] SCANNER A1-2 POWER". Look at the commands screen to see that the commands were received by the instrument {the state of those commands should go from NO (OFF) to YES (ON)}. Both A1-1 and A1-2 scan motors should now be scanning. Record the status on TDS 8.
- 18. Enter the STE command "[12] ANTENNA WARM CAL MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES and the state of ANTENNA IN FULL SCAN MODE should go from YES to NO). Both A1-1 and A1-2 scan motors should have moved to the warm calibration position. Record the status on TDS 8.
- 19. Enter the STE command "[14] ANTENNA NADIR MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES and the state of ANTENNA WARM CAL MODE should go from YES to NO). Both A1-1 and A1-2 scan motors should have moved to the nadir position. Record the status on TDS 8.
- 20. Enter the STE command "[13] ANTENNA COLD CAL MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES and the state of ANTENNA NADIR MODE should go from YES to NO). Both A1-1 and A1-2 scan motors should have moved to the cold calibration 1 position (LSB=0, MSB=0). Record the status on TDS
- 21. Enter the STE command "[19] COLD CAL POSITION 4". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES. Also, the state of ANTENNA COLD CAL MODE should stay YES). Both A1-1 and A1-2 scan motors should have moved slightly to the cold calibration 4 position. Record the status on TDS 8.
- 22. Enter the STE command "[18] COLD CAL POSITION 3". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES. Also, the state of ANTENNA COLD CAL MODE should stay YES). Both A1-1 and A1-2 scan motors should have moved slightly to the cold calibration 3 position. Record the status on TDS 8.
- 23. Enter the STE command "[17] COLD CAL POSITION 2". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES. Also, the state of ANTENNA COLD CAL MODE should stay YES). Both A1-1 and A1-2 scan motors should have moved slightly to the cold calibration 2 position. Record the status on TDS 8.
- 24. Enter the STE command "[16] COLD CAL POSITION 1". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES. Also, the



state of ANTENNA COLD CAL MODE should stay YES). Both A1-1 and A1-2 scan motors should have moved slightly to the cold calibration 1 position. Record the status on TDS 8.

- 25. Enter the STE command "[15] PLO POWER". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from PLO #1 to PLO #2 or vice versa depending on its starting state. Record the status on TDS 8. Leave this step with PLO #1 active (if PLO #2 is active enter STE command "[15] PLO POWER" to make PLO #1 active).
- 26. Enter the STE command "[20] RESET C&DH PROCESSOR". Look at the bottom of the commands screen to see that SA28 resets and starts counting from 1. Record the status on TDS 8.
- 27. Leave the unit powered and the setup intact for paragraph 3.3.5.2 testing.
- 3.3.5.3 Science and engineering data verification. The engineering data in the engineering packet is also found embedded in the science data packet. The STE does a comparison between the data in the engineering packet and the same data located in the science data packet. If there is total agreement between the two data sets then a message "ENGR OK" appears at the bottom of the STE screen. Because of the fact that the two packets agree with respect to engineering data, this test validates both science and engineering data by verifying the data in the science data packet for each of the following instrument modes. Look at ENGINEERING DATA, also UNPOWERED THERMISTORS prior to starting these modes.
 - 1. Full Scan Mode (3.3.5.3.1)
 - 2. Warm Cal Mode (3.3.5.3.1)
 - 3. Cold Cal Mode (3.3.5.3.3)
 - 4. Nadir Mode (3.3.5.3.4)
- 3.3.5.3.1 Full scan mode. The full scan mode science and engineering data is verified as follows:
 - From the STE command screen shown in Figure 11, enter the STE command "[11] ANTENNA FULL SCAN MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES). Record the status on TDS 9.
 - 2. Look to see that "ENGR OK" message is displayed in bottom left corner of screen. Record the status on TDS 9.
 - Look to see that the unit is operating in full scan mode. Enter the observed result on TDS 9.
 - 4. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 9.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (use data from procedure AE-26002/1 TDS 5 and 6 for required position data) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)



- g. PRT temperature data (elements 1090 1180, page 7 of printout) Refer to Table IV for PRT data description.
- h. Status (page 8 of printout)
- i. Engineering data (page 8 of printout)
- 5. Attach the printout to TDS 9.

3.3.5.3.2 Warm cal mode. The warm cal mode science and engineering data is verified as follows:

- 1. From the STE command screen shown in Figure 11, enter the STE command "[12] WARM CAL MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES). Record the status on TDS 10.
- 2. Look to see that "ENGR OK" message is displayed in bottom left corner of screen. Record the status on TDS 10.
- 3. Look to see that the unit reflectors have moved to warm cal position. Enter the observed result on TDS 10.
- 4. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 10.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (use data from procedure AE-26002/1 TDS 5 and 6 for required position data for warm cal position) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)
 - h. Status (page 8 of printout)
 - i. Engineering data (page 8 of printout)
- 5. Attach the printout to TDS 10.

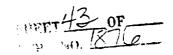
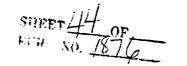


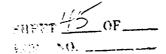
Table IV PRT Data Description

PRT				
Number	Description			
1	Scan Motor A1-1 Temperature			
2	Scan Motor A1-2 Temperature			
3	Feedhorn A1-1 Temperature			
4	Feedhorn A1-2 Temperature			
5	RF Mux - A1-1 Temperature			
6	RF Mux - A1-2 Temperature			
7	Local Oscillator - Channel 3 Temperature			
8	Local Oscillator - Channel 4 Temperature			
9	Local Oscillator - Channel 5 Temperature			
10	Local Oscillator - Channel 6 Temperature			
11	Local Oscillator - Channel 7 Temperature			
12	Local Oscillator - Channel 8 Temperature			
13	Local Oscillator - Channel 15 Temperature			
14	Phase Locked Oscillator No. 2 Temperature			
15	Phase Locked Oscillator No. 1 Temperature			
16	S.P. (1553 Interface) Temperature			
17	Mixer/IF Amplifier - Channel 3 Temperature			
18	Mixer/IF Amplifier - Channel 4 Temperature			
19	Mixer/IF Amplifier - Channel 5 Temperature			
20	Mixer/IF Amplifier - Channel 6 Temperature			
21	Mixer/IF Amplifier - Channel 7 Temperature			
22	Mixer/IF Amplifier - Channel 8 Temperature			
23	Mixer/IF Amplifier - Channel 9/14 Temp			
24	Mixer/IF Amplifier - Channel 15 Temperature			
25	IF Amp - Channel 11/14 Temperature			
26	IF Amp - Channel 9 Temperature			
27	IF Amp - Channel 10 Temperature			
28	IF Amp - Channel 11 Temperature			
29	DC/DC Converter Temperature			
30	IF Amp - Channel 13 Temperature			
31	IF Amp - Channel 14 Temperature			
32	IF Amp - Channel 12 Temperature			
33	RF Shelf - A1-1 Temperature			
34	RF Shelf - A1-2 Temperature			
35	Detector/Preamplifier Temperature			
36	A1-1 Warm Load 1 Temperature			
37	A1-1 Warm Load 2 Temperature			
38	A1-1 Warm Load 3 Temperature			
39	A1-1 Warm Load 4 Temperature			
40	A1-1 Warm Load Center Temperature			
41	A1-2 Warm Load 1 Temperature			
42	A1-2 Warm Load 2 Temperature			
43	A1-2 Warm Load 3 Temperature			
44	A1-2 Warm Load 4 Temperature			
45	A1-2 Warm Load Center Temperature			



3.3.5.3.3 Cold cal mode. The cold cal mode science and engineering data is verified as follows:

- 1. From the STE command screen shown in Figure 11, enter the STE command "[13] COLD CAL MODE". Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES). Record the status on TDS 11.
- 2. Look to see that "ENGR OK" message is displayed in bottom left corner of screen. Record the status on TDS 11.
- 3. Look to see that the unit reflectors have moved to cold cal position 1. Enter the observed result on TDS 11. PENTER THE STE GAMMAN [3] to obtain a full printout. Use this did a in Step 6,
- 4. From the STE command screen shown in Figure 11, enter the STE command"(10] ANTENNA FULL SCAN MODE". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).
- 5. From the STE command screen shown in Figure 11, enter the STE command" [12] ANTENNA COLD CAL MODE". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).
- 6. Enter the STE command [3] to obtain a full printout. Review the following data and record the results on TDS 11.
 - a.) packet ID (elements 1 and 2, page I of printout) (from 5tep 3 kgull print)
 - b.) packet length (elements 3 and 4, page I of printout) Grom Step 3 & Full print)
 - c.) unit serial number (element 5 and 6, page I of printout) (From Step 3 full print)
 - d.) Instrument/ mode status (element 7 and 8, page 1 of printout) (from Step 3 full print)
 - e.) reflector positions (use data from procedure AE-26002/2 TDS 2 for required position data for cold cal position 1) (page 1 and 2 of printout) (STED 3 Full print used for TDS 11 Sheet 4) Step 6 full print used for TDS 11 Sheet 4
 - f.) radiometric scene data (pages I and 2 of printout) (From step 3 Sull print)
 - g.) PRT temperature data (elements 262 300, page 2 of printout) from step 3 fill print)
 - h.) status (page 3 of printout) (From Step 3 fill print)
 - i.) engineering data (page 3 of the printout) (From Step 3 full print)
- 7. Attach the printout to T. DS 11
- 8. From the STE command screen shown in Figure 11 enter the STE command [15] COLD CAL POSITION 2". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES). Record status on TDS 11
- Look to see that "ENGR OK" message is displaced in the bottom left comer of screen. Record status on TDS 11
- 10. Look to see that the unit reflector has moved to cold cal position 2. Enter the results on TDS 11. THIER THE STE COMMAND COLD TO OBTAW A FULL PRINTAIN, USE THIS DATA IN STEP 14.
- 11. From the STE command screen shown in Figure 11, enter the STE command"[10] ANTENNA FULL SCAN MODE". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).



- 12. From the STE command screen shown in Figure 11, enter the STE command"[12] ANTENNA COLD CAL MODE". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).
- 13. Look to see that the unit reflector has moved to cold cal position 2
- 14. Enter the STE command [3] to obtain a full printout. Review the following data and record the results on TDS I 1
 - a.) Instrument/mode status (element 7 and 8, page 1 of printout) from step 10 full print)
 - b.) status (page 3 of printout) (from step 10 full print)
 - c.) reflector positions (use data from procedure AE-26002/2 TDS 2 for required position data for cold cal position 1) (page 1 and 2 of printout) (Step 10 Rell print for TDS 11 Sheet 4)

 Attach the printout to TDS 1.1
- 15. Attach the printout to TDS I 1
- 16. From the STE command screen shown in Figure 1-1, enter the STE command" [16] COLD CAL POSITION 3". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES). Record status on TDS 11.
- 17. Look to see that "ENGR OK" message is displayed in the bottom left comer of screen. Record status on TDS 11.

18. Look to see that the unit reflector has moved to cold cal position 3. Enter the results on TDS 11. FATER THE STE COMMAND (3) TO OBTAIN A FULL PRINTENT. USE THIS DATA IN STEP 22,

- From the STE command screen shown in Figure I 1, enter the STE command [IO] ANTENNA FULL SCAN MODE. Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).
- 20. From the STE command screen shown in Figure 11, enter the STE command"[12] ANTENNA COLD CAL MODE". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).
- 21. Look to see that the unit reflector has moved to cold cal position 3.
- 22. Enter the STE command [3] to obtain a full printout. Review the following data and record the results on TDS 11.
 - a.) Instrument/mode status (element 7 and 8, page 1 of printout) (from Step 18 foll print)
 - b.) status (page 3 of printout) (From step 18 FM Print)
 - c.) reflector positions (use data from procedure AE-26002/2 TDS 2 for required position data for cold cal position 1) (page 1 and 2 of printout) (Step 18 Full print For TDS 11 Step 22 Full print For TDS 11 Step 4)
- 23. Attach the printout to TDS 11
- 24. From the STE command screen shown in Figure II, enter the STE command"[17] COLD CAL POSITION 4". Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES). Record status on TDS 11.

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25. Look to see that "ENGR OK" message is displayed in the bottom left corner of screen. Record status, on TDS 11.

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Look to see that the unit reflector has modified to cold cal position 4. Enter the results on TDS 11. EMER THE STE COMMAND [3] TO OBTAIN A FULL PRIMOUT, USE THE DOTA IN STEP 301

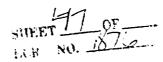
From the STE command screen shown in Figure [11] enter the STE command [10] ANTENNA FULL SCAN MODE. Look at the command screen to see that the command was received by the instrument (the state of the command should go from NO to YES).

28. From the STE command screen shown in Figure (II) enter the STE command" [12] NNA COLD CAL MODE". Look at the command screen to see that the command -, vas received by the instrument (the state of the command should go from NO to YES).

- 29. Look to see that the unit reflector has moved to cold cal position 4.
- 30. Enter the STE command [3] to obtain a full printout. Review the following data and record the results on TDS X 11
 - a.) Instrument/mode status (element 7 and 8, page 1 of printout) FAST (From Step 26 full print
 - b.) status (page 3 of printout) (From Step 26 Full print)
 - c.) reflector positions (use data from procedure AE-26002/2 TDS 2 for required position data for cold cal position 1) (page 1 and 2 of printout) step 26 full print used for TDS 11 step 4 3, step 30 full print used for TDS 11 step 4
- 31. Attach the printout to TDS 11.

3.3.5.3.4 Nadir mode. The nadir mode science and engineering data is verified as follows:

- From the STE command screen shown in Figure 11, enter the STE command "[14] NADIR MODE".
 Look at the commands screen to see that the command was received by the instrument (the state of that command should go from NO to YES). Record the status on TDS 12.
- Look to see that "ENGR OK" message is displayed in bottom left corner of screen. Record the status on TDS 12.
- 3. Look to see that the unit reflectors have moved to nadir position. Enter the observed result on TDS 12.
- 4. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 12.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (use data from procedure AE-26002/1 TDS 5 and 6 for nadir position data) (pages 1 6 of printout).
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)



- h. Status (page 8 of printout)
- i. Engineering data (page 8 of printout):
- 5. Attach the printout to TDS 12.

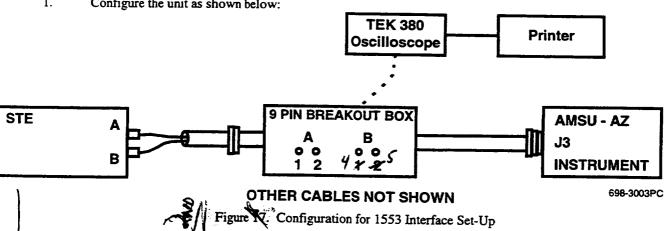
3.3.5.3.5 Noisy Bus current measurement during warm cal, cold cal and nadir mode.

- Place instrument in warm cal by repeating paragraph 3.3.5.3.2(1).
- 2. Record Noisy Bus current from STE noisy bus power supply display on TDS 13.
- 3. Command A1-1 scanner to "off" and record current
- 4. Command A1-1 scanner to "on" and A1-2 scanner to "off" and record current.
- 5. Command A1-1 and A1-2 scanner to "on."
- Place instrument in cold cal by repeating paragraph 3.3.5.3.3 (1). Repeat step number 2. 6.
- 7. Place instrument in Nadir by repeating paragraph 3.3.5.3.4 (1). Repeat step number 2.

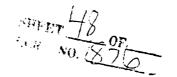
3.3.5.4 1553 Bus interface test.

The 1553 Bus interface shall be verified by observing its operation during full scan operation. The interface test shall be accomplished by the following steps:

1. Configure the unit as shown below:



- Insure all switches are closed on the 9-pin breakout box. 2.
- Connect Ocsilloscope to J3-1 (Hi) and J2-3 (lo) to measure 1553 Interface A data. A representative 3. waveform is shown in Figure A. Set the vertical to 5 volts; horizontal to 5 us, DC coupling: Trig-GH1. Print hard copy and attach to TDS 19.
- Using the Vertical and Horizontal bars, measure the amplitude and rise time of the instrument response. 4. Record these on TDS 19.
- Repeat steps 3 and 4 for Interface B. Attach and record data on TDS 19. Connect to J3-4 (Hi) and J3-5 5. (Lo). Note: Figure B shows a typical rise-time measurement.



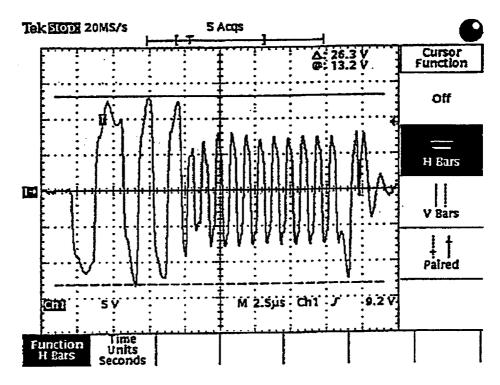


Figure A. Typical 1553 Bus Wave Form (Instrument Response)

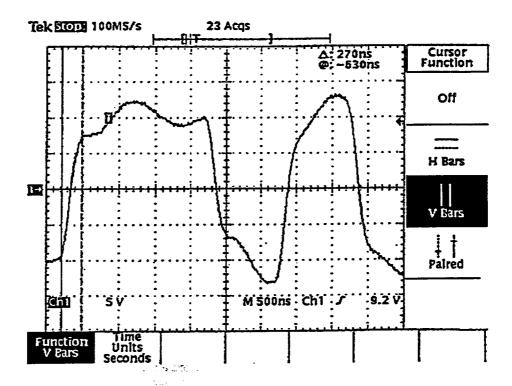


Figure B. Typical Rise-Time Measurement

3.3.6 Test point interface test. The purpose of this test is twofold:

- 1. Verify the following test point signals:
 - a. Intentionally left blank
 - b. 8 second sync pulse test point (3.3.6.2)
 - c. Integrate/hold and dump test points (3.3.6.3)
 - d. Channel 3 through 15 analog output test points (3.3.6.4)
 - e. PLO #1 and PLO #2 lock test points (3.3.6.5).
- 2. Verify the following GSE mode operations:
 - a. GSE-1 mode (3.3.6.6)
 - b. GSE-2 mode (3.3.6.7)
 - c. GSE-3 mode (3.3.6.8)
 - d. GSE-4 mode (3.3.6.9)
 - e. GSE-5 mode (3.3.6.10)
 - f. GSE-7 mode (3.3.6.11).

The test point interface connector (J4) is not used during spacecraft configuration and is covered with a cover plate when the unit is operating in the flight configuration. The above test points and GSE modes are used only by Aerojet during test and evaluation of instrument performance and do not meet any system level requirements.

3.3.6.1 Intentionally left blank. Perform the following procedures.

3.3.6.2 8 second sync pulse test point verification. Perform the following procedures.

- 1. Connect channel 1 of the oscilloscope to pins J4-2 (High) and J4-21 (Low).
- 2. Plot the oscilloscope display and record the information indicated on TDS 14. Attach the plot to TDS 14.

3.3.6.3 Integrate/hold and dump test point verification. Perform the following procedures.

- 1. Connect channel 1 of the oscilloscope to pins J4-6 (High) and J4-5 (Low).
- 2. Connect channel 2 of the oscilloscope to pins J4-24 (High) and J4-5 (Low).
- 3. Set the scope to trigger internally on channel 1. Optimize time and amplitude for best resolution. The desired display should look similar to the top two traces shown in Figure 18.
- 4. Plot the oscilloscope display and record the information indicated on TDS 15. Attach the plot to TDS 15.

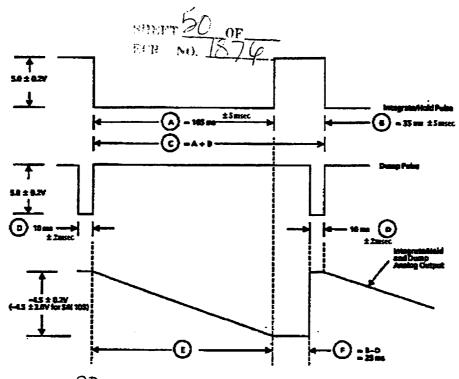
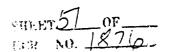


Figure 18. Integrate/Hold, Dump, and Analog Out Test Point Signals

3.3.6.4 Radiometer channel analog output test point verification. Perform the following procedures.

- 1. Connect channel 1 of the oscilloscope to pins J4-8 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 2. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 3 and attach the plot to TDS 16.
- 3. Connect channel 1 of the oscilloscope to pins J4-9 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 4. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 4 and attach the plot to TDS 16.
- 5. Connect channel 1 of the oscilloscope to pins J4-10 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 6. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 5 and attach the plot to TDS 16.
- 7. Connect channel 1 of the oscilloscope to pins J4-11 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 8. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 6 and attach the plot to TDS 16.
- 9. Connect channel 1 of the oscilloscope to pins J4-12 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 10. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 7 and attach the plot to TDS 16.



- 11. Connect channel 1 of the oscilloscope to pins J4-13 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 12. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 8 and attach the plot to TDS 16.
- 13. Connect channel 1 of the oscilloscope to pins J4-14 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 14. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 9 and attach the plot to TDS 16.
- 15. Connect channel 1 of the oscilloscope to pins J4-27 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 16. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 10 and attach the plot to TDS 16.
- 17. Connect channel 1 of the oscilloscope to pins J4-28 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 18. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 11 and attach the plot to TDS 16.
- 19. Connect channel 1 of the oscilloscope to pins J4-29 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 20. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 12 and attach the plot to TDS 16.
- 21. Connect channel 1 of the oscilloscope to pins J4-30 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 22. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 13 and attach the plot to TDS 16.
- 23. Connect channel 1 of the oscilloscope to pins J4-31 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 24. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 14 and attach the plot to TDS 16.
- 25. Connect channel 1 of the oscilloscope to pins J4-32 (High) and J4-26 (Low). Optimize time and amplitude for best resolution. The desired display should look similar to the bottom trace shown in Figure 18.
- 26. Plot the oscilloscope display and record the information indicated on TDS 16. Label the plot Channel 15 and attach the plot to TDS 16.

3.3.6.5 PLO #1 and PLO #2 lock test point verification. Perform the following procedures.

Look to see that PLO #1 is enabled. If not enter STE command "PLO POWER". Wait for the screen to show PLO #1 active.

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2. Connect channel 1 of the oscilloscope to pins J4-22 (High) and J4-4 (Low). If the PLO is locked, the voltage observed will be less than or equal to 1 volt. If the PLO is unlocked, the voltage observed will be approximately 10 volts.

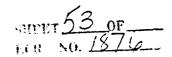
3. Plot the oscilloscope display and record the information indicated on TDS 17. Label the plot PLO#1 Lock,
Detect and attach the plot to TDS 17. Record Voltage on TDS 17

- Enter STE command "PLO POWER". Wait for the screen to show PLO #2 active.
- 5. Connect channel 1 of the oscilloscope to pins J4-3 (High) and J4-4 (Low). If the PLO is locked, the voltage observed will be less than or equal to 1 volt. If the PLO is unlocked, the voltage observed will be approximately 10 volts.
- 6. Plot the oscilloscope display and record the information indicated on TDS 17. Label the plot PLO #2 Lock
 Detect and attach the plot to TDS 17. Record Vollege on TDS 17

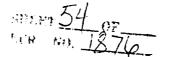
3.3.6.6 GSE-1 mode verification. This test mode positions the reflectors at beam position 6 for 10 integration periods, then to the cold calibration position for 10 integration periods, and finally to the warm cal position for 10 integration periods. This process is then repeated. To verify this mode, perform the following procedures. Look at ENGINEERING DATA, also UNPOWERED THERMISTORS prior to starting these modes.

- 1. Enter a "1" on the mode switch located on the front of the STE test panel (refer to Figure 2 for test panel location).
- 2. From the STE command screen shown in Figure 11, enter the STE command "[21] GSE MODE".
- 3. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
- 4. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (1st 10 at beam position 6, 2nd 10 at cold cal position, 3rd 10 at warm cal position, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)
 - h. Status (page 8 of printout)
 - i. Engineering data (page 8 of printout).
- 5. Attach the printout to TDS 18. There is no Pass/Fail criteria.
- 3.3.6.7 GSE-2 mode verification. This test mode positions the reflectors at beam position 1 for 30 integration periods. This process is then repeated. To verify this mode, perform the following procedures.

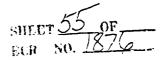
7:



- 1. Enter a "2" on the mode switch located on the front of the STE test panel.
- 2. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
- 3. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (30 positions at beam position 1, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)
 - h. Status (page 8 of printout)
 - i. Engineering data (page 8 of printout).
- 4. Attach the printout to TDS 18. There is no Pass/Fail criteria.
- 3.3.6.8 GSE-3 mode verification. This test mode positions the reflectors at each beam position for 30 integration periods incrementing the beam position to the next beam position each 8 seconds. This process is then repeated. To verify this mode, perform the following procedures.
 - 1. Enter a "3" on the mode switch located on the front of the STE test panel.
 - 2. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
 - 3. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (30 positions at beam position when printout obtained, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)



- g. PRT temperature data (elements 1090 1180, page 7 of printout)
- h. Status (page 8 of printout)
- i. Engineering data (page 8 of printout).
- 4. Attach the printout to TDS 18. There is not Pass/Fail criteria.
- **3.3.6.9** GSE-4 mode verification. This test mode positions the reflectors at beam position 30 for 30 integration periods. This process is then repeated. To verify this mode, perform the following procedures.
 - 1. Enter a "4" on the mode switch located on the front of the STE test panel.
 - 2. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
 - 3. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (30 positions at beam position 30, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)
 - h. Status (page 8 of printout)
 - i. Engineering data (page 8 of printout).
 - 4. Attach the printout to TDS 18. There is no Pass/Fail criteria.
- 3.3.6.10 GSE-5 mode verification. This test mode positions the reflectors at beam position 6 for 39 integration periods. This process is then repeated. To verify this mode, perform the following procedures.
 - 1. Enter a "5" on the mode switch located on the front of the STE test panel.
 - 2. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
 - 3. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)



- c. Unit serial number (elements 5 and 6, page 1 of printout)
- d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
- e. Reflector positions (30 positions at beam position 6, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
- f. Radiometer scene data (pages 1 6 of printout)
- g. PRT temperature data (elements 1090 1180, page 7 of printout)
- h. Status (page 8 of printout)
- i. Engineering data (page 8 of printout).
- 4. Attach the printout to TDS 18. There is no Pass/Fail criteria.
- 3.3.6.11 GSE-7 mode verification. This test mode is used in conjunction with GSE-3 mode to pause the reflector at the current beam position for 30 integration periods. This process is then repeated. To verify this mode, perform the following procedures.
 - 1. Enter a "7" on the mode switch located on the front of the STE test panel.
 - 2. Wait 18 seconds, and look to see that the unit is performing the scan pattern described. Enter the observed result on TDS 18.
 - 3. Enter the STE command "[3]" to obtain a full printout. Review the following data and record the results on TDS 18.
 - a. Packet ID (elements 1 and 2, page 1 of printout)
 - b. Packet length (elements 3 and 4, page 1 of printout)
 - c. Unit serial number (elements 5 and 6, page 1 of printout)
 - d. Instrument mode/.status (elements 7 and 8, page 1 of printout)
 - e. Reflector positions (30 positions at current beam position, ignore cold cal and warm cal positions on the printout) (pages 1 6 of printout)
 - f. Radiometer scene data (pages 1 6 of printout)
 - g. PRT temperature data (elements 1090 1180, page 7 of printout)
 - h. Status (page 8 of printout)
 - i. Engineering data (page 8 of printout).
 - 4. Attach the printout to TDS 18. There is no Pass/Fail criteria.
- 3.3.7 Radiometer functional performance test. The purpose of this test is to verify the radiometric performance of the 'AMSU-A1 instrument at the system level. This test consists of:
 - 1. PLO frequency measurements (3.3.7.1)



3.3.7.1 PLO frequency measurements. Perform the following procedures.

2.

- 1. The unit should still be powered and configured as shown in Figure 12. The measurement feedhorn and mixer (items 16 and 17 from Table I) should be positioned looking into the Al-1 reflector when the reflector is looking at the cold scene position.
- 2. Enter the STE command "[11] ANTENNA COLD CAL MODE". Wait 18 seconds before issuing the next command.
- 3. Both reflectors should be positioned looking at the cold calibration position and PLO #2 should be active.

 Wait at least 1 hour for the instrument to stabilize.
- 4. Record the frequency measured for PLO #2 on TDS 19. Attach a plot of the spectrum analyzer display labeled PLO #2 to TDS 19. A sample display is shown in Figure 19.
- 5. Enter the STE command "[15] PLO POWER".
- 6. Both reflectors should be positioned looking at the cold calibration position and PLO #1 should now be active. Wait at least 1 hour for the instrument to stabilize.
- 7. Record the frequency measured for PLO #1 on TDS 19. Attach a plot of the spectrum analyzer display labeled PLO #1 to TDS 19.

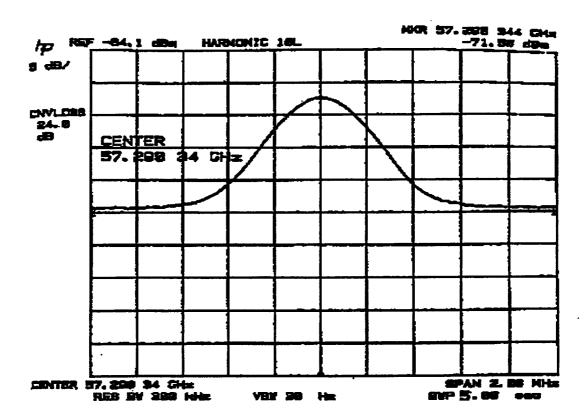


Figure 19. Typical Spectrum Analyzer Plot of PLO Frequency

8. Remove the feedhorn, mixer, and spectrum analyzer leaving the rest of the setup powered and intact for the next section.

3.3.7.2 Relative radiometer NE \(\Delta T \) measurements. The purpose of this test is to perform a preliminary evaluation of the radiometer NE \(\Delta T \) at the system level. Since the STE is not in the thermal vacuum configuration, no temperature readings from the cold load are available. To compute the NE \(\Delta T \) for this test, the temperature used for the cold load temperature shall be 80 K.

The data obtained from this test are considered as relative NEAT and are to be used as a diagnostic tool to verify proper operation of each radiometer channel from antenna input to the spacecraft interface. The equation to determine relative NEAT is as follows:

$$NE\Delta T = \frac{\left[SD*\left(T_h - T_c\right)\right]}{M - N}$$

where

SD = Standard deviation of 120 radiometric samples looking at the warm load

 T_h = Physical temperature of the warm load (300 K)

 T_c = Physical temperature of the cold target (80 K)

M = Average of the radiometric readings in counts viewing the warm load (120 samples)

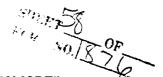
N = Average of the radiometric readings in counts viewing the cold target (30 samples).

Perform the following procedures:

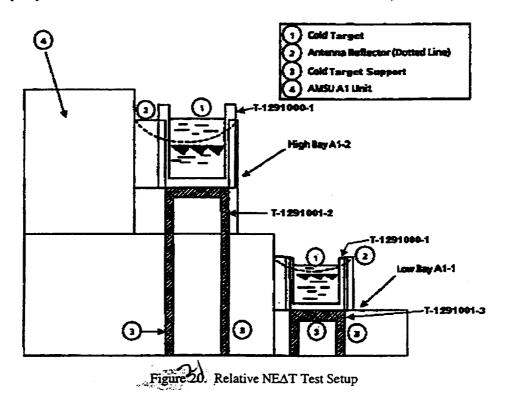
WARNING

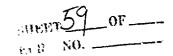
The use of liquid nitrogen in a confined poorly ventilated area can cause asphyxiation and death due to lack of oxygen (oxygen concentration below 20 percent). Accidental contact with liquid nitrogen will cause severe frostbite to the eyes or skin. When handling liquid nitrogen, personnel shall observe the following safety precautions:

- a. Ensure that the work area is well ventilated to prevent excessive gas buildup.
- b. To protect your eyes always wear a face shield or safety goggles (safety glasses without side shields do not provide adequate protection).
- c. To protect exposed skin, always wear an apron when pouring LN2 and whenever exposed to LN2, always wear a lab coat, gloves made for cryogenic work, cuffless trousers (worn outside the boots or shoes), and safety shoes.
- Do not fill target fuller than 1.0 inch from the top.
 Fill target at the floor level, away from unit.
- e. Do not move filled target without cover in place.
- 1. The unit should still be powered and configured as shown in Figure 12. The unit should already be in a stabilized state with PLO #1 active.



- 2. Enter the STE command "[11] ANTENNA FULL SCAN MODE".
- 3. After the unit is stabilized (minimum of 30 minutes required), fill the cold targets with liquid nitrogen and position them as shown in Figure 20.
- 4. Enter the STE command "[1] RETURN" twice to return to the EOS/AMSU-A1 STE main screen shown in Figure 9.
- 5. From the Main screen, enter the STE command "[13] FUNCTIONAL TEST".
- 6. The STE then asks for "COLD TARGET POSITION... ENTER C=COLD, N=NADIR". Enter "C" for cold.
- 7. No additional operator input is needed as the computer will automatically display the results. There is typically a 40 second delay after executing a functional test before the results are displayed. A typical screen is shown in Figure 21.
- 8. Obtain a screen printout by issuing the STE command "[2]".
- 9. Repeat steps 5 through 8 four more times obtaining four additional screen printouts. Average the NEΔT readings from the five printouts for each channel and enter those averages on TDS 20. Attach the printouts to TDS 20.
- 10. Go to the Commands screen on the STE. From the main screen shown in Figure 9, enter a [2] and then a [14].



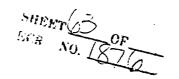


СН	WARM TEMP.	WARM COUNTS	COLD COUNTS	GAIN	DELTA-T
3	297.45	16558.0	13752.0	0.069	0.623
4	297.44	16317.0	13108.0	0.061	0.556
5	•	•	•	-	•
6	•	•	•	•	•
7	•		•	•.*	· • .
8	. •	•	•	•	
9	•	*	•	-	
10	•	•	•	•	•
11	•	•	•	•	
12		,	•	•	*
13	•	•	•		•
14	•	•	•	•	•
15	•	•	•	•	•

Figure 21. Typical Screen Display Following a Functional Test

- 11. Enter the STE command "[15] PLO POWER" to change from PLO #1 to PLO #2. Wait at least 30 minutes to allow the unit to stabilize after changing to PLO #2.
- 12. Change out the cold targets for newly filled cold targets if available. If new targets are unavailable, refill the old targets and wipe off the front surface of the targets to remove any ice or water before proceeding.
- 13. Repeat steps 5 through 9 to obtain data using PLO #2.
- 14. Remove the cold loads and associated hardware.
- 15. Turn the STE power supply panel N/PULSE switch off (refer to Figure 3).
- 16. Turn the STE power supply panel Q/MAIN switch off (refer to Figure 3).
- 17. Turn the STE power supply panel MAIN POWER switch off (refer to Figure 3).

- 3.3.8 Channel Identification Test. The purpose of the Channel Identification Test is to verify the proper final configuration/assembly of each radiometer channel from antenna input to the spacecraft interface.
 - 1. Configure the unit and test equipment as shown in Figure 3
 - 2. Connect the STE to instrument using the following STE interface cables.
 - a. STE interface cable J1 (1356648-1)
 - b. STE interface cable J2 (1356648-2)
 - c. STE interface cable J3 (1356648-3)
 - d. STE interface cable J4 (1356648-4)
 - 3. Turn the STE main power switch ON. From the A1 directory, and at the "\$" prompt, enter the command to the STE "RUN EI." The A1 software program should be running as evidenced by the STE screen shown in Figure 9.
 - 4. Turn the STE power supply panel main power switch ON (refer to Figure 3).
 - 5. Turn the STE power supply panel Q/Main switch ON (refer to Figure 3).
 - 6. Turn the STE power supply panel N/pulse switch ON (refer to Figure 3).
 - 7. From the main screen shown in Figure 9, enter the STE command [2] "MONITOR ONLY." The screen should now be as shown in Figure 10. Enter the STE command "[14] COMMANDS." The screen should now be as shown in Figure 11.
 - 8. Enter the STE command "SCANNER A1-1 POWER." Wait 18 seconds before issuing the next command.
 - 9. Enter the STE command "SCANNER A1-2 POWER." Wait 18 seconds before issuing the next command.
 - 10. Enter the STE command "ANTENNA COLD CAL." Wait 18 seconds before issuing the next command. Both reflectors should scan to the cold calibration beam position.
 - 11. Enter the STE command "[1] RETURN" to return to the monitor only screen shown in Figure 10.
 - 12. Enter the STE command "[10] SCIENCE DATA." The STE should now display the science data screen shown in Figure 22. From this screen enter the STE command "[9] BEAM POSITION NN-ALL CHANNELS.



- 13. The STE then asks "ENTER BEAM POSITION NO (1 TO 30)." Enter "30" to show the radiometric counts data for channels 3-15. The STE should now display the radiometric data screen shown in Figure 23; except with a different set of count data.
- 14. Allow the instrument to stabilize for approximately 20 minutes. Enter the STE command "[2]" to obtain a screen only printout.
- 15. Configure the unit and test equipment as shown in Figure 24. Turn ON the sweeper and allow to warm up approximately 10 minutes. Make sure that the RF power is OFF during sweeper warm up.

CAUTION

Extreme care must be used when turning on RF power. When RF power is first applied the multiplier/gain horn should be approximately three to four feet from the unit. The RF power setting should be no greater than -20 dBm.

- 16. Set the sweeper frequency to 50.35 ± .01 GHz and set the RF power level to -20 dBm. Position the multiplier/gain horn three to four feet from the instrument so that the A1-2 antenna and gain horn are approximately aligned. Rotate the gain horn, if needed, to the vertical polarization position.
- 17. Turn ON the RF power making sure the power level is set to -20 dBm. Allow the multiplier to warm up approximately five minutes.
- 18. At the STE screen compare the radiometric data counts of channel 3 to the counts printed out at step 14. Enter the STE command "[2]" to obtain a screen only printout.
- 19. From the printouts obtained in steps 14 and 18 verify that the radiometric data counts for channel 3 have increased significantly approximately 1000 more, and that the other channels data counts have remained relatively unchanged, less than 300 counts.
- 20. Record the counts difference on TDS 21 of channel 3 from the printouts obtained X steps 18 and 14 and attach printouts to TDS 21.
- 21. Repeat steps 16 through 20 for the frequencies and polarizations listed on TDS 21.



- 22. After all A1 channels have been identified, turn OFF the RF power. Return the reflectors to the warm cal position.
- 23. Turn the STE Q/Main and N/Pulse switches to OFF.
- 24. Turn the STE power supply panel main power switch OFF.

A1-03 E1.EXE;31 COLD CAL MODE P1 5-JUN-98 09:36:59 SCAN NUMBER 34 **EOS** ELEMENT 0000 [5] SCIENCE DATA [6] CONTROLISTATUS ELEMENT **ENGINEERING** [7] DATA STREAM (64 VALUES) [8] BEAM POSITION NN - ALL CHANNELS [9] **CHANNEL NN - ALL BEAAM POSITIONS** [10] WARM CALIBRATE [11] **COLD CALIBRATE** [12] **REFLECTOR POSITIONS** [13] TEMPERATURE DATA (16 VALUES) [14] CHECKSUM IN 15A1 SA28 **POWER** ENGR OK SCREEN ONLY [2] PRINT [3] FULL [I] RETURN **SELECT BUTTON 2** FIGURE 22. **SCIENCE DATA SCREEN** A1-03 E1.EXE;31 COLD CAL MODE P1 5-JUN-98 09:49:07 SCAN NUMBER 11 SCIENCE DATA ELEMENT 0000 [5] CONTROL/STATUS ELEMENT [6] ENGINEERING ELEMENT [7] RADIOMETRIC DATA **BEAM POSITION 30** DATA CH DATA CH DATA 15414 13 16029 10 15102 -15661 16413 15639 11 18044 12 15817 [22] DOWN [21] UP IN DF5D CALC DFSD SA2 ON **CHECKSUM** ENGROK POWER **SELECT BUTTON 2** FIGURE 23. RADIOMETRIC DATA SCREEN P.K. Patel 7/14/98

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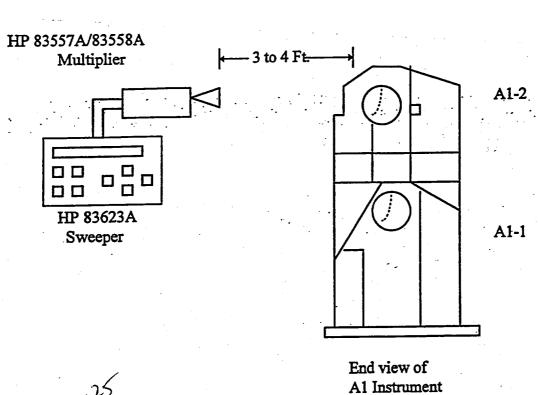


FIGURE CHANNEL IDENTIFICATION SET UP

QC 55 e p. k. Patel 7/4/98 2.

SPORET 60 OF SON NO. 1876

(Add para 3.3.8) See pages 55a Wrough 55e

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1/14/90 QC 1/14/90

4. QUALITY ASSURANCE PROVISIONS

SHEET GF ____

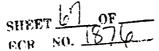
- 4.1 Responsibility for inspection. Aerojet Quality Assurance shall inspect in accordance with the requirements of this test procedure, S-480-80 and S-480-79. Quality Control shall verify all test set-ups prior to start of test. Bonded software shall be used for all tests and shall be obtained from Quality Control. Quality Control shall review all test data for conformance to success criteria. The test data shall include test limits. For tests that satisfy requirements from S-480-80 on protoflight and flight units, customer representatives shall be invited to monitor tests and shall be invited to review the data and show approval on the test data sheets.
- 4.1.1 Test facilities. Unless otherwise specified, the examinations and tests described herein shall be conducted at Aerojet, Azusa Operations, Azusa, California.
- 4.1.2 Electrostatic device (ESD) handling. All electronic hardware shall be handled in accordance with Aerojet Standard STD-2454.
- 4.2 Monitoring procedures. All tests in this procedure shall be witnessed by Quality Control.
- 4.2.1 Test equipment. Test equipment calibration procedures shall comply with the requirements of MIL-STD-45662.
- 4.2.2 Software. Bonded software shall be used at all times.
- 4.3 Monitoring procedures for materials. Not applicable.
- 4.4 Certification. Certification for handling ESD sensitive equipment is required for all personnel working on the assembly and test of the AMSU-A instrument.

4.5 Test methods

4.5.1 Accept-reject criteria. The accept-reject criteria for each examination or test shall be as specified in the data sheets included in each phase of the applicable test procedure. The test results shall be recorded on the data sheets to demonstrate compliance with the applicable specification requirements. Methods of analysis shall be appropriate for the parameters being inspected. It shall be the responsibility of Aerojet to review the test data and determine conformance of the unit under test to the performance requirements contained in S-480-80 and this specification.

In the event of a failure during any phase of this test procedure, the test activity shall record the required information on the Test Anomaly Report and alert the design assurance and quality engineers. Except for failures which only represent a limited out-of-tolerance condition for a particular parameter and are not expected to interfere with the balance of the testing and which are non-destructive, the testing must be stopped until a complete description of the observed anomaly failure is documented and a Failure Analysis Strategy (FAS) is formulated, documented, and implemented to preclude loss of information or evidence that may facilitate determining the failure cause. The full set of data from the referenced tests are required in order to formulate a plan of action. The cognizant reliability engineer, quality assurance engineer, and the system or responsible test engineer shall jointly develop the FAS which must be approved by Design Assurance and Quality Assurance. Analysis and reporting shall be performed in accordance with Aerojet procedures.

4.5.2 General. Separate test reports shall be prepared in accordance with 4.5.2.1.1 for each series which has successfully completed testing. This report shall include all data sheets associated with the tests on the unit plus the data reduction and analysis of specific parameters required by each applicable test procedure specification obtained from screen printouts and plots, oscilloscope photographs, or magnetic recordings. During tests in which a CRT screen is to be printed or plotted and retained as a data sheet, the following annotation shall be applied:



Test/Systems Engineer:	
Quality Control:	(Signature)
Customer Representative:	(Signature)
(Flight hardware only)	(Signature)
Date:	
Test Paragraph No.:	
Subassembly/Assembly Serial No.	

The report shall also include a certification statement. A complete copy of the report shall be included in the shop order package.

4.5.2.1 Acceptance test reports

- 4.5.2.1.1 Format. The acceptance test report shall be prepared and shall include, as a minimum, the following:
 - a. Title page
 - b. Summary
 - c. Requirements satisfied (if any)
 - d. Discrepancy reports (if any)
 - e. Test data
- 4.5.2.1.2 Test data. The test data included in the report shall be that which was obtained during performance of the tests specified herein and recorded on the Test Data Sheet(s) (TDS) (see Appendix A) and on printouts and plots.

5. NOTES

STE

SHEET NO. 187

5.1 Intended use. The intended use of this process specification is to establish the requirements for the comprehensive and limited performance testing of the Advanced Microwave Sounding Unit - A1 System.

5.2 Abbreviations and acronyms

AMSU	Advanced Microwave Sounding Unit
BW	Bandwidth
C CAL CCA CH CPT	Celsius Calibration Circuit Card Assembly Channel Comprehensive Performance Test
DMM DRB DVM	Digital Multimeter Decade Resistor Box Digital Voltmeter
ESD	Electrostatic Discharge
F FAS	Fail Failure Analysis Strategy
GND GPIB GSFC	Ground General Purpose Interface Bus Goddard Space Flight Center
HP HTR	Hewlett-Packard Heater
I/O IF	Input/Output Intermediate Frequency
K	Degrees Kelvin
LO LPT	Local Oscillator Limited Performance Test
max MUX	Maximum Multiplexer
NF	Noise Figure
P P/N PRT	Pass Part Number Platinum Resistance Transducer
RF RTN	Radio Frequency Return
S/N	Serial Number

Special Test Equipment

TDS TLM TAR

Test Data Sheet

Telemetry
Test Anomaly Report

SHEET TO OF TEST DATA SHEETS

10. APPENDIX A

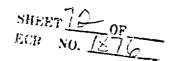
10.1 Scope. This appendix contains the test data sheets for all tests and inspections listed in section 3.

TDS	·	Page
1	Grounding Interface Test	A-2
2	Quiet Power Bus Operational Power Test	A-8
3	Quiet Power Bus Operational Power Test (LPT)	
4	Ouiet Power Bus Turn On Transient Test	A-10
5	Noisy Power Bus Operational Power Test	A-11
6	Noisy Power Bus Turn On Transient Test	A-12
7	Passive Analog Interface Test	A-13
8	Instrument Commanding Test	A-14
9	Science and Engineering Data Test (Full Scan Mode)	A-15
10	Science and Engineering Data Test (Warm Cal Mode)	
11	Science and Engineering Data Test (Cold Cal Mode)	
12	Science and Engineering Data Test (Nadir Mode)	A-25
13	Test Point Interface Test (1.248 MHz Clock TP)	A-27
14	Test Point Interface Test (8 Second Sync Pulse TP)	A-29
15	Test Point Interface Test (Integrate/Hold and Dump TPs)	A-30
16	Test Point Interface Test (Radiometer Channel Analog Output TPs)	A-31
17	Test Point Interface Test (PLO #1 and PLO #2 Lock TPs)	A-32
18	Test Point Interface Test (GSE Modes)	A-33
19	Radiometer Functional Performance Test (PLO Frequency Measurements)	A-34
20	Radiometer Functional Performance Test (Relative NEAT Measurements*)	A-35

TEST DATA SHEET NO. 1 (Sheet 1 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

From Chassis		acecraft Interface		
Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
J1-1	+29V QUIET PWR BUS	> 1M		
J1-2	+29V QUIET PWR BUS	> 1M		
J1-3	29V QUIET BUS RTN	> 1M		
J1-4	29V QUIET BUS RTN	> 1M		
J1-5	+29V NOISY PWR BUS	> 1M		
J1-6	+29V NOISY PWR BUS	> 1M		
J1-7	29V NOISY BUS RTN	> 1M		
J1-8	29V NOISY BUS RTN	> 1M		
J1-9	SURVIVAL PWR BUS A	> 1M		
J1-10	SURVIVAL BUS A RTN	> 1M		
J1-11	SURVIVAL PWR BUS A	> 1M		
J1-12	SURVIVAL BUS A RTN	> 1M		
J1-13	CHASSIS GROUND	<1		
J1-14	+29V QUIET PWR BUS	> 1M		
J1-15	+29V QUIET PWR BUS	> 1M		
J1-16	29V QUIET BUS RTN	> 1M		
J1-17	29V QUIET BUS RTN	> 1M		· ·
J1-18	+29V NOISY PWR BUS	> 1M		·
J1-19	+29V NOISY PWR BUS	> 1M		
J1-20	29V NOISY BUS RTN	> 1M		
J1-21	29V NOISY BUS RTN	> 1M		
J1-22	SURVIVAL PWR BUS B	> 1M		 -
J1-23	SURVIVAL BUS B RTN	> 1M		
J1-24	SURVIVAL PWR BUS B	> 1M		
J1-25	SURVIVAL BUS B RTN	> 1M		

EOS/AMSU- Circle Test:		P/N 1356008 Final CPT	Shop Order:Sub CPT	S/N: LPT	
				Test Systems Engineer	Date
Customer Rep	resentative	Date		Quality Control	Date

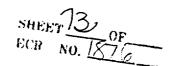


TEST DATA SHEET NO. 1 (Sheet 2 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

From Chassis	Pin Description	cecraft Interface Required Resistance	Measured Value	
Ground to		(Ohms)	(Ohms)	Pass/Fai
J2-1	A1-1 MOTOR TEMP HI	> 1M	(0/	14331 41
J2-2	A1-1 MOTOR TEMP LO	> 1M		
J2-3	A1-1 RECEIVER TEMP 1 HI	> 1M	-	
J2-4	A1-1 RECEIVER TEMP 1 LO	> 1M		
J2-5	A1-1 WARM LOAD TEMP HI	> 1M		
J2-6	A1-1 WARM LOAD TEMP LO	> 1M		
J2-7	A1-2 MOTOR TEMP HI	> 1M		
J2-8	A1-2 MOTOR TEMP LO	> 1M		
J2-9	A1-2 RECEIVER TEMP 1 HI	> 1M		
J2-10	A1-2 RECEIVER TEMP 1 LO	> 1M		
J2-11	A1-2 WARM LOAD TEMP HI	> 1M		
J2-12	A1-2 WARM LOAD TEMP LO	> 1M		
J2-13	No Connection	> 1M		
J2-14	No Connection	> 1M		
J2-15	No Connection	> 1M		
J2-16	No Connection	> 1M		
J2-17	No Connection	> 1M		
J2-18	No Connection	> 1M		
J2-19	No Connection	> 1M		
J2-20	No Connection	> 1M		
J2-21	No Connection	> 1M		
J2-22	A1-1 RECEIVER TEMP 2 HI	> 1M		
J2-23	A1-1 RECEIVER TEMP 2 LO	> 1M		
J2-24	No Connection	> 1M		
J2-25	No Connection	> 1M		
J2-26	No Connection	> 1M		
J2-27	No Connection	> 1M		
J2-28	A1-2 RECEIVER TEMP 2 HI	> 1M		
J2-29	A1-2 RECEIVER TEMP 2 LO	> 1M		
J2-30	No Connection	> 1M		
J2-31	No Connection	> 1M		
J2-32	No Connection	> 1M		
J2-33	No Connection	> 1M	· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·
J2-34	No Connection	> 1M		
J2-35	No Connection	> 1M		
J2-36	No Connection	> 1M		
J2-37	No Connection	> 1M		·

		Test Systems Engineer	Date
Circle Test: 1 st	•	b CPT LPT	
EOS/AMSU-A1	System P/N 1356008 Sh	hop Order: S/N:	· · · · · · · · · · · · · · · · · · ·
J2-37	No Connection	> 1M	
J2-36	No Connection	> 1M	
	No Connection	> 1M	
J2-35	No Commercia	> 1M	

TEST DATA SHEET NO. 1 (Sheet 3 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)



_	J3 of Spacec	raft Interface		
From Chassis Ground to	Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	D # ::
J3-1	1553 INTERFACE DATA A HI	> 100K	(Omis)	Pass/Fai
J3-2	1553 INTERFACE DATA A LO	> 100K		
J3-3	No Connection	> 1M	<u></u>	
J3-4	1553 INTERFACE DATA B LO	> 100K		-
J3-5	1553 INTERFACE DATA B HI	> 100K		
J3-6	1553 INTERFACE DATA A SHIELD	<1		
J3-7	No Connection	> 1M		
J3-8	No Connection	> 1M		
J3-9	1553 INTERFACE DATA B SHIELD	<1		

EOS/AMSU-A1 System 1 Circle Test: 1st CPT	P/N 1356008 Final CPT	Shop Order:	S/N: LPT	
	•			
			Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 1 (Sheet 4 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

From Chassis	Pin Description	Required Resistance	Measured Value	
Ground to		(Ohms)	(Ohms)	Pass/Fail
J4-1	CHASSIS GROUND	<1		
J4-2	8 SECOND SYNC PULSE TP	>25K > 100K &		· · · · · · · · · · · · · · · · · · ·
J4-3	PLO #2 LOCK TP	> 100K		
J4-4	PLO LOCK RTN (7/8)	<1		
J4-5	I/H & DUMP RTN (2/3)	<1		
J4-6	DUMP COMMAND TP	> 100K		
J4-7	No Connection	> 1M		
J4-8	CH 3 ANALOG OUT TP	> 100K		
J4-9	CH 4 ANALOG OUT TP	> 100K		
J4-10	CH 5 ANALOG OUT TP	> 100K		
J4-11	CH 6 ANALOG OUT TP	> 100K		
J4-12	CH 7 ANALOG OUT TP	> 100K		
J4-13	CH 8 ANALOG OUT TP	> 100K		
J4-14	CH 9 ANALOG OUT TP	> 100K		
J4-15	No Connection	> 1M		•
J4-16	No Connection	> 1M		
J4-17	GSE COMMAND LSB	> 5K		
J4-18	GSE COMMAND MSB-1	> 5K		
J4-19	No Connection	> 1M		
J4-20	1.248 MHz CLOCK TP	> 100K		
J4-21	1.248 MHz CLOCK RTN (1)	< 1		
J4-22	PLO #1 LOCK TP	>20K > 100K		
J4-23	No Connection	> 1M		
J4-24	I/H COMMAND TP	> 100K		
J4-25	No Connection	> 1M		
J4-26	ANALOG OUT RTN (2/3)	<1		
J4-27	CH 10 ANALOG OUT TP	> 100K		
J4-28	CH 11 ANALOG OUT TP	> 100K		
J4-29	CH 12 ANALOG OUT TP	> 100K		
J4-30	CH 13 ANALOG OUT TP	> 100K		
J4-31	CH 14 ANALOG OUT TP	> 100K		
J4-32	CH 15 ANALOG OUT TP	> 100K		
J4-33	No Connection	> 1M		
J4-34	No Connection	> 1M		· · · · · · · · · · · · · · · · · · ·
J4-35	GSE COMMAND MSB	>5K		
J4-36	GSE COMMAND RTN (1)	< 1		
J4-37	No Connection	> 1M		
S/AMSU-A1	System P/N 1356008 Shop Ord	er: S/N:		

EOS/AMSU-A1 System P/N Circle Test: 1st CPT F	1356008 Final CPT	Shop Order: Sub CPT		
			Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 1 (Sheet 5 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

	-		Required	Measured Value	
Source	Destination	Source Pin Description	Resistance (Ohms)	(Ohms)	Pass/Fail*
J1-1	J1-2	+29V QUIET PWR BUS	<1		
J1-1	J1-14	+29V QUIET PWR BUS	<1		
J1-1	J1-15	+29V QUIET PWR BUS	<1		
J1-3	J1-4	29V QUIET BUS RTN	<1		
J1-3	J1-16	29V QUIET BUS RTN	<1		
J1-3	J1-17	29V QUIET BUS RTN	<1		
J1-5	J1-6	+29V NOISY PWR BUS	<1		
J1-5	J1-18	+29V NOISY PWR BUS	<1		
J1-5	J1-19	+29V NOISY PWR BUS	<1		
J1-7	J1-8	29V NOISY BUS RTN	<1		
J1-7	J1-20	29V NOISY BUS RTN	<1		
J1-7	J1-21	29V NOISY BUS RTN	<1		
J1-9	J1-11	SURVIVAL PWR BUS A	<1		
J1-10	J1-12	SURVIVAL BUS A RTN	<1		
J1-22	J1-24	SURVIVAL PWR BUS B	<1		
J1-23	J1-25	SURVIVAL BUS B RTN	<1		
J1-1	J1-5	+29V QUIET PWR BUS	> 1M		
J1-1	J1-7	+29V QUIET PWR BUS	> 1M		
J1-1	J1-9	+29V QUIET PWR BUS	> 1M		
J1-1	J1-10	+29V QUIET PWR BUS	> 1M		
J1-1	J1-22	+29V QUIET PWR BUS	> 1M		
J1-1	J1-23	+29V QUIET PWR BUS	> 1M		
J1-3	J1-5	29V QUIET BUS RTN	> 1M		
J1-3	J1-7	29V QUIET BUS RTN	> 1M		
J1-3	J1-9	29V QUIET BUS RTN	> 1M		

EOS/AMSU-A1 System F Circle Test: 1 st CPT	P/N 1356008 Final CPT	Shop Order:	S/N: LPT	
Customer Representative	Date		Test Systems Engineer Quality Control	Date Date

TEST DATA SHEET NO. 1 (Sheet 6 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

	D	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fail
Source	Destination			(Ошь)	1 433/1 411
J1-3	J1-10	29V QUIET BUS RTN	> 1M		
J1-3	J1-22	29V QUIET BUS RTN	> 1M		
J1-3	J1-23	29V QUIET BUS RTN	> 1M		
J1-5	J1-9	+29V NOISY PWR BUS	> 1M		
J1-5	J1-10	+29V NOISY PWR BUS	> 1M		
J1-5	J1-22	+29V NOISY PWR BUS	> 1M		
J1-5	J1-23	+29V NOISY PWR BUS	> 1M		
J1-7	J1-9	29V NOISY BUS RTN	> 1M		
J1-7	J1-10	29V NOISY BUS RTN	>1M		
J1-7	J1-22	29V NOISY BUS RTN	> 1M		
J1-7	J1-23	29V NOISY BUS RTN	> 1M		
J1-9	J1-22	SURVIVAL PWR BUS A	> 1M		
J1-9	J1-23	SURVIVAL PWR BUS A	> 1M		
J1-10	J1-22	SURVIVAL BUS A RTN	> 1M		
J1-10	J1-23	SURVIVAL BUS A RTN	> 1M		
J1-13	J1 OUTER SHELL	CHASSIS GROUND	<1		
J1-13	J2 OUTER SHELL	CHASSIS GROUND	<1		
J1-13	J3 OUTER SHELL	CHASSIS GROUND	<1		
J1-13	J4 OUTER SHELL	CHASSIS GROUND	<1	,	
J3-1	J3-5	1553 INTERFACE DATA A HI	> 100K		
J3-1	J3-4	1553 INTERFACE DATA A HI	> 100K		
J3-2	J3-5	1553 INTERFACE DATA A LO	> 100K		
J3-2	J3-4	1553 INTERFACE DATA A LO	> 100K		<u> </u>

EOS/AMSU-A1 System P/I Circle Test: 1st CPT	N 1356008 Final CPT	Shop Order:	S/N: LPT	
			Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 2 Quiet Power Bus Operational Power Test (Paragraph 3.3.3.1.1)

	Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	PLO	Maximum Peak Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Peak Power (QBV x QBI) (Watts)	Pass/Fail
	26.95 - 27.05		#1		<u>≤</u> 94		
eg	28.95 - 29.05		#1		≤94		
	30.95 - 31.05		#1		≤94		
6	26.95 - 27.05		#2		≤94		
81	28.95 - 29.95		#2		≤94		
	30.95 - 31.05		#2		≤94		

	Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	PLO	Average Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Average Power (QBV x QBI) (Watts)	Pass/Fail
	26.95 - 27.05		#1		≤88 <8£		
K	28.95 - 29.05	·	#1		≤88		
	30.95 - 31.05		#1		≤88 ∠86		
	£ 26.95 - 27.05		#2		≤88 ∠86		
I	28.95 - 29.05		#2		≤88 ∠86		
	30.95 - 31.05		#2		₹88 ₹ <i>€</i> €		

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EOS/AMSU-A1 System I Circle Test: 1 st CPT	P/N 1356008 Final CPT	Shop Order:	S/N:	
			Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 3 Quiet Power Bus Operational Power Test (LPT) (Paragraph 3.3.3.1.2)

er j				•	·	
Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	Average Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Average Power (QBV x QBI) (Watts)	Pass/Fail	3 J
28.95 - 29.05			≥88 86			
			7-14	the fly france	፞	
: :						
AMSU-A1 System I	P/N 1356008	Shop Order:		S/N:		

Date

Customer Representative

Quality Control

Date

TEST DATA SHEET NO. 4
Quiet Power Bus Turn On Transient Test (Paragraph 3.3.3.1.3)

Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	Amps	<10.6 Amps	
Pulse Width (Steady State)	ms	< 250 mscc <150 ms	
Rate of Change(slope): dI/dT	ma/µs	<677 mA/µs	
F[Pulse Width (transient)	ms	< 100 M Sec	
+29 Volts			
Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	Amps	<10.6 Amps	
Pulse Width (Steady State)	ms	< 250 msec <150 ms	
Rate of Change(slope): dl/dT	ma/µs	<677 mA/μs	
Pulse width (transient)	ms	< 100 msec	
+27 Volts			
Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	Amps	<10.6 Amps	
Pulse Width (strady State)	ms	< 250 M Sec <150 ms	
Rate of Change(slope): dI/dT	ma/μs	<677 mA/μs	
Pulse width (transient)	ms	<100 msec	
EOS AMSU- AL P/N'. <u>B</u>	26.00 2N:		
6:	_		
ircle Test 1 st CPT Final	CPT Sub CPT		
	Test	Erstems Engineer	·

TEST DATA SHEET NO. 5 Noisy Power Bus Operational Power Test (Paragraph 3.3.3.2.1)

Required Noisy Bus Voltage NBV (Volts)	Measured NBV (Volts)	Required Peak Current (Amps)	Maximum Peak Noisy Bus Current NBI (Amps)	Required Peak Power (Watts)	Calculated Peak Power (NBV x NBI) (Watts)	Pass/Fail
26.95 - 27.05		≤1		<u>≤</u> 40		
28.95 - 29.05		≤1		<u>≤</u> 40		
30.95 - 31.05		≤l		≤40		

Required Noisy Bus Voltage NBV (Volts)	Measured NBV (Volts)	Average Noisy Bus Current NBI (Amps) Sec	Required Average Power (Watts)	Calculated Average Power (NBV x NBI) (Watts)	Pass/Fail
26.95 - 27.05			<u><</u> ¢ 8		
28.95 - 29.05			≤¢ 8		
30.95 - 31.05			≤¢ 8		<u> </u>

Required Noisy Bus Voltage NBV (Volts)	Measured NBV (Volts)	Bus Current During the I/H, D. Period	Pass/Fail
26-75 - 27.05		ma *	Not Applicable
28.75 - 29.05		MR *	Not Applicable
30.75 — 31.05		ma de	Not Applicable

* between bears

BEIDEST - TO			•
EOS/AMSU-A1 System P/N 1356008 Circle Test: 1 st CPT Final CPT	Shop Order:Sub CPT	S/N:	
		Test Systems Engineer	Date
Customer Representative Date		Quality Control	Date

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ECR	NO.	1875	~

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TEST DATA SHEET NO. 6 pisy Power Bus Turn On Transient Test (Paragraph 3.3.3.2.2)	ECR	ľ

+31 Volts

Parameter	Measured/Calculated	Required	Pass/Fail	
Peak Current	Amps	<11.5 Amps		
Pulse Width	ms	<100 ms		
Rate of Change(slope): dI/dT	ma/µs	<744 mA/μs		

+29 Volts

Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	Amps	<11.5 Amps	
Pulse Width	ms	<100 ms	
Rate of Change(slope): dI/dT	ma/µs	<744 mA/μs	

+27 Volts

Parameter	Measured/Calculated	Pass/Fail	
Peak Current	Amps	<11.5 Amps	
Pulse Width	ms	<100 ms	
Rate of Change(slope): dI/dT	ma/μs	<744 mA/μs	

EOS ANSU 5/0':		/;	2N;	<u> </u>
Circle Test	1 ST CPT	Final CAT	Sub GPT	
			Test Trotems Engineer	Oute:
Castomer K	erresentativ	ie	Quality: CONTROL	

TEST DATA SHEET NO. 7 Passive Analog Interface Test (Paragraph 3.3.4)

Number	Thermistor	Required Temperature (Celsius)	Measured Temperature (Celsius)	Pass/Fail
1	A1-1 SCAN MOTOR	* ± 5°		
2	A1-2 SCAN MOTOR	*±5°		
3	A1-1 RF SHELF#1	*±5°		
4	A1-2 RF SHELF#1	* ± 5°		
5	A1-1 WARM LOAD	*±5°		
6	A1-2 WARM LOAD	*±5°		
7	A1-1 RF SHELF # 2	* ± 5°		
8	A1-2 RF SHELF # 2	*±5°		

^{*} is the measured temperature of the unit environment

EOS/AMSU-A1 System I		Shop Order: Sub CPT		
			Test Systems Engineer	Date
Customer Representative	Date		Ouality Control	Date

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TEST DATA SHEET NO. 8 Instrument Commanding Test (Paragraph 3.3.5.2)

stomer Represen	tative	Date Quality Co	ontrol	Date
		Test Syste	ms Engineer	Date
Circle Test: 1 st		N 1356008 Shop Order: S/N: S/N: S/N:		
EOS/AMSU-A1	System D/I	V 1356008 Show Onder		
Y.	25 s = Pass	Did C&DH processor reset? No = Fail		
		Did PLO toggle?		
j		Reflectors positioned looking at cold cal 1?		
		Reflectors positioned looking at cold cal 2?		
		Reflectors positioned looking at cold cal 3?		
· · · · · · · · · · · · · · · · · · ·	21 22	Reflectors positioned looking at cold cal 4?		
		Reflectors positioned looking at cold cal 1?		
	20	Reflectors positioned looking at nadir?		
	19	Reflectors positioned looking at warm loads?		
	18	Are both motors scanning?		
	17	Did A1-2 motor stop scanning?		
	15	Is A1-2 motor scanning?		
	14	Did A1-1 motor stop scanning?		
	13	Is A1-1 motor scanning?		
	12	Full Scan Mode command received?		
	Step	Instrument Status	(Y)es / (N)o	
	l Stan		1 1	

TEST DATA SHEET NO. 9 (sheet 1 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

Step	Instrument Status	(Y)es / (N)o
1	Full Scan Mode command received?	(=)00. (2.)0
2	ENGR OK message seen?	
3	Unit (both reflectors) running in full scan mode?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000101	
4b	3-4	Packet Length		0000001010111111	
4c	5-6	Unit Serial Number		0000001100000000	
4d	7-8	Instrument Mode/ Status		10/1101000000010	

			0	K. HENZbein
<u></u>	RADIOMETE	R SCENE DATA		
Step	Description	Required Counts	П	(P)ass/(F)ail
4f	Review All Scene Data	12500-20500		(- /

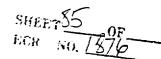
		PRT TEMPERATURE	DATA	
Step	Element	Description	Required	(P)ass/(F)ail
4g	1090-1178	Review All PRT Data**	10-40 degrees C	(-)== (-)==:
4g	1180	Temperature Sensor Reference		

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		YES	(-)(-)
	Antenna in Warm Cal Mode		NO	· · · · · · · · · · · · · · · · · · ·
	Antenna in Cold Cal Mode		NO	
	Antenna in Nadir Mode		NO	
	Cold Cal Position LSB		ZERO	· · · · · · · · · · · · · · · · · · ·
4h	Cold Cal Position MSB		ZERO	
	PLO Redundancy		PLO #1	
	Scanner A1-1 Power	· · · · · · · · · · · · · · · · · · ·	ON	
	Scanner A1-2 Power		ON	
	PLO #1 Lock		YES	
	PLO #2 Lock		OFF	
	ADC Latchup Flag		ONE	

* Rewriting printout data on	this data sheet	is optional.
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EOS/AMSU-A1 System P/N 1356 Circle Test: 1 st CPT Final (_	
		Test Systems Engineer	Date
Customer Representative D	ate	Quality Control	Date

^{**} Refer to Table IV for PRT Data Description



TEST DATA SHEET NO. 9 (sheet 2 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

		A1-1 REF	LECTOR		IONS (Step	4e) A1-2 REF	I ECTOR	
BP	Element	Position (*)	Required (**) ± 5	(P)ass/ (F)ail	Element	Position (*)	Required	(P)as
1	14		<u> </u>	(1)411	16	(-)	$(**) \pm 5$	(F)a
2	48				50			
3	82				84			
4	116				118			
5	150				152			
6	184				186			
7	218				220			
8	252				254			
9	286				288			
10	320				322			
11	354				356			
12	388				390			
13	422				424			
14	456				458			
15	490				492			
16	524				526			
17	558				560			
18	592				594			
19	626				628			
20	660				662			
21	694				696			
22	728				730			
23	762				764			
4	796				798			
5	830				832			
6	864				866			
7	898				900			
8	932				934			
9	966				968			
0	1000				1002			
c T	1034				1036			
С	1186				1188			
	* Actual co ** Required	unts from pr	intout. Rewri	ting count	s on this dat	a abast is a	<u> </u>	

EOS/AMSU-A Circle Test: 1	l System CPT	P/N 1356008 Final CPT	Shop Order:Sub CPT	S/N: LPT	•
				Test Systems Engineer	Date
Customer Represe	entative	Date		Quality Control	Date

TEST DATA SHEET NO. 9 (sheet 3 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

	. ENGIN	EERING DATA	•	
Step	Description	Measured*	Required	(P)ass/(F
	Signal Processor (+5 VDC)	·	+4 to +6 volts	
	Signal Processor (+15 VDC)		+14 to +16 volts	
	Signal Processor (-15 VDC)		-14 to -16 volts	
-	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	
	PLO (-15 VDC)	·	-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
ļ	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
Ĺ	Quiet Bus Current		≤ 3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	
	A1-2 Noisy Bus Current		≤ 125 milliamps	

EOS/AMSU-A1 System P/N 13560 Circle Test: 1 st CPT Final CP		S/N: LPT	
		Test Systems Engineer	Date
Customer Representative Dat	e	Quality Control	Date

TEST DATA SHEET NO. 10 (Sheet 1 of 2)

Science and Engineering Data Test (Warm Cal Mode) (Paragraph 3.3.5.3.2)

	Step	Instrument Status	(Y)es / (N)o
L	-1	Warm Cal Mode command received?	(1)037 (11)0
Г	2	ENGR OK message seen?	
Г	3	Both reflectors positioned at warm loads?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	
4b	3-4	Packet Length		0000001010111111	
4c	5-6	Unit Serial Number		0000001100000000	
4d	7-8	Instrument Mode/ Status		10/1101000000100	

			U	K Heuricu
	RADIOMETE		ें ते	
Step	Description	Required Counts	Т	(P)ass/(F)ail
4f	Review All Scene Data	12500-20500	\vdash	(1)455 (1)441

PRT TEMPERATURE DATA					
Step	Element	Description	Required	(P)ass/(F)ail	
4g	1090-1178	Review All PRT Data**	10-40 degrees C	(1 /ass/(1 /aii	
4g	1180	Temperature Sensor Reference	23244-26317 counts		

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		NO	(1)ass/(1)all
	Antenna in Warm Cal Mode		YES	
	Antenna in Cold Cal Mode		NO	
	Antenna in Nadir Mode		NO	
	Cold Cal Position LSB		ZERO	
4h	Cold Cal Position MSB		ZERO	
	PLO Redundancy		PLO #1	
	Scanner A1-1 Power	Power		
	Scanner A1-2 Power		ON ON	
	PLO #1 Lock		YES	
	PLO #2 Lock			
	ADC Latchup Flag		OFF ONE	

^{*} Rewriting printout data on this data sheet is optional.
** Refer to Table IV for PRT Data Description

EOS/AMSU-A1 System P/I Circle Test: 1st CPT	N 1356008 Final CPT	Shop Order: Sub CPT	S/N: LPT	
·			Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 10 (sheet 2 of 2) Science and Engineering Data Test (Warm Cal Mode) (Paragraph 3.3.5.3.2)

_ -		-1 REFLECTOR		A1	-2 REFLECTOR	
P	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail	Position Range	Required (**) ± 5 counts	(P)ass/ (F)ail
0						(1)411
0	Rewriting cor	(min to max) of counts on this data sh	ounts from p	rintout (Only beam	positions 1-30).	

	ENGI	VEERING DATA		
Step	Description	Measured***	Required	(P)ass/(F)ail
	Signal Processor (+5 VDC)		+4 to +6 volts	(= /== (= /=)
	Signal Processor (+15 VDC)		+14 to +16 volts	
	Signal Processor (-15 VDC)		-14 to -16 volts	
	Scan Drive (+5 VDC)		+4 to +6 volts	· · · · · · · · · · · · · · · · · · ·
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	-
~	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current			
	A1-1 Noisy Bus Current		≤ 125 milliamps	
	A1-2 Noisy Bus Current		≤ 125 milliamps	

*** Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System P/N 1356 Circle Test: 1st CPT Final C		S/N: LPT	
		Test Systems Engineer	Date .
Customer Representative Da	ate	Quality Control	Date

TEST DATA SHEET NO. 11 (Sheet 1 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

Step	Instrument Status	(Y)es / (N)o
1	Cold Cal Mode command received?	(, , , , , , , , , , , , , , , , , , ,
2	ENGR OK message seen?	
3	Both reflectors positioned at cold cal position 1?	
6	Cold Cal Position 2 command received?	
7	ENGR OK message seen?	
8	Both reflectors positioned at cold cal position 2?	
11	Cold Cal Position 3 command received?	
12	ENGR OK message seen?	
13	Both reflectors positioned at cold cal position 3?	-
16	Cold Cal Position 4 command received?	
17	ENGR OK message seen?	
18	Both reflectors positioned at cold cal position 4?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	
4b	3-4	Packet Length		0000001010111111	
4c	5-6	Unit Serial Number		0000001100000000	
4d	7-8	Instrument Mode/ Status		1071101000001000	
9a	7-8	Instrument Mode/ Status		10/1101000101000	·
14a	7-8	Instrument Mode/ Status		1071101001001000	
19a	7-8	Instrument Mode/ Status		1011101001101000	

0 RHutiberg 7-13-98 RADIOMETER SCENE DATA Required Counts (P)ass/(F)ail

10-40 degrees C

23244-26317 counts

	4f	Review All Scene Data	12500-20500	
-		PRT TEMPER	ATURE DATA	
Step	Element	Description	Requir	ed (P)acc/(F)ail

* Rewriting printout data on this data sheet is optional.

Description

Review All PRT Data**

Temperature Sensor Reference

** Refer to Table IV for PRT Data Description

Step

1090-1178

1180

EOS/AMSU-A1 System P/N 13560 Circle Test: 1 st CPT Final CF		S/N: LPT	
			•
		Test Systems Engineer	Date
Customer Representative Date	e	Quality Control	Date

TEST DATA SHEET NO. 11 (sheet 2 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		NO	
	Antenna in Warm Cal Mode		NO	
	Antenna in Cold Cal Mode		YES	
	Antenna in Nadir Mode		NO	
	Cold Cal Position LSB		ZERO	
4h	Cold Cal Position MSB		ZERO	
	PLO Redundancy		PLO#1	
	Scanner A1-1 Power		ON	
	Scanner A1-2 Power		ON	
	PLO #1 Lock	_	YES	
	PLO #2 Lock		OFF	
	ADC Latchup Flag		ONE	
9ь	Cold Cal Position LSB		ONE	
	Cold Cal Position MSB		ZERO	
i4b	Cold Cal Position LSB		ZERO	
	Cold Cal Position MSB		ONE	
19b	Cold Cal Position LSB		ONE	
	Cold Cal Position MSB		ONE	

^{*} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System Circle Test: 1 st CPT	P/N 1356008 Final CPT	Shop Order:Sub CPT	S/N: LPT	
			Test Systems Engineer	Date

SHEET I OF ECR NO. 1876

TEST DATA SHEET NO. 11 (sheet 3 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

mer Rep	resentative	Date		Quality Contr	ol	Date
-				Test Systems	Engineer	Date
S/AMSU le Test:	J-A1 System P/N 1 1st CPT Fin	356008 Shop O al CPT Sub CP	rder: T	S/N: LPT		
		•				
* A	ctual range (min to ewriting counts on	max) of counts from this data sheet is o	om printout (ptional.	Only beam position	s 1-30).	
	vedanea conu	is from AE26002/	TDS 5&6 +	/- 5 counts for Cold	Cal Position #4	
30	** Pagyind as	4. C				
1-	(*)	± 5 counts	(F)ass/ (F)ail	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail
BP	Position Range	REFLECTOR Required (**)	(P)ass/	A1-2 REFLECTOR		
ļ	A 4	REFLECT	OR POSITION	ONS (Step 19e)		
					u Cai Position #3	
	** Required cou	nts from AE26002	1 TDS 5&6	+/- 5 counts for Col	1015	
1- 30			<u> </u>	()	± 5 counts	(F)ail
	(*)	± 5 counts	(F)ass/ (F)ail	Position Range (*)	Required (**)	(P)ass/
BP	Position Range	Required (**)	(P)ass/	A1-	2 REFLECTOR	
	A1	-1 REFLECTOR	TOK POSITI	ONS (Step 14c)		
				22 C		
	vedanten col	inis from AE26002	11 TDS 5&6	+/- 5 counts for Co	ld Cal Position #2	·
30	** Required and	Into from A Economic	//			
1- 30					counts	(F)ail
- -	(*)	±5 counts	(F)ail	(*)	± 5 counts	(P)ass/
BP	Position Range	Required (**)	(P)ass/	Position Range	-2 REFLECTOR Required (**)	(D)
_	A	1-1 REFLECTOR			2 DEEL POTOS	
		REFLE	CTOR POST	14 C. TIONS (Step 9c)		
			<u>-11003&</u>		old Cal Position #	1
		unts from AF2600	2/1 TDC 594	5 +/- 5 counts for C		
1- 30				T		(F)ail
<u> </u>	(*)	±5 counts	(F)ail	* (*)	Required (**) ± 5 counts	(P)ass/
BP	Position Range	Required (**)	(P)ass/	Position Range	1-2 REFLECTOR	
ı	A	1-1 REFLECTOR		TIONS (Step 46)	100000	
			3C.11.1K PI 151			

MIN (12) (227)

TEST DATA SHEET Nº 11 (sheet 4 of 5) SCIENCE AND ENGINEERING DATA TEST (COLD CAL MODE) (PARGRAPH 33.5.3.3)

						SHEET
BEAM	V 41=1 6	REFLECTOR P	BSITIONS (STEP (L)		ECB /6:
POSITION	ACTUAL POSICION#	REGULERON	(P) ASS/	AI-	Z REFLECTOR	(MET)
COLD CAL		# 5 COUNTS	LEAIL	בי נפאונטים	REQUIRED**	(FIALL
<u> -'</u>	74131	4129	P	3779_	3777	R
Į, i	*	ACTUAL COUNT	FRUM ARIN	TOT COLD CAL	1 BEAM ROLL	(m)
3.2	**	MEQUIRED CIVI	UT FROM AC	E-21002/16 TD	BEAM ASIC S 6 ±5 coyuts	FOR COLD CALL

REFLECTOR POSITIONS (STEP 140) REFLECTOR AL-1 REFLECTOR RESTION ACTUAL POSITIONS REQUIRED STEP (P) ASS/ E) ALL ACTUAL RESTIONS REQUIRED STEP (P) ASS/						
PESTION COLD CAL	ACTUAL POSITIONAR	REQUIRED	(F) ASS/	ACTUAL POSITIONS	REQUIRED **	PIA'S/
2	4050	4033	·p	3703-	3701	12
	** **	REQUIRED C	OUT FROM A	RINTAUT COLP CAN AE-ZIGOZIC TI	L 2 BEAM POS 05 G ±5 CONT	S FOR COLD CAL 2

1 A1-1	KU CI / C		S (STEP 22C)	<u>-</u>	
COLD (A)	* REQUIRED **		ACTUAL RESIDENT	REFLECTOR REDUIRED **	(PAGS/
3 3976	3977	1-10-	3622	3625	PAIL
× ××	RECURBO CO	nt from pour train	RINTOUT COLD C		FOR COLD CAL 3

- 1			_					_1
	BEAM BESTOWN	AI-1 ATTAL PSISPUX	REFLECTOR REFLECTOR	Glass	(STEP 30 C)			F
	CTLD CAL 4	1 2040	3320	12	2/15/	REGULAD **	(PASS/	İ
1		**	ACTUAL COUN REQUIRES C	JEFROM PRIM DUNT FROM A	JET WED CAR	3474 4 BEAM COSITION	P COXD CAX 4	-
						- S COUNTS FOR	COXD CAX 4	

CIRCLE TEST (STEP) PIN 1356008 SHOPOROR: 29856 SIN: 202 CIRCLE TEST (STEP) FINAL CPT SUB OF NIA LOT NIA

TEST SYSTEMS ENGINEER DATE

CUSTOMER REPRESENTATIVE DATE

QUALITY CONTROL

14 14 1938 OATE

himid

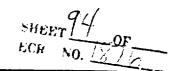
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PET NO. TEST DATA SHEET NO. 7-1348

TEST DATA SHEET NO. 11 (sheet 4 of 5)
Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.)

•
Fail
ail
au.
ail
ail
nil
nil

TEST DATA SHEET NO. 11 (sheet 5 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)



	ENGIN	EERING DATA		
Step	Description	Measured*	Required	(P)ass/(F)ail
,	Signal Processor (+5 VDC)		+4 to +6 volts	
	Signal Processor (+15 VDC)		+14 to +16 volts	
	Signal Processor (-15 VDC)		-14 to -16 volts	-
	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	· · · · · · · · · · · · · · · · · · ·
	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	· · · · · · · · · · · · · · · · · · ·
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
[LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current		≤3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	
ſ	A1-2 Noisy Bus Current		≤ 125 milliamps	

* Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System P/N 1356008 Shop Order: _____ S/N: ____
Circle Test: 1st CPT Final CPT Sub CPT _____

Test Systems Engineer Date

Customer Representative Date

Quality Control Date

AE-20 18 Jun

TEST DATA SHEET NO. 12 (Sheet 1 of 2) Science and Engineering Data Test (Nadir Mode) (Paragraph 3.3.5.3.4)

Step	Instrument Status	(Y)es/(N)o
1	Nadir Mode command received?	(2)007 (21)0
2 .	ENGR OK message seen?	
3	Both reflectors positioned at nadir position?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	
4b	3-4	Packet Length		0000001010111111	
4c	5-6	Unit Serial Number		0000001100000000	
4d	7-8	Instrument Mode/ Status	†	1001101000010000	

RADIOMETER SCENE DATA						
Step	Description	Required Counts	(P)ass/(F)ail			
4f	Review All Scene Data	12500-20500	(= /== (= /===			

		PRT TEMPERATURI	DATA	
Step	Element	Description	Required	(P)ass/(F)ail
4g	1090-1178	Review All PRT Data**	10-40 degrees C	(1)433/(1)411
4g	1180	Temperature Sensor Reference	23244-26317 counts	

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		NO	(1 /433/(1 /41)
	Antenna in Warm Cal Mode		NO	
	Antenna in Cold Cal Mode		NO	
	Antenna in Nadir Mode		YES	
	Cold Cal Position LSB		ZERO	
4h	Cold Cal Position MSB		ZERO	
	PLO Redundancy		PLO#1	
	Scanner A1-1 Power		ON ON	
	Scanner A1-2 Power		ON	
	PLO #1 Lock		YES	
	PLO #2 Lock			
	ADC Latchup Flag		OFF ONE	

* Reumiting printout date	· a- ebi-	4-41		
* Rewriting printout data	i on inis	cata sr	ieet ic	Contional
U 1				vruviiai.

EOS/AMSU-A1 System Circle Test: 1 st CPT	n P/N 1356008 Final CPT	Shop Order: Sub CPT	S/N: LPT	
			Test Systems Engineer	Date•
Customer Representative	Date		Quality Control	Date

^{**} Refer to Table IV for PRT Data Description

TEST DATA SHEET NO. 12 (sheet 2 of 2)

Science and Engineering Data Test (Nadir Mode) (Paragraph 3.3.5.3.4)

		REFLE	CTOR POST	HONS (Step 4e)		
BP	A	-1 REFLECTOR		A1	-2 REFLECTOR	
БР	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail	Position Range	Required (**)	(P)ass/
1-			(= /	 	± 5 counts	(F)ail
30						
				rintout (Only beam al. +/- 5 counts for "rr		

	ENGIN	EERING DATA		·
Step	Description	Measured***	Required	(A) (A)
	Signal Processor (+5 VDC)			(P)ass/(F)ail
	Signal Processor (+15 VDC)		+4 to +6 volts	
	Signal Processor (-15 VDC)		+14 to +16 volts	
	Scan Drive (+5 VDC)		-14 to -16 volts	
	Scan Drive (+15 VDC)		+4 to +6 volts	
	Scan Drive (-15 VDC)		+14 to +16 volts	
	PLO (+15 VDC)		-14 to -16 volts	
	PLO (-15 VDC)		+14 to +16 volts	
	Receiver (+8 VDC)		-14 to -16 volts	
4i			+7 to +9 volts	
••	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
- 1	LO Channel 3		+9 to +11 volts	·
ŀ	LO Channel 4		+9 to +11 volts	
ļ	LO Channel 5		+9 to +11 volts	
- 1	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
L	Quiet Bus Current			
L	A1-1 Noisy Bus Current		≤ 3 Amps	
	A1-2 Noisy Bus Current		≤ 125 milliamps	
			≤ 125 milliamps	

^{***} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System Circle Test: 1 st CPT	Final CPT	Shop Order:	_ S/N: LPT	
	•	the state of the s	Test Systems Engineer	Date
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 19° 1553 Bus Interface Test (Paragraph 3.3.5.4)

	ATTACH BUS A	WAVE FORM	
Bus A Amplitude Bus A Rise Time		: 18.0 – 27.0 VP-P : 100 – 300 nsec	P/F
	ATTACH BUS B W	AVE FORM	
			·
			P/F
Bus B Amplitude Bus B Rise Time		: 18.0 – 27.0 VP-P : 100 – 300 nsec	
Bus B Amplitude Bus B Rise Time CPT:; Final CFO:	PTT		

SHEET 98 OF NO. 1500

TEST DATA SHEET NO. 13 Noisy Bus Current Measurement During Warm Cal, Cold Cal and Nadir

	•		Test Systems E	ngineer	Date
/Ai le]	MSU-A1 System P/N 1356008 Test: 1 st CPT Final CPT	Shop Order:Sub CPT	S/N: _ LPT		
				No Applic	
	adir 1-1 & A1-2 Scanner N			ļ	
A O	old Cal 1-1 & A1-2 Scanner N				
A	11-2 Scanner / Al-1 Scanner OFF ON				
A O	Varin Cal A1-1 & A1-2 Scanner ON A1-1 Scanner / A1-2 Scanner				lot icable
77	Instrument Mode Varm Cal	Noisy Bus		Pas	s/Fail

TEST DATA SHEET NO. 14 Test Point Interface Test (8 Second Sync Pulse TP) (Paragraph 3.3.6.2)

8 SECOND SYNC PULSE TEST POINT

Attach Photograph or Plot Here or to Back of TDS

	8 SE	COND SYNC PULSE	TEST POINT	
Step	Parameter	Measured	Required	(P)ass / (F)ail
2	Pulse Length	seconds	8 seconds +/- 10%	
. 2	Amplitude	volts	3 – 5 volts	

EOS/AMSU-A1 System P/N 1356008 Circle Test: 1 st CPT Final CPT		Shop Order:Sub CPT			
				Test Systems Engineer	_
				Quality Control	_

SHEET/O/ OF ECH NO. 1870

TEST DATA SHEET NO. 16 Test Point Interface Test (Radiometer Channel Analog Output TPs) (Paragraph 3.3.6.4)

RADIOMETER CI	HANNEL ANALOG OUTPUT	TEST POINTS

Attach Photographs or Plots Here or to Back of TDS

	RAI	DIOMETER CH	IANNEL ANA	LOG OUTPI	UT TEST POIN	VTS	
<i>c</i> . ,	Integration	Integration	Hold	Hold	Dump	Dump /	
Channel	Time.	Time	Time	Time	Time	Time	(P)ass /
ł	Measured	Required	Measured	Required	Measured	Required	(F)ail
	(E)*	(ms) לאן	(F)*	(ms)	(F)*	(ms)	(-)
3	ms	158 ± 5 ms	ms	23-27			
4	ms	158 ± 5 ms	ms	23-27	H.S.		
5	ms	158 ± 5 ms	ms	23-27	mx	9-15	
6	ms	1.58 ± 5 ms	ms	23-27	o ms	9-15	
7	ms	158 ± 5 ms	ms	23-27	P ms	9-15	
8	ms	158 ± 5 ms	ms	23-27	/ ms	9-15	
9	ms	158 ± 5 ms	ms	23-27	ms	9,15	
10	ms	158 ± 5 ms	ms	23-27	/ ms	9-15	
11	ms	188 ± 5 ms	ms	23-27	/ ms	9-15	
12	ms	1.58 ± 5 ms	ms	23-27	/ ms	9-15\	
13	ms	158 ± 5 ms	ms	23-27	ms	9-15 \	
14	ms	158 ± 5 ms	ms	23-27	ms	9-15	
15	ms .	\$158 ± 5 ms	ms	23-27	ms	9-15	

Shop Order: ____

Sub CPT ___

S/N: ____

Test Systems Engineer

Quality Control

Date

Date

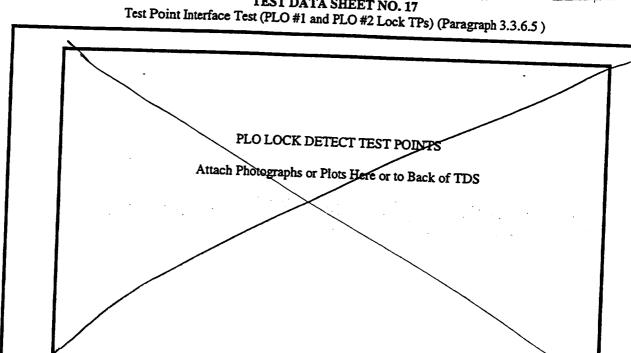
EOS/AMSU-A1 System P/N 1356008

Final CPT

Circle Test: 1st CPT

V

TEST DATA SHEET NO. 17



-		CK DETECT TEST P	OINTS	•
Step	Parameter	Measured	Required	(P)ass / (F)ail
3	PLO #1 Lock Detect*	volts	rvolt و خ	(1/4357 (1/411
6	PLO #2 Lock Detect**	volts	- Ot volt	

^{*} When PLO #1 is selected

EOS/AMSU- Circle Test:		Shop Order:Sub CPT	S/N:	
* • •				
			Test Systems Engineer	Date
			Quality Control	Date

^{**} When PLO #2 is selected

SBEET (03) OF AE-26156/9
ECR NO. /876 18 June 1998

TEST DATA SHEET NO. 18 Test Point Interface Test (GSE Modes) (Paragraphs 3.3.6.6 - 3.3.6.11)

		 	1	MODES	· · · · ·		_
	1	2	3	4	5	7	
		MOD	E OBSE	RVED? (Y	ES/NO)	3 .	
·							
		DATA	REVIE	WED? (Y	ES/NO)		
Printout data							
Packet ID			_				1
Packet Length							1
Unit Serial Number					1		
Instrument Mode/Status							Ì
Reflector Positions							
Radiometer Scene Data							
PRT Temperature Data							
Engineering Data							
AMSU-A1 System P/N 1356008 Test: 1 st CPT Final CPT	Shop Or	rder:		S/N:			
			Ĩ	est System	s Engineer	-	Date
			5	uality Con	trol		Date

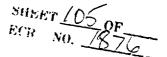
TEST DATA SHEET NO. 19

Radiometer Functional Performance Test (PLO Frequency Measurements) (Paragraph 3.3.7.1)

	PLO FREQUE	NCY MEASUREMENTS	
PLO	Measured Frequency (GHz)	Required Frequency (GHz)	Pass/Fail
#1		57.290294 - 57.290394	
# 2		57.290294 - 57.290394	

P = Pass F = Fail

EOS/AMSU Circle Test:	Al System	P/N 1356008 Final CPT	Shop Order:	_ S/N:	
				Test Systems Engineer	Date
Customer Repr	esentative	Date	· · · · · · · · · · · · · · · · · · ·	Quality Control	Date



TEST DATA SHEET NO. 20 (Sheet 1 of 2) Radiometer Functional Performance Test (Relative NEAT Measurements*) (Paragraph 3.3.7.2) PLO #1 Turned On

RELA	TIVE NEAT MEASUR	EMENTS (PLO #1 A	CTIVE)
Channel	Average NEAT	Required**	
Number	for 5 Data Sets	NEAT	Pass/Fail
	(K)	(K)	
3		0.40	
4		0.25	
5		0.25	
6		0.25	
7		0.25	
8		0.25	
9		0.25	
10		0.40	
11		0.40	
12		0.60	·
13		0.80	
14		1.20	
15		0.50	

P = Pass F = Fail

EOS/AMSU-A1 System P/N 135 Circle Test: 1st CPT Final	S56008 Shop Order:	S/N: _ LPT	•
Customer Representative D		Test Systems Engineer	Date
Customer Representative L	Date	Quality Control	Date

^{*} Baseline data for acceptance tests. Use 1st CPT data along with specification value for pass/fail criteria.

^{**} For reference only

TEST DATA SHEET NO. 20 (Sheet 2 of 2)

Radiometer Functional Performance Test (Relative NEAT Measurements*) (Paragraph 3.3.7.2) PLO #2 Turned On

RELA	ATIVE NEAT MEASU	REMENTS (PLO #2	ACTIVE)
Channel	Average NEAT	Required**	
Number	for 5 Data Sets	NEΔT	Pass/Fail
<u> </u>	(K)	(K)	
3		0.40	()
4	1000	0.25	1: able
<u> </u>	1 plice	8,25	Apple
6	1/2/1/2	0.23	10,1
7	N	0.25	7
8		0.25	
9		0.25	
10		0.40	
11		0.40	
12		0.60	
13		0.80	
14	MICOR	1.20	
= Pass F = Fail	NOT BEEN CABL	0.50	NET APPLICATE TO

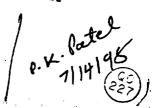
EOS/AMSU-A1 System P/N 1356008 Circle Test: 1st CPT Final CPT	Shop Order:Sub CPT	S/N: LPT	
		Test Systems Engineer	Date
Customer Representative Date		Quality Control	Date

^{*} Baseline data for acceptance tests. Use 1st CPT data along with specification value for pass/fail criteria.

^{**} For reference only

06/18/98

TEST DATA SHEET 2 **Channel Identification Test** (para 3.3.8)



	Channel Number	Antenna Location	Sweeper Freq. Setting (GHz)	Polarization (H/V)	Radiometric Data Counts A Counts	Channel Verified
-	3	A1-2	50.35	V	+ totale	
-					1000	YES
F	4	A1-2	52.85	V	9162	2157
-	5				11000	YES
-	3	A1-2	53.70	H	15635	MEC
\vdash	6				13000	YES
-		A1-1	54.45	Н	15681	HES
\vdash	7					763
 -		A1-1	54.99	V	14598	YES
	8	A1-2				103
-		A1-2	55.55	H	15873	465
-	9	A1-1	- 55.0.1			100
-		A1-1	57.34	H	76056	YES
\vdash	10	A1-1	57.50			1
		Al-i	57.50	Н	15953	YES
	11	A1-1	57.564			102
			37.364	Н	15903	WES
_	12	A1-1	57.59			
			31.39	H	15557	455
	13	A1-1	57.602			
			57.002	Н	1564B -	485
	14	A1-1	57.608			
			27.000	H	15237	485
	15	A1-1	89.55	v		
				<u>v</u>	16257 1	15-5

GOS/AMSU-AI System P/N 1356008 shor order 7985 Final CPT Sub CPT

Rep. Date

ChIDTest.doc

DOCUMENT APPROVAL SHEET



Process Specification EOS/AMSU-A1 System Comprehensive and Limited Performance Tests Test Procedure	AE-26156/ 18 June 19	9
P. Patel DATE CDRL: SPECIFICATION ENGINEER:	<u>L </u>	DATE
CHECKED BY: DATE JOB NUMBER:		DATE
74 TROVED SIGNATURES	DEPT. NO.	DATE
Specifications Engineering (J. Kirk) Ry Hauerwas 9110 Product Team Leader (A. Nieto) P. R. Patel Systems Engineer (R. Platt) Raff Safety (W. Neighbors) For Theholour for Design Assurance (E. Lorenz) Alack Solicions Quality Assurance (R. Taylor) RM Tuylon Technical Director/PMO (R. Hauerwaas) RV Hauerwass Configuration Management (J. Cavanaugh)	8631 8341 8311 8331 8331 7831 4001 8361	6/30/98 6/30/98 6/30/98 6/30/98 6/30/98 6-30-98 6/39/98 7/1/98
P. Patel DATE CDRL: 409 SPECIFICATION ENGINEER: P. Patel DATE JOB NUMBER: DEFECKED BY: D		
By my signature, I certify the above document has been reviewed by me and concurs with the technical requirements related to my area of responsibility. RELEASE (Data Center) FINAL alla Marks Jurida 15-66-98		•

APPENDIX A

TEST DATA SHEETS

10. APPENDIX A

10.1 Scope. This appendix contains the test data sheets for all tests and inspections listed in section 3.

TDS		Page
1	Grounding Interface Test	A-2
2	Oniet Power Bus Operational Power Test	
3	Oujet Power Bus Operational Power Test (LPT)	
4	Quiet Power Bus Turn On Transient Test	A-10
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6	Noisy Power Bus Turn On Transient Test	A-12
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11	Science and Engineering Data Test (Cold Cal Mode)	A-20
12	Science and Engineering Data Test (Nadir Mode)	
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14	Test Point Interface Test (8 Second Sync Pulse TP)	A-23
15	Test Point Interface Test (Integrate/Hold and Dump TPs)	A-3t
16	Test Point Interface Test (Radiometer Channel Analog Output TPs)	
17	Test Point Interface Test (PLO #1 and PLO #2 Lock TPs)	A-34
18	Test Point Interface Test (GSE Modes)	A-33
19	Radiometer Functional Performance Test (PLO Frequency Measurements)	
20	Radiometer Functional Performance Test (Relative NEAT Measurements)	A-35

TEST DATA SHEET NO. 1 (Sheet 1 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

From Chassis	Pin Description	acecraft Interface Required Resistance	Measured Value	
Ground to		(Ohms)	(Ohms)	Pass/Fai
J1-1	+29V QUIET PWR BUS	> 1M	> I MEG	P
J1-2	+29V QUIET PWR BUS	> 1M	1	1
J1-3	29V QUIET BUS RTN	> 1M		
J1-4	29V QUIET BUS RTN	> 1M		
J1-5	+29V NOISY PWR BUS	> 1M		
J1-6	+29V NOISY PWR BUS	> 1M		
J1-7	29V NOISY BUS RTN	> 1M		
J1-8	29V NOISY BUS RTN	> 1M		
J1-9	SURVIVAL PWR BUS A	> 1M		
J1-10	SURVIVAL BUS A RTN	> 1M		
J1-11	SURVIVAL PWR BUS A	> 1M		
J1-12	SURVIVAL BUS A RTN	> 1M		
J1-13	CHASSIS GROUND	< 1	1350	
J1-14	+29V QUIET PWR BUS	> 1M	.135sc >1mc6	
J1-15	+29V QUIET PWR BUS	> 1M	1	
J1-16	29V QUIET BUS RTN	> 1M		
J1-17	29V QUIET BUS RTN	> 1M		
J1-18	+29V NOISY PWR BUS	> 1M		
J1-19	+29V NOISY PWR BUS	> 1M		
J1-20	29V NOISY BUS RTN	> 1M		
J1-21	29V NOISY BUS RTN	> 1M		
J1-22	SURVIVAL PWR BUS B	> 1M		
J1-23	SURVIVAL BUS B RTN	> 1M		
J1-24	SURVIVAL PWR BUS B	> 1M		
J1-25	SURVIVAL BUS B RTN	> 1M	1	$-\pm$

EOS/AMSU-A1 System P/N 1356008 Circle Test: (1st CPT) Final CPT	Shop Order: 298561 S/N: 202 Sub CPT LPT /	,
Cystomer Representative Date	Test Systems Engineer Quality Control	7/13/38 1 Date 10N 1 4 1898

TEST DATA SHEET NO. 1 (Sheet 2 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

		ecraft Interface	· · · · · · · · · · · · · · · · · · ·	
From Chassis	Pin Description	Required Resistance	Measured Value	
Ground to		(Ohms)	(Ohms)	Pass/Fai
J2-1	A1-1 MOTOR TEMP HI	> 1M	TIMEG	<u> </u>
J2-2	A1-1 MOTOR TEMP LO	> 1M		
J2-3	A1-1 RECEIVER TEMP 1 HI	> 1M		
J2-4	A1-1 RECEIVER TEMP 1 LO	> 1M		
J2-5	A1-1 WARM LOAD TEMP HI	> 1M		
J2-6	A1-1 WARM LOAD TEMP LO	> 1M		
J2-7	A1-2 MOTOR TEMP HI	> 1M		
J2-8	A1-2 MOTOR TEMP LO	> 1M		
J2-9	A1-2 RECEIVER TEMP 1 HI	> 1M		
J2-10	A1-2 RECEIVER TEMP 1 LO	> 1M		
J2-11	A1-2 WARM LOAD TEMP HI	> 1M		
J2-12	A1-2 WARM LOAD TEMP LO	> 1M		
J2-13	No Connection	> 1M		
J2-14	No Connection	> 1M		
J2-15	No Connection	> 1M		
J2-16	No Connection	> 1M		
J2-17	No Connection	> 1M		
J2-18	No Connection	> 1M		
J2-19	No Connection	> 1M		
J2-20	No Connection	> 1M		
J2-21	No Connection	> 1M		
J2-22	A1-1 RECEIVER TEMP 2 HI	> 1M		
J2-23	A1-1 RECEIVER TEMP 2 LO	> 1M		
J2-24	No Connection	> 1M		
J2-25	No Connection	> 1M		
J2-26	No Connection	> 1M		
J2-27	No Connection	> 1M		
J2-28	A1-2 RECEIVER TEMP 2 HI	> 1M		
J2-29	A1-2 RECEIVER TEMP 2 LO	> 1M		
J2-30	No Connection	> 1M		
J2-31	No Connection	> 1M		
J2-32	No Connection	> 1M		
J2-33	No Connection	> 1M		
J2-34	No Connection	> 1M		
J2-35	No Connection	> 1M		
J2-36	No Connection	> 1M		
J2-37	No Connection	> 1M	4	7

TEST DATA SHEET NO. 1 (Sheet 3 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

	J3 of Spacec	raft Interface	_	
From Chassis Ground to		Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fai
J3-1	1553 INTERFACE DATA A HI	> 100K	>100K	PKSS
J3-2	1553 INTERFACE DATA A LO	> 100K	>10014	1 1
J3-3	No Connection	> 1M	>IM	
J3-4	1553 INTERFACE DATA B LO	> 100K	>100K	
J3-5	1553 INTERFACE DATA B HI	> 100K	71001	
J3-6	1553 INTERFACE DATA A SHIELD	<1	0.25	
J3-7	No Connection	> 1M	> IH	
J3-8	No Connection	> 1M	ZIH	1
J3-9	1553 INTERFACE DATA B SHIELD	<1	0.76	PKSS

EOS/AMSU-Al System P/N 1356008 Shop Order: 29856 S/N: 202
Circle Test: (1st CPT) Final CPT Sub CPT LPT

Test Systems Engineer Date

Customer Representative Date

Quality Control Date

TEST DATA SHEET NO. 1 (Sheet 4 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

		acecraft Interface	1	<u> </u>	┨
From Chassis	Pin Description	Required Resistance	Measured Value	Dans/Fail	ı
Ground to		(Ohms)	(Ohms)	Pass/Fail	┨
J4-1	CHASSIS GROUND	1/ 620_1 . ~ 1	11852	Tass	┨
J4-2	8 SECOND SYNC PULSE TP	1 2 10011	7/00K		┨
J4-3	PLO #2 LOCK TP	>201 >100K-1114	22K		1
J4-4	PLO LOCK RTN (7/8)	<1 Athles.	7 +04sc		42
J4-5	I/H & DUMP RTN (2/3)	<1 7/13/18	,29/01/		₹,
J4-6	DUMP COMMAND TP	> 100K	449W70	k	┨′
J4-7	No Connection	> 1M	>1mEG		4
J 4-8	CH 3 ANALOG OUT TP	> 100K	7/00K		4
J 4-9	CH 4 ANALOG OUT TP	> 100K			4
J4-10	CH 5 ANALOG OUT TP	> 100K	l		1
J4-11	CH 6 ANALOG OUT TP	> 100K			1
J4-12	CH 7 ANALOG OUT TP	> 100K			1
J4-13	CH 8 ANALOG OUT TP	> 100K			_
J4-14	CH 9 ANALOG OUT TP	> 100K	V		1
J4-15	No Connection	> 1M	>/mEG		
J4-16	No Connection	> 1M	>/mEG		
J4-17	GSE COMMAND LSB	> 5K	フグド		
J4-18	GSE COMMAND MSB-1	> 5K	75K		
J4-19	No Connection	> 1M	>/mEG		7
J4-20	1.248 MHz CLOCK TP	> 100K (227)	>100 K		7
J4-21	1.248 MHz CLOCK RTN (1)	<1	.328.1		1
J4-22	PLO #1 LOCK TP	>20K > 100K	22 K		1
J4-23	No Connection	> 1M	DIMEG		1
J4-24	I/H COMMAND TP	> 100K	7/00K		1
J4-25	No Connection	> 1M	>/mEG		7
J4-26	ANALOG OUT RTN (2/3)	<1	.280,2		7
J4-27	CH 10 ANALOG OUT TP	> 100K	7/00 K		7
J4-28	CH 11 ANALOG OUT TP	> 100K			7
J4-29	CH 12 ANALOG OUT TP	> 100K			7
J4-30	CH 13 ANALOG OUT TP	> 100K			1
J4-31	CH 14 ANALOG OUT TP	> 100K			1
J4-32	CH 15 ANALOG OUT TP	> 100K	₩		7
J4-32 J4-33	No Connection	> 1M	>/mEG		1
J4-34	No Connection	> 1M	>imeg		1
J4-34 J4-35	GSE COMMAND MSB	> 5K	>5K		1
J4-36	GSE COMMAND RTN (1)	<1	,3/2		1
J4-30 J4-37	No Connection	> 1M	TIMEG	Pass	7
	System P/N 1356008 Shop Ord		202	1/13/38	ـ ۲
10	1 D 1-22-90	Test Sys	tems Engineer	Date	
4-X-2-4	entative Date	Quality (Date	_

TEST DATA SHEET NO. 1 (Sheet 5 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

Source	Destination	Source Pin Description	Required Resistance (Ohms)	Measured Value (Ohms)	Pass/Fan
J1-1	J1-2	+29V QUIET PWR BUS	<1	.2572	1 233 7 2/1
J1-1	J1-14	+29V QUIET PWR BUS	<1	.2682	
J1-1	J1-15	+29V QUIET PWR BUS	<1	,2612	
J1-3	J1-4	29V QUIET BUS RTN	<1	,259.2	
J1-3	J1-16	29V QUIET BUS RTN	<1	,258-2	 - -
J1-3	J1-17	29V QUIET BUS RTN	<1	1256-2	
J1-5	J1-6	+29V NOISY PWR BUS	<1	.247s	
J1-5	J1-18	+29V NOISY PWR BUS	<1	.2692	
J1-5	J1-19	+29V NOISY PWR BUS	<1	.268_re	
J1-7	J1-8	29V NOISY BUS RTN	<1	,240sc	
J1-7	J1-20	29V NOISY BUS RTN	<1	.260R	
J1-7	J1-21	29V NOISY BUS RTN	<1	,259_R	
J1-9	J1-11	SURVIVAL PWR BUS A	<1	.226 /2	
J1-10	J1-12	SURVIVAL BUS A RTN	<1	12232	
J1-22	J1-24	SURVIVAL PWR BUS B	<1	122802	
J1-23	J1-25	SURVIVAL BUS B RTN	<1	12282	
J1-1	J1-5	+29V QUIET PWR BUS	> 1M	7/MEQ	
J1-1	J1-7	+29V QUIET PWR BUS	> 1M	TIMEG	
J1-1	J1-9	+29V QUIET PWR BUS	> 1M		
J1-1	J1-10	+29V QUIET PWR BUS	> 1M	7/mEG	
J1-1		+29V QUIET PWR BUS		7/mE6	
J1-1		+29V QUIET PWR BUS	> 1M	7/mEG	
J1-3		29V QUIET BUS RTN	> 1M	7/meg	
J1-3		29V QUIET BUS RTN	> 1M	7/mea	
J1-3		29V QUIET BUS RTN	· · · · · · · · · · · · · · · · · · ·	71 MEG	

EOS/AMSU-A1 System P/N 1356008 Shop Order: 29856 S/N: 207
Circle Test: (1st CPT) Final CPT Sub CPT N/A LPT N/A

Test Systems Engineer Date

Costomer Representative Date

Output

Outp

TEST DATA SHEET NO. 1 (Sheet 6 of 6) Grounding Interface Test (Paragraph 3.3.2, Step 2)

	<u> </u>		Required	Measured Value	
Source	Destination	Source Pin Description	Resistance (Ohms)	(Ohms)	Pass/Faul
J1-3	J1-10	29V QUIET BUS RTN	> 1M	>/mEG	1
J1-3	J1-22	29V QUIET BUS RTN	> 1M	7/mEG	
J1-3	J1-23	29V QUIET BUS RTN	> 1M	>1 mEG	
J1-5	J1-9	+29V NOISY PWR BUS	> 1M	7/mEG	
J1-5	J1-10	+29V NOISY PWR BUS	> 1M	7/m=6	
J1-5	J1-22	+29V NOISY PWR BUS	> 1M	>1m=G	
J1-5	J1-23	+29V NOISY PWR BUS	> 1M	>1mEG	
J1-7	J1-9	29V NOISY BUS RTN	> 1M	>1MEG	
J1-7	J1-10	29V NOISY BUS RTN	> 1M	7/MEG.	
J1-7	J1-22	29V NOISY BUS RTN	> 1M	7/mEG	
J1-7	J1-23	29V NOISY BUS RTN	> 1M	SIMEG	
J1-9	J1-22	SURVIVAL PWR BUS A	> 1M	7/MEG	
J1-9	J1-23	SURVIVAL PWR BUS A	> 1M	>1MEG	
J1-10	J1-22	SURVIVAL BUS A RTN	> 1M	>1mEG	
J1-10	J1-23	SURVIVAL BUS A RTN	> 1M	7/MEG	
J1-13	JI OUTER	CHASSIS GROUND	< 1	/3/2	1
	SHELL			./3/52	ļ <u>i</u>
J1-13	J2 OUTER	CHASSIS GROUND	<1 .	/30 .	
	SHELL			-1301	-
J1-13		CHASSIS GROUND	< 1	.13/2	
	SHELL			. 19102	 -
J 1-13	I	CHASSIS GROUND	< 1	1130n	
	SHELL		10077		
J3-1	J3-5	1553 INTERFACE DATA A HI	> 100K	>/mEG	
J 3-1	J3-4	1553 INTERFACE DATA A HI	> 100K	DIMEG	
J3-2	J3-5	1553 INTERFACE DATA A LO	> 100K	ZIMEC	
J3-2	J3-4	1553 INTERFACE DATA A LO	> 100K	> IMEG.	

EOS/AMSU-A1 System P/N 1356008 Shop Order: 29856 S/N: 202
Circle Test: (1st CPT) Final CPT Sub CPT //W LPT //W LPT //W Date

Test Systems Engineer Date

Quality Control Date

TEST DATA SHEET NO. 2 Quiet Power Bus Operational Power Test (Paragraph 3.3.3.1.1)

Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	PLO	Maximum Peak Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Peak Power (QBV x QBI) (Watts)	Pass/Fail
26.95 - 27.05	27.03V	#1	2.5848 A.	≤94	69.86 MAG	P
28.95 - 29.05	29.01V	#1	2.388 A	≤94	69.27 Watt	P
30.95 - 31.05		#1	2.229 A	≤94	69.12 abts	ρ
26.95 - 27.05	27.03 V	#2	2.586 A	≤94	69.89 abls	A
28.95 - 29.05	29.09 V	#2	2.40 A	<u>≤</u> 94	69.81 White	P
30.95 - 31.05	31.044	#2	2.2486 A	≤94	69079 Walts	P

	Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	PLO	Average Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Average Power (QBV x QBI) (Watts)	Pass/Fail
	26.95 - 27.05	27.03V	#1	2.555 A	58886	69.06 Watter	ρ
	28.95 - 29.05	29.01V	#1	2.3855 A	58886	69.20 UST	P
	30.95 - 31.05	31.01 V	#1	2.218 A	≤88′86	68.78 White	P
łL	26.95 - 27.05	27.03 V	#2	2.612 A	≤8886	70.6 Watts	P
3	28.95 - 29.95	29.09 V	#2	2.407 A	≤88/86	70.01 white	P
L	30.95 - 31.05	31.04 V	#2	2.218 A	<8886	67.66 Watts	A

20 7/14/78 227 //14/78

EOS/AMSU-AL System P/N 1356008 Circle Test: 1st CPT Final CPT Shop Order: <u>298561</u> Sub CPT <u>MA</u>

S/N: <u>202</u>

Test Systems Engine

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Date

Quality Control

Date

QC 227

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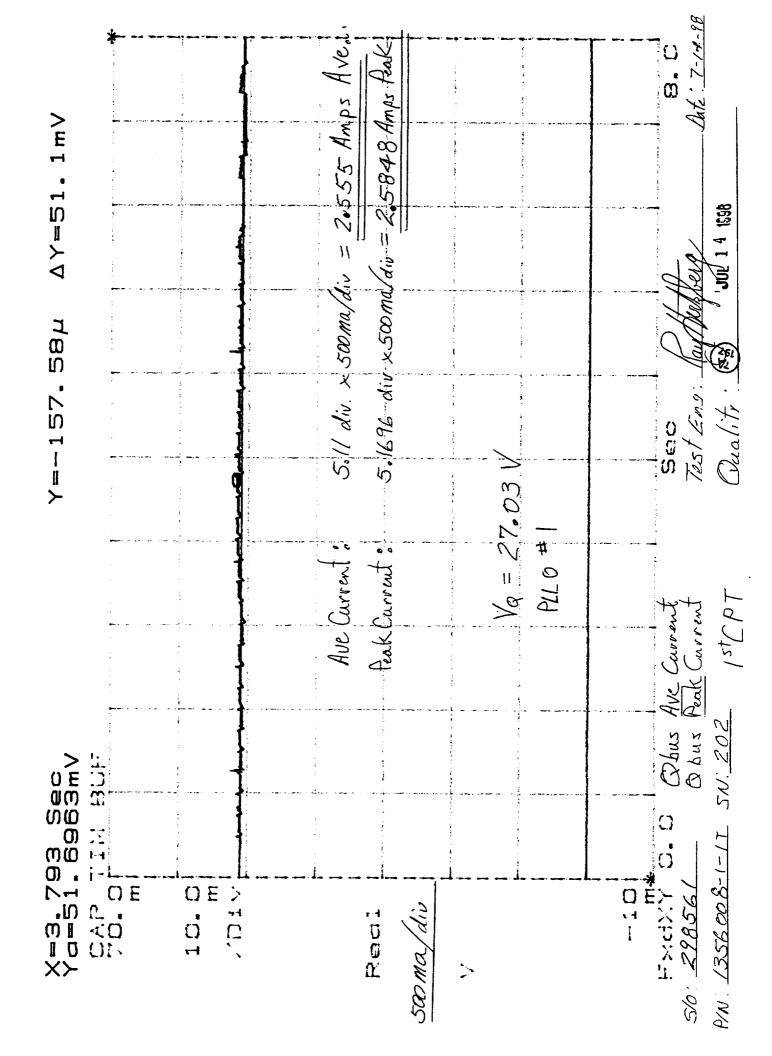
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				500				56: 52 P.N. 1.

ΔY=44.36mV			2.218 AMP AVE. 1.				C	JUL 1 4 1998
$\forall =-157.58\mu$			Ave Current: 4.436 div × 500 ma/div = Peak Current: 4.4972 div × 500 ma/div =		Va = 31.04 V	P(L0 #2		Whas Ave Current Test Eng' lay the
X=3.793 Sec Yo=44.972EV	0.0 0.0 0.0 0.0	DE >	£ 55 € £	Seo malair				SN: 20

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Ave Current ? Peak Current o	4.436 div × 500 ma/div = 4.4589 div × 500 ma/div =	(div = 2,218 Amps Ave div = 2,229 Amps Pak	125 Ave.
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TEST DATA SHEET NO. 3 Quiet Power Bus Operational Power Test (LPT) (Paragraph 3.3.3.1.2)

Required Quiet Bus Voltage QBV (Volts)	Measured QBV (Volts)	Average Quiet Bus Current QBI (Amps)	Required Power (Watts)	Calculated Average Power (QBV x QBI) (Watts)	Pass/Fail
28.95 - 29.05	29.0	2.27	≥88	65.83	4

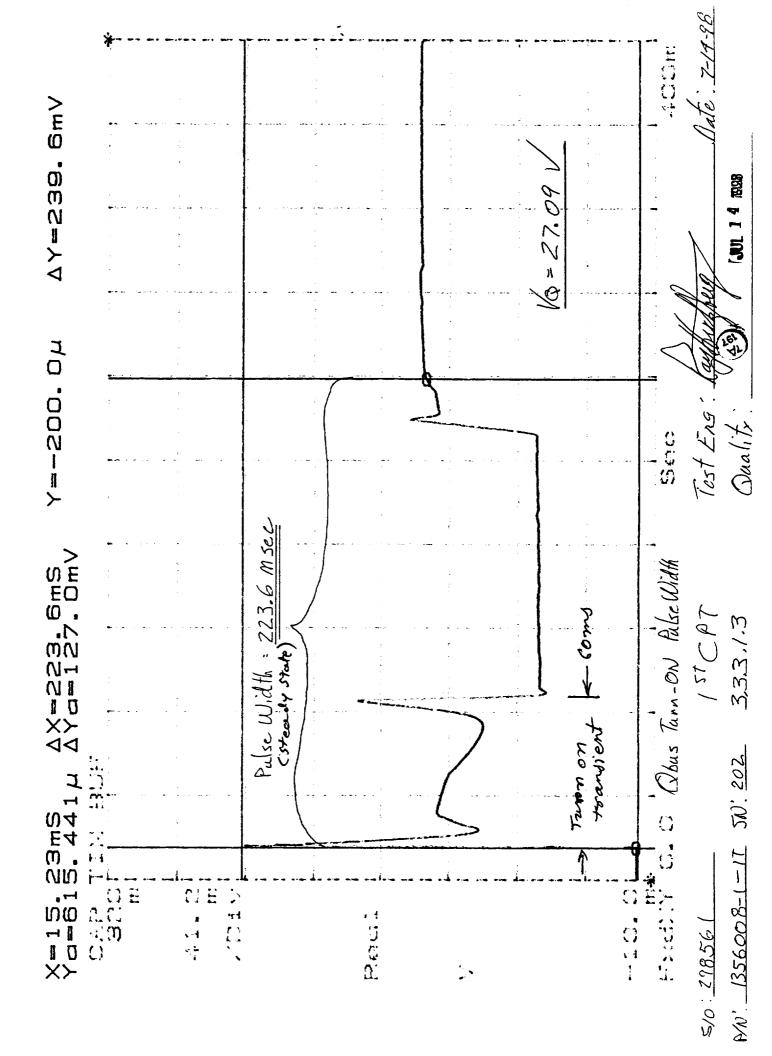
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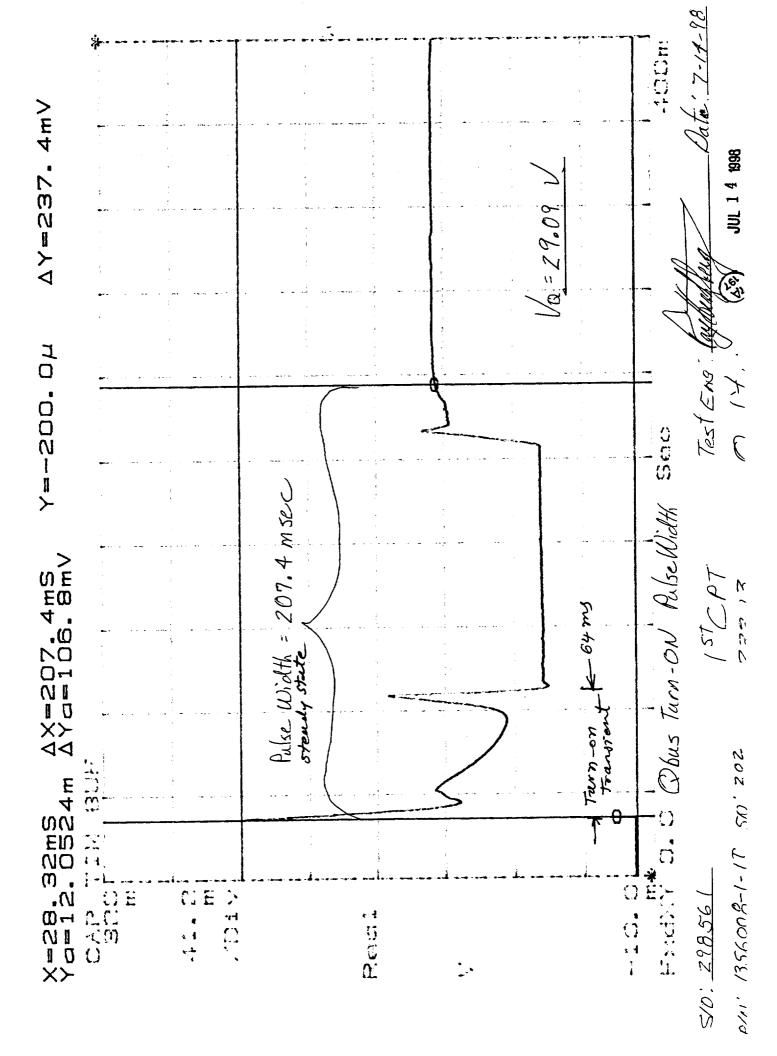
TEST DATA SHEET NO. 4 Quiet Power Bus Turn On Transient Test (Paragraph 3.3.3.1.3)

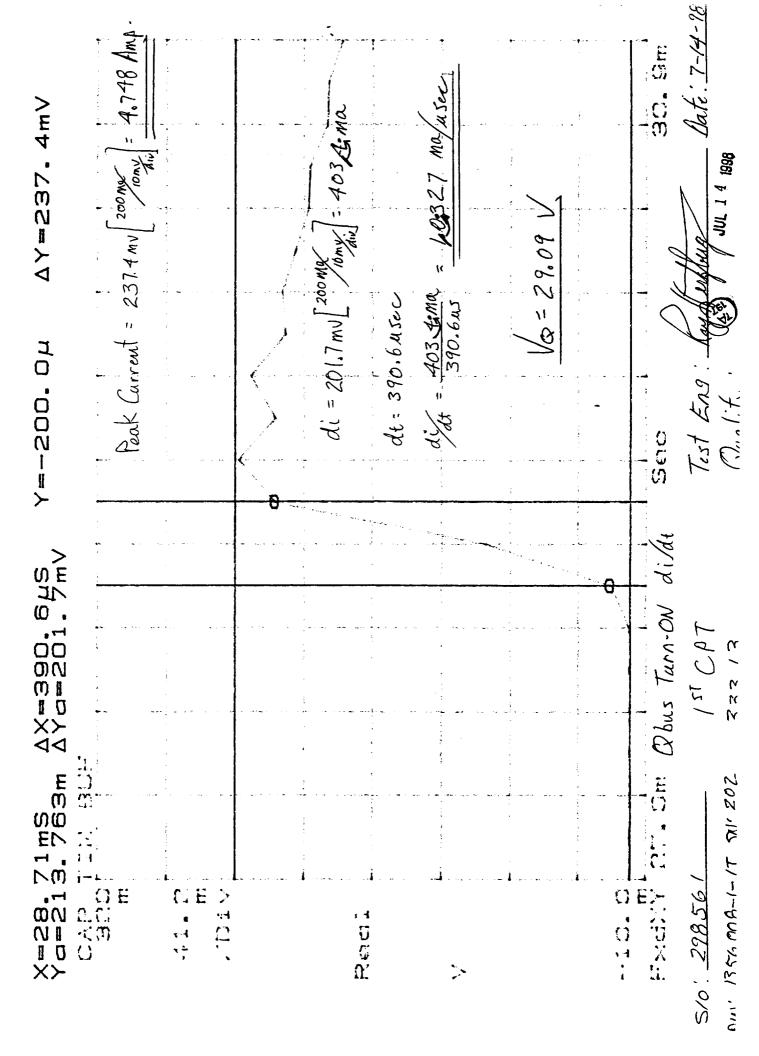
+31 Volts Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	10 1 5.194 Am	ups <10.6 Amps	P
Pulse Width (Steady S		£ 250 m 5	P
Rate of Change(slope): dl/d	T LZ347 8.9%ma/	μs <677 mA/μs	P
Palse Width (Trans	ient) (4 mb = 64 and.	/100 ms.	P
+29 Volts	All Math 1-12-Pot of		
Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	4,748 Am	•	P
Pulse Width (Steady	State 207.4	< 250 ms ≤150 ms	P
Rate of Change(slope): dI/d		us <677 mA/μs	P
False with (Fran	ROS Ket DAZ Putel	<100 ms	P
+27 Volts	Rafflet p. 12. Pertel		
Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	4.792 Amp	os <10.6 Amps	P
Pulse Width (Stocoly St	sefe) 223.6 m	<250 ms <150 ms	P
Rate of Change(slope): dI/d	1 12110	<677 mA/μs	P
Prelse Width (Tran	sing 60ms.	100ms	P /
EOS/AMSUL-AI PIN: <u>135</u> EVO: <u>298561</u> Enrele Test (1 ^{SI} CPT) FII	6008-1-1T SN:_	202 LPT	-1.46
1 Sanfacol	7-22-98	Systems Engine	T/14/90 CEP PATE

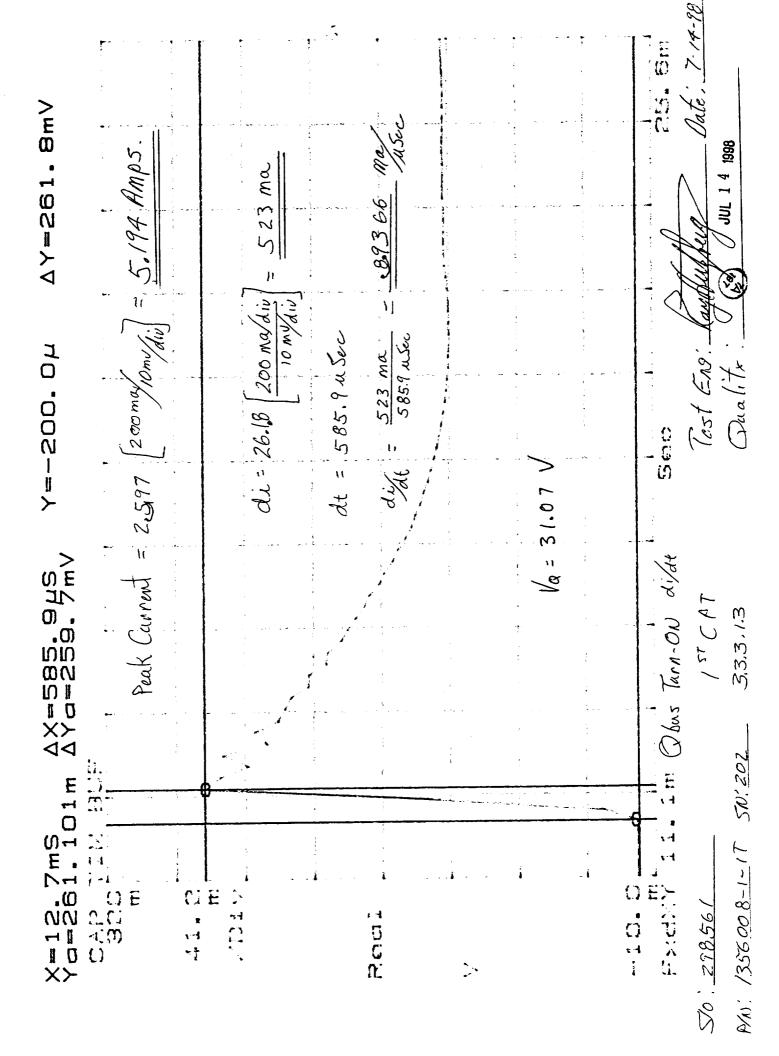
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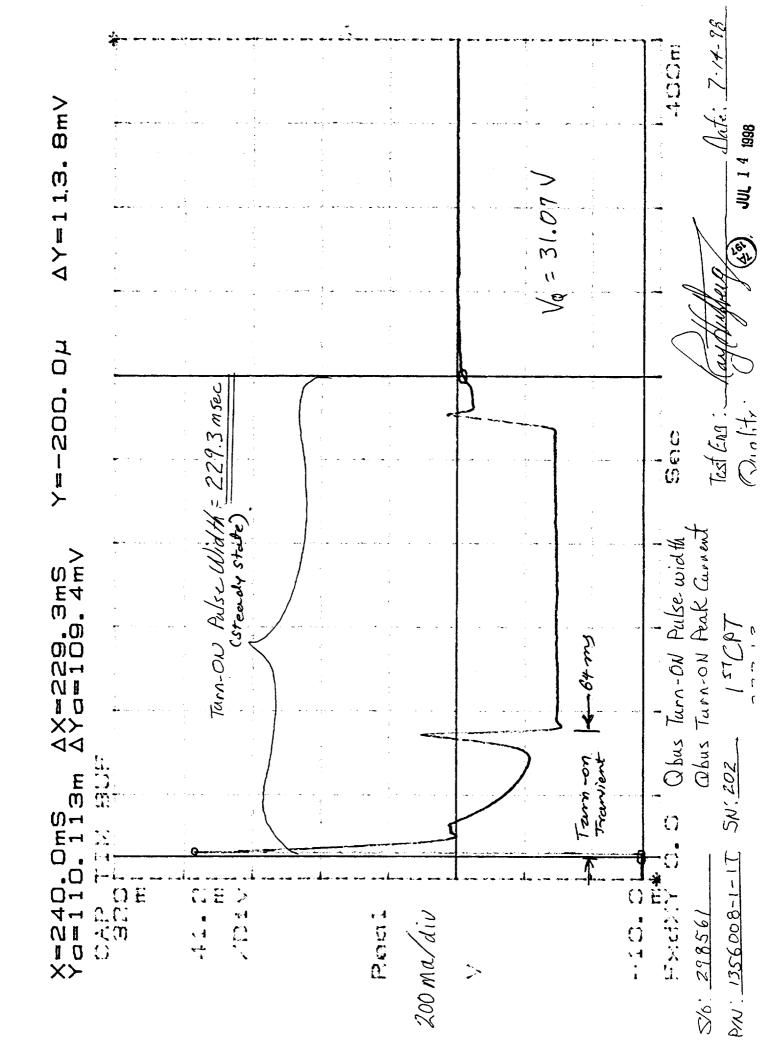
۵a <	4.792 Amps.							18.9m 0.4.7-14.99	<i>Jaile</i> . ————————————————————————————————————
∆Y=239.6	my air		2496 ma		2.78 mayusec			1.18.0	JUL 1 4 1998
Y=-200. 0μ	Current = 239.6 mv [200 mg		di = 124.8 may [200 ma/		2496 Ma = 1			Sec	(05) Cho . 100 pt 100 Chall . 120
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TEST DATA SHEET NO. 5 Noisy Power Bus Operational Power Test (Paragraph 3.3.3.2.1)

Required Noisy Bus Voltage NBV (Volts)	Measured NBV (Volts)	Required Peak Current (Amps)	Maximum Peak Noisy Bus Current NBI (Amps)	Required Peak Power (Watts)	Calculated Peak Power (NBV x NBI) (Watts)	Pass/Fail
26.95 - 27.05	27.01	<u>≤</u> 1	.989	<u>≤</u> 40	26.7	?
28.95 - 29.05	2.01	≤1	.984	≤40	28.6	P
30.95 - 31.05	31.01	<u>≤</u> 1	. 983	≤40	30.5	?

Required Noisy Bus Voltage NBV (Volts)	Measured NBV (Volts)	Average Noisy Bus Current NBI (Amps) SEC	Required Average Power (Watts)	Calculated Average Power (NBV x NBI) (Watts)	Pass/Fail
26.95 - 27.05	27.01	.119	≤ \$ 8	5,6	P
28.95 - 29.05	29.01	.130	\$8	68	A.
30.95 - 31.05	31.01	.133	≤\$8	7.1	P

Bus Current During Measured NBV Required Pass/Fail the I/H, D. Period (Volts) Noisy Bus Voltage **NBV** (Volts) 30.4 ma 15,32 ma 27.01 Not Applicable 26.95 - 27.05 32 ma \$ Not Applicable 28.95-29.05 29.01 16.66 Max 35.2 ma * 30.95 - 31.05 Not Applicable 31.01 18.1 ma #

* between beams

EOS/AMSU-A1 System P/N 1356008 Circle Test: (1st CPT Final CPT Shop Order: <u>24856</u>/ Sub CPT _____ sn: 202

Test Systems Engineer

Date

omer Representative

Date

Quality Control

Date

TEST DATA SHEET NO. 6

Noisy Power Bus Turn On Transient Test (Paragraph 3.3.3.2.2)

+31	Volt	ÌS
-----	------	----

Parameter	Measured/Calculated	Required	Pass/Fail	
Peak Current	16.2 Amps	<11.5 Amps	F *	
Pulse Width	Oil ms	<100 ms	P	
Rate of Change(slope): dI/dT	/856 ma/μs	<744 mA/μs	FX	

+29 Volts

Parameter	Measured/Calculated	Required	Pass/Fail	
Peak Current	15.2 Amps	<11.5 Amps	F*	
Pulse Width	O, 1 ms	<100 ms	P	
Rate of Change(slope): dI/dT	11 58 ma/µs	<744 mA/μs	FX	

+27 Volts

Parameter	Measured/Calculated	Required	Pass/Fail
Peak Current	14.2 Amps	<11.5 Amps	FX
Pulse Width	O i ms	<100 ms	P
Rate of Change(slope): dI/dT	/7/0 ma/μs	<744 mA/μs	F *

TAR # 3190

EOS/AMSU-AI	P/N: 1356008-1-1T	SN: 202
=1. + 000 t=/ 1		

Circle Test 1st CFT Final CFT Sub CFT N/A

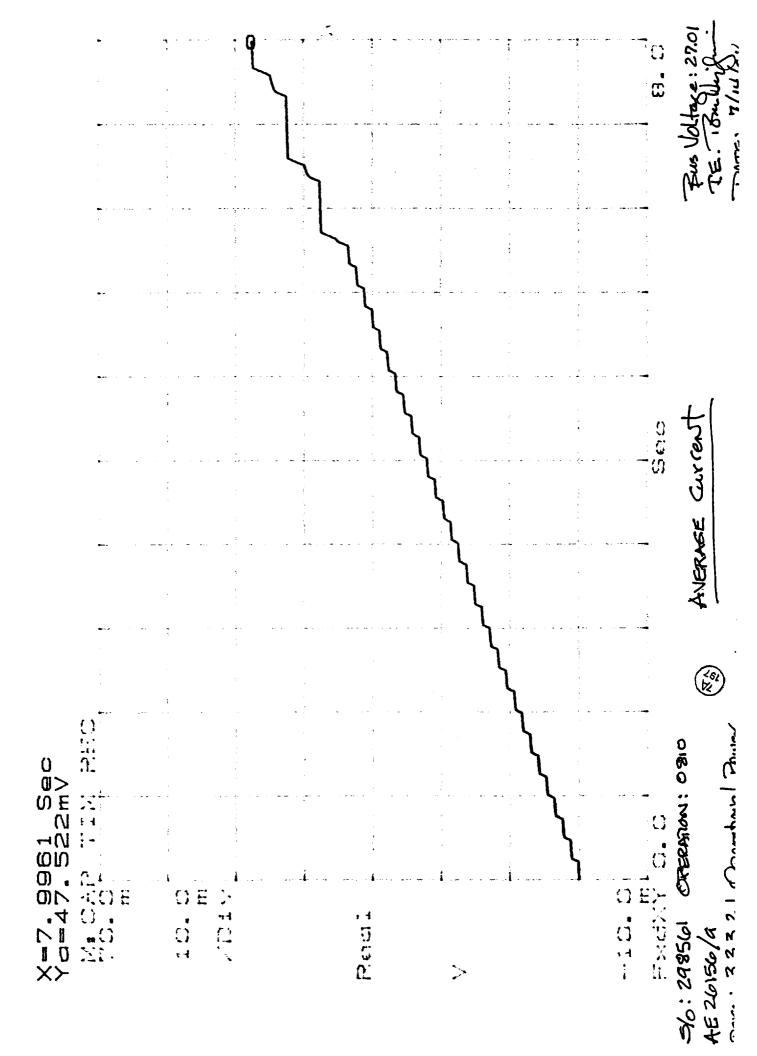
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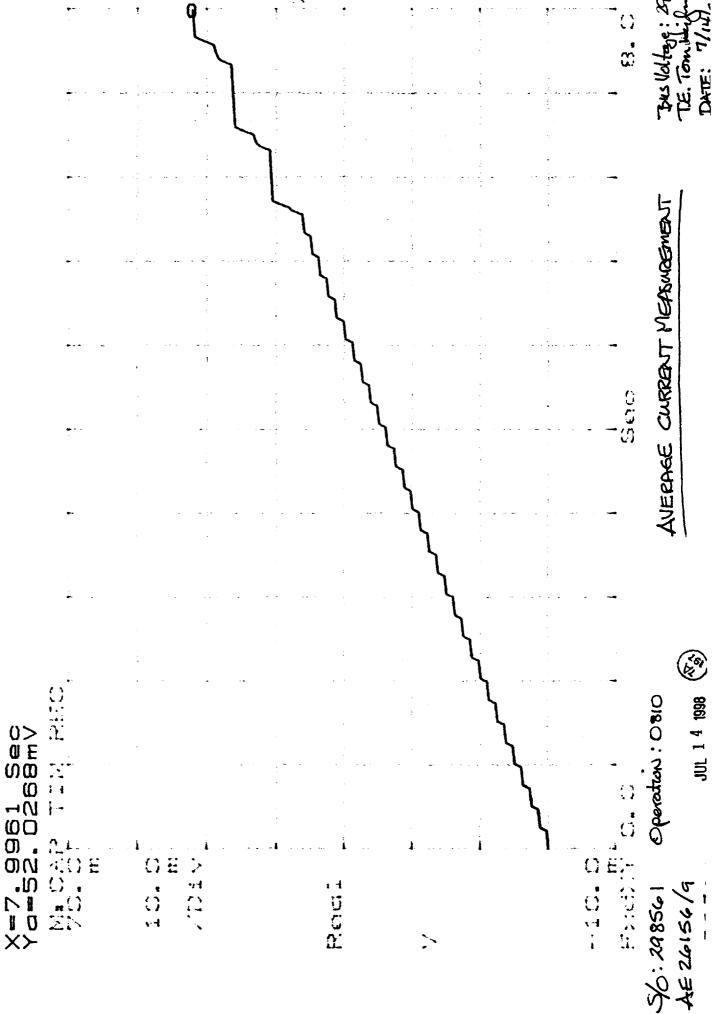
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Quality Control

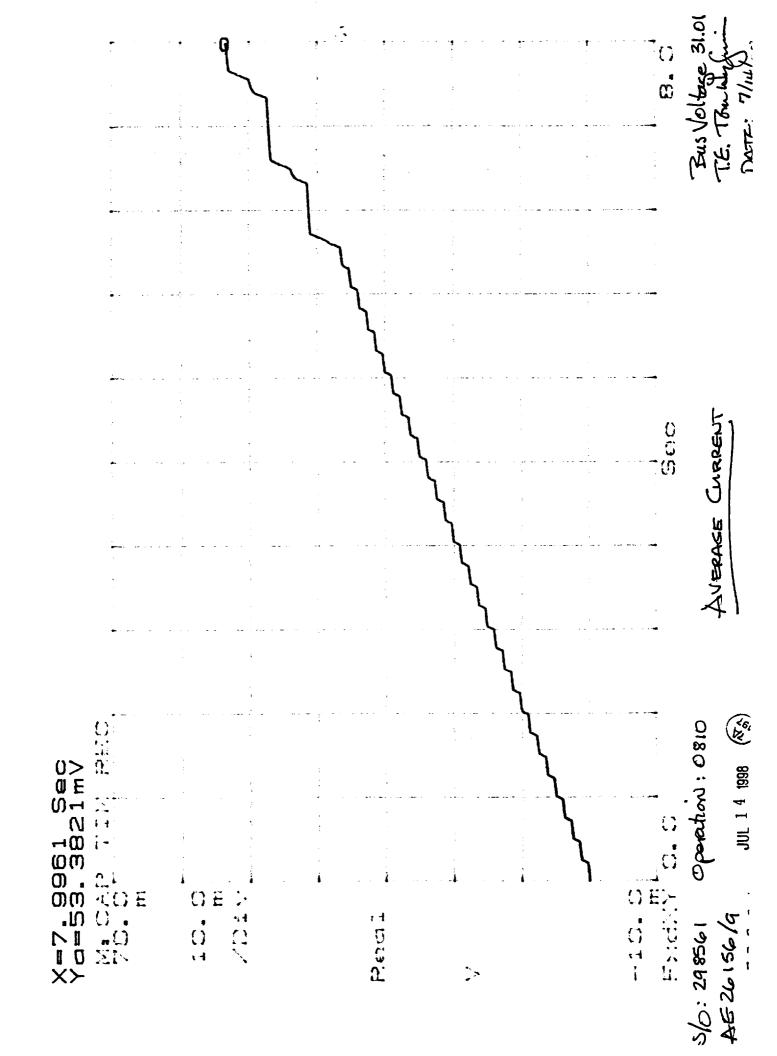
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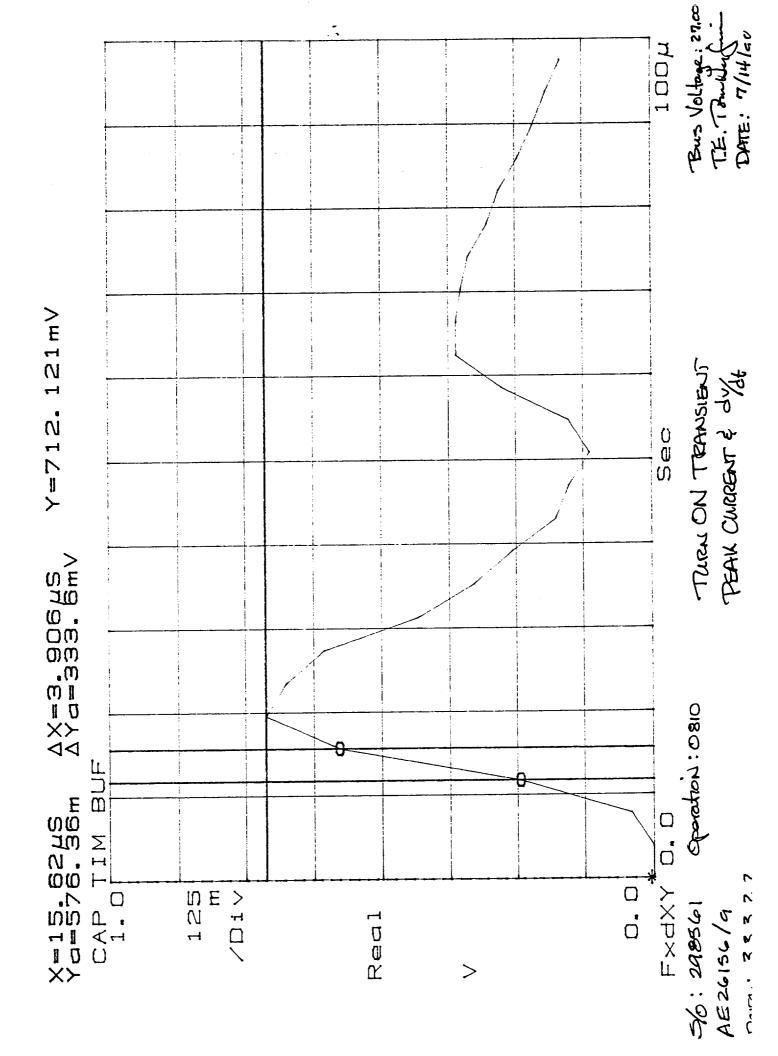
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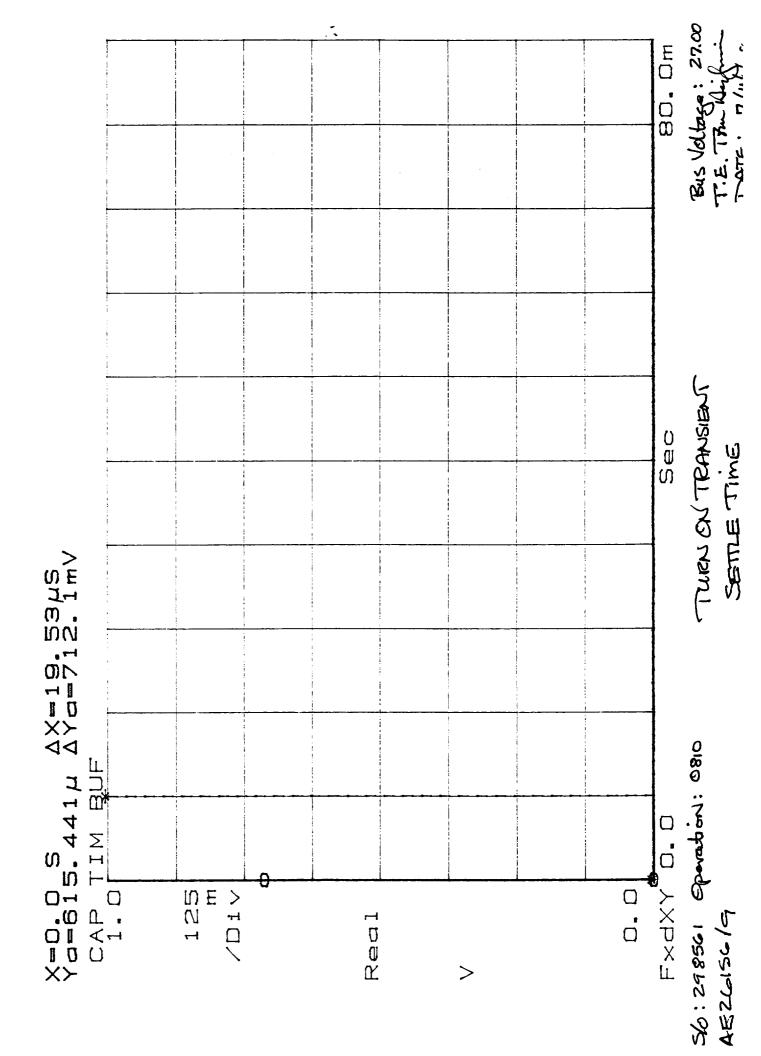


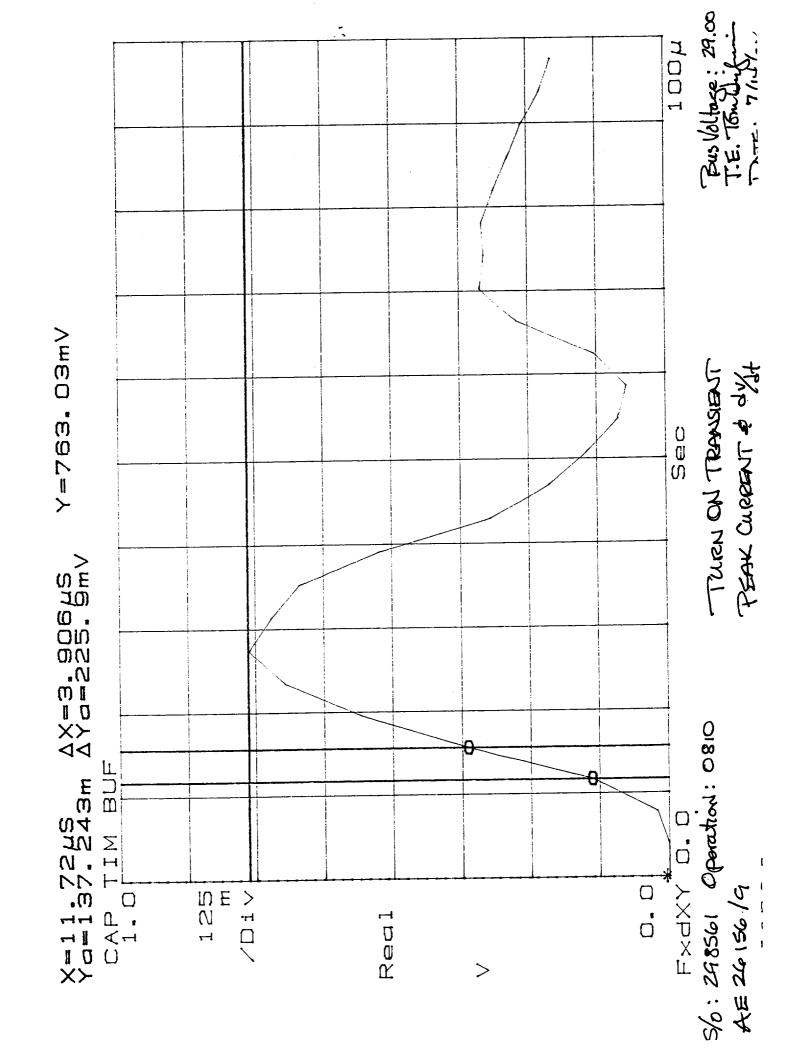


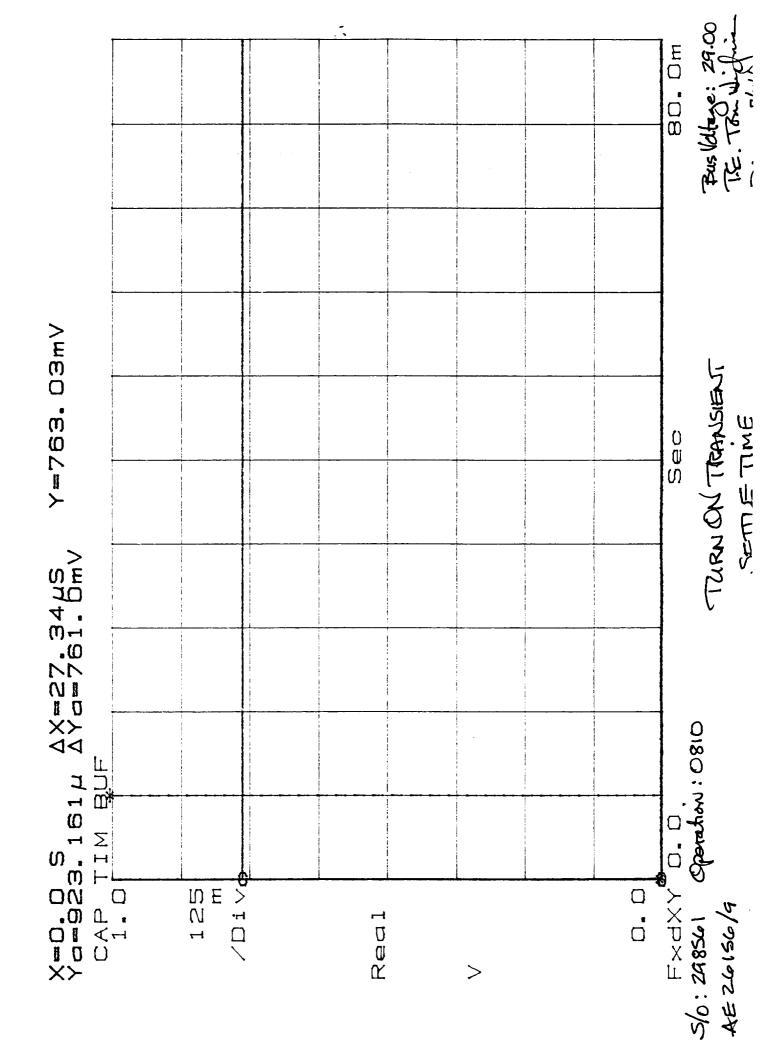
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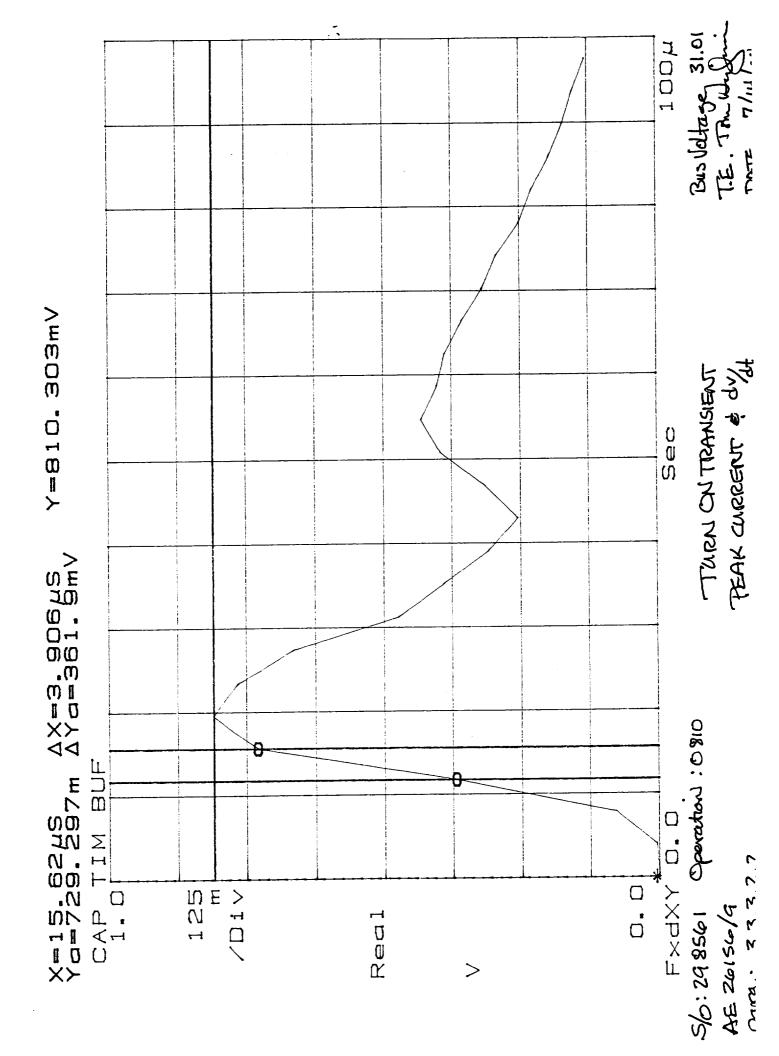


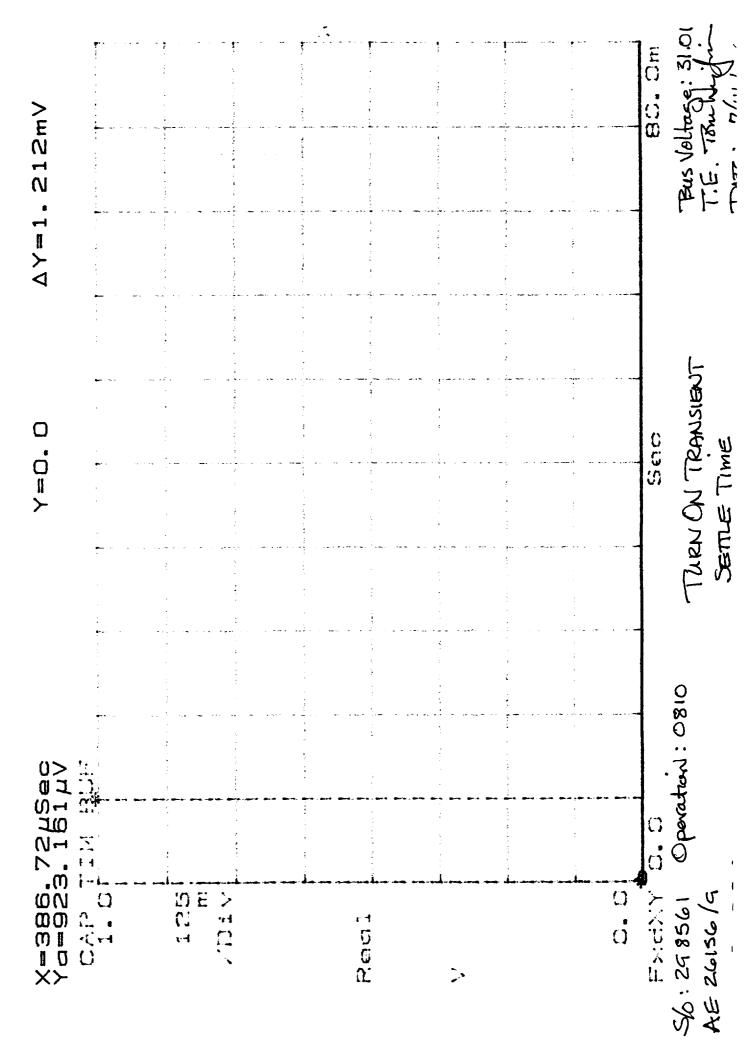












TEST DATA SHEET NO. 7 Passive Analog Interface Test (Paragraph 3.3.4)

Number	Thermistor	Required Temperature (Celsius)	Measured Temperature (Celsius)	Pass/Fail
1	A1-1 SCAN MOTOR	20.9 * ± 5°	Z2/8	ρ
2	A1-2 SCAN MOTOR	20.9 * ± 5°	23.49	P
3	A1-1 RF SHELF#1	20.9 *±5°	24.41	P
4	A1-2 RF SHELF # 1	20.9 * + 5°	25.79	P
5	A1-1 WARM LOAD	70.9 * ± 5°	23.89	ρ
6	A1-2 WARM LOAD	209 * ± 5°	24.08	P
7	A1-1 RF SHELF#2	<u>20.9</u> * ± 5°	24.28	ρ
8	A1-2 RF SHELF#2	26,9 * ± 5°	25:57	P

^{*} is the measured temperature of the unit environment

EOS/AMSU-A1 System P/N 1356008 Shop Order: 29856/ S/N:
Circle Test: 1st CPT Final CPT Sub CPT N/H LPT N/H

Test Systems Engineer Date

Ouality Control Date

TEST DATA SHEET NO. 8 Instrument Commanding Test (Paragraph 3.3.5.2)

Step	Instrument Status	(Y)es / (N)o
12	Full Scan Mode command received?	Y
13	Is A1-1 motor scanning?	Y
14	Did A1-1 motor stop scanning?	Y
15	Is A1-2 motor scanning?	Y
16	Did A1-2 motor stop scanning?	Y
17	Are both motors scanning?	Y
18	Reflectors positioned looking at warm loads?	Y
19	Reflectors positioned looking at nadir?	Y
20	Reflectors positioned looking at cold cal 1?	Y
21	Reflectors positioned looking at cold cal 4?	Y
22	Reflectors positioned looking at cold cal 3?	Y
23	Reflectors positioned looking at cold cal 2?	Y
24	Reflectors positioned looking at cold cal 1?	Y
25	Did PLO toggle?	Y
25	Did C&DH processor reset?	Y

Yes = Pass No = Fail

EOS/AMSU-A1-System P/N 1356008 Circle Test: (1st CPT) Final CPT	Shop Order: 298521 Sub CPT	S/N: 20Z LPT W/A	-1.
	}	18m Hermin	7/13/98
I lanfol ME 23 9	3	Test Systems Engineer JUI 1 4 190	Date
Costomer Representative Date		Quality Control	Date

SCAN NUMBER					0 SA29 [1] RETURN
			TEMP C	00000000000000000000000000000000000000	SA28] FULL
14-JUL-98 21:41:04 ELEMENT 0000	ELEMENT 00	ELEMENT 00	UNPOWERED THERMISTORS DATA	A1-1 SCAN MOTOR TEMPERATURE A1-1 RF SHELF TEMPERATURE #1 A1-1 WARM LOAD TEMPERATURE A1-2 SCAN MOTOR TEMPERATURE A1-2 RF SHELF TEMPERATURE A1-1 RF SHELF TEMPERATURE A1-2 RF SHELF TEMPERATURE A1-2 RF SHELF TEMPERATURE #2 A1-2 RF SHELF TEMPERATURE #2	OFF CHECKSUM IN CALC SCREEN ONLY [2] PRINT [3] FULL
A1-03 E1.EXE;35] SCIENCE DATA] CONTROL/STATUS] ENGINEERING	ON	⊣ αω4πον∞	POWER

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TEST DATA SHEET NO. 9 (sheet 1 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

Step	Instrument Status	(Y)es / (N)o
1	Full Scan Mode command received?	P
2	ENGR OK message seen?	I V
3	Unit (both reflectors) running in full scan mode?	Y

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000101	P
4b	3-4	Packet Length		0000001010111111	P
4c	5-6	Unit Serial Number		0000001100000000	P
4d	7-8	Instrument Mode/ Status		10/11010000000010	P
	L			O PROHPLANT	(QC) 227)

	RADIOMETE	7/13/98	
Step	Description	Required Counts	(P)ass/(F)ail
4f	Review All Scene Data	12500-20500	و

PRT TEMPERATURE DATA						
Step Element Description Required (P)ass/(F)ail						
4g	1090-1178	Review All PRT Data**	10-40 degrees C	P		
4g	1180	Temperature Sensor Reference	23244-26317 counts	ρ		

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
<u> </u>	Antenna in Full Scan Mode		YES	ρ
	Antenna in Warm Cal Mode		NO	
	Antenna in Cold Cal Mode		NO	P
	Antenna in Nadir Mode		NO	Ρ
	Cold Cal Position LSB		ZERO	P
4h	Cold Cal Position MSB		ZERO	Ρ
•••	PLO Redundancy		PLO #1	Ρ
	Scanner A1-1 Power		ON	<u></u>
	Scanner A1-2 Power		ON	19
	PLO #1 Lock		YES	<u> </u>
	PLO #2 Lock		OFF	LP
	ADC Latchup Flag		ONE	- ρ

* Rewriting printout data on this data sheet is optional.
** Refer to Table IV for PRT Data Description

EOS/AMSU-A1 System P/N 1356008 Circle Test: (1st CPT) Final CPT	Shop Order: 298561 Sub CPT <u>1/1</u>	S/N: 202 LPT 7/P	
		R OH blatt	7/3/98
		Test Systems Engineer	Date
Sustomer Representative Date		Quality Control	Date

TEST DATA SHEET NO. 9 (sheet 2 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

REFLECTOR POSITIONS (Step 4e) A1-1 REFLECTOR A1-2 REFLECTOR									
22			· · · · · · · · · · · · · · · · · · ·		A1-2 REFLECTOR				
BP	Element	Position	Required	(P)ass/	Element	Position	Required	(P)ass/	
		(*)	(**) ± 5	(F)ail	ļ	(*)	$(**) \pm 5$	(F)ail	
1_	14	A Kur	14520	PASS	16	PASS	14168	Poss	
2	48		14672		50	57 41.78	14320		
3	82		14824		84	5114/08	14472		
4	116		14975		118	1111111	14623		
5	150		15127		152		14775		
6	184		15279		186		14972		
7	218		15430		220		15078		
8	252		15582		254		15230		
9	286		15734		288		15382		
10	320		15885		322		15533		
11	354		16037		356		15685		
12	388		16189		390	11	15837		
13	422		16340		424	C.K. Val	15988	1	
14	456		108		458	1111	16140		
15	490		260		492	7/7/8	16292		
16	524		411		526	3	59		
17	558		563		560	CTET	211		
18	592		715		594		363		
19	626		866		628	DAIDH	514		
20	660		1018		662	W 41.60	666		
21	694		1170		696	7/160	818		
22	728		1321		730	(221)	969		
23	762		1473		764	1	1121		
24	796		1625		798		1273		
25	830		1776	7	832		1424	/	
26	864		1928		866		1516		
27	898		2080		900		1728	-1-	
28	932		2231		934		1879	$\overline{}$	
29	966		2383		968		2031		
30	1000	<u> </u>	2535		1002	7/14/48	2183	 	
CC	1034		4129	 	1036	1-R West	3717		
VC	1186		8528	PASS	1188	PASS		Puss	

Actual counts from printout. Rewriting counts on this data sheet is optional.
 Required counts from AE26002/1 TDS 5&6 +/- 5 counts

	<u> </u>											
)												
	150			15]	16]	NO [17]	NO [18]	[61] ON	[20]	[21]	168 TEN	
	P1 13-JUL-98 23:36:49 SCAN NUMBER			PLIC#1 [15]	YES [16	8	2	2			84 SA29 168	- -
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•	5:49				N 1 =	2 =	3	N 4 =	ESSOR		SA28	}
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	ECS A1-03 EL.EXE;35 FULL SCAN MODE [5] SCIENCE DATA ELEMENT 0000	[6] CONTROL/STRIUS ELEMENT 00	[7] ENGINEERING		[9] SCANNER A1-1 POMER =	[10] SCANNER A1-2 FOMER =	[11] ANTENNA FULL SCAN MODE =	×	מ	Ż	ц	TICN 3
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	ECS]	[9]	[7]		[9]	[10	[11	[12]	[13]	[14]	ENGR OK	SE

∺	VALUE	16641 16518 16511	16292 16620	16877	720 720 366 714 15920	16671 16671 16636 16514	16289 16436 16892 16892	517 517 513 513 15921	16672 16872 15975 16639	16504 16289 16605 16884 16884	16279 1022 1017 664 15917 16671 16897
3-JUL-98 23:36:53 PACH	r description	SCENE DATA BP 17 CH 8 CH 9 CH 10 CH 10	:13:	2.9.6 2.4.1	REFLECTOR 1 POSITION 18 REFLECTOR 2 POSITION 18 REFL 1 POS 18 2ND LOOK REFL 2 POS 18 2ND LOOK SCENE DAILA BP 18 CH 3	36699999999999999999999999999999999999	988888 311114	RETECTOR 1 ROSTITION 19 RETECTIOR 2 ROSTITION 19 RETECTION 19 2ND LOOK RETECTION 19 2ND LOOK RETECTION 19 2ND LOOK SCENE DATA BP 19 CH 3 CH 4	, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0,	355555 355555	REFLECTOR 1 POSITION 20 REFL 1 POS 20 AND LOOK REFL 2 POS 20 AND LOOK SCENE DATA BP 20 CH 3 SCENE DATA BP 20 CH 3 CH 4
	EI EMENT	_	578 580 80	38°		600000 6004080	2000 2000 2000 2000 2000 2000 2000 200		4588939 458864	444666 4446666	656 665 665 666 666 666 666 666 666 666
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GH 7 16.

GH 10 1 16.

GH 11 16.

GH 12 16.

GH 13 14.

GH 13 14.

GH 14 15.

REFIECTOR 2 POSITION 23.

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MET 2 POS 23. 2ND LOCK

MET 3 POS 23. 2ND LOCK

MET 4 POS 23. 2ND LOCK

MET 5 POS 23. 2ND LOCK

MET 6 POS 23. 2ND LOCK

MET 7 POS 24. 2ND LOCK

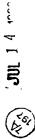
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MET 7 POS 25. 2ND LOCK

MET 7 POS 25. 2

RETLECTOR 1 POSITION 6
RETLECTOR 2 POSITION 6
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RETL 2 POS 6 2ND LOOK
SCENE DATA BP 6 GH 3
GCENE DATA BP 6 GH 3

REFLECTOR 1 REFL 1 FOS REFL 2 FOS SCENE DATA REFL 1 FOS 4 REFL 1 FOS 4 REFL 2 FOS 4 SCENE DATA E



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SCIENCE DATA 1 FULL SCAN MODE	
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SCIENCE DATA FULL SCAN MODE	VALUE	166992 165981 165966 165766 165766 165766 165982 165982 165933 16593 1
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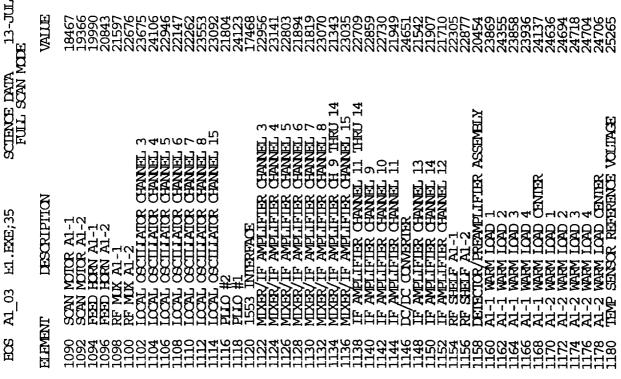
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TEMPERATURE DEG C WALLE

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PAGE
23:36:53
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MODE & STATUS FULL SCAN MODE
El.EXE;35
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TEST DATA SHEET NO. 9 (sheet 3 of 3) Science and Engineering Data Test (Full Scan Mode) (Paragraph 3.3.5.3.1)

	ENGIN	EERING DATA		
Step	Description	Measured*	Required	(P)ass/(F)ai
	Signal Processor (+5 VDC)		+4 to +6 volts	P
	Signal Processor (+15 VDC)		+14 to +16 volts	
	Signal Processor (-15 VDC)		-14 to -16 volts	
	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	
	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current		≤ 3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	4
	A1-2 Noisy Bus Current		≤ 125 milliamps	P

^{*} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System I Circle Test: (1 st CPT)	P/N 1356008 Final CPT	Shop Order: <u>29856</u> / Sub CPT <u>Y</u>	S/N: <u>2</u> 02 LPT <u>r/A</u>	
			BH Plat	7/14/98
1.2	JUL 22 '98		Test Systems Engineer	Date ### 1 4 mas
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 10 (Sheet 1 of 2) Science and Engineering Data Test (Warm Cal Mode) (Paragraph 3.3.5.3.2)

Step	Instrument Status	(Y)es / (N)o
-1	Warm Cal Mode command received?	V
2	ENGR OK message seen?	
3	Both reflectors positioned at warm loads?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	D
4b	3-4	Packet Length		0000001010111111	R
4c	5-6	Unit Serial Number		0000001100000000	D
4d	7-8	Instrument Mode/ Status		101/1101000000100	

			O HOFFILEH
	RADIOMETE	ER SCENE DATA	1113/98
Step	Description	Required Counts	(P)ass/(F)ail
4f	Review All Scene Data	12500-20500	P

PRT TEMPERATURE DATA					
Step	Element	Description	Required	(P)ass/(F)ail	
4g	1090-1178	Review All PRT Data**	10-40 degrees C	P	
4g	1180	Temperature Sensor Reference	23244-26317 counts	P	

STATUS					
Step	Description	Status*	Required Status	(P)ass/(F)ail	
	Antenna in Full Scan Mode		NO	P	
	Antenna in Warm Cal Mode		YES	n	
	Antenna in Cold Cal Mode		NO		
	Antenna in Nadir Mode		NO		
	Cold Cal Position LSB		ZERO		
4h	Cold Cal Position MSB		ZERO		
	PLO Redundancy		PLO#1		
	Scanner A1-1 Power		ON		
	Scanner A1-2 Power		ON		
	PLO #1 Lock		YES		
	PLO #2 Lock		OFF		
	ADC Latchup Flag		ONE	P	

^{*} Rewriting printout data on this data sheet is optional.
** Refer to Table IV for PRT Data Description

EOS/AMSU-A1 System P/N 1356008 Circle Test: (1st CPT) Final CPT	Shop Order: 29856 Sub CPT V/A	S/N: 202 LPT W/A	
		Test Systems Engineer	7/14/98 Date
Constance Representative Date		JUL 14	1998
Instomer Representative Date		Quality Control	Date

EOS A1-03 E1.EXE;35 WARM CAL MODE [5] SCIENCE DATA ELEMENT 0000	n.exe;35	WARM CAL	MODE 2000	P1 14-JUL-98 01:06:42 SCAN NIMBER	an numer	206
[6] CONTROL/STATUS	//SIMIUS	EL EMENT	00			
[7] ENGINEERING	RING	EL EMENT	8			
		COMMENDS	MDS	PILO POMER =	PLIO#1 [15]	15]
[9] SCANNER A1-1 FOWER =	R A1-1 P	OMER =	8	COLD CAL POSITION 1 =	YES [16]	16]
[10] SCANNER A1-2 FOWER =	R A1-2 P	OWER =	8	2 =	NO [17]	17]
[11] ANTENNA FULL SCAN MODE =	A FULL S	CAN MODE:	Q I	3 =	<u>Q</u>	[18]
[12]	WARM CAL		= YES	COLD CAL POSITION 4 =	2	[19]
[13]	COLD CAL		Q =	RESET CATH PROCESSOR	<u> </u>	[20]
[14]	NADIR	••	Q	CSE MODE	_	[21]
ENGR OK	POWER		HECKSUM	ON CHECKSIM IN SPEE CALC SPEE SA28	759 SA29 1517	1517
SELECT BUTTON 3	3 ° ×	N CKEECK		FIXTRE [2] FORTE	7777 [+]	

∺	VALUE	16552 16552	
14-JUL-98 01:06:49 四任	Д	RETLECTOR 1 POSITION 18 REFL 1 POS 18 REFL 2 POS 18 REFL 2 POS 19 REFL 2 POS 19 REFL 2 POS 19 REFL 1 POS 19 REFL 2 POS 19 REFL 2 POS 19 REFL 1 POS 19 REFL 2 POS 19 REFL 1 POS 19 REFL 2 POS 20 REFL 2	
	ELEMENT	7477 888 888 888 888 888 888 888 888 888	
SCIENCE DATA FULL SCAN MODE	VALUE	24200000000000000000000000000000000000	
A1_03 EL.EXE;35	ENI DESCRIPTION	PACKET I ID PACKET LENGTH UNIT SERIAL NUMBER INSTRUMENT MODE/STRAIUS REFLECTOR 1 POSITION 1 REFLECTOR 2 POSITION 2 REFLECTOR 1 POSITION 2 REFLECTOR 2 POSITION 3 REFLECTOR 1 POSITION 3 REFLECTOR 1 POSITION 3 REFLECTOR 2 POSITION 3 REFLECTOR 3 POSITION 3 REFLECTOR 4 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 7 POSITION 3 REFLECTOR 6 POSITION 3 REFLECTOR 7 POSITI	1998 3)
EOS	EL EMENT	HUW4N0L@3J4143074%86W%%8444448074788094686747889848889	

ELEMENT

7	VALUE	165907 16
EXE;35 SCHENCE DATA 14-JUL-98 01:06:49 PACE. FULL SCAN MODE	ם	Charlest
	ELEMENT	546688888888888888888888888888888888888
	VALLE	11655912 1650403 16504
	DESCRIPTION	日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日日
년 전 - 03		REFLECTOR 1 REFLECTOR 2 REFLECTOR 2 REFLECTOR 2 REFLECTOR 2 REFLECTOR 1 REFLECTOR 1 REFLECTOR 2 REFLECTOR 3 REFLETOR 3
ZĀ.	EM	KKKKO KKKKO KKKKO KKKKO

е	VALLE	165915 165917 16460 16460 165038 161033 16103 16543 16543 16543 16543 16533 16503 16503 16503 16503 16503 16503 16503 16503 16510 16	8525 8171 8171 15886 16541 16591 16592 16457 16457 16164 16164 16164 16164	15155 8525 8171 8525 8171 15886 16239
01:06:49 PME	DESCRIPTION	1 POSITIO 2 POSITIO 24 28D 8P 24D	2 DOSITION 25 25 25 25 25 25 25 25 25 25 25 25 25	1 ROSITION 26 2 ROSITION 26 S 26 ZND LOOK S 26 ZND LOOK A BP 26 CH 3
14-JUL-98	ELEMENT DEX	772 774 776 778 782 782 786 798 798 798 861 800 800 800 811 811 811 811 811 811 81	826 RETLECTOR 828 RETLECTOR 832 RETL 1 FO 834 SCENE DATA 844 844 855 856 856	860 REFLECTOR 862 REFLECTOR 864 REFL 1 PO 866 REFL 2 PO 868 SCENE DAID 870
SCIENCE DATA FULL SCAN MODE	VALUE E	165933 165933 165933 165933 165933 165933 165933 165933 16553 16553 16553		16239 16239 16239 16239
A1_03 E1.EXE;35 SCTI	NI DESCRIPTION	0/800112542/ XX 6400/800112543	RETECTOR 1 POSITION 8 REFL 1 POS 8 2ND LOOK REFL 2 POS 8 2ND LOOK SCENE DATA BP 8 CH 3 SCENE DATA BP 8 CH 4 7 CH 10 CH 10 CH 11 CH 11	REFLECTOR 1 POSITION 9 REFLECTOR 2 POSITION 9 REFL 1 POS 9 2ND LOOK REFL 2 POS 9 2ND LOOK SCENE DATA BP 9 CH 3
EDS	EL EMENT	4880048801411102222222222222222222222222	40000000000000000000000000000000000000	2228822 2228822 22288622 22288622

16560 16187 16187 16533 16533 16531 16591 16460 16460 16338 16338 16338 16338	16572 16188 8525 8171 8525 8171 16241 16547 16337	16592 16457 16450 16503 16164 16164 16186 171 18171 16186 16186 16186 16186 16186 16186 16186
788 790 RETLECTOR 1 POSITION 24 794 RETLECTOR 2 POSITION 24 795 RETL 1 POS 24 24 796 RETL 2 POS 24 24 800 SCENE DATA BP 24 CH 3 806 806 806 806 806 811 818 820 CH 6 808 818 820 CH 6 818 820 CH 6 820 CH 7 820 CH 6 820 CH 7 820 CH 7 820 CH 8 820 CH 9 820 CH 10 820 CH 10	REFLECTOR 1 POSITION 25 REFLECTOR 2 POSITION 25 REFL 1 POS 25 2ND LOOK REFL 2 POS 25 2ND LOOK SCENE DATA BP 25 CH CH	HE H
	x x x x x x x x x x x x x x x x x x x)

ELEMENT

	DESCRIPTION	CH 5 CH 6 CH 9 CH 10 CH 10 CH 11 CH 11 CH 12 CH 12 CH 13 CH 13 CH 14 CH 15 CH 15 CH 15 CH 16 CH 16 CH 17 CH 17 CH 16 CH 17 CH 11 CH 11		HET ECTOR 1 POSITION 29 REFLECTOR 2 POSITION 29 REFL 1 POS 29 REFL 2 POS 29 REFL 3 POS 29 REFL 2 POS 29 REFL 2 POS 29 REFL 3 POS
д 1	ENI			
	EL EMENT	68888888888888888888888888888888888888	<i></i>	44488888888888
FULL SCAN MODE	VALLE	165449 165461 1656461 166461 166461 166461 166461 166461 166462 16638 16638 16638 16638 16638 16638	165110 16567 16567 16567 8525 8525 8171 8171 16540 16550 16550	16555 16460 16460 16402 161333 16184 8525 8171 8525 8171 8171
US ELLEARITOS SAL	DESCRIPTION	· · · · · · · · · · · · · · · · · · ·	REFLECTOR 1 POSITION 11 REFLECTOR 2 POSITION 11 REFL 1 POS 11 2ND LOOK REFL 2 POS 11 2ND LOOK SCENE DWIN BP 11 CH 3 CONE DWIN BP 11 CH 4 CH 5	RETECTOR 1 POSITION 12 REEL 1 POS 12 AND LOOK REEL 2 POS 12 AND LOOK SCHEEL 2 POS 12 AND LOOK
₹`	Ħ	密度 配成 の と と と と と と と と と と と と と と と と と と 	現成的対象	アアアアの



2	VALLE	16524 16524 16539		
EOS AL_03 EL.EXE;35 SCIENCE DATA 14-JUL-98 01:06:49 PACE: FULL SCAN MODE	DESCRIPITON	PSS 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	55555	
	ELEMENT DES	972 974 976 977 986 987 987 987 988 988 998 887 1000 887 1000 887 1000 887 1000 887 1010 1010 1010 1010 1020 1032 1033 1034 1040 1040 1040 1040 1040 1040 1040 1040 1050 1040 1040 1050 1060 1070	064 COLD CAL DATA 066 070	
	ALLE	66529391105539111391125113911528855391113911139115391153911539115391153		
	DESC	REFLECTOR 1 POSITION 13 SCENE DATA BY 14 SCENE DATA BY 14 SCENE DATA BY 15 SCENE DATA BY 14 SCENE DATA BY 15 SCENE DATA BY 14 SCENE DATA BY 15	REFLECTOR 1 POSITION 15 REFLECTOR 2 POSITION 15 REFL 1 POS 15 2ND LOOK REFL 2 POS 15 2ND LOOK	
	EI EMENT	WWW44444444444444444444444444444444444	44444 49864 49864	

9	VALUE	00000000000000000000000000000000000000
14-JUL-98 01:06:49 PACE	DESCRIPTION	RETECTOR 1 WARM CAL POS RETECTOR 2 WARM CAL DATA 2 CH 11 WARM CAL DATA 1 CH 2 CH 12 WARM CAL DATA 2 CH 12 CH
SCIENCE DATA 14-7 FULL SCAN MODE	VALUE ELEMENT	15888 1072 16549 1076 16549 1076 16549 1076 16549 1076 16549 1076 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16594 1080 16595 1080 1080 1080 1080 1080 1080 1080 108
EOS A1_03 E1.EXE;35 SCI	ELEMENT DESCRIPTION	494 SCENE DATA BP 15 GH 4 4 4 4 998 500 500 600 600 600 600 600 600 600 600

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TEMPERATURE

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SCAN MOTOR A1-1
FEED HORN A1-2
FEED HORN A1-1
FAMELIFIER CHANNEL 3
MUXERY IF AMELIFIER CHANNEL 3
MUXERY IF AMELIFIER CHANNEL 15
FEED HORN A1-1
FAMELIFIER CHANNEL 11
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01:06:49 PAGE 8
14-JUL-98 E
MODE & STATUS FULL SCAN MODE
EL.EXE;35
8

01:06:49			70									
14-JUL-98		<u>.</u>	DEG C 23.4 28.7 28.7 25.2 32.5 32.5 32.6 32.6 AMPS/VOLITE	4.9 15.1		1.41. 1.8.2.	10.0	000	00000	0.00	2283.2 00.3 0.2	
SCAN MODE	STATUS	SHANGER SHOW	DATA	22070 21840	21799 22161 22128	22455 22455 22071			32767 21256 21205			
MODE & S FULL		ш	SINEERING ATURE #1 ATURE #1 #1 #2 #2	+5 VDC +15 VDC	-15 VDC +15 VDC +15 VDC	115 V	φ ⁴ 5			110 VIC		
ECS Al_03 El.EXE;35	DESCRIPTION	ANTENNA IN FULL SCAN MODE ANTENNA IN WARM CAL MODE ANTENNA IN COLD CAL MODE COLD CAL. POSITION ISB COLD CAL. POSITION NSB PLO REJUNDANCY SCANNER AL-1 ROWER SCANNER AL-2 ROWER PILO #1 IOCK PILO #2 IOCK ACCITATION PILO #2 IO	A1-1 SCANNER MOTOR TEMPERATURE # A1-1 RF SHELF TEMPERATURE # A1-2 SCANNER MOTOR TEMPERATURE # A1-2 RF SHELF TEMPERATURE A1-1 RF SHELF TEMPERATURE A1-2 RF SHELF TEMPERATURE A1-2 RF SHELF TEMPERATURE # A1-2 RF SHELF TEMPERATURE	SIGNAL PROCESSOR	SCAN DRIVE	PLO	RECEIVER MIXER/IF AMPLIFIER A1-1	IO CHANNEL 6	SPARE (10 CFANNEL 3	റയപ്	KRENI OMER BUS OMER BUS	



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TEST DATA SHEET NO. 10 (sheet 2 of 2) Science and Engineering Data Test (Warm Cal Mode) (Paragraph 3.3.5.3.2)

		REFLEC	CTOR POSI	TONS (Step 4e)		
A1-1 REFLECTOR A1-2 REFLECTOR						
BP	Position Range	Required (**) ± 5 counts	(P)ass/ (F)ail	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail
1- 30	852 5	4528	ρ	8171	8176	P_

* Actual range (min to max) of counts from printout (Only beam positions 1-30). Rewriting counts on this data sheet is optional.

^{**} Required counts from AE26002/1 TDS 5&6 ± 5 counts for warm calibration position

	ENGIN	EERING DATA		
Step	Description	Measured***	Required	(P)ass/(F)ail
	Signal Processor (+5 VDC)		+4 to +6 volts	P
	Signal Processor (+15 VDC)		+14 to +16 volts	
	Signal Processor (-15 VDC)		-14 to -16 volts	
	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	
	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
ł	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
}	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current		≤ 3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	4
]	A1-2 Noisy Bus Current		≤ 125 milliamps	<u> </u>

*** Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System P/N 13 Circle Test: (1st CPT) Fina		op Order: <u>29856</u> CPT <u>N/Y</u>	S/N: 202 LPT <u>V/A</u>	
		•	R OH Reals	7/4/98
0. Q_ 6. S	IL 28 '98		Test Systems Engineer J	ULI4 1985
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 11 (Sheet 1 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

Step	Instrument Status	(Y)es / (N)o
1	Cold Cal Mode command received?	F
2	ENGR OK message seen?	V
3	Both reflectors positioned at cold cal position 1?	V
6	Cold Cal Position 2 command received?	
7	ENGR OK message seen?	
8	Both reflectors positioned at cold cal position 2?	P
11	Cold Cal Position 3 command received?	7
12	ENGR OK message seen?	
13	Both reflectors positioned at cold cal position 3?	V
16	Cold Cal Position 4 command received?	V
17	ENGR OK message seen?	V
18	Both reflectors positioned at cold cal position 4?	

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	P
4b	3-4	Packet Length		0000001010111111	P
4c	5-6	Unit Serial Number		0000001100000000	7
4d	7-8	Instrument Mode/ Status		10/11101000001000	P
9a	7-8	Instrument Mode/ Status		10,71101000101000	
14a	7-8	Instrument Mode/ Status		1071101001001000	
19a	7-8	Instrument Mode/ Status		1021101001101000	P

	RADIOMETE	R SCENE DATA	
Step	Description	Required Counts	(P)ass/(F)ail
4f	Review All Scene Data	12500-20500	P

PRT TEMPERATURE DATA						
Step	Element	Description	Required	(P)ass/(F)ail		
4g	1090-1178	Review All PRT Data**	10-40 degrees C	P		
4g	1180	Temperature Sensor Reference	23244-26317 counts	P		

* Rewriting printout data on this data sheet is optional.
** Refer to Table IV for PRT Data Description

EOS/AMSU-A1 System I	P/N 1356008 Final CPT	Shop Order: <u>29856</u> Sub CPT <u>K/A</u>	/ S/N: Z°Z LPT <u>~/A</u>	
			RAS Plat	7/14/98
J. Sanfard	10.7 55 ,23		Test Systems Engineer 1	Date 4 190s
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 11 (sheet 2 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

		STATUS	-	
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		NO	9
	Antenna in Warm Cal Mode		NO	
	Antenna in Cold Cal Mode		YES	
	Antenna in Nadir Mode		NO	
	Cold Cal Position LSB		ZERO	
4h	Cold Cal Position MSB		ZERO	
	PLO Redundancy	-	PLO #1	
	Scanner A1-1 Power		ON	
	Scanner A1-2 Power		ON	
	PLO #1 Lock		YES	
	PLO #2 Lock		OFF	
	ADC Latchup Flag		ONE	•
9b	Cold Cal Position LSB		ONE	
	Cold Cal Position MSB		ZERO	
14b	Cold Cal Position LSB		ZERO	
	Cold Cal Position MSB		ONE	
19b	Cold Cal Position LSB		ONE	V
	Cold Cal Position MSB		ONE	P

^{*} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1-System P/N 1356008	Shop Order: 298561	S/N: 202	
Circle Test: (1st CPT) Final CPT	Sub CPT I	PT - 7/A - 01/1	
Cystomer Representative Date		Test Systems Engineer Date Quality Control Date Date	

TEST DATA SHEET NO. 11 (sheet 3 of 5)
Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

				Cal Mode) (Paragr		
		REFLF	CTOR POSIT	TIONS (Step 46)	7/13/18/22	5) ;
	A¹	1-1 REFLECTOR			1-2 REFLECTOR	9} -
BP	Position Range	Required (**)	(P)ass/	Position Range	Required (**)	(P)ass/
	(*)	± 5 counts	(F)ail	(*)	± 5 counts	
1- 30	4128	4129	P	3779	3777	(F)ail
	** Required cor	unts from AE2600	12/1 TDS 5&6	+/- 5 counts for Co	old Cal Position #	1
				14C		
		REFLE	CTOR POSIT	TIONS (Step 9c)	7/3/48 (22)	2)
,	Al	I-1 REFLECTOR	<u> </u>		1-2 REFLECTOR	
BP	Position Range	Required (**)	(P)ass/	Position Range	Required (**)	(P)ass/
	(*)	±5 counts	(F)ail	(*)	± 5 counts	(F)ass/ (F)ail
1- 30	4050	4053	P	3703	370/	(F)all
	1	ints from AE2600	2/1 TDS 5&6	+/- 5 counts for Co	,	
		no nom.	21 100 000		40.74.77	
		REFLEC	דוופחם פחד	IONS (Step 140)	7/13/18/22	}
	A1	-1 REFLECTOR	.TOK I COIII.		-2 REFLECTOR	<i>[]</i>
BP	Position Range	Required (**)	(P)ass/	Position Range		
	(*)	± 5 counts	(F)ass/	(*)	Required (**) ± 5 counts	(P)ass/
1-			P			(F)ail
30	** Required cou	3977	1	3626	3625	P
	Vedanea com	ALS FOR ALL 20002	<u>71 1D2 280 +</u>	+/- 5 counts for Co		
		DEET EC	TOP PORTE	30C,)}
	Δ1	-1 REFLECTOR	TOK PUSITIN	ONS (Step 196)	7/13/98 22	7))
BP	Position Range	Required (**)	(B)acc/		-2 REFLECTOR	
P.	(*)	± 5 counts	(P)ass/	Position Range	Required (**)	(P)ass/
1-			(F)ail	(*)	± 5 counts	(F)ail
30	3827	382C	P	3476	3474	P
	** Kequirea cour	nts from AEZOUUZ	/1 TDS 5&6 +	+/- 5 counts for Col	ld Cal Position #4	
* A R	ctual range (min t ewriting counts on	to max) of counts in this data sheet is	from printout (optional.	(Only beam position	ns 1-30).	
	SU-A1 System P/N t: 1st CPT Fi	1356008 Shop (inal CPT Sub C	Order: <u>298</u> CPT <u>W/A</u>	LPT N	ns Engineer	7/14/98 Date
mer Ré	presentative	Date		Quality Con		Date
	*			Q	.1401	Date



TEST DATA SHEET Nº 11 (sheet 4 of 5) SCIENCE AND ENGINEERING DATA TEST (COLD CAL MODE) (PARGRAPH 33,5,3,3)

	Po	EFLECTOR	PACITIONS	(STEP BE)		
DEAM 1	A1-1 R	EFLECTOR			-Z REFLECTUR	1 (PASS)
POSITION !	ACTUAL POSICION#	REGULADO #	(P) ASS/	ASSIVES *	REQUIRED ***	(FAIL
COLD CAS	4131	4129	P	3779	3777	P
<u> </u>			T FROM AR	INTOT COLD CP	L I BEAM ROLL	CION COL COLD CALL
	≯ ₹ 	REQUIRED G	OUT FROM	AE-2:002/1C .T	056 ± 5000	S POR CERT
	DE	FLECTOR P	ON ITIONS	(STEP 14C)		
Beam	A1-1 R	FFIECTOR	_ {	AI.	2 REFLECTO	<u>e</u>
PETRON	ACTUAL POSITIONA	REQUIREDAN	(P) 455/	ACTUAL POSITION*	REQUIRED**	(P) ASS/
	1				7701	
COLD CAL 2	4050	4053	p	3703	3701	P
	<u> </u>	1	1 =	2 - 1-0F D (6	1 2 REAM POS	ורטוט
	يالا	DECLIRED O	SOLUT FROM	AE-ZEDOZIC T	DS 6 ±5 Cant	FOR COLD CALZ
	***	Recourses a			<u>-</u>	
	0.	GELGCTOR S	POSITIONS	(STEP 22C)		
BEAM) AI-1K	efiector		A1-2		(AKS /
POSITION	ACTUAL POSITION *	REGULRED **	(P) 455/	ACTUAL ROSITION #	REQUIRED **	(PAS)
COLD CAL			(F)AIL			- WALL
3	3976 1	3977	P	3627	3625	(P)
- -	1 - 1		X 500ml 91	PUNEOUT OUR	AL 3 BEAM PO	115002
1000	R	EFLECTOR :	POSITIOUS	(STEP 30 C)		
BEAM	ACTUAL POGGONINE	DECHOCOST	DIAGE /	A1-2	REFLECTOR	
8615W	AI-I ACTUAL RESTRONUM	W. CONKELL	CAALL	ACTUAL POSITION*	REQUIRED **	(Plass/
CE 10 CAL	38276	3826	·P	3476	3474	MAIL
<u> </u>	1 00 0 1		FROM PRI		4 BEAM DOSITION	I P
ģ	₹ <i>F</i>	account cons	W. From	DE-7602/11 TIS	+ BEAM POSITA	ON
** REQUIRED LOUNT FROM AE-ZCOOZIL TOSE + 5 COUNTS FOR COLD CAC 4						
ESS/F CIRCL	HMSU-AI SYS E TESTI (1 ^{SI}	ION PW CPT FIN	1356008 IAL CPT (SHOP OR OCK:	298561 5/A	1: 20Z
Jan	La faction	## 22 ATIVE DAT		TEST SYST	ENGINEDA ENGINEDA ENGINEDA ENTROL	7/14/98 2 DATE JUL 1 4 199 0ATE

Replaced By new TOS 11 ABA

7/13/98 AE-26156/9
18 June 1998

TEST DATA SHEET NO. 11 (sheet 4 of 5)

Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

	A 1 PEELEC	TOR POSITION (Step 4e)	1
Beam	Actual Beam Count*	Required ** Beam Count	Pass/Fail
Position	Actual Beam Count	(± 5 counts)	
		(± 5 counts)	
Cold Cal 1	* Actual count from printout (Or	als been positions Cold Col 1)	
•		02/IC TDS 6 ± 5 counts for Cold C	'al 1
	Reguled Count Holli AE-2000	JAIC 103 0 ± 3 counts for cold c	
	A1 REFLEC	TOR POSITION (Step 4e)	
Beam	Actual Beam Count*	Required ** Beam Count	Pass/Fail
Position	Actual Beam Count	(± 5 counts)	
		(± 5 counts)	
Cold Cal 1	* Actual count from printou (Or	nly beam positions Cold Cal 1)	
		02/IC TDS 6 ± 5 counts for Cold C	'al 1
	** Required Count from AE-2000	OZIC 1D3 0 ± 3 counts for cold c	
	A1 REFI EC	TOR POSITION (Step 4e)	
Beam	Actual Beam Count*	Required ** Beam Count	Pass/Fail
Position	Actual Beath Count	(± 5 counts)	
		(£ 5 counts)	
Cold Cal 1	* Actual count from printout (Or	nly beam positions Cold Cal 1)	
•		02/IC TDS 6 ± 5 counts for Cold C	'al 1
·	Required count from AL-2000	OZIC IDS 0 2 3 comms for cone c	
	A1 REFLEC	TOR POSITION (Step 4e)	
Beam	Actual Beam Count*	Required ** Beam Count	Pass/Fail
Position	Actual Beam Count	(± 5 counts)	
		(25 002.115)	
Cold Cal 1	* Actual count from printout (O	nly beam positions Cold Cal 1)	
	+ Actual count from printout (O	02/IC TDS 6 ± 5 counts for Cold C	-1
	** Required count from AE-200	OZIC IDS 0 ± 3 coulds for cold c	
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			\
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OS/AMSU-A		rder: S/N: T LPT	
ircle Test: 1	S CPT Final CPT Sub CP	I LFI	- \
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. 5	Data	Quality Control	Date
tomer Repres	sentative Date	Quanty Condo	- Dance
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TEST DATA SHEET NO. 11 (sheet 5 of 5) Science and Engineering Data Test (Cold Cal Mode) (Paragraph 3.3.5.3.3)

·	ENGIN	EERING DATA		
Step	Description	Measured*	Required	(P)ass/(F)ai
	Signal Processor (+5 VDC)		+4 to +6 volts	ρ
	Signal Processor (+15 VDC)		+14 to +16 volts	ŀ
	Signal Processor (-15 VDC)		-14 to -16 volts	
	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	
	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current		≤3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	1
	A1-2 Noisy Bus Current		≤ 125 milliamps	P

^{*} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System P/N 135 Circle Test: (1st CPT) Final	6008 Shop Order: CPT Sub CPT	29856/ S/N: ZOZ <u>N/A</u> LPT <u>N/A</u>	•
Je Somfail July	L 22 70 Date	Test Systems Engineer 14	7/12//98 Date
Customer Representative 1	Date	Quality Control	Date

	308	2K	レモニエ	SUPPORT DATA 1125 12	ÿ	KO 17 012 CY	2	7
ECS A1-03 E1.EXE;35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	135 COLD CAL TA ELEMENT	MODE 0000	P1 14-J	P1 14-JUL-98 00:08:18 SCAN NUMER	SCAN NUM	HER 68		
[6] CONTROL/STATUS ELEMENT 00	US ELEMENT	8						
[7] ENGINEERING	ELEMENT 00	8						
	8	COMMENDS	PLIO	PLIO POWER =	PLLO	PLIC#1 [15]		
[9] SCANNER A1-1 FOWER =	1 POWER =	O	ON COLD	COLD CAL POSITION 1 =		YES [16]		
[10] SCANNER A1-2 FOWER =	2 POWER =	O	8		2	NO [17]		
[11] ANTENNA FULL SCAN MODE =	L SCAN MODE	Q =	0	Ë	2	NO [18]		
[12] WAR	WARM CAL	Z II	OID ON	COLD CAL POSITION 4 =		[19]		
[13]	COID CAL	¥.	YES RESET	RESET CATH PROCESSOR		[20]		
[14] NADIR	景	Z II	NO GSE MODE	DE		[21]		
ENGR OK FOW	ER ON		M IN 33B9 (FOMER ON CHECKSIM IN 33B9 CALC 33B9 SA28	3 321 \$	321 SA29 641		
SELECT BUTTON 3		-1	TIME T		1			

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A1_03 E1.EXE;35	DESCRIPTION	PACKET ID PACKET LENGIH UNIT SERIAL NUMBER	NETLECTOR 1 POSITION 1 REFLECTOR 2 POSITION 1 REFLECTOR 2 POSITION 1 REFL 2 POS 1 2ND LOOK REFL 2 POS 1 2ND LOOK SCENE DATA BP 1 CH C	RETLECTOR 1 POSITION 2 REFL 1 POS 2 2ND LOOK REFL 2 POS 2 2ND LOOK SCENE DATA BP 2 CH 3 SCENE DATA BP 2 CH 3 SCENE DATA BP 2 CH 3 CH 10 CH 10 CH 11 CH 12 CH 12	REFLECTOR 1 POSITION REFLECTOR 2 POSITION REFL 1 POS 3 2ND LOO SCENE DATA BP 3 CH AND
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ECS AL_03 E1.EXE;35	DESCRIPTION	ANTENNA IN FULL SCAN MODE ANTENNA IN WARM CAL MODE ANTENNA IN COLD CAL MODE ANTENNA IN NADIR MODE COLD CAL. POSITION NEB PLO REDUNDANCY SCANNER A1-1 POWER SCANNER A1-2 POWER PLIO #1 LOCK PLIO #2 LOCK PLIO #2 LOCK ADC LAICHUP FLAG	DESCRIPTION A1-1 SCANNER MOTOR TEMPERA A1-1 RF SHELF TEMPERATURE A1-2 WARM LOAD TEMPERATURE A1-2 RF SHELF TEMPERATURE A1-2 WARM LOAD TEMPERATURE A1-2 RF SHELF TEMPERATURE	SIGNAL PROCESSOR SCAN DRIVE PLO RECEIVER MIXER/IF AMPLIFIER A1-1 10 CFANNEL 6 SPARE 10 CFANNEL 3 10 CFANNEL 3 5 6 15 CUIET BUS CURRENT A1-1 NOISY FOWER BUS CUI A1-2 NOISY FOWER BUS CUI A1-2 NOISY FOWER BUS CUI



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P1 14-JUL-98 00:13:06 SCAN NUMBER 104

HOS A1-03 E1. EXE; 35 COLD CAL MODE [5] SCIENCE DATA HIEMENT 0000

		PLIO#1 [15]	NO [16]	YES [17]	NO [18]	[19] ON	[20]	[21]	356 SA29 712	ו ד ו עפורשיי
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[6] CON	[7] ENGINEERING		[6]	[10] SC	[11] AN	[12]	[13]	[14]	ENGR OK	SELECT BUTTON 3

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SCIENCE DATA FULL SCAN MODE	VALLE	000001001 000000101 000000101 000000101 000000	
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IN 14-C	EL EMENT	2624
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EOS A1_03 E1.EXE;35 SC	ELEMENT DESCRIPTION	94 98 98 98 98 98 98 98 98 98 98 98 98 98

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EOS A1_03 E1.EXE;35	DESCRIPTION	ANTENNA IN FULL SCAN MANIENNA IN WARM CAL MO ANTENNA IN COLD CAL MO ANTENNA IN NADIR MODE COLD CAL. POSITION ISB PLO RECENDANCY SCANNER A1-1 POWER SCANNER A1-2 POWER PLO #2 LOCK PLO #3 L	DESCRIPTION A1-1 SCANNER MOICR TEMPERA A1-1 RF SHEIF TEMPERATURE A1-2 WARM ICAD TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 WARM ICAD TEMPERATURE A1-2 WARM ICAD TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-1 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE	SCAN IRIVE PLO RECEIVER MILIFIER AL IO CHANNEL 6 SPARE IO CHANNEL 3 IO CHANNEL 3 SPARE IO

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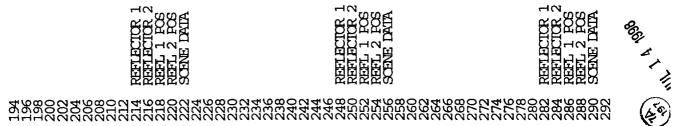
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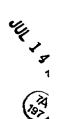
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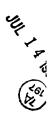
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SCIENCE DATA FULL SCAN MODE	VALUE	16921 16567 16519 16519 16519 16240 16240 16557 16557 16556 16556 16487 16487		
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4	VALUE	165569 16
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RETL 2 COLD CAL POS

RETL 3 COLD CAL POS

RETL 4 COLD CAL POS

RETL 6 COLD CAL POS

RETL 7 COLD CAL POS

RET PACE 00:21:07 DESCRIPTION 14-JUL-98 ELEMENT SCIENCE DATA
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REFL 2 FOS 14 2ND LOOK
SCENE DATA BP 14 CH 3
CH 5
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CH 12
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9	VALUE	15917 165568 164842 166568 16678 16523 16523 16609 16609 16695
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A1_03	Ħ	REFLECTION
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ECS AL 03	EL EMENT	1000 SCAN MO 1000 RF MAX 1100 RF MAX 11100 R





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		PLIO POWER =	COLD CAL POSITION 1 =	2	ε 11	COLD CAL ROSITION 4 =	RESET CATH PROCESSOR	CSE MODE	ON CHECKSIM IN TEFT CALC TEFT SA28 SCREEN ONLY [2] PRINT [3] FILL.	
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[6] CONTROL/STATUS ELEMENT 00	[7] ENGINEERING		[9] SCANNER A1-1 FOWER =	[10] SCANNER A1-2 FOWER =	[11] ANTENNA FULL SCAN MODE =	[12]	[13]	[14]	ENGR OK	SELECT BUTTON 3



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4	VALUE	165546 16
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A 14	EL EMENT	247478888888888888888888888888888888888
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SCIENCE DATA FULL SCAN MODE	VALUE	16525 165455 16546 16546 16552 16552 16552 16535
03 EL.EXE;35	DESCRIPTION	RETECTOR 1 POSITION 13 RETECTOR 1 POSITION 13 RETECTOR 1 POSITION 13 RETECTOR 1 POSITION 14 RETECTOR 1 POSITION 15
ECS AL	ELEMENT	後後後の4444444444444444444444444444444444

9	VALUE	15921 16574 16483 16483 16203 16203 16203 16203 16213 16214 16213
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Ž.	ET EMENT	01101010101111111111111111111111111111
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MODE & S			ENGINEERING DATA N TEMPERATURE RATURE #1 FERATURE TEMPERATURE REATURE #2 FERATURE #2 FERATURE #2 FERATURE #2	+5 VDC +15 VDC	-15 VDC +15 VDC		1 _φ 4.				
ECS A1_03 E1.EXE;35	DESCRIPTION	ANIENNA IN FULL SCAN MODE ANIENNA IN WARM CAL MODE ANIENNA IN COLD CAL MODE COLD CAL. POSITION ISB COLD CAL. POSITION NSB PLO REDUNDANCY SCANNER AL-1 POWER SCANNER AL-2 POWER PILO #1 LOCK ADC LAICHUP FLAG	DESCRIPTION A1-1 SCANNER MOICR TEMPERA A1-1 RF SHEIF TEMPERATURE A1-2 SCANNER MOICR TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 WARM ICAD TEMPERATURE A1-1 RF SHEIF TEMPERATURE A1-1 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE A1-2 RF SHEIF TEMPERATURE	SIGNAL PROCESSOR	SCAN IRIVE	PIO	RECEIVER MIXER/IF AMPLIFIER A1-1	IO CHANNEL 6	SPARE '	4 tv ထ [†]	88



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ECS A1_03 E1.EXE;35	PRC TEMPERATURES VARIABLE TARFET	FIXED TARGET	PASEPI ATE	THERNOCOUPLE TEMPERATURES FIXED TARGET SHROLD VARIABLE TARGET NZ VARIABLE TARGET NZ HEYIER NZ FIXED TARGET FLOW METER VARIABLE TARGET FLOW METER VARIABLE TARGET FLOW METER VARIABLE TARGET FLOW METER PASEPLATE HEYTER NZ BASEPLATE NZ



TEST DATA SHEET NO. 12 (Sheet 1 of 2) Science and Engineering Data Test (Nadir Mode) (Paragraph 3.3.5.3.4)

Step	Instrument Status	(Y)es / (N)o
1	Nadir Mode command received?	P
2	ENGR OK message seen?	TP
3	Both reflectors positioned at nadir position?	V

Yes = Pass No = Fail

Step	Element	Description	Measured Value* (Binary)	Required Value (Binary)	(P)ass/(F)ail
4a	1-2	Packet ID		0000100100000011	P
4b	3-4	Packet Length		0000001010111111	٩
4c	5-6	Unit Serial Number		0000001100000000	P
4d	7-8	Instrument Mode/ Status		1001101000010000	P

RADIOMETER SCENE DATA							
Step	Description	Required Counts	(P)ass/(F)ail				
4f	Review All Scene Data	12500-20500	P				

PRT TEMPERATURE DATA						
Step Element Description Required (P)ass/(F						
4g	1090-1178	Review All PRT Data**	10-40 degrees C	P		
4g	1180	Temperature Sensor Reference	23244-26317 counts	P		

		STATUS		
Step	Description	Status*	Required Status	(P)ass/(F)ail
	Antenna in Full Scan Mode		NO	P
	Antenna in Warm Cal Mode		NO	P
	Antenna in Cold Cal Mode		NO	P
	Antenna in Nadir Mode		YES	P
	Cold Cal Position LSB		ZERO	P
4h	Cold Cal Position MSB		ZERO	P
	PLO Redundancy		PLO #1	e
	Scanner A1-1 Power	<u> </u>	ON	P
	Scanner A1-2 Power		ON	$\overline{\varphi}$
	PLO #1 Lock		YES	
	PLO #2 Lock		OFF	P
	ADC Latchup Flag		ONE	P

* Rewriting printout data on this data sheet is optional.
** Refer to Table IV for PRT Data Description

EOS/AMSU-A1 System P	/N 1356008	Shop Order: <u>24856</u>	S/N: 202	
Circle Test: (1st CPT)	Final CPT	Sub CPT //	LPT <u>~/A</u>	
	i 🚱)	R No Ratt	7/4/98
19 June	MF 88) 5 ·	Test Systems Engineer	JUL 1 4 1998
Customer Representative	Date		Quality Control	Date

TEST DATA SHEET NO. 12 (sheet 2 of 2) Science and Engineering Data Test (Nadir Mode) (Paragraph 3.3.5.3.4)

	A1-1 REFLECTOR			A1-2 REFLECTOR		
BP	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail	Position Range (*)	Required (**) ± 5 counts	(P)ass/ (F)ail
1- 30	333	3 <i>3</i> 6	P	16369	16368	P

^{**} Required counts from AE26002/1 TDS 5&6 +/- 5 counts for "true" nadir position.

	ENGIN	EERING DATA		
Step	Description	Measured***	Required	(P)ass/(F)ail
	Signal Processor (+5 VDC)		+4 to +6 volts	P
	Signal Processor (+15 VDC)		+14 to +16 volts	Ţ,
	Signal Processor (-15 VDC)		-14 to -16 volts	
	Scan Drive (+5 VDC)		+4 to +6 volts	
	Scan Drive (+15 VDC)		+14 to +16 volts	
	Scan Drive (-15 VDC)		-14 to -16 volts	
	PLO (+15 VDC)		+14 to +16 volts	
	PLO (-15 VDC)		-14 to -16 volts	
	Receiver (+8 VDC)		+7 to +9 volts	
4i	Mixer/IF Amplifier A1-1 (+10 VDC)		+9 to +11 volts	
	Mixer/IF Amplifier A1-2 (+10 VDC)		+9 to +11 volts	
	LO Channel 6		+9 to +11 volts	
	LO Channel 7		+9 to +11 volts	
	LO Channel 3		+9 to +11 volts	
	LO Channel 4		+9 to +11 volts	
	LO Channel 5		+9 to +11 volts	
	LO Channel 8		+9 to +11 volts	
	LO Channel 15		+14 to +16 volts	
	Quiet Bus Current		≤ 3 Amps	
	A1-1 Noisy Bus Current		≤ 125 milliamps	V
	A1-2 Noisy Bus Current		≤ 125 milliamps	P

^{***} Rewriting printout data on this data sheet is optional.

EOS/AMSU-A1 System P/N 1356008 Circle Test: (1st CPT) Final CPT	Shop Order:	S/N: <u>202</u> LPT <u>V/A</u>	
# 22 M		Test Systems Engineer	7/14/98 Date
Customer Representative Date		Quality Control	1 4 1998 Date

301			15]	YES [16]	NO [17]	18]	M [19]	[20]	[21]	471 URN		
CAN NAMEER			PLIO#1 [15]	YES [8	NO [18]	2	_		236 SA29 471 [1] REIURN		
Pl 13-JUL-98 23:56:57 SCAN NAMEER			PLIO POMER =	COLD CAL POSITION 1 =	2	æ.	COLD CAL POSITION 4 =	RESET CATH PROCESSOR	CSE MODE	ON CHECKSIM IN 8617 CALC 8617 SA28 SCREEN ONLY [2] PRINT [3] FULL		
0	8	8	g	8	g	2	2	2	YES	XSUM X [2		
MODE INT 000	O IN	ELEMENT 00	COMMENDS		11	TE =	11	11	11	AS SE		
NADI	H. EM	EL EIM	O	OMER =	OMER.	CAN M	M.	Æ		FOWER ON SCREE		
EOS A1-03 Ed.EXE;35 NADIR MODE [5] SCIENCE DATA ELEMENT 0000	[6] CONTROL/STRIUS FLEMENT 00	INEERING		[9] SCANNER A1-1 FOWER =	[10] SCANNER A1-2 FONER =	[11] ANIENNA FULL SCAN MODE =	WARM CAL	COLD CAL	NADIR	FOWER	JUICN 3	
EOS A1-([6] con	[7] ENGINEERING		[6]	[10] 823	[11] AN.	[21]	[13]	[14]	ENGR OK	SELECT BUTION 3	

(Feb.)

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SCIENCE DATA FULL SCAN MODE	VALUE	00000000000000000000000000000000000000
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	VALUE	15958 16613 16613 16708 16726 16323 16323 16323 16301 16869 16869 16615 16615	16227 16227 16227 16227 16227 16223 16623 16614 15954 15954 16614	5210 5244 5244 5236 5336 5311 5311 5311 5311 5311
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Al_03L.EXE;35	ELEMENT DESCRIPTION	REFLECTOR 1 REFL 1 FOS REFL 2 FOS REFL 2 FOS SCENE DATA	REFLECTOR REFL 1 POE REFL 2 POE SCENE DATE	CH 11 CH 12 CH 12 CH 13 CH 13 CH 13 CH 14 CH 15
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æ	VALLUE		16369 15333 163014 16621 16616 16526 16526 16526 16786 16787	1333 1333 1333 15918 15918
Г-98 23:57:04 В.	DESCRIPTION	PS 24 PS SELECTION OF SELECTION	22 POSITION 25 POS 25 200 LOOK 25 200 LOOK 25 CHOOK 25 CHOOK 26 CHOOK 27 CHOOK 28 CHOOK 28 CHOOK 29 CHOOK 20 CHOOK	TOR 1 POSITION 26 TOR 2 POSITION 26 POS 26 2ND LOOK POS 26 2ND LOOK POS 26 2ND LOOK POS 30 CH 3
SCIENCE DATA 13-JUL-98 FULL SCAN MODE	VALLIE ELEMENT	664466 66443	3369 3333 3333 3333 3321 3321 334 336 336 337 336 337 336 337 337 338 338 338 338 338 338 338 338	
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N MODE	ELEMENT	2472808888888888888888888888888888888888
SCIENCE DATA FULL SCAN MODE	VALUE	16299 16299 16499 16499 16499 16309 16309 16309 16499 16499 16499 16499 16309 16309 16309 16499 16309
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ECS

<u>ჅჁჿჁჾჾჇႯႯႯႯႷჿჿႿႿჾჃჁჿჁჾჾჇႯႯႯႯႯჽჽჽჽჽ</u>๛ჅჁჿჁჾჇჇႯႯႯႯႯ Н $^{\circ}$ COLD CAL DATA COLD CAL DATA 1038 1040 1040 1040 1044 1060 1060 1066 1068 1068 15923 16301 16457 16457 16457 16529 16529 16369 16369 16369 SCENE DATA BP 14 CH 3
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9	VALUE	00000000000000000000000000000000000000
23:57:04 PAGE	PITON	######################################
13-JUL-98 23	IT DESCRIPTION	REFLECTOR 1 WARM CAL DATA WARM CAL DATA WARM CAL DATA
MODE	EL EMENT	01010101010101010101010101010101010101
TIENCE DATA FULL SCAN MODE	VALUE	11.05.000
A1_03 E1.EXE;35 SC	ENT DESCRIPTION	SCENE DATA BP 15 GH 4 4 GH 10 GH 11 GH 12 GH 13 GH 14 GH 12 GH 14 GH 15 GH 14 GH 15
ECS	EL EMENT	4,44,000,000,000,000,000,000,000,000,00





23:57:04 F	IEMPERATURE DE	222222 8222228888888888888888888888888
A 13-JUL-98 MODE	VALUE	22222222222222222222222222222222222222
SCIENCE DATA FULL SCAN MODE		CHANNEL 3 CORNNEL 5 CORNNEL 6 CORNNEL 6 CORNNEL 15 CORNNEL 15 CORNNEL 15 CORNNEL 14 ER CORNNEL 5 ER CORNNEL 6 ER CORNNEL 15 ER C
ਦਾ.EXE;35	DESCRIPTION	R AI-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1-1
ECS AL_03	ELEMENT	1090 SCAN MOTOR 1099 SCAN MOTOR 1099 SCAN MOTOR 1099 RF MIX AL- 1100 RF MIX AL- 1100 RF MIX AL- 1100 RF MIX AL- 1110 ICCAL OSCI- 1110 ICCAL 1110 ICCAL OSCI- 1110 ICCAL 1110 IC



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23:57:04	
13-711-98	
MODE & STATUS	FULL SCAN MODE
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23:57:04			70	
13-JUL-98		т **	DEG C 23.4 28.7 25.2 32.5 28.6 32.6 AMS/ACLIS	22 4.7.7.4.4.7.4.7.7.000000000000000000000
SCAN MODE	STATUS	SSSEE SSEE SSSEE S	DATA	22078 221739 221739 22156 22156 221457 221434 221347 221347 221347 221347 221347 221347 221347
MODE & S FULL		BBBB	ENGINEERING DATA N TEMPERATURE RATURE #1 ERATURE #1 ERATURE #1 ERATURE #2 ERATURE #2 ERATURE #2	
EOS AL_03 E1.EXE;35	DESCRIPTION	ANTENNA IN FULL SCAN MODE ANTENNA IN WARM CAL MODE ANTENNA IN COLD CAL MODE ANTENNA IN NADIR MODE COLD CAL. POSITION LES COLD CAL. POSITION MES PLO RECENDANCY SCANNER AL-1 POWER SCANNER AL-2 POWER PLIO #1 IOCK PLIO #2 IOCK ADC LATCHUP FLAG	A1-1 SCANNER MOIGR TEMPERA A1-1 RF SHELF TEMPERALURE A1-2 WARM LOAD TEMPERALURE A1-2 RF SHELF TEMPERALURE A1-2 WARM LOAD TEMPERALURE A1-2 WARM LOAD TEMPERALURE A1-2 WARM LOAD TEMPERALURE A1-1 RF SHELF TEMPERALURE A1-1 RF SHELF TEMPERALURE A1-2 RF SHELF TEMPERALURE A1-2 RF SHELF TEMPERALURE A1-2 RF SHELF TEMPERALURE	SIGNAL PROCESSOR SCAN IRIVE PLO RECEIVER MICHEL Al-1 IO CHANNEL 6 SPARE IO CHANNEL 3 10 CHANNEL 3 15 CUIET BUS CURRENT Al-1 NOISY FOWER BUS CUIALING STATES CONTRACT Al-1 NOISY FOWER BUS CUIALING STATES CUICKLE STATE



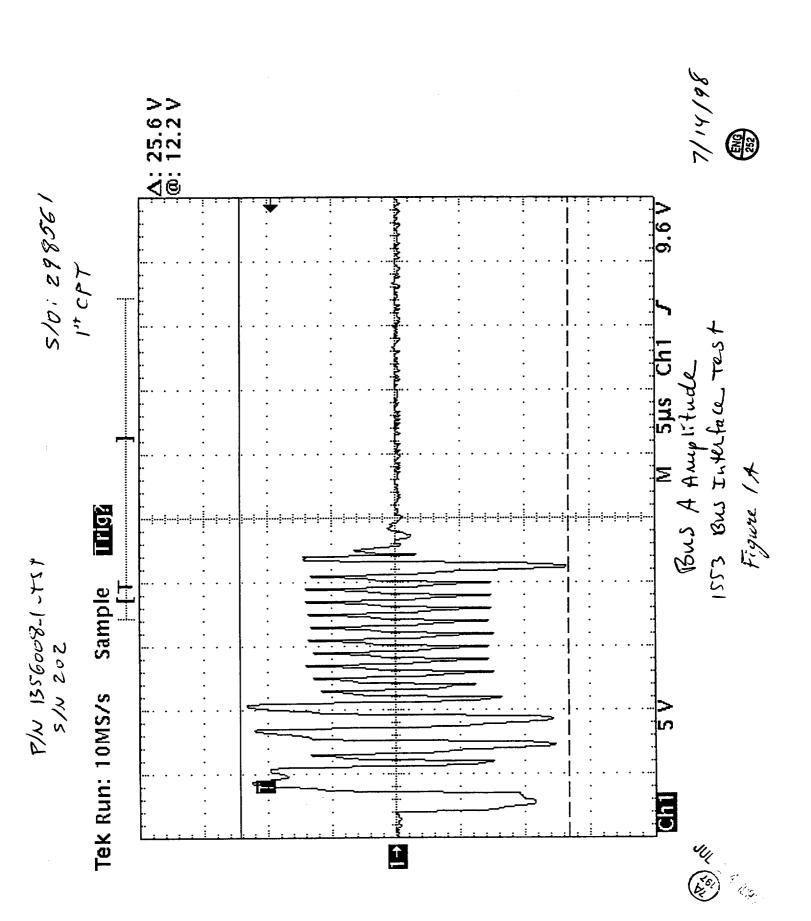
AZONIX DATA 13-JUL-98 FULL SCAN MODE	A1-1	H 44444	621 622 623 623 624 624 625 52.00 626 625 53.00 627 627	68.00 71.00 26.00	A1-1	400		57.00 57.00	0000 00000 000000	900 900 900	ER 508 63.00 510 510 510 510 510 510 504 504 513 64.00 513 513 520 520 520 520 520 520 520 520 520 520	78.55 90.55
ECS A1_03 .EXE;35	PRT TEMPERATURES	VARLABLE TARGET	FIXED TARGET	BASEPLATE	THERMOCOUPLE TEMPERATURES	FIXED TARGET SHROLD	VARIABLE TARGET SHROLD	FIXED TARGET NZ	VARIABLE TAREST NZ	HEMTER NZ	GET FLOW MET TARGET FLOW HEATER NZ	ERSEPLATE IN BASEPLATE FLOW METER ADJUNCT RADIATORS

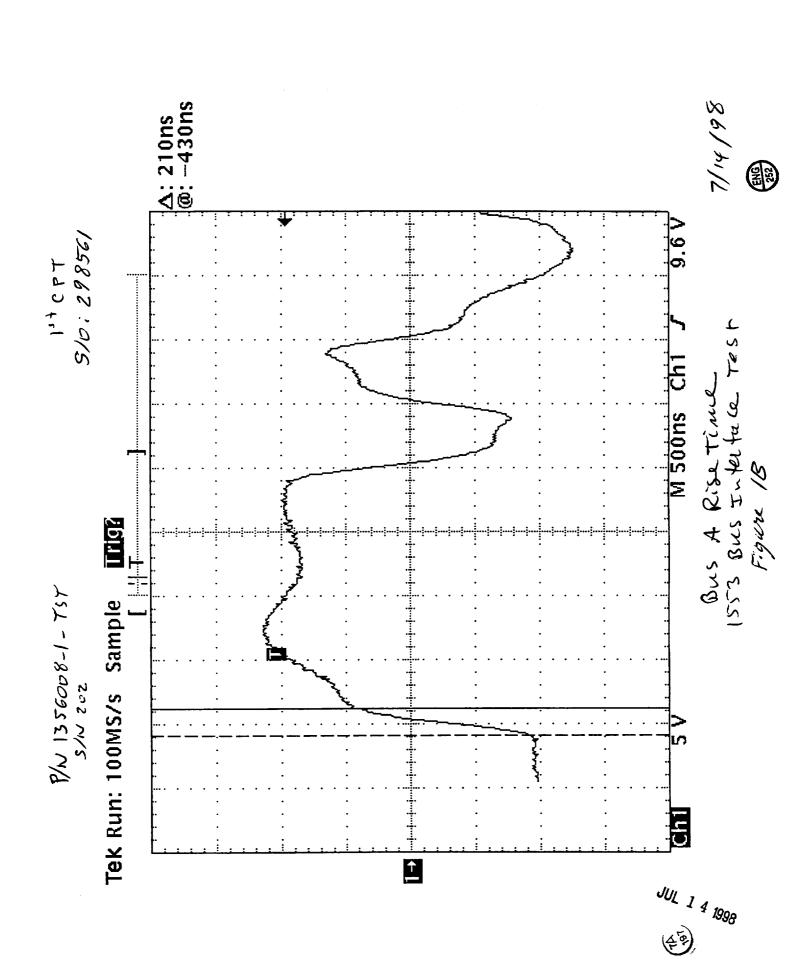
TEST DATA SHEET NO. 19` 1553 Bus Interface Test (Paragraph 3.3.5.4)

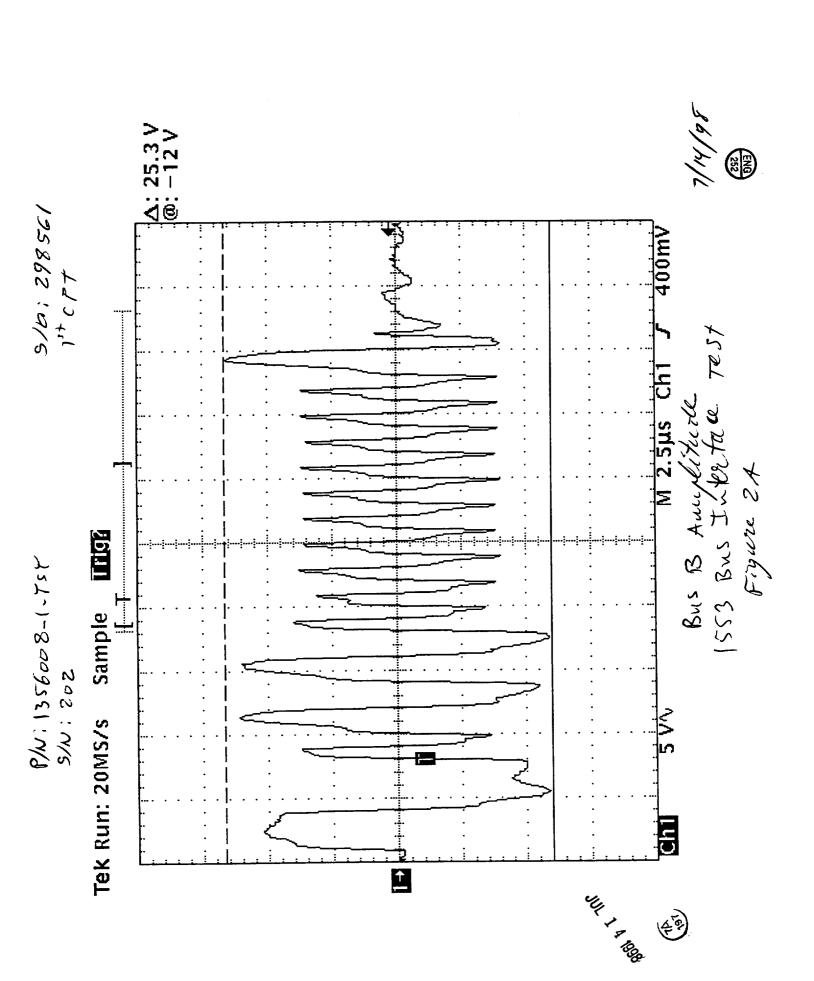
	ATTACH BUS A WAVE FORM See attached figures 14 + 1B	
	Bus A Amplitude Bus A Rise Time P/F 25.6V: 18.0 - 27.0 VP-P Pan 260	
	ATTACH BUS B WAVE FORM See Attached figures 2 A & 2 B	
S/O:	Bus B Amplitude Bus B Rise Time 25.3 \(\text{ : 18.0 - 27.0 VP-P } \) Bus B Rise Time 220 \(\text{ is 100 - 300 nsec} \) CPT: \(\text{ : 100 - 300 nsec} \) CPT: \(: 100 - 300 nsec	, (-

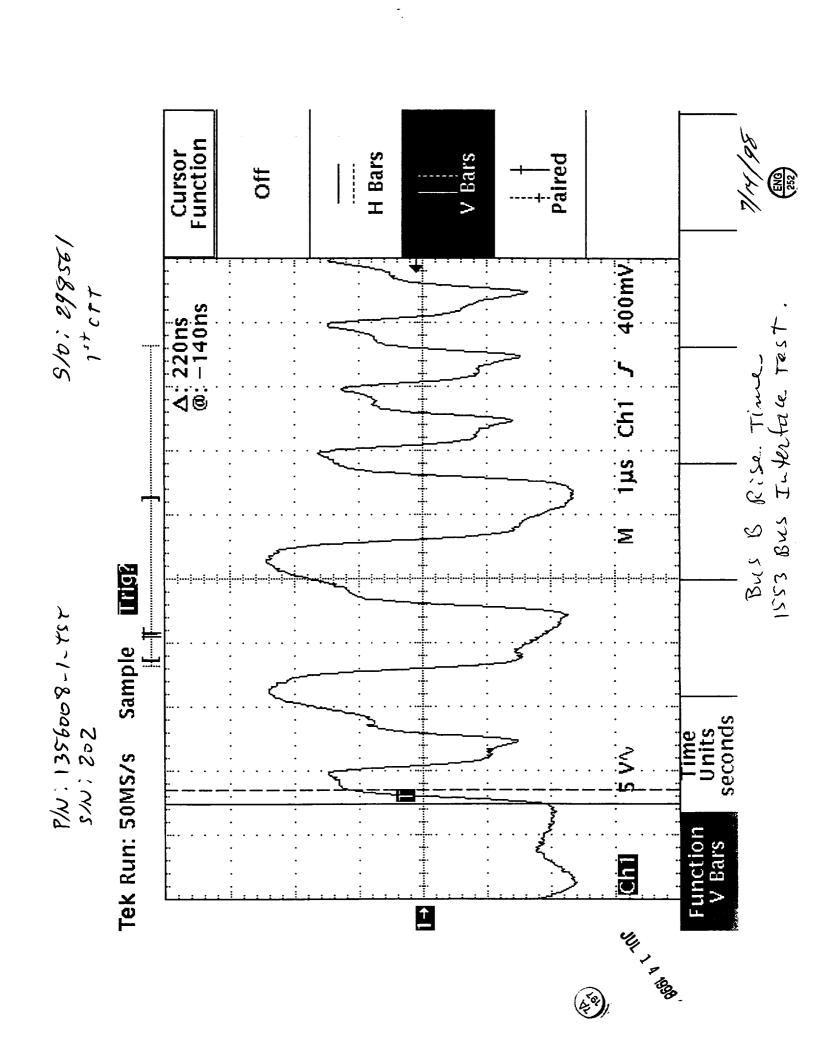
TEST DATA SHEET NO. 13 Noisy Bus Current Measurement During Warm Cal, Cold Cal and Nadir

Instrument Mode	Noisy Bus Current (mA)	Pass/Fail
Warm Cal A1-1 & A1-2 Scanner ON	.27	Not
Al-1 Scanner /Al-2 Scanner OFF	21	Applicable
A1-2 Scanner A1-1 Scanner OFF A1-1 SANNER A1-2 SWINE	21	
Cold Cal A1-1 & A1-2 Scanner ON	30	
Nadir Al-1 & Al-2 Scanner		
ON	2/	
		Not Applicable
S/AMSU-A1 System P/N 1356008 S cle Test: 1 st CPT Final CPT S	hop Order: 29856/ S/N:	202 A- D 11
<i>L</i> 10 41 22 38	Test System	s Engineer 1 4 1998 Date









TEST DATA SHEET NO. 14 Test Point Interface Test (8 Second Sync Pulse TP) (Paragraph 3.3.6.2)

8 SECOND SYNC PULSE TEST POINT

Attach Photograph or Plot Here or to Back of TDS

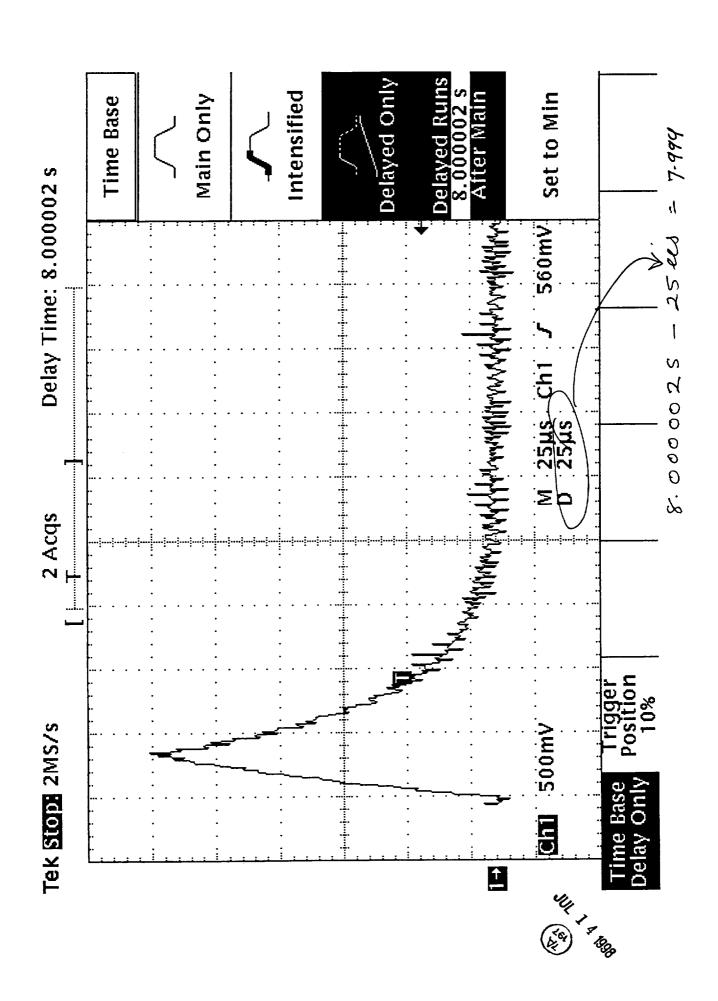
	8 SECOND SYNC PULSE TEST POINT						
Step	Parameter	Measured	Required	(P)ass / (F)ail			
2	Pulse Length	7- 999 seconds	8 seconds +/- 10%	\mathcal{P}			
-2-	Amplitude	-volts	-3-5 volts	a/A			

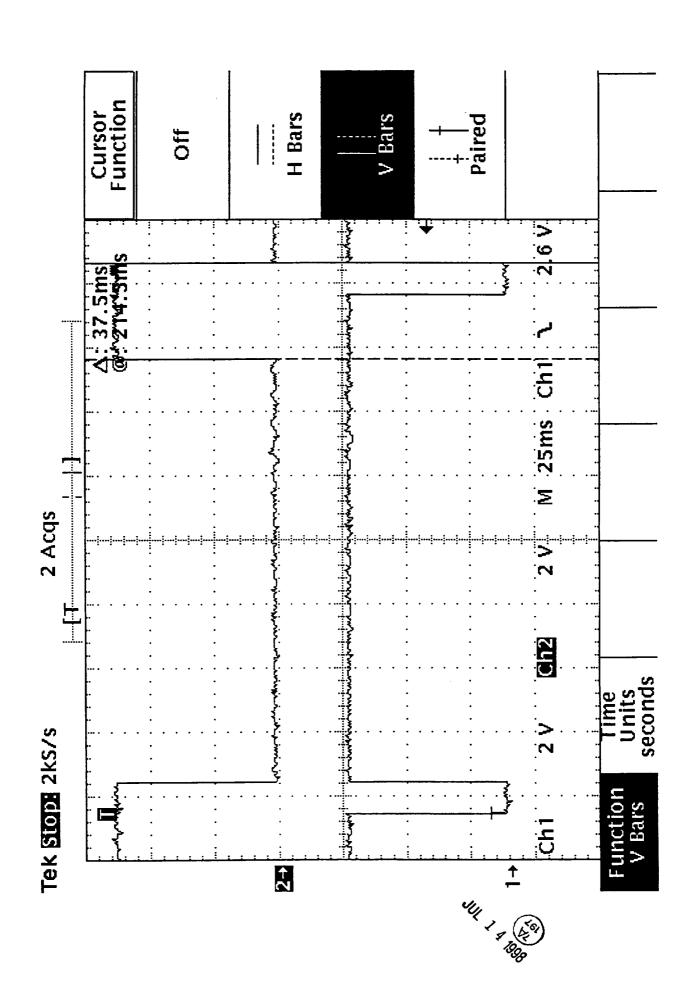
EOS/AMSU-A1 System P/N 1356008 Shop Order: 29856 Circle Test: 1st CPT Final CPT Sub CPT Final CPT

Quality Control

TEST DATA SHEET NO. 15 Test Point Interface Test (Integrate/Hold and Dump TPs) (Paragraph 3.3.6.3)

	Г—		tegrate/Hold and Dump	3.07	
			E/HOLD AND DUMP?		
		INTEGRA	TE/HOLD SIGNAL TE	EST POINT	21/18 (227)
į	Step	Parameter	Measured	Regulired	(P)ass / (F)ail
ł	4	Time Measured (A)*	1//	1 / D /	
	-		/65 milliseconds	158 ± 5 ms	P
207)	4	Time Measured (B)*	37.5 milliseconds	158 ± 5 ms 32 -38 ms	P
207) 227)	4	Time Measured (B)* Time Measurement (A+B)*	37.5 milliseconds		
200	 	Time Measured (B)*	37.5 milliseconds	32 -38 ms	P
25 test	 	Time Measured (B)* Time Measurement (A+B)*	37.5 milliseconds	32 -38 ms 200 ± 5 ms	P
27) 21/4/9	4	Time Measured (B)* Time Measurement (A+B)* Amplitude	37.5 milliseconds	32 -38 ms 200 ± 5 ms 4-6 volts	P
2027) 21/4/9	 	Time Measured (B)* Time Measurement (A+B)* Amplitude	37.5 milliseconds 02.5 milliseconds volts	32 -38 ms 200 ± 5 ms 4-6 volts	P
077 de 1419	4	Time Measured (B)* Time Measurement (A+B)* Amplitude DUM	37.5 milliseconds 02.5 milliseconds volts	32 -38 ms 200 ± 5 ms 4-6 volts	P
0227) 21019	Step 4	Time Measured (B)* Time Measurement (A+B)* Amplitude DUM Parameter Time Measured (D)* Amplitude	37.5 milliseconds 02.5 milliseconds volts AP SIGNAL TEST PO Measured /2-5 ms volts	32 -38 ms 200 ± 5 ms 4-6 volts INT Required	(P)ass / (F)ail
Ridd Ridd Ridd Ridd Ridd Ridd Ridd Ridd	Step 4	Time Measured (B)* Time Measurement (A+B)* Amplitude DUN Parameter Time Measured (D)* Amplitude Refer to Figure 18 for Waveford 1 System P/N 1356008 Shop	37.5 milliseconds 02.5 milliseconds volts MP SIGNAL TEST PO Measured /2-5 ms volts orm Definition Order: 29856/ CPT	32 -38 ms 200 ± 5 ms 4-6 volts INT Required 9-15 ms	(P)ass / (F)ail P 7/14/48





TEST DATA SHEET NO. 16 Test Point Interface Test (Radiometer Channel Analog Output TPs) (Paragraph 3.3.6.4)

RADIOMETER CHANNEL ANALOG OUTPUT TEST POINTS

Attach Photographs or Plots Here or to Back of TDS

RADIOMETER CHANNEL ANALOG OUTPUT TEST POINTS							
/	Integration	Integration	Hold	Hold	Dump	Dump /	
Channel	Time	Time	Time	Time	Time	Time /	(P)ass /
	Measured	Required	Measured	Required	Measured	Required	(F)ail
	(E)*	165 (ms)	(F)*	(ms)	(F)*	(ms/	<u>^</u>
3	165 ms	1258 ± 5 ms	25 ms	23-27	ms	9-15	ρ
4	165 ms	158 ± 5 ms	25 ms	23-27	ms	9/15	
5	165 ms	158 ± 5 ms	25 ms	23-27	ns	9-15	
6	165 ms	158 ± 5 ms	25 ms	23-27	ms	9-15	
7	165 ms	158 ± 5 ms	25 ms	23-27	m x \	9-15	
8	165 ms	158°± 5 ms	25 ms	23-27	∂ \max_s	9-15	
9	165 ms	158 ± 5 ms	25 ms	23-27) ms	9-15	
10	165 ms	158 ± 5 ms	25 ms	23-27	, ms	\9-15	
11	165 ms	158 ± 5 ms	25 ms	23-27	(0' /ms	Q -15	
12	165 ms	158 ± 5 ms	25 ms	23-27	ms	9-\15	
13	165 ms	158 ± 5 ms	25 ms	23-27	ms	9-15	
14	165 ms	$\sqrt{158 \pm 5}$ ms	25 ms	23-27	/ ms	9-15	V
15	165 ms	3 58 ± 5 ms	25 ms	23-27	ms	9-15	ρ

* Refer to Figure 18 for Waveform Definition

EOS/AMSU-A1 System P/N 1356008 S Circle Test: (1st CPT) Final CPT

Shop Order: <u>29856</u>/ Sub CPT <u>r/A</u> S/N: 202

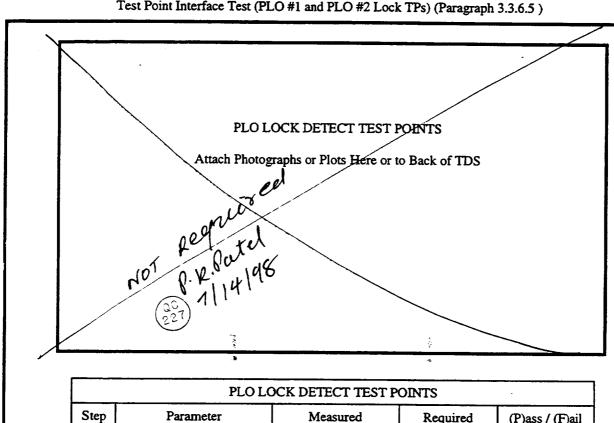
Test Systems Engineer

Quality Control

Date

7/14/58

TEST DATA SHEET NO. 17 Test Point Interface Test (PLO #1 and PLO #2 Lock TPs) (Paragraph 3.3.6.5)



	PLO LO	OCK DETECT TEST PO	INTS	•
Step	Parameter	Measured	Required	(P)ass / (F)ail
3	PLO #1 Lock Detect*	0.012myolts	0-1-volt	P
6	PLO #2 Lock Detect**	-2-80 mvolts	0-1 volt	1

* When PLO #1 is selected

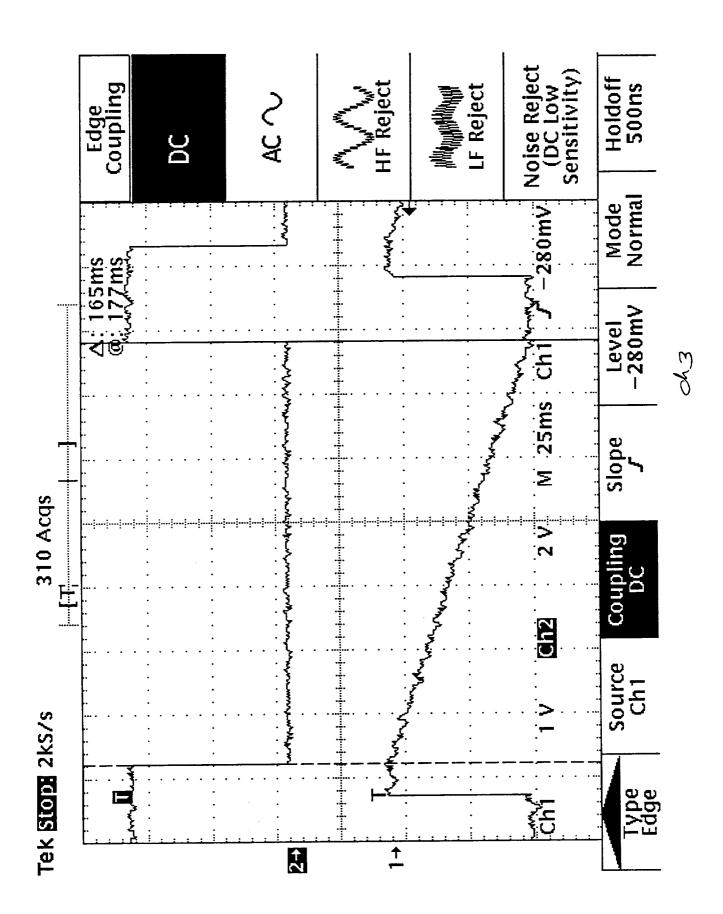
** When PLO #2 is selected

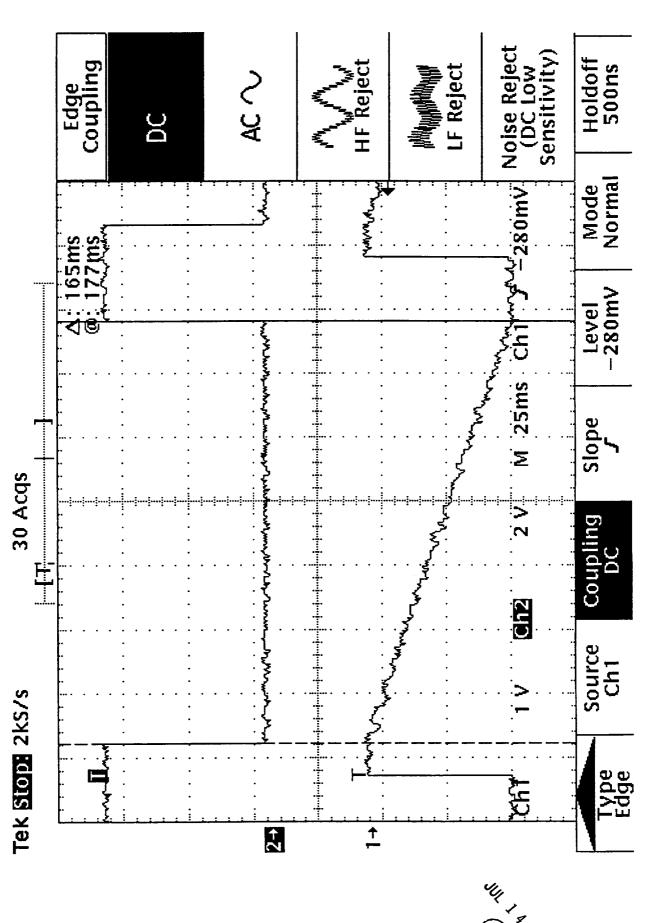
Test Systems Engineer

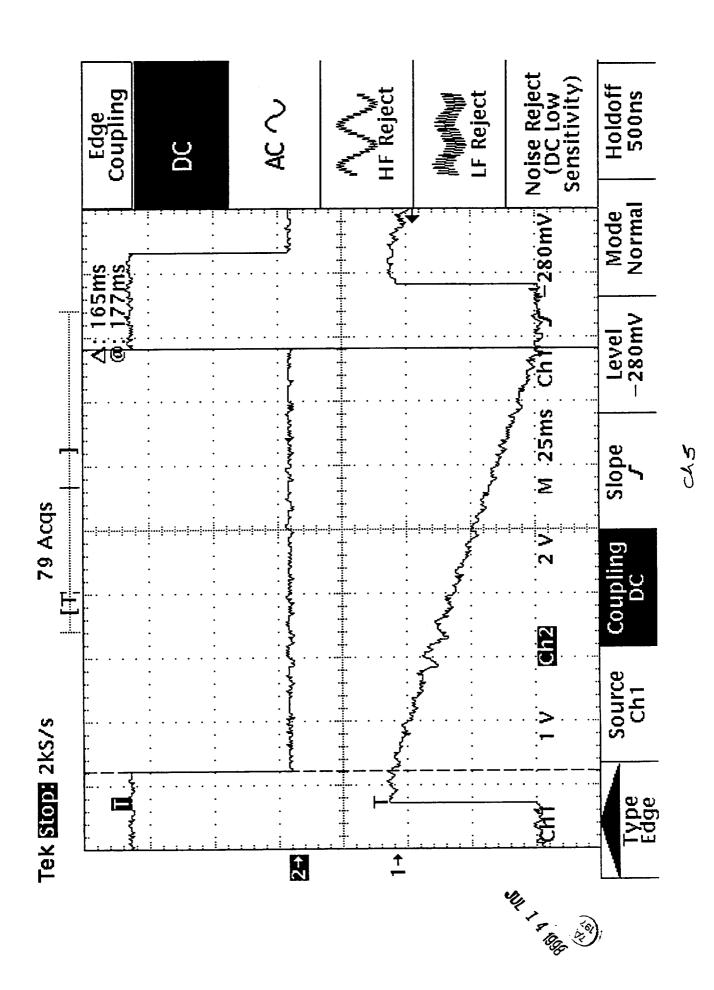
Date

Quality Control

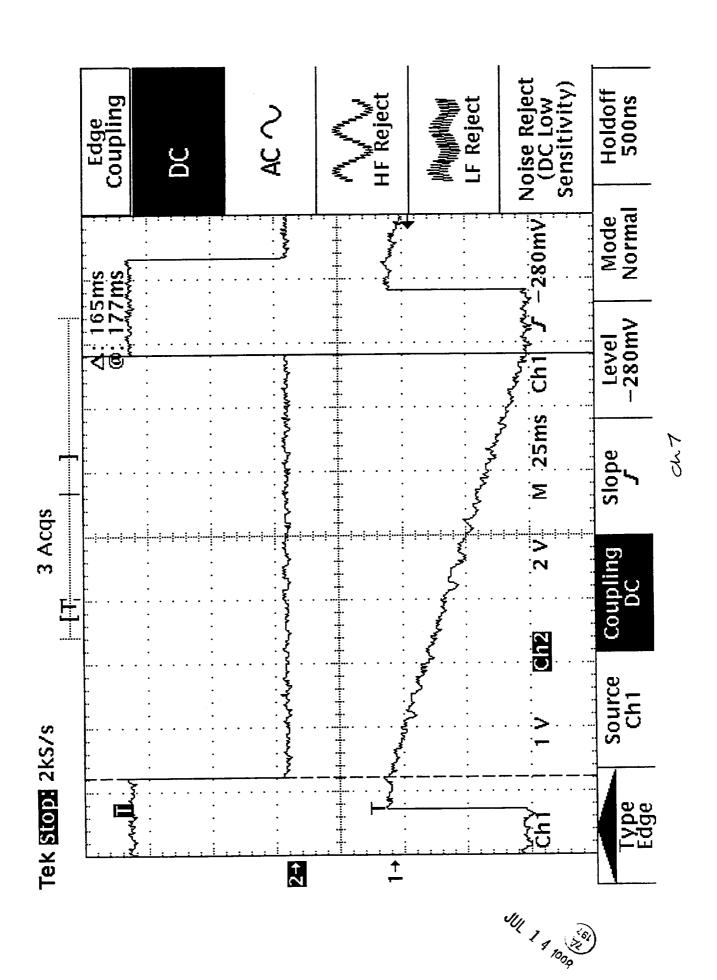
Date

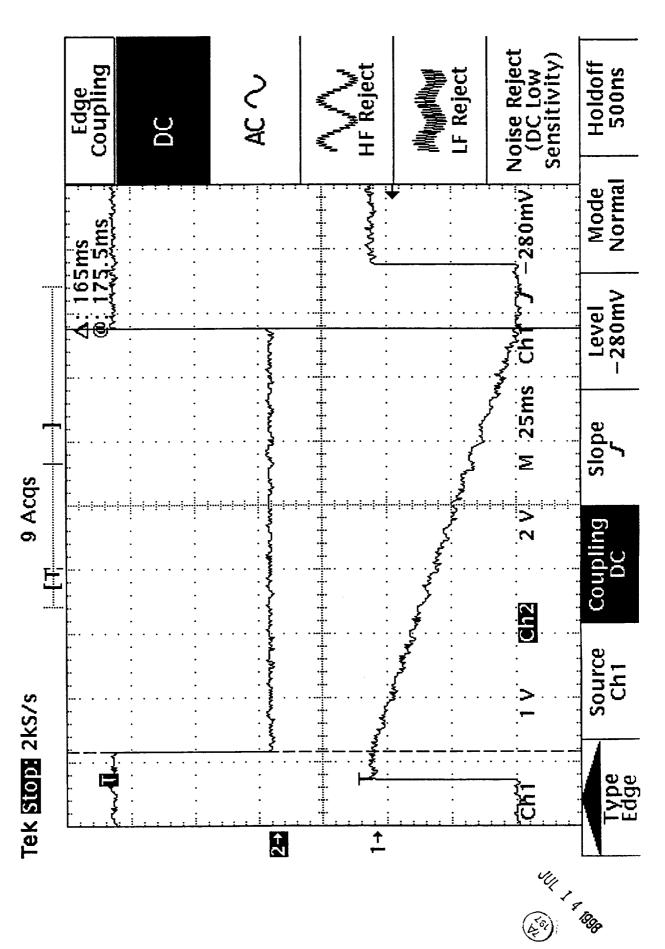


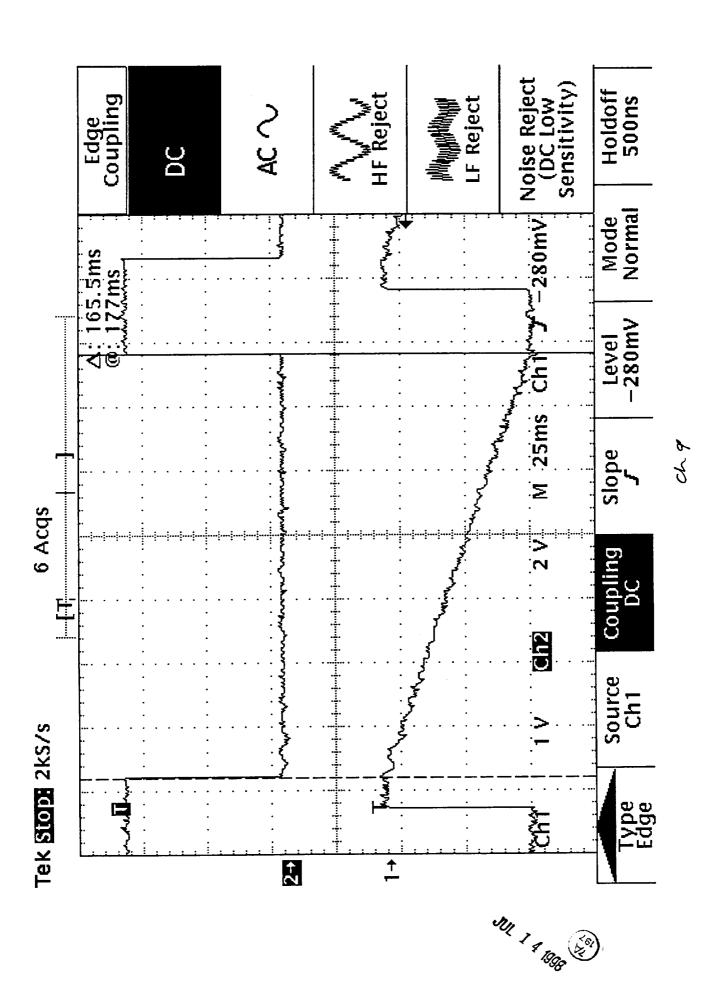


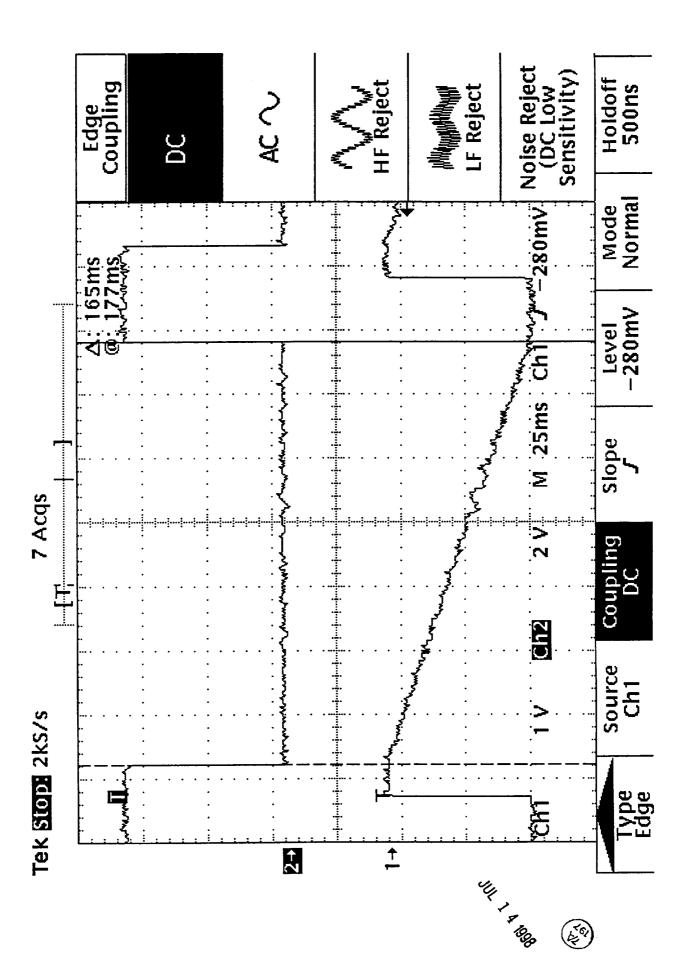


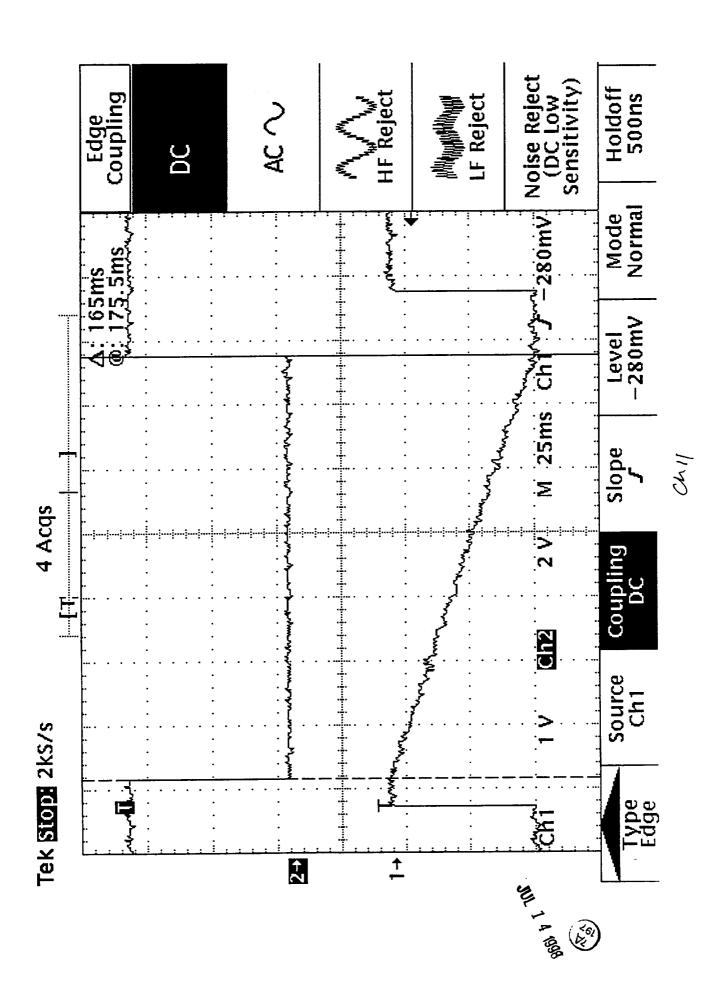
G46

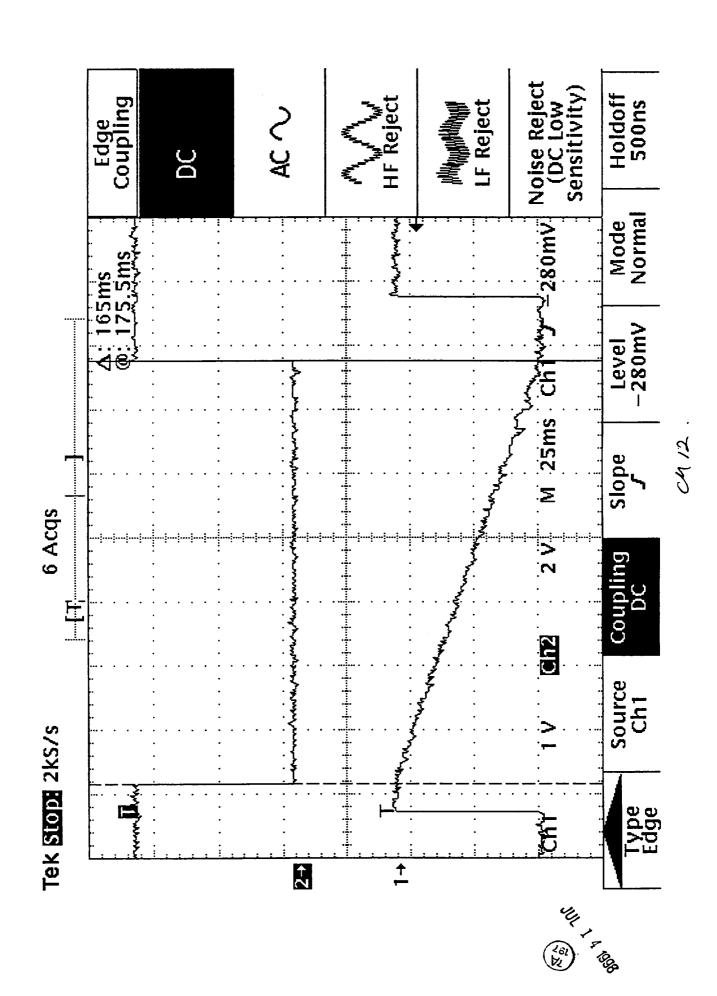


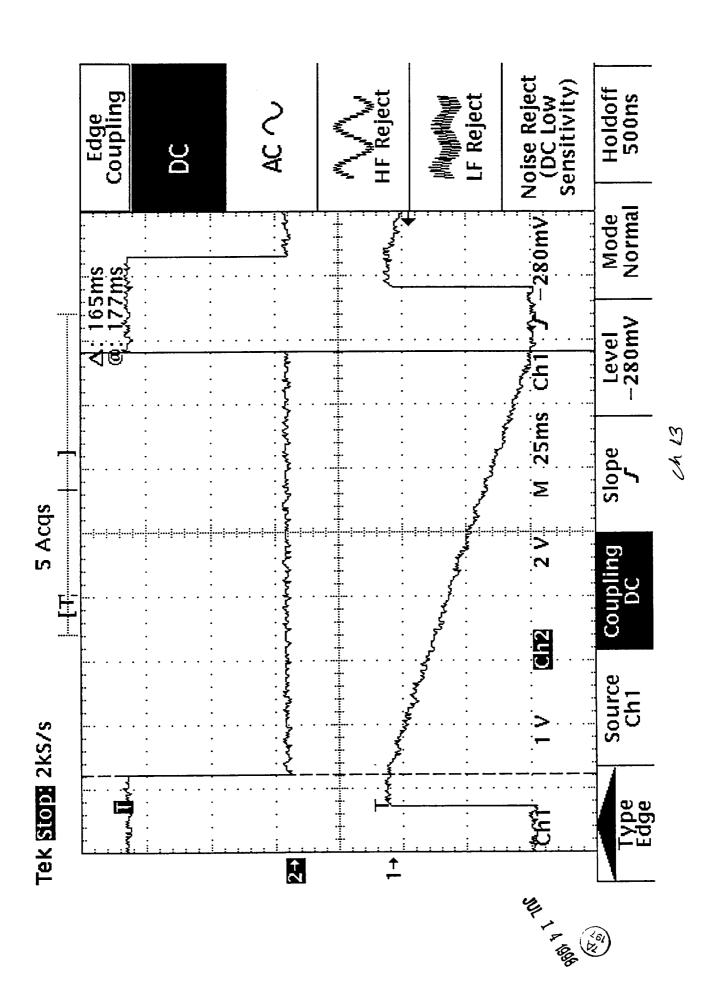


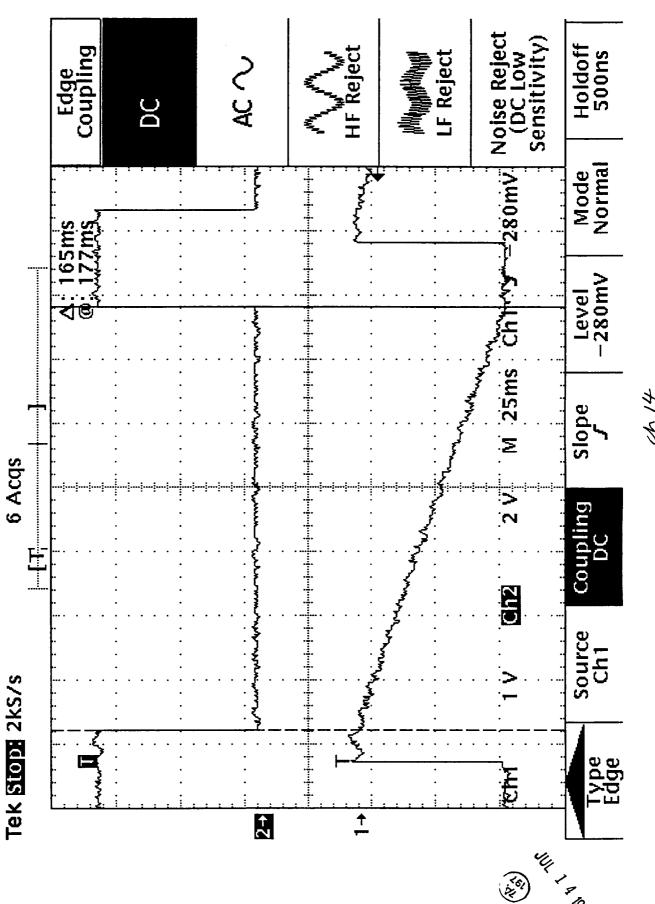




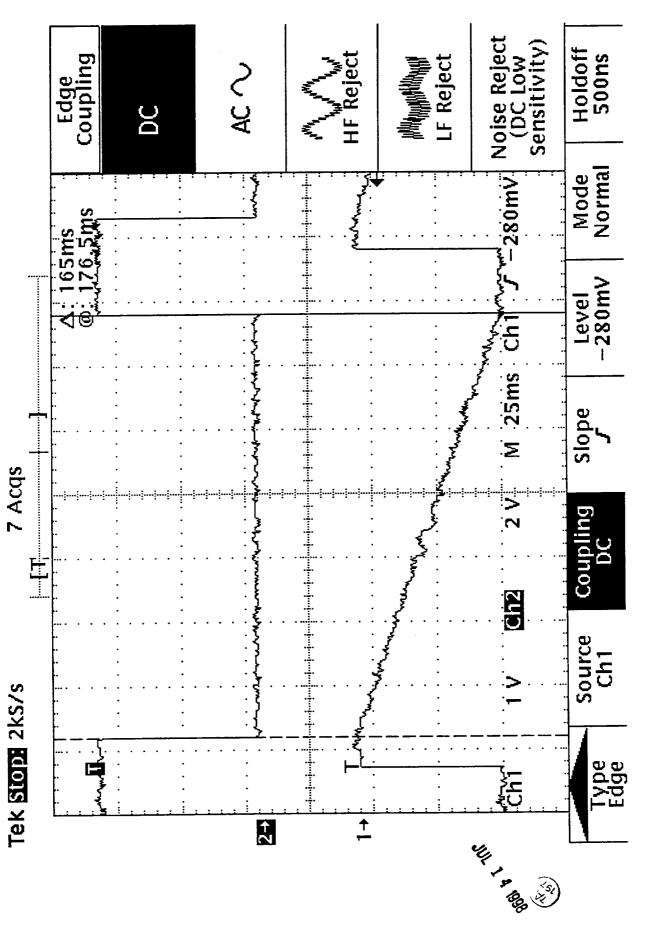








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TEST DATA SHEET NO. 18 Test Point Interface Test (GSE Modes) (Paragraphs 3.3.6.6 - 3.3.6.11)

					(GSE M	IODI	ES				
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	Y	ËS	۲	65	۲	E5	Y	(65	4	165	Ye	35
				DATA	A RE	VIEW	ÆD?	(YE	S/NC))		
Printout data	V	65	- {	(ES	Y	5	Y	ES	4	ES	4	25
Packet ID			,						•			
Packet Length												
Unit Serial Number				!								
Instrument Mode/Status												
Reflector Positions												
Radiometer Scene Data											·	
PRT Temperature Data	V	,		r				1	•	<i>y</i>	V	/
Engineering Data	YE	5	Y	ES	Y	55	Ye	15	Y	25	KE	S

EOS/AMSU-A1 System P/N 1356008 Shop Order: 29656]
Circle Test: (1st CPT Final CPT

Test Systems Engineer

Quality Control

4 1998 Date

TEST DATA SHEET NO. 19

Radiometer Functional Performance Test (PLO Frequency Measurements) (Paragraph 3.3.7.1)

	PLO FREQUEN	ICY MEASUREMENTS	
PLO	Measured Frequency (GHz)	Required Frequency (GHz)	Pass/Fail
#1	57.290322	57.290294 - 57.290394	PASS
# 2	57,290330	57.290294 - 57.290394	PASS

P = Pass F = Fail

EOS/AMSU-Al System P/N 1356008 Shop Order: 29856 S/N: 207
Circle Test: (1st CPT) Final CPT

Test Systems Engineer Date

Ustomer Representative Date

Quality Control

270
SCAN NUMBER
P1 14-JUL-98 00:35:13
3 A1-03 EL:EXE;35 GSE MODE 1 6-C-W P1 14-JUL-98 00:35:13 5 SCIENCE DATA ELEMENT 0000

8	8
EL EMENT	EL EMENT
CONTROL/STATUS	ENGINEERING
9	7

0
ELEMENT
ENGINEERING
_
7
_

			8	COMMANDS		PLIO POWER =	PLIC#1 [15]	
6	9] SCANNER AL-1 FOWER =	1-1 FOW	됐		8	COLD CAL FOSITION 1 =	YES [16]	
10]	10] SCANNER A1-2 FOWER =	1-2 FOW	# ₩		g	2 ==	NO [17]	
11]	11] ANTENNA FULL SCAN MODE	ULL SON	N MOD	[] [1]	8	e E	NO [18]	
12]	M	WARM CAL		11	8	COLD CAL POSITION 4 =	NO [19]	
13]	מ	COLD CAL	_	11	8	RESET CATH PROCESSOR	[20]	
14]	Ž	NADIR		II	8	GSE MODE	[21]	
NGR OK		FOWER (NS S		SCM	ON CHECKSIM IN DECS CALC DECS SA28	523 SA29 1045	
SELECT	SELECT BUTTON 3		מראבים		7	FIGURE [S] FORTH	L J NELLONG	

r l	VALLE	11645 1245 1265
00:35:20 PAGE	NOEE	SOS SOS SEL CONTROLLO SOS SOS SOS SOS SOS SOS SOS SOS SOS
	DESCRIPTION	REFLECTOR 1 FOSI REFLECTOR 1 FOSI REFLECTOR 2 FOSI 18 2 COLD CAL SAMPLE COLD CAL SAMPLE REFLECTOR 1 FOSI REFLECTOR 1 FOSI REFLECTOR 2 FOSI REFLECTOR 3 FOSI REFLECTOR 4 FOSI REFLECTOR 4 FOSI REFLECTOR 4 FOSI REFLECTOR 5 FOSI REF
TENCE DATA 14-JUL-98 10-10-10 CAL MODE	EL EMENT	CCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCCC
SCIENCE DATA 10-10-10 C	VALUE	00000000000000000000000000000000000000
A1_03 E1.EXE;35	r description	PACKET ID PACKET LENTH UNIT SERIAL NUMBER ONE INSTRUMENT MODE/STRUCK NEET, 1 POS 1 2ND LOOK NEET, 2 POS 2 2ND LOOK NEET, 2 POS 2 2ND LOOK NEET, 2 POS 2 2ND LOOK NEET, 2 POS 3 2ND LOOK NEET, 3 POS 3 2ND LOOK NEET,
EDS /	EL EMENT	11/11/11/11/11/11/11/11/11/11/11/11/11/

PAGE
00:35:20
14-JUL-98 MODE
SCIENCE DATA 10-10-10 CAL
El.EXE;35
1_03

2	VALUE	11046433 110564433 11056423 11056433 1105643 1	10000000000000000000000000000000000000	8527 8177 8527 8177 15895 16586
SCIENCE DATA 14-JUL-98 00:35:20 PACE 10-10-10 CAL MODE	DESCRIPTION	<u> </u>	CAL SAMPLE 2	ACTOR 1 POSITION 2 POSITION 2 POSITION 23 EN CAL SAMPLE
	EL EMENT	7222 7222 7222 7223 7223 7233 7233 7233 7233 7233 7234 7235 7235 7235 7335		758 RETLE 760 RETLE 764 REFL 766 WARM (
	VALUE			15281 14935 15281 14934 15857 16224 16540
ECS A1_03 E1.EXE;35 SC	ELEMENT DESCRIPTION	ET ECTOR 1 FOST EFT 1 FOS 4 2 SET 2 FOS 4 2 OS #6 SAMPLE SET ECTOR 1 FOST EFT ECTOR 1 FOST EFT ECTOR 2 FOST	REFLICATION A POSITION AND POSI	REFLECTOR 1 POST REFL 1 POS 6 2 REFL 2 POS 6 2 REFL 2 POS 6 2 POS #6 SAMPLE
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EL. EXE; 35 A1 03 **E**

15858 155308 155309 155 CH 14 CH 15 CH 15 CH 15 I POSITION 26 F ZND LOOK 8E m PAE 00:35:20 DESCRIPTION SCIENCE DATA 14-JUL-98 10-10-10 CAL MODE ELEMENT VALLE 16913 165545 1655466 1655466 1655466 165546 L POSITION 9 2 POSITION 9 2 POSITION 9 3 9 AND LOOK METE 9 GH SOSTITUTE SOSTIT DESCRIPTION RETLECTOR 1 FOR REFL 1 FOS 7 REFL 2 FOS 7 FOS 7 FOS 7 FOS 7 FOS 7 FOS 7 FOS 1 FOS 1 FOS 1 FOS 4 ROS SAMPLES 9 REFLECT REFL 1 REFL 2 ROS #6 EL EMENT



00:35:20 PA
14-JUL-98 MODE
SCIENCE DATA 10-10-10 CAL MC
EL.EXE;35
S A1_03

	回	woddaaoowaoccccaaaacaadowaoccccaaaaaaaaaa	<i>2222</i>
4	VALUE	16583 16583 16602 16602 16602 16623	852 817 852 817 1585
UL-98 00:35:20 PACE	DESCRIPTION	ECTOR 1 POSITION 27 L1 POS 27 2ND LOOK M OLL SWIELE 8 CH 12 L2 POS 27 2ND LOOK M OLL SWIELE 8 CH 12 L2 POS 28 2ND LOOK T 1 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 12 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 12 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 12 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 11 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 11 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 11 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 11 POS 28 2ND LOOK M OLL SWIELE 8 CH 12 CH 12 POS 28 2ND LOOK M OLL SWIELE 8	CTOR 1 POSITION CTOR 2 POSITION 1 POS 29 ZND 1 2 POS 29 ZND 1 CAL SAMPLE 9 (
14-J	ENT	REFT BY WARM WARM WARM WARM WARM WARM WARM WARM	REFLE REFL REFL WARM
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10] SCANNER A1-2 FOWER =	2 POWER =		8	2	[71] ON	7]
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3 ENGINEERING DATA

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		[9] SCANNER	10 SCANNER A1-2 FOWER =	11 JANTENNA	[12]	[13]	[14]	NGR OK	CET EPTP DETERMINED



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01:02:12 PAGE	TICN	#7 SWITE 17 GH 8 16543 GH 10 16430 GH 11 16076 GH 12 16364 GH 12 16364 GH 12 16364 GH 13 16364 GH 14 13 16365 GH 15 100 16430 GH 16 16565 GH 17 1607 GH 17 1607 GH 18 200 100 K GH 19 16565 GH 19 16665 GH 19 16434 GH 19 16665 GH 10 16665 GH 10 16667 GH 11 16667 GH 12 16566 GH 10 16667 GH 12 16566 GH 13 16566 GH 14 16566 GH 16 16667 GH 16 16667 GH 17 16667 GH 18 16566 GH 19 16667 GH 19 16669 GH 19 16667 GH 19 1667 GH 19 167 GH 19 167 GH 19 167 GH 19	
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_03 El.EXE;35	DESCRIPTION	PACKET ID UNIT SERIAL NUMBER UNIT SERIAL NUM	
ECS A1	ELEMENT	4.0 × 4.0 × 0.0 ×	1110



2 VALLE			15281 14934 15281 14934 15850 16192 16500
14-JUL-98 01:02:12 PAGE NT DESCRIPTION	RET ECIOR 1 POSITION 21 RET ECIOR 2 POSITION 21 RET 2 POS 21 2ND LOOK RET 2 POS 21 2ND LOOK RET 2 POS 21 2ND LOOK GSE #7 SAMPLE 21 GH 3 GGE #7 SAMPLE 21 GH 6 GGE #7 SAMPLE 21 GH 7 GGE #7 SAMPLE 21 G	RETECTOR 1 POSITION 22 15281 REFL 1 POS 22 2ND LOOK 15281 REFL 2 POS 22 2ND LOOK 14934 SSE #7 SWIPLE 22 CH 3 15847 SSE #7 SWIPLE 22 CH 3 15847 CH 4 16503 CH 6 16693 CH 10 16432 CH 11 16073 CH 12 16536 CH 13 16128 CH 13 16128 CH 14 16581	RETECTOR 1 POSITION 23 RETECTOR 2 POSITION 23 RETE 1 POS 23 2ND LOOK RETE 2 POS 23 2ND LOOK GSE #7 SAMPLE 23 CH 3 CH 5
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El.EXE;35
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01:02:12 PAGE	DESCRIPTION	### ### ##############################
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_ _03 E1.EXE;35 SC	DESCRIPTION	RETLECTOR 1 POSITION 7 REFL 1 FOS 7 2ND LOOK GEE #7 SAMPLE 7 CH 14 REFL 1 FOS 7 2ND LOOK GEE #7 SAMPLE 7 CH 3 GET #1 SAMPLE 10 CH 14 REFL 1 FOS 8 2ND LOOK GEE #7 SAMPLE 8 CH 15 REFL 1 FOS 8 2ND LOOK GEE #7 SAMPLE 8 CH 15 REFL 1 FOS 8 2ND LOOK GEE #7 SAMPLE 8 CH 15 REFL 1 FOS 9 2ND LOOK REFL 2 FOS 9 2ND LOOK GEE #7 SAMPLE 9 CH 3 AN A SAMPLE 9 CH 3 A A A SAMPLE 9 CH 3
ECS A1	EL EMENT	22222222222222222222222222222222222222

4,	VALUE	16503 1658303 165849 165949 165949 165949 165933 165933 165933 165933 165933 165933 165943 165933 165943 165933 16594 165943 165
14-JUL-98 01:02:12 PACE	DESCRIPTION	CH 5 16593 CH 6 6 16890 CH 7 15949 CH 10 6 6689 CH 11 6 6689 CH 12 6 6689 CH 12 6 6689 CH 12 6 6689 CH 13 6667 CH 14 6667 CH 15 6689 CH 16 6689 CH 16 6689 CH 17 6689 CH 18 6689 CH 19 6689 CH 19 6689 CH 19 6689 CH 10
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	ELEMENT	2476799999999999999999999999999999999999
SCIENCE DATA GSE MODE 7	VALLE	16594 165889 16584 165889 16584 165889 16588
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ECS A1_03	EL EMENT	494 494 496 496 496 496 496 496





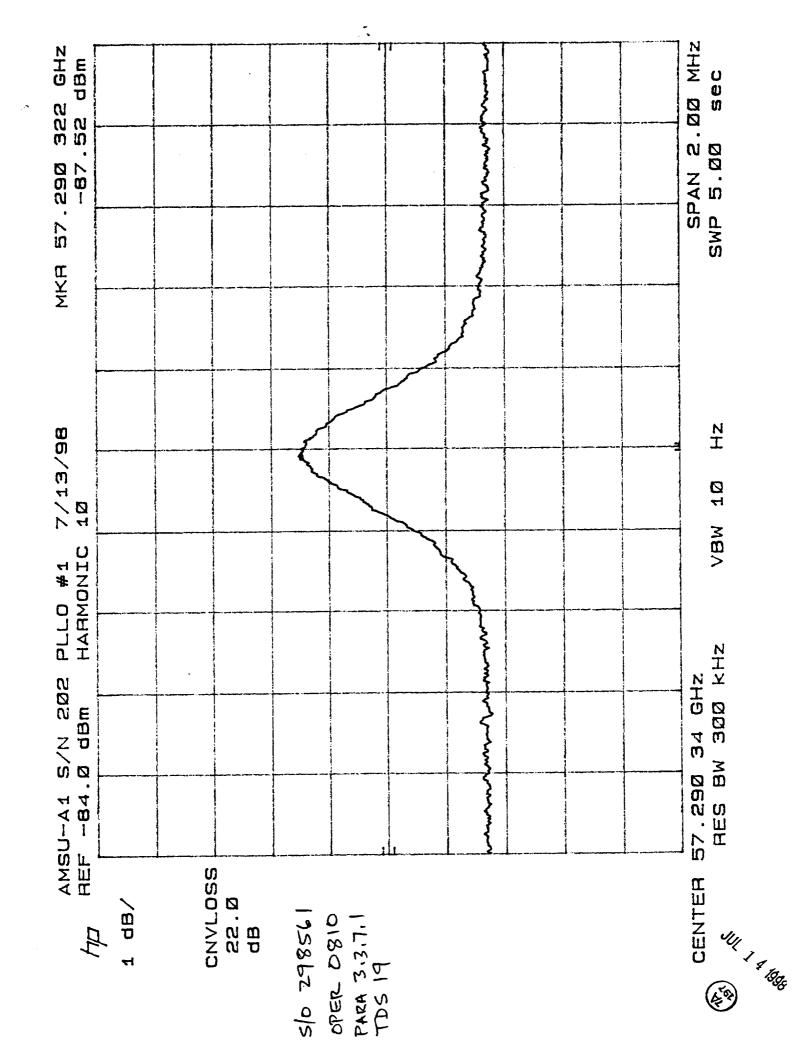
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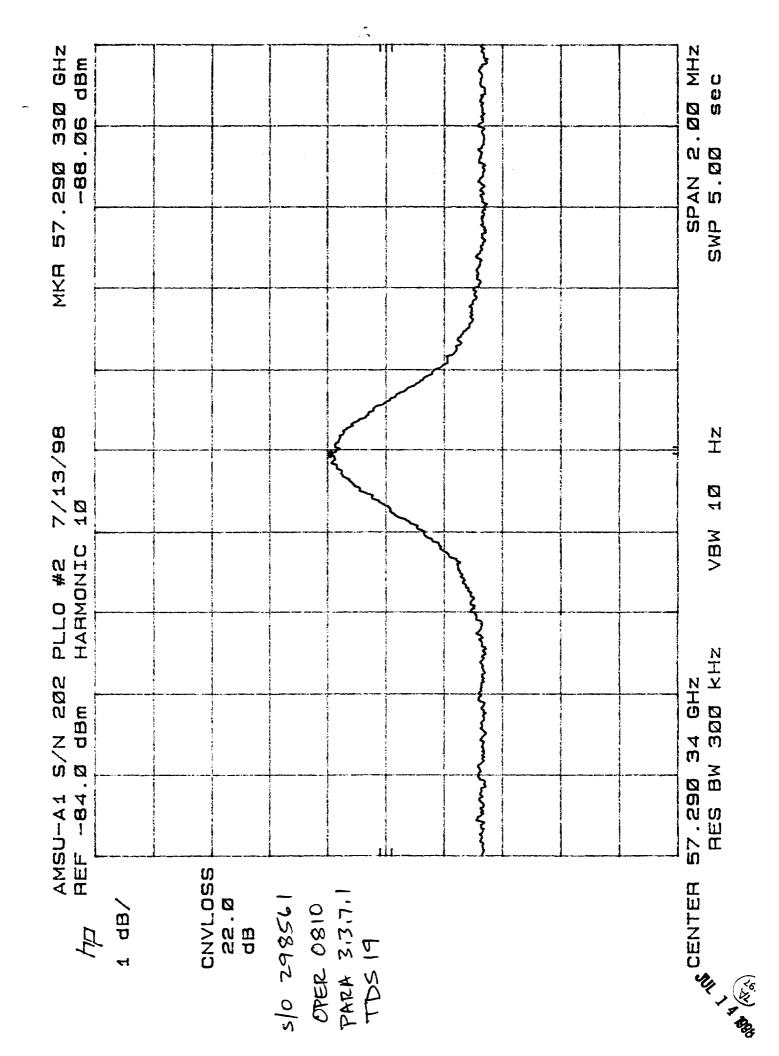
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ń	A1-1 DEG 42.0 43.0	4444 000000000000000000000000000000000	25 26 26 26	1		0/-0	270	22%	346	104wo	. 826
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EL.EXE;35	ekt temperatures iarcet			THERMOCOUPLE TEMPERATURES	ROTT	SHROUD		2		FIOW METER ET FIOW ME IER NO	FIOW METER DIATORS
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TEST DATA SHEET NO. 20 (Sheet 1 of 2)

Radiometer Functional Performance Test (Relative NEAT Measurements*) (Paragraph 3.3.7.2) PLO #1 Turned On

RELATIVE NEAT MEASUREMENTS (PLO #1 ACTIVE)							
Channel	Channel Average NE∆T Required**						
Number	for 5 Data Sets	ΝΕΔΤ	Pass/Fail				
	(K)	(K)					
3	.258	0.40	PASS				
4	.142	0.25	PASS				
5	.169	0.25	PASS				
6	.152	0.25	PASS				
7	.175	0.25	PASS				
8	.173	0.25	PASS				
9	.169	0.25	PASS				
10	. 226	0.40	PASS				
11	. 250	0.40	· PASS				
12	. 365	0.60	PA55				
13	. 503	0.80	PASS				
14	. 806	1.20	PASS				
15	.131	0.50	PASS				

P = Pass F = Fail

EOS/AMSU-A1 System P/N 1356008 Shop Order: 29856 S/N: 202
Circle Test: (1st CPT) Final CPT Sub CPT N/A LPT O/A

Test Systems Engineer Date

Quality Control

Date

Quality Control

^{*} Baseline data for acceptance tests. Use 1st CPT data along with specification value for pass/fail criteria.

^{**} For reference only

		AI-1 channels PLO#1 active RUN#1	
!	DELIPA T	3.022 22.117472 0.1377 0.220 0.250 0.654 0.138	[4] PRINT HISTOGRAM
19:14:27	GAIN	00000000000000000000000000000000000000	[4] F
	COLD COUNTS	15803.0 16123.0 13406.0 13313.0 13522.0 13511.0 13535.0 1401.0	3] PRINT RAW DATA
NAL TEST RESULTS 13-JUL-98	WARM COUNTS	15860.0 16182.0 16500.0 16807.0 16802.0 16422.0 16246.0 16458.0	[3] PRI
	WARM TEMP	22222222222222222222222222222222222222	[2] PRINT SCREEN
A1 I	Ð	64700080011111111111111111111111111111111	[2]

REIURN [1]

SELECT BUTTON 2



		AI-I channels Pro#1 Active Run #2	
1			₩.
	DELITA T	2.808 1.518 0.168 0.168 0.144 0.230 0.336 0.336 0.346 0.168 0.168	PRINT HISTOGRAM
19:16:28	GAIN	00000000000000000000000000000000000000	
	COLD COUNTS	15805.0 16124.0 13319.0 13344.0 13552.0 13552.0 1420.0 14395.0	[3] PRINT RAW DATA
NAL TEST RESULTS 13-JUL-98	WARM COUNTS	15860 161820 1650320 15894 163461 163461 164590 164590 164590	[3] PRID
FUNC	WARM TEMP	22222222222222222222222222222222222222	2] PRINT SCREEN
A1 E1.EXE;35	B		[2]

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTION 2

THE 1 TOOM SE

	AI-I CHANNELS PLOHI ACTIVE RUN #3
DELITA T	2.874 1.962 1.962 1.962 0.220 0.220 0.5098 0.5098 0.133
GAIN	000000000000000000000000000000000000000
COLD COUNTS	15805.0 16122.0 16446.0 13315.0 13315.0 13524.0 13524.0 13415.0 13763.0
WARM COUNTS	15859.0 16182.0 16805.0 16805.0 15894.0 16443.0 16247.0 16247.0 16463.0
WARM TEMP	00000000000000000000000000000000000000
끙	w4r0r0001111111

19:19:08

A1 FUN. JAPL TEST RESULTS 13-JUL-98

RETURN [1]

[4] PRINT HISTOGRAM

[3] PRINT RAW DATA

[2] PRINT SCREEN

SELECT BUITON 2



AI-I CHANNELS PLO # I ACTIVE RUN #4

DELITA T	2.779 1.966 0.159 0.159	6.5373 0.227 0.2860 0.2860 0.2860 0.2860 0.2860 0.2860 0.2860 0.2860	0.807 0.136	PRINT HISTOGRAM
GAIN	1111000 0001000 0000000000000000000000	0.001 0.081 0.083 0.081	0.085 0.124	[4] F
COLD COUNTS	15804.0 16123.0 16447.0 13331.0	18946.0 13741.0 13537.0 13557.0	13774.0 13774.0 14388.0	NT RAW DATA
WARM COUNTS	15861.0 16182.0 16501.0 16804.0 15894.0	164443.0 16443.0 15366.0 15966.0	16012.0 16463.0 16159.0	[3] PRINT
WARM TEMP	301.07 301.07 301.07 299.31 299.31		299.31 299.31 299.31	PRINT SCREEN
ਲ	W4N0 C0	, 12156	1 4 7	2] F

19:21:01

A1 FUN. JABL TEST RESULTS E1.EXE;35 REIURN [1]

SELECT BUITON 2

AI-I channels PLO # 1 ACTIVE RUN # S

DELITA T	22.100 22.1000 22.200 22.200 23.78 24.43 25.45 2	PRINT HISTOGRAM
GAIN	11100010000000000000000000000000000000	_
COLD COLNIS	15803.0 16425.0 13897.0 13311.0 13352.0 13314.0 13408.0 13757.0	NI RAW DAIPA
WARM COUNTS	15862 1618320 165000 165000 158920 164430 164430 164430 164650 1646120	[3] PRINT
WARM TEMP	3,22,22,22,22,22,22,22,22,22,22,22,22,22	PRINT SCREEN
픙	64600 80012847	•

19:23:58

A1 FUN. JABL TEST RESULTS 13-JUL-98 RETURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTION 2

401 / 4 1896 (pt)

	DELITA T	0.00.02.00.02.0.0.4.0.0.1 641.10.00.01.02.0.0.0.0.0.0.0.0.0.0.0.0.0.	PRINT HISTOGRAM
18:55:28	GAIN	111111110000	[4]
	COLD COLNIS	13259.0 13263.0 13684.0 15896.0 16444.0 16375.0 1629.0 16484.0	PRINT RAW DATA
TEST RESULTS 13-JUL-98	WARM COUNTS	15859.0 16185.0 16502.0 16811.0 15899.0 16447.0 16378.0 16264.0 16481.0	[3] PRI
FUNC. JAPL	WARM TEMP	201330 201330 2013330 2013332 20132 20132 2013	PRINT SCREEN
A1 E1.EXE;35	Ħ	6470C805HZZZZ	[2]

A1-2 channels Plo4 (active Ron #1

RETURN [1]

[5] RINT DISTRIBUTION GRAPH SELECT BUTTON 2

MI 1 1 1898

A1-2 channets Plot 1 ACTIVE RUN #2

DELITA T	0.277 0.139 0.153	2.323 0.203 1.981		4: /25 6:084 1:081	PRINT HISTOGRAM
N N	0.086	0.000	11111 000 000	9000 1444	
COLD COUNTS	13299.0 13304.0 13724.0	15896.0 13636.0 14444.0	16372.0 15975.0	16225.0 16475.0 16163.0	NT RAW DATA
WARM COUNTS	15861.0 16183.0 16500.0	15898.0 16602.0	16376.0 15976.0	16255.0 16025.0 16477.0 16165.0	[3] PRINT
WARM TEMP	301.35 301.35 301.35	299.31 299.31 301.35	2000 2000 311 311 311 311 311 311 311 311 311	299.31 299.31 299.31	PRINT SCREEN
ਲ	w4r0	00 م	, 212,	32 4 5	-

18:58:24

A1 FUN. JAML TEST RESULTS 13-JUL-98 RETURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2

411 / 5 1686 (EQ) Al-2 channels PLO# 1 ACTIVE RUN#3

DELITA T	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	PRINT HISTOCRAM
GAIN	00000000000000000000000000000000000000	[4] I
COLD COUNTS	13312.0 133212.0 137381.0 15894.0 15894.0 16443.0 16250.0 16468.0	nt raw data
WARM COUNTS	15860.0 16181.0 16500.0 16809.0 16603.0 16445.0 16375.0 16256.0 16471.0	[3] PRINT
WARM TEMP	28888888888888888888888888888888888888	PRINT SCREEN
ਲ	646000001111111111111111111111111111111	[2]

19:01:29

A1 FUN. JAPL TEST RESULTS 13-JUL-98 RETURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2

Al-2 channels PLO# 1 PRTIVE PLO# 1

DELTA T	0.00 0.22 0.01 0.01 0.01 0.02 0.03 0.03 0.03 0.03 0.03 0.03 0.03	PRINT HISTOGRAM
GAIN	000000000000000000000000000000000000000	[4]
COLD COLNIS	13337.0 13345.0 13760.0 15894.0 15894.0 16442.0 16371.0 16017.0 16471.0	PRINT RAW DATA
WARM COUNTS	15859.0 16180.0 16499.0 15896.0 16602.0 16445.0 16253.0 16469.0	[3] PRI
WARM TEMP	301.26 301.26 301.26 301.26 301.27 2999.37 2999.37 2999.37 2999.37	PRINT SCREEN
Ð	647078901177747	[2] F

19:03:21

A1 FUNC NAL TEST RESULTS 13-JUL-98 REIURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2



A1-2 dannels PLO# 1 ACTIVE ROW#5

DELITA T	0.262 0.152 0.176 2.422 2.422 1.899 1.899 10.497 1.220	PRINT HISTOGRAM
SP IN	00000000000000000000000000000000000000	[4]]
COLD COUNTS	13352.0 13357.0 13773.0 16803.0 16441.0 16369.0 16248.0 1613.0	VI RAW DATA
WARM COUNTS	15858 161758.0 164789.0 16601.0 16444.0 16373.0 16252.0 16466.0	[3] PRINT
WARM TEMP	301 301 301 301 301 301 302 303 303 303 303 303 303 303 303 303	PRINT SCREEN
ਲ	K420C8004444	2] P

19:06:58

A1 FUN. JAPL TEST RESULTS E1.EXE;35 [5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2

RETURN [1]

VU 1 4 1998

Al-1 chamels	PLO#2 ACTIVE	/# NOS

DELITA T	3.181 0.1392 0.177 0.173 0.226 0.3696 0.3699 0.145	PRINT HISTOGRAM
GAIN CAIN	000000000000000000000000000000000000000	[4]F
COLD COLNIS	15799.0 16109.0 16378.0 13485.0 13732.0 13732.0 13579.0 14524.0	nt raw data
WARM COUNTS	15859.0 16174.0 16438.0 16787.0 16599.0 16434.0 15987.0 15987.0	[3] PRINT
WARM TEMP	301.45 301.45 301.45 2999.27 2999.27 2999.27 2999.27 2999.27	PRINT SCREEN
Ŧ	646968901177 846968901177	[2]

20:28:24

A1 FUN JAL TEST RESULTS 13-JUL-98 SELECT BUILDN 2

RETURN [1]



		AI-1 channels Pro#2 Acture Run#2	
10	DELIFA T	3.191 1.794 2.002 0.183 0.183 0.222 0.275 0.275 0.366 0.496 0.134 0.134	
20:30:56	GAIN.	1.000 1.0000 0.0080 0.0086 0.0087 0.133	
	COLD COUNTS	15797.0 16108.0 16376.0 14062.0 13455.0 13893.0 13701.0 13698.0 13553.0 13553.0	
TEST RESULIS 13-JUL-98	WARM COUNTS	15859.0 16174.0 16787.0 16787.0 15868.0 16349.0 16334.0 16438.0 16438.0 16144.0	
AL FUN. JAL 7 EXE;35	WARM TEMP	301.47 301.47 301.47 299.25 299.25 299.25 299.25 299.25 299.25 299.25 299.25 299.25 299.25	
A E1.EXE;3	¥	64700080511511411 C C C C C C C C C C C C C C C C C	

RETURN [1] SELECT BUTTON 2



		AI-I channels PLO# 2 ACTIVE RUN#3	
	DELITA T	000 3.006 000 2.048 080 0.172 091 0.191 000 2.217 082 0.244 088 0.256 0.086 0.399 0.090 0.399 0.097 0.885 1.33 0.149	
20:32:49	GAIN	1.000 1.000	•
	COLD COLNIS	15797.0 16109.0 16377.0 13459.0 13459.0 13698.0 13698.0 13696.0 13552.0 13552.0 14495.0	
NAL TEST RESULTS 13-JUL-98	WARM COUNTS	15858.0 16174.0 16436.0 16787.0 15866.0 16434.0 16369.0 15990.0 16441.0	
z Z	WARM TEMP	301.48 301.48 301.48 301.48 299.22 299.22 299.22 299.22 299.22 299.22 299.22	ייייי זייייי
A1 FEL.EXE;35	8	. 4470789511111141 - c	

RETURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2

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Al FIN. ANL TEST RESILIS

EL.EXE;35

CH WARM TEMP WARM COUNTS COLD COUNTS CAIN DELITA T

3 301.49 15860.0 15798.0 1.000 1.742

5 301.49 16436.0 16109.0 1.000 1.742

6 299.18 16787.0 14023.0 0.079 0.178

8 301.49 16601.0 16535.0 1.000 2.000

9 299.18 16435.0 13857.0 0.085 0.239

11 299.18 1635.0 13664.0 0.085 0.239

12 299.18 16235.0 13664.0 0.085 0.329

13 299.18 16442.0 13874.0 0.085

14 299.18 16442.0 13874.0 0.085

15 299.18 16442.0 13874.0 0.085

15 299.18 16442.0 13874.0 0.085

15 299.18 16442.0 13874.0 0.085

15 299.18 16442.0 13874.0 0.085

AI-I Chounels PLOHZ ACTIVE RUN #4

REIURN [1]

[4] PRINT HISTOGRAM

[3] PRINT RAW DATA

[2] PRINT SCREEN

[5] PRINT DISTRIBUTION GRAPH SELECT BUTTON 2

VU 1 1 1000

	DELITA T	2.145 2.1245 0.1542 0.1542 0.242 0.3375 0.3375 0.489	PRINT HISTOGRAM
20:37:22	GAIN	0.000000000000000000000000000000000000	[4]
•	COLD COLNIS	15799.0 16107.0 16377.0 13428.0 13579.0 13679.0 13575.0 13534.0 13534.0	PRINT RAW DATA
TEST RESULTS 13-JUL-98	WARM COUNTS	15861.0 16174.0 16437.0 15864.0 16434.0 16368.0 15951.0 15951.0 16442.0	[3] PRI
A1 FUN. JNAL 7,35	WARM TEMP	222999144445050 222999145050 2229991445050 223991445050 2414445050	PRINT SCREEN
A1 1 E1.EXE;35	픙	K4207803113241	[2]

AI-I dannels PLO#2 ACTIVE RUN #5

REIURN [1]

[5] PRINT DISTRIBUTION GRAPH SELECT BUTION 2

JU 1 4 100gg

Channel Identification Test (para 3.3.8)



Channel Number	Antenna Location	Sweeper Freq. Setting (GHz)	Polarization (H/V)	Radiometric Data -Counts A Counts	Channel Verified (Yes/No)
3	A1-2	50.35	V	16666	YES
4	A1-2	52.85	v	9162	YES
5	A1-2	53.70	Н	15635	YES
6	A1-1	54.45	Н	15681	YES
7	A1-1	54.99	V	14598	YES
8	A1-2	55.55	Н	15873	YES
9	A1-1	57.34	Н	16056	YES
10	A1-1	57.50	Н	15953	YES
11	A1-1	57.564	Н	15903	YES
12	A1-1	57.59	Н	15557	YES
13	A1-1	57.602	Н	15640	YES
14	A1-1	57.608	Н	15237	YES
15	A1-1	89.55	V	16252	YES 202

COS/AMSU-AI System P/N 1356008 Shor order 298561

le CPT Final CPT Sub CPT _____ / PT ______...

ChIDTest.doc

92

COLD CAL MODE ELEMENT 0000 00 ELEMENT A1-03 E1.EXE;35 SCIENCE DATA CONTROL/STATUS EOS [5] 9

ELEMENT ENGINEERING [7]

00

RADIOMETRIC DATA

30 BEAM POSITION

6450

ON CHECKSUM IN F459 CALC F459 SA28 SCREEN ONLY [2] PRINT [3] FULL [22] DOWN

69 SA29 137 [1] RETURN

N SELECT BUTTON

POWER

[21] UP ENGR OK

166									318 PN	
SCAN NUMBER									160 SA29	
COLD CAL MODE P1 14-JUL-98 15:38:38 8	ELEMENT 00	ELEMENT 00	RADIOMETRIC DATA	BEAM POSITION 30	CH DATA CH DATA CH DATA	3 32767 8 16802 13 16914 4 16554 9 16633 14 17335 5 16997 10 16708 15 16424	16998 11 16138 12	[22] DOWN	ON CHECKSUM IN AAIF CALC AAIF SA28 SCREEN ONLY [2] BRINT [3] FILL.	SCREEN ONLY [2] TATAL
					O				POWER	2
EOS A1-03 E1.EXE;35 [5] SCIENCE DATA	[6] CONTROL/STATUS	[7] ENGINEERING						[21] UP	ENGR OK	SELECT BUTTON 2



m									
178									343 URN
TOMBER									172 SA29 [1] RET
SCAN NUMBER									
Pl 14-JUL-98 15:40:14									BB51 SA28 FULL
4 -JUL-98			ATA		DATA	16915 17351 16446			BB51 CALC BB51 PRINT [3] FULI
7			C D	30	CH	113 144 154			BB
			RADIOMETRIC DATA	BEAM POSITION 30	DATA	16820 16642 16733	17051	OWN	$\begin{bmatrix} UM & IN \\ 2 & J \end{bmatrix}$
MODE 0000	00	00	RADI	M Pos	CH	800-	12	[22] DOWN	HECKS
COLD CAL MODE ELEMENT 0000	ELEMENT	ELEMENT		BEA	DATA	16080 155814 16996	16106	[2:	ON CHECKSUM SCREEN ONLY [;
COL	ELE	ELE			CH	W4104			OS
A1-03 E1.EXE;35 SCIENCE DATA	/sratus	RING							POWER
A1-03 E] SCIENCE	6] CONTROL/STATUS	7] ENGINEERING						[21] UP	ENGR OK
EOS 1	9]	[7						[2]	ENGF

SELECT BUTTON 2



ER 208									29 402 PETITEN	
SCAN NUMBER									202 SA29	- - -!
P1 14-JUL-98 15:44:14 SYNC SEQUENCE BYTE 1			RADIOMETRIC DATA	BEAM POSITION 30	CH DATA	13 17032 14 17334 15 16349		[22] DOWN	ADD99 CALC AD99 SA28 PRINT [3] FULL	
MODE 1	00	00			CH DATA	8 16739 9 16618 10 16637			IECKSUM IN	
COLD CAL ELEMENT	ELEMENT	ELEMENT			CH DATA	3 16033 4 16497 5 32767			ON CHECKSI SCREEN ONLY	
A1-03 E1.EXE;35 SCIENCE DATA	6] CONTROL/STATUS	7] ENGINEERING						[21] UP	OK POWER	SELECT BUTTON 2
EOS [5]	[9]	[7]						[21	ENGR OK	SEL



214									413	OKIN
SCAN NUMBER 00000000									207 SA29 4	1 1 KE1
P1 14-JUL-98 15:45:02 SYNC SEQUENCE BYTE 1			IC DATA	30	CH DATA	13 16864 14 17378 15 16387			N C81D CALC C81D SA28	FALINI (3) FOLD
MODE E	00	0.0	RADIOMETRIC DATA	BEAM POSITION 30	CH DATA	8 16768 9 16663 10 16672	1—] DOWN	CHECKSUM IN	N
COLD CAL ELEMENT	ELEMENT	ELEMENT		BEAM	CH DATA	3 16045 16515 3 16949	,—	[22]	ON CH	SCREEN O
A1-03 E1.EXE;35 SCIENCE DATA	6] CONTROL/STATUS	7] ENGINEERING						0.	POWER	SUTTON 2
EOS A1-	[6] CON	[7] ENG						[21] UP	ENGR OK	SELECT BUTTON 2

(1) 1 1886

230								446	
SCAN NUMBER 00000000								224 SA29	יייאי ו יי
P1 14-JUL-98 15:47:10 SYNC SEQUENCE BYTE 1			IC DATA	30	CH DATA	13 19987 14 20696 15 16350		N 6815 CALC 6815 SA28	TOTIC I TOTI
MODE 1	00	00	RADIOMETRIC DATA	BEAM POSITION 30	CH DATA	8 16736 9 19242 10 19793	[22] DOWN	ON CHECKSUM IN 6	
COLD CAL	ELEMENT	ELEMENT		BEAN	CH DATA	3 16113 16483 16869	[22	ON CE	O CREEN
A1-03 E1.EXE;35 SCIENCE DATA	6] CONTROL/STATUS	7] ENGINEERING] UP	OK POWER	SELECT BUTTON 2
EOS [5]	[9]	[7]					[21] UP	ENGR OK	SELE



220									425 URN
SCAN NUMBER									213 SA29 [1] RET
P1 14-JUL-98 15:45:50 SYNC SEQUENCE BYTE 1			RADIOMETRIC DATA	N 30	ra ch data	57 13 16888 29 14 17334 31 15 16422	93		IN 12D9 CALC 12D9 SA28] PRINT [3] FULL
MODE 1	00	00	RADIOMET	BEAM POSITION 30	CH DATA	8 3276 9 1662 10 1670 11 1666	21	22] DOWN	CHECKSUM ONLY [2
COLD CAL	ELEMENT	ELEMENT		BEAN	CH DATA	2 16050 165229 169260 169269	1608	[22	ON CH SCREEN C
EOS A1-03 E1.EXE;35 [5] SCIENCE DATA	[6] CONTROL/STATUS	[7] ENGINEERING						[21] UP	ENGR OK POWER SELECT BUTTON 2

P1 14-JUL-98 15:31:34 COLD CAL MODE ELEMENT 0000

00 ELEMENT CONTROL/STATUS 9

A1-03 E1.EXE;35 SCIENCE DATA

EOS [5]

ELEMENT ENGINEERING 7]

RADIOMETRIC DATA

30 BEAM POSITION

DATA CH DATA CH DATA

W4100C

[22] DOWN

ON CHECKSUM IN FOOD CALC FOOD SA28 SCREEN ONLY [2] PRINT [3] FULL

106 SA29 211 [1] RETURN

7 SELECT BUTTON

POWER

[21] UP

ENGR OK

UII 1 4 1890

9 ~

00 ELEMENT CONTROL/STATUS 9

EOS [5]

00 ELEMENT ENGINEERING 7] RADIOMETRIC DATA

BEAM POSITION 30

DATA CH DATA GH DATA

[21] UP

ENGR OK

[22] DOWN

ON CHECKSUM IN BD55 CALC BD55 SA28 SCREEN ONLY [2] PRINT [3] FULL

122 SA29 243 [1] RETURN

N SELECT BUTTON

POWER

A1-03 E1.EXE;35 SCIENCE DATA

EOS]

00

ELEMENT

CONTROL/STATUS

6]

00

ELEMENT ENGINEERING [7] RADIOMETRIC DATA

30 BEAM POSITION DATA CH CH DATA DATA

[22] DOWN

POWER

[21] UP

ENGR OK

ON CHECKSUM IN 8FEF CALC 8FEF SA28 SCREEN ONLY [2] PRINT [3] FULL

132 SA29 262 [1] RETURN

 α SELECT BUTTON

VU 7 28 1884

157									299 JRN	
SCAN NUMBER									150 SA29 299 [1] RETURN	•
SCAN										•
P1 14-JUL-98 15:37:26									FOOD CALC FOOD SA28	
-JUL-			TA		DATA	14946 32767 16452	1 1 1		OD CAL	
1 14			C DZ	30	CH	114 844	1		N FO	; !
Д			RADIOMETRIC DATA	BEAM POSITION 30	DATA	16821 12593	13452 14394	DOWN	UM IN	1
MODE 3000	00	00	RADI	A POS	CH	ω <i>ω</i> (777	2] D	HECKS	
COLD CAL MODE ELEMENT 0000	ELEMENT	ELEMENT		BEAN	DATA	16060	19317	[22]	ON CHECKSI	
	ELE	ELEI			CH	W417			NO.)
EXE;35 DATA	STATUS	ING							POWER	1 2
EOS A1-03 E1.EXE;35 [5] SCIENCE DATA	[6] CONTROL/STATUS	7] ENGINEERING						[21] UP	ENGR OK	SELECT BUTTON 2
soa]	9]	7						[2:	ENGI	SE

VU 1 1 1888

251								487 FURN
SCAN NUMBER								244 SA29 [1] RETU
P1 14-JUL-98 15:49:58 SYNC SEQUENCE BYTE 1			RIC DATA	N 30	A CH DATA	4 13 16846 9 14 17303 4 15 32767 3		IN 7187 CALC 7187 SA28] PRINT [3] FULL
, MODE	00	00	RADIOMETRIC DATA	1 POSITION 30	CH DATA	8 16774 9 16619 10 16694 11 16643 12 16843	[22] DOWN	0
COLD CAL	ELEMENT	ELEMENT		BEAM	CH DATA	3 16014 4 16511 5 16873 7 16988 16074	[22	ON CHECKSUM SCREEN ONLY [
EOS A1-03 E1.EXE;35 [5] SCIENCE DATA	CONTROL/STATUS	ENGINEERING] UP	NGR OK POWER SELECT BUTTON 2
EOS [5]	[9]	[7]					[21] UP	ENGR OK SELECT

AMSU-A1-1 MOTOR - BEAM POINTING POSITIONS FOR INFO ONLY

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	HEX Values	0540	E300	E560	E7C0	EA1C	EC7C	EEDC	F138	F398	F5F8	F854	FAB4	5044	FF70	200	0100	0420	2890	OSEC	UB4C	0DA8	1008	1268	14C4	1724	1984	1BE0				255C	27BC	3800	3E2C	3F5C	408C		
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`	Relative Position	200	330 -1856	1204	1011	-1332	-1401-	-1249	1097	-946	-794	-642	-491	-339	-187	-36	116	268	419	571	723	874	1026	1178	1329	1481	1633	1784	1936	2088	2230	2209	2331	2000	3828	3979	4055	4131	8530
\	Relative Differential	rosition	0000	7,500	761	.	151	152	152	151	152	152	151	152	152	151	152	152	151	152	152	151	150	152	15.	157	132	154	152	160	132	151	7¢L	152	3490	3641	3717	3793	8192
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Page 1

Decimal Printout 14168 15230 15382 15533 15685 15837 15988 1424 1576 1728 1879 3474 363 818 FOR INFO ONLY HEX Values 05AC 0808 0A68 0CC8 0CC8 13E4 1640 1BA0 1BBC 1FBC 221C 3648 38A4 F9D0 FC30 00EC 034C E6DC E93C EB98 EDF8 F2B4 F514 F774 E47C F058 0,: 00 0 0 0 10 0 0 ¥ • -AMSU-A1-2 MOTOR - BEAM POINTING POSITIONS Binary Weights o Relative Position 1273 1424 1576 1728 2031 2183 3474 3701 3777 -1609 -2216 -2064 -1912 -1457 -1761 Differential Relative Position 3717 3793 8192 2200 152 152 151 CC 3 Position NADIR 13 2 4 15 6 ထ G ည ဖ

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DOCUMENT APPROVAL SHEET



TITLE			BOCOMENT NO.	
Process Specification			AE-26156/9	
EOS/AMSU-A1 System Compreher	sive and Limited Po	erformance Tests	18 June 1998	
Test Procedure				ļ
,				
INPUT FROM: DATE	CDRL:	SPECIFICATION ENGINEER:		DATE
P. Patel	409			
CHECKED BY:	DATE	JOB NUMBER:		DATE
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ARRENTED GLOVATURES	·		DEPT. NO.	DATE
APPROVED SIGNATURES			<u> </u>	
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Specifications Engineering (J. Ki	rk) 11 V Davie	nwws /	0031	0/21/
	0 (Patel		
Product Team Leader (A. Nieto)	8-K.6	ares	8341	6/30/48
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Contamo Engineer (D. Dlott)	K UIX KL	\mathcal{H}	8311	6/30/98
Systems Engineer (R. Platt)	7 70 10		00	
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Safety (W. Neighbors)	- Justin	in the	8331	6/30/10
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Design Assurance (E. Lorenz) _1	Allet Sal	indin_	8331	6/30/98
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(RM7		7831	6-30-98
Quality Assurance (R. Taylor)	! In Jugan		7031	
	/	1		4 4 -
Technical Director/PMO (R. Hau	erwaas) <u>4.00 %</u>	auenous	4001	6/30/98
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Configuration Management (J. C	avanaugh)	Mulling	8361	7/1/98
Configuration Management (6: 6	Availadgil)	Con Consumo		''
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This Revision incorporated appro	oved Macter Mark	Line dated 18 June		
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By my signature, I certify the above document h	as been reviewed by me a	nd concurs with the technical	İ	
requirements related to my area of responsibility	<u>'. </u>			<u> </u>
RELEASE (Data Center) FINAL				
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P.O. Box 296 Azusa, CA 91702-0296 CAGE/Facility Ident: 70143

GENCORP AEROJET

AE-26600B 23 June 1998

Superseding AE-26600A 15 January 1998

S.O. # 29856| 15J Op. 810 AE-26156/9 para 3.3.5./.

PROCESS SPECIFICATION

Earth Observing System/ Advanced Microwave Sounding Unit-A (EOS/AMSU-A) Firmware Test Procedures (CDRL: 415)

Contract No. NAS5-32314

Prepared for: NASA/Goddard Space Flight Center Greenbelt Road Greenbelt, MD 20771

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1. INTRODUCTION

- 1.1 Identification. These are the Firmware Test Procedures for the Instrument Control and Command and Data Handling firmware. This document is submitted in response to Contract NAS 5-32314, CDRL 415.
- 1.2 Scope. This document describes the steps necessary to test the Instrument Control and Command and Data Handling firmware and establish that all requirements in the Firmware Requirements Specification have been satisfied.
- 1.3 Purpose and objectives. The purpose of this document is to provide the steps to verify that the requirements of the firmware have been accomplished. Through the use of tables, procedure steps, and test data sheets, the user of this document will be able to determine that the requirements of the specification for the firmware have been correctly implemented as described in the Firmware Test Plan.
- 1.4 Document status and schedule. The requirements traceability and expected results tables are complete and are included in this revision.
- 1.5 Documentation organization. The EOS/AMSU-A Software Documentation Tree is as shown in Figure 1.

Document	Document No.	CDRL No.
Software Management Plan	10339	800
Acquisition Activities Plan	10341	508
Software Standards and Procedures		402
Software Assurance Plan	10428	309
Configuration Management Plan	9803	005
Software Product Specifications	•	306
Software Concept Document	10432	306-1a
Software Requirements Specification	10457	306-2a
Software Architectural Design	10464	306-3a
Software Detailed Design Document	10463	306-5a
Firmware Support Manual	10466	306-7
Version Description Document	10467	306-8a
User's Guide	10443	306-10a
Firmware Product Specification		306
Firmware Concept Document	10436	306-1b
Firmware Requirements	10458	306-2b
Firmware Architectural Design	10460	306-3b
Firmware Detail Design Document	10387	306-5b
Firmware Version Description	10976	306-8b
Software/Firmware Test Plan	10369/10352	033
Software Test Procedures	AE-26602	415
Software Test Reports	10975	217
Firmware Test Procedures	AE-26600	415
Firmware Test Reports	10974	217

Figure 1. EOS/AMSU-A Software Documentation Tree

2. RELATED DOCUMENTATION

- 2.1 Parent documents. The firmware test plan is the parent document to this test procedure as indicated in Figure 1.
- Applicable documents. The following documents are referenced or applicable to these test procedures. Unless otherwise specified, the latest issue is in effect.

NATIONAL AERONAUTICS And SPACE ADMINISTRATION

NASA-DID-A200	Test Procedures Data Item Description
GSFC 422-10-	Earth Observing System (EOS) Instrument Project Software Acquisition Management Plan
422-11-12-01	General Interface Requirements Document (GIRD)
AEROJET DOCUMENTS	
Report 10352	EOS/AMSU-A Firmware Test Plan
Report 10458	Earth Observing System/Advanced Microwave Sounding Unit-A (EOS/AMSU-A) Firmware Requirements
Report 10974	Earth Observing System/Advanced Microwave Sounding Unit-A (EOS/AMSU-A) Firmware Test Report

2.3 Information documents

EOS/AMSU-A Project Plan, Including Project Organization Chart, Report 10345 WBS Diagram, and Task Description

(Copies of Aerojet documents should be obtained from Aerojet, CAGE 70143, P.O. Box 296, Azusa, California 91702-0296.)

3. TEST IDENTIFICATION AND OBJECTIVE

The tests described in this document are the EOS/AMSU-A test of the Instrument Control and Command and Data Handling firmware. The firmware requirements are specified in Report 10458, EOS/AMSU-A Firmware Requirements Specification and the plan for testing the firmware is described in Report 10352, the EOS/AMSU-A Firmware Test Plan. Table I provides traceability between the Firmware Requirements Specification and the test procedures found in this section. The objective of this test is to verify that the firmware meets the requirements specified in Report 10458.

Some requirements are shown to be satisfied through inspection of listings, equipment, or other indirect means. Those requirements are indicated in Table I under the test procedure step number heading.

Table I. Firmware Requirements Traceability

Firmware Requirement Specification Paragraph Number	Firmware Test Plan Paragraph Number	Requirements Description	Test Case Num- ber	Test Procedure Step Number	Test Data Sheet
5.1.1	4.1.4.1	Shall read the following data (see table)	2	All	
5.1.1	4.1.4.1	Shall write the following data (see table)	2	Ail	
5.1.1(1)	4.1.4.1	Upon power up, initialize the system	*	•	
5.1.1(2)	4.1.4.1	Upon receipt of an 8 second, begin the processing and output the data.	2	6	
5.1.1(2)a	4.1.4d	Output a data header including all instrument status and housekeeping data to the FIFO memory.	2	4a	
5.1.1(2)b1	4.1.4c	Operate the scanner, determine mode	1	2	
5.1.1(2)b2	4.1.4c	Operate the scanner, sequence the antenna	*	*	
5.1.1(2)b3	4.1.4f	Operate the scanner, test antenna position, and set error bit true or false	•	•	
5.1.1(2)b4	4.1.4e	Read and place antenna position data into FIFO memory.	2	4k	
5.1.1(2)b5	4.1.4e	Read and place radiometer data into FIFO memory.	2	Ali	
5.1.1(2)c	4.1.4d	Read all housekeeping data	2	4a	
5.1.1(2)d	4.1.4a	Read the commands from the input ports	1	2	
5.1.1(2)e	4.1.4a, b	For hardware control commands, send the appropriate pulse, or level to the output ports	*	•	
5.1.1(2)f	4.1.4a, b	Read instrument power commands from the input port and turn on or off appropriate scanner, PLL, and main power as indicated.	•	•	
5.1.2	4.1.4.1	The instrument control firmware shall read the following data (see table)	2	All	
5.1.2	4.1.4.1	The instrument control firmware shall write the following data (see table)	2	All	
5.1.2	All	The instrument control firmware shall provide identical functions forA2 as for A1	All	All	
5.1.3	4.2.4.1	The input data shall come from two sources	All	All	
5.1.3	4.2.4.1	the output shall go to the same 2 destinations.	All	All	

^{*} The power and antenna position commands cannot be verified at this hardware level. They will be verified during system integration testing per AE-26156/9, 10 paragraph 3.2.4.3.3

Table I. Firmware Requirements Traceability (continued)

Firmware Requirement Specification Paragraph Number	Firmware Test Plan Paragraph Number	Requirements Description	Test Case Num- ber	Test Procedure Step Number	Test Data Sheet
5.1.3c	4.2.4c	The command and data handling firmware shall read the FIFO data and the FIFO status (see table)	2	4, 5	
5.1.3d	4.2.4a	The command bit locations shall match the bit locations	1	2	
5.1.3e	4.2.4a	and shall be sent 8 bits at a time	1	2	
5.1.3.1a	4.2.4a	Upon receipt of an instrument command, the firmware shall read and read the command word.	1	2	
5.1.3.1b	4.2.4a	If it is a valid input, restructure the bits of the command to match	1	2	
5.1.3.2	4.2.4b	Upon receipt of Synchronize with Data Time Mark Command interrupt, the firmware shall read the low- order bits	2	5, 6	
5.1.3.2	4.2.4b	This shall also be used to initiate all data output functions of the C&DH (command and data handling) firmware.	2	5, 6	
5.1.3.3	4.2.4c	Read Subaddress #28 and #29 to determine if previous data have been	2	5, 7	
5.1.3.4	4.2.4e	Create the CCSDS header for the low-rate science data and place this header into the RAM location assigned	2	4a	
5.1.3.5	4.2.4e	Create the CCSDS header for the engineering data and place this header into the RAM	2	4a	
5.1.3.6	4.2.4f	Read the time code data from the RAM memory addresses associated	2	4a	
5.1.3.7	4.2.4c	Acquire the data from the EOS/AMSU-A instrument FIFO memory and place them into the RAM memory	2	4a-4l	
5.1.3.8	4.2.4c	Place the instrument data allocated to the engineering data output stream into the RAM memory	2	4a	
5.1.3.9	4.2.4c	When a full block of data has been created in RAM memory, the ready flag shall be incremented in the corresponding subaddress memory location; subaddress	2	5	
5.1.3.10	4.2.4d	Upon receipt of the last data word for a full scan from the AMSU-A instrument is read, a checksum shall be placed	2	5	
5.2.1	4.1.1.3	Each of the three CSCI shall not exceed 4096 bytes		Inspect listing	
5.2.1	4.1.1.3	and not utilize more than 4096 bytes of RAM.		Inspect listing	
5.2.2	4.1.4.1	The Instrument Control Firmware CSCI shall be capable of completing all their required tasks with the 8 seconds	2	6	

Table I. Firmware Requirements Traceability (continued)

Firmware Requirement Specification Paragraph Number	Firmware Test Plan Paragraph Number	Requirements Description	Test Case Num- ber	Test Procedure Step Number	Test Data Sheet
5.2.2	4.2.4.1	The Command and Data Handling firmware CSCI shall be capable of receiving all sensor data within the same 8-second time frame	2	6	
5.2.2		and capable of outputting the data at the required 3.2 kbps data rate		Analysis	
5.2.3		The Instrument Control firmware shall be capable of recovering from any data error condition by restarting with each 8-second sync pulse interrupt.		Analysis	
5.2.3	4.2.4a	The Command and Data Handling Firmware shall validate all incoming data packets by their headers and reject any invalid packets.	1	2	
5.3		N/A			
5.4		N/A			

The firmware tests will be conducted utilizing the AMSU-A Special Test Equipment (STE).

The STE block diagram is shown in Figure 2. It aids in the understanding of the test setup and the movement of data during the tests.

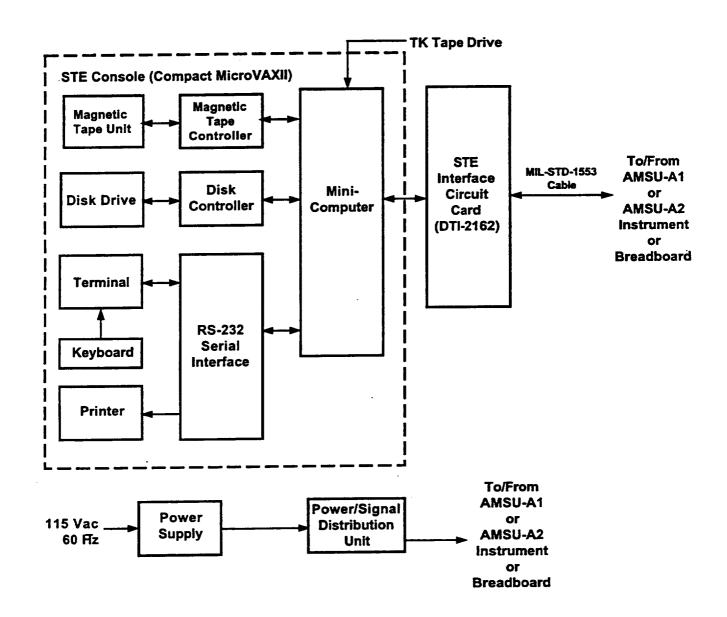


Figure 2. EOS/AMSU-A Special Test Equipment Block Diagram

4. DETAIL PROCEDURES

The following detailed procedure steps shall be utilized for both the AMSU-A1 and AMSU-A2 firmware. The media containing the respective firmware shall be labeled with the CSCI numbers and the test results shall be identified correspondingly. If a test failure or anomaly occurs during testing, the process described in paragraph 4.1.1.4 of the Software Test Plan, Report 10369, shall be followed.

- 4.1 Load Bonded Software. The tape labeled E1.EXE shall be loaded into the VAX STE computer using the VAX Backup utility for use in testing AMSU-A1 firmware CSCIs N7 and N8. The tape labeled E2.EXE shall be loaded into the VAX STE for use in testing AMSU-A2 firmware CSCIs N11 and N12. The tape loaded shall be noted on Test Data Sheet, (TDS) 1. The tape labeled N7 for AMSU-A1 or N11 for AMSU-A2 shall be loaded into the HP64000 development system and then downloaded into the DATA-IO prom burner. The two EEPROMS shall be burned and installed in the breadboard CPU board. The tape loaded shall be noted on TDS 1. The tape labeled N8 for AMSU-A1 or N12 for AMSU-A2 shall be loaded into the HP64000 development system and then downloaded into the DATA-IO prom burner. The two EEPROMS shall be burned and installed in the breadboard MIL-STD-1553 board. The tape loaded shall be noted on TDS 1.
- 4.2 Configure the Test Environment. Perform steps 1 through 5 only if not already configured. Verify setup and power status.
 - 1. Verify that the STE and instrument power are OFF.
 - 2. Interconnect the signal processor breadboard and STE using the special test MIL-STD-1553 cable as shown in Figure 2.
 - 3. On the STE computer control panel, set the MAIN POWER switch to ON. (Switch indicator lit).
 - 4. Turn on printer power.
 - 5. Observe power-up messages on the terminal screen until the message SYTEM JOB TERMINATED AT (DATE) AND (TIME) is displayed (approximately three minutes). Press ENTER key on the terminal keyboard.
 - 6. When prompted for the user name, enter E1 or E2 according to the unit under test. (If steps 1 through 5 are not performed, press RETURN key on terminal keyboard to obtain username prompt.)
 - 7. The message PERFORMING INITIALIZATION PLEASE WAIT will be displayed followed by the display of the Main Menu, which displays the heading message EOS AMSU A1 (or A2) WHAT TYPE OF TEST?"
 - 8. Turn on all power to the signal processor breadboard and wait until power "on" indication on Main menu display.
 - 9. From the Main menu, select the MONITOR ONLY menu option.
 - 10. From the Monitor Only menu, select the COMMANDS menu option.
- 4.3 Test case 1. The test objective is to verify implementation of instrument commands, including the ability of the software to receive command data within the 3.2 kbps rate, to read and interpret commands, to control sensor power, to operate the scanner in accordance with the scanner mode input commands, to initiate prescribed command activities, and to process the input commands.
 - 1. From the Commands menu, select the PRINT SCREEN ONLY menu option to output the displayed sensor housekeeping status to the printer prior to making changes through the issuing of commands.

- 2. From the Commands menu, issue the following commands by selecting their menu option. Wait a minimum of twenty seconds after each selection and print the displayed results by selecting the SCREEN ONLY menu option prior to issuing the next command in the sequence. After each print, type: 1. Press <ENTER> (return to the Monitor Only menu). Select the SCIENCE DATA menu. From the Science Data menu, select the DATA STREAM menu. When prompted for Element Number, select element: 1. Select PRINT SCREEN ONLY. Type: 1. Press <ENTER> to return to the Science Data menu. Type: 1. Press <ENTER> to return to the Monitor Only menu. Then select the COMMANDS menu and select the next command.
 - Reset C&DH Processor
 - b. Cold Cal
 - c. Cold Cal Position 4
 - d. Cold Cal Position 3
 - e. Cold Cal Position 2
 - f. Cold Cal Position 1
 - g. Nadir
 - h. Warm Cal
 - i. Antenna Full Scan Mode
- 3. From the Commands menu, type: 1. Press <ENTER> (return to the Monitor Only menu).
- 4. From the Monitor Only menu, type: 1. Press <ENTER> (return to the Main menu).
- 5. Compare actual versus expected data values. Record the results on TDS 1. The equality of the actual and expected results verify that all the objectives of this test, as stated in paragraph 4.3, have been successfully met. This verifies requirements 5.1.1.2b1), 5.1.1.2d), 5.1.3.1, and part of 5.2.3. The remainder of 5.2.3 will be verified in test case 2, step 8. Requirements 5.1.1.1, 5.1.1.2e and 5.1.1.2f cannot be verified until system level testing.
- 4.4 Test case 2. The following steps test proper acquisition and display of "Low Rate Science" data. The objectives are to acquire Low Rate Science" and "Engineering" data and verify "Low Rate Science" and "Engineering" data timing performance. The test steps will acquire and format sensor data into MIL-STD-1553 data packets. The firmware must receive all sensor data, format, and output sensor data within the required 3.2 kbps maximum rate and within the 8-second time frame for each sensor scan. Since "Engineering" data is a sub-set of the "Low Rate Science" data, the test equipment (STE) has no software routines to specifically process or display "Engineering" data except to acquire "Engineering" data packets and validate these data by comparing them with the corresponding "Low Rate Science" data. The result of this comparison is indicated by the operator terminal displayed "ENGR OK" or "ENGR FAIL". An indication of "ENGR OK" verifies requirement 5.1.1, 5.1.2, 5.1.3, 5.1.3.5 and 5.1.3.8.
 - 1. From the Main menu, select the MONITOR ONLY option.
 - 2. From the Monitor Only menu, select the SCIENCE DATA menu option.
 - 3. For the first selected sub-menu below (paragraph 4.4.4a), select PRINT FULL (generates full scan printout and screen display printout). For the remainder of the selected sub-menus (paragraphs 4.4.4b 4.4.4l) select SCREEN ONLY printouts. This will produce a hard copy of test results for evaluation and comparison purposes to verify that all "low rate science" data are properly output by the firmware.

4. Select from the Science Data menu:

- a. DATA STREAM menu that displays raw input data stream values. When prompted for Element Number, select element: 1. Scan header information is included in the first twenty bytes of the data stream.
- b. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of the first sixteen bytes in the output with the expected results verifies requirement 5.1.1.2a and 5.1.3.4 and 5.1.3.6.
- c. BEAM POSITION NN ALL CHANNELS menu which displays channel data for operator selected beam position. When prompted for Beam Position, select beam position: 1.
- d. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of this data with the expected results verifies requirement 5.1.1.2b5 and 5.1.3.7. Requirements 5.1.1.2b2 & 5.1.1.2b3 cannot be verified until system level testing.
- e. CHANNEL NN ALL BEAM POSITIONS menu which displays beam position data for operator selected channel. When prompted for Channel Number, select channel: 3 (for A1) 1 (for A2).
- f. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of this data with the expected results further verifies requirement 5.1.1.2b5 and 5.1.3.7.
- g. WARM CALIBRATE menu which displays warm calibration data for all channels. The matching of this data with the expected results further verifies requirement 5.1.1.2b5 and 5.1.3.7.
- h. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of this data with the expected results further verifies requirement 5.1.1.2b5 and 5.1.3.7.
- i. COLD CALIBRATE menu which displays cold calibration data for all channels.
- j. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of this data with the expected results further verifies requirement 5.1.1.2b5 and 5.1.3.7.
- k. REFLECTOR POSITIONS menu which displays reflector positions and error code for A1-1 or A1-2 antenna as selected by the operator. A1-1 antenna data is displayed initially. An "E" beside a value indicates a reflector position error outside threshold values. The presence or absence of an "E" are both acceptable for purposes of this test.
- 1. After print is selected, type: 1. Press <ENTER> (return to the Science Data menu). The matching of this data with the expected results further verifies requirement 5.1.1.2b4 and 5.1.3.7.
- 5. On the display, observe that the received (IN) checksum and the computed (CALC) checksum of the data match. Also observe that the sub-address (SA) #28 increments at the same rate as the scan count and that the sub-address (SA) #29 increments at the same rate as the scan count for the A2 sensor and at twice the rate as the scan count for the A1 sensor. This verifies requirement 5.1.3.3 and 5.1.3.9 and 5.1.3.10.
- 6. Observe that the scan number increments at a rate of 8.0 ± 0.5 seconds per scan average after ten minutes of observation. Record a beginning scan number and the corresponding time. After ten minutes, record the scan number. Divide the 600 seconds by the scan number difference. The result is the average scan time which shall be within the above tolerance. Record this result on TDS 2. This verifies requirement 5.1.3.2
- 7. Select the STE procedure that skips the transmission of one of the one-second time marks during each eightsecond scan so as to test that the C&DH firmware adapts to this anomalous condition by observing that the

scan count, checksums and SA counters all function as previously noted. Verify that selecting 7 as the number to skip causes all data functions to stop updating and that selecting 0 as the number to skip restores the firmware to normal operation.

- 8. Select the STE procedure to send an invalid command APID. From the Commands menu, select one or more commands (with a one minute wait between commands). Verify that the STE reports that the command was not accepted and that the previous sensor mode is unchanged.
- 9. Evaluate all displayed and printed low rate science data. Compare actual versus expected data values.
- 10. Record the results on TDS 2.

5. EVALUATION CRITERIA

The evaluation criteria for test cases 1 and 2 are contained in Tables II and III.

These tables describe criteria used to evaluate how each test satisfies requirements. The sources of actual and expected results relevant to each requirement are also shown.

Table II. Expected Results, Test Case 1

Step Number	Expected Results		Data Stream Byte No. 20	Error Condition
4.3(2a)	Subaddress #28 & 29 V	alues Reset		Values Do Not Reset
4.3(2b)	Cold Cal Position	= YES	8	= NO
4.3(2c)	Cold Cal Position 4	= YES	104	= NO
4.3(2d)	Cold Cal Position 3	= YES	72	= NO
4.3(2e)	Cold Cal Position 2	= YES	40	= NO
4.3(2f)	Cold Cal Position 1	= YES	8	= NO
4.3(2g)	Nadir	= YES	16	= NO
4.3(2h)	Warm Cal	= YES	4	= NO
4.3(2i)	Full Scan	= YES	2	= NO

Table III. Expected Results, Test Case 2

Step Number	Expected Results
4.4.4a	Packet ID, Packet Sequence, Length according to GIRD
4.4.4c	Channel 1 changes to 28800 ±500 with applied 2 volts, Channel 3 changes to 30000 ±500 with applied 2 volts
4.4.4e	Channel 1 changes to 28800 ±500 with applied 2 volts, Channel 3 changes to 30000 ±500 with applied 2 volts
4.4.4g	Channel 1 changes to 28800 ±500 with applied 2 volts, Channel 3 changes to 30000 ±500 with applied 2 volts
4.4.4i	Channel 1 changes to 28800 ±500 with applied 2 volts, Channel 3 changes to 30000 ±500 with applied 2 volts
4.4.4k	(no criteria at this hardware level)
4.4.5	Checksums match -SA 28 and 29 increment
4.4.6	8.0 ±0.5 sec scan time
4.4.7	Skip 1-6 cause no disruption; 7 causes stop; 0 restarts
4.4.8	Commands not accepted

6. EXPECTED RESULTS

The expected results for test cases 1 and 2 are contained in Tables II and III. These tables show the expected results from test sources described in the tables.

7. ACTUAL RESULTS

The actual results of the firmware test will be included in the Test Report (Report 10974).

8. ABBREVIATIONS AND ACRONYMS

AMSU	Advanced Microwave Sounding Unit
C&DH CCSDS CDR CDRL CSCI CTE	Command and Data Handling Consultative Committee for Space Data Systems Critical Design Review Contract Data Requirements List Computer Software Configuration Item Calibration Test Equipment
EOS	Earth Observing System
FIFO	First In First Out
GIRD	General Interface Requirements Document
KBPS	Kilobytes per second
NASA	National Aeronautics and Space Administration
PWR	Power
RAM	Random Access Memory
STE	Special Test Equipment
WBS	Work Breakdown Structure

9. GLOSSARY

None.

10. NOTES

None.

11. TEST DATA SHEETS

AE-26600B 23 June 1998

TEST DATA SHEET 1 Test Case 1 (Paragraph 4.3)

Unit Tested (AMSU-A1 or AMSU-A2)	EOS/AMSU-AI
STE Tape Loaded EI, EXE;	
Instrument Control Tape Loaded	
Control and Data Handling Tape Loaded	NONE - FLIGHT PROMS

Procedure Step	Requirement Description	Specification Reference	Requirement Satisfied ? yes or no	HardCopy Test Data Attached ?	Test Data on Tape ?	Related Discrepany Reports
4.3.2a	Reset C&DH	5.1.1.2b,d 5.1.3.1	res	YES	No	N/A
4.3.2b	Cold Cal	5.1.1.2b,d 5.1.3.1	YES	YES	NO	N/A
4.3.2c	Cold Cal Position 4	5.1.1.2b,d 5.1.3.1	YES	YES	No	NA
4.3.2d	Cold Cal Position 3	5.1.1.2b,d 5.1.3.1	res	Yes	No	N/A
4.3.2e	Cold Cal Position 2	5.1.1.2b,d 5.1.3.1	Yes	Yes	No	N/A
4.3.2f	Cold Cal Position 1	5.1.1.2b,d 5.1.3.1	YES	YES	No	N/A
4.3.2g	Nadir	5.1.1.2b,d 5.1.3.1	YES	Yes	NO	N/A
4.3.2h	Warm Cal	5.1.1.2b,d 5.1.3.1	YES	res	No	N/A
4.3.2i	Full Scan	5.1.1.2b,d 5.1.3.1	YES	res	No	N/A

	5.1.3.1	163	ر <u>ب ا</u>	<u> </u>	
Comments:					
Authentication:	211 M	, , , , , , , , , , , , , , , , , , , ,	•	/ //	
Aerojet System Test:	bat A Katt		Date: 💆	14/98	
Aerojet Quality Assurance:		898	Date:	T M M	
Customer Representative:			Date:	JUL 23 '98	
Other Witness (optional): _			Date:		

TEST DATA SHEET 2 Test Case 2 (Paragraph 4.4)

Unit Tested (AMSU-A1 or AMSU-A	A2) EOS/AMSU-AI
	(E; 35 EIX.EXE; 31
	NONE - FLIGHT PROMS
	oaded None - FLIGHT PROMS

Procedure Step	Requirement Description	Specification Reference	Requirement Satisfied ? yes or no	HardCopy Test Data Attached ?	Test Data on Tape?	Related Discrepany Reports
4.4.4a	Data Stream	5.1.1.2a, 5.1.3.4,5.1.3.6	res	YES	No	N/A
4.4.4c	Beam Position NN	5.1.1.2b5 5.1.3.7	Yes	YES	No	NA
4.4.4e	Channel NN	5.1.1.2b5 5.1.3.7	YES	YES .	No	N/A
4.4.4g	Warm Calibrate	5.1.1.2b5 5.1.3.7	YES	YES.	No	N/A
4.4.4i	Cold Calibrate	5.1.1.2b5 5.1.3.7	Y85	HES	No	NA
4.4.4k	Reflector Positions	5.1.1.2b4 5.1.3.7	YES	YES	No	NÍA
4.4.5	Checksum sub-address	5.1.3.3,5.1.3.9 5.1.3.10	<i>YES</i>	YES	No	N/A
4.4.6	8 Sec Scan	5.1.3.2	res	SEE BELOW	NO	NA
4.4.7	Skip Time Mark	No Req't	YES	NO	No	NA
4.4.8	Invalid APID	5.2.3	YES	NO	No	N/A

Comments: PARAGRAPH 4.	4.6 START SCAN	366 STAK	RT11MC 20:08:3
	EMD SCAN	441 STOP	O TIME 20:18:31
***	75	SCANS PER	600 SECONDS
			SAN/8 SEGN
Authoration	. 1		
Authentication: Aerojet System Test: H H H H H H H H H H H H H	Vatt .	Date: 7/14/	9 8
	893	Date:	
Aerojet Quality Assurance:			
Customer Representative:	(Date:	
Other Witness (optional):		Date:	

DOCUMENT APPROVAL SHEET

GENEDRP AEROJET

TITLE	· · · · · · · · · · · · · · · · · · ·		DOCUMENT NO.	
Earth Observing System/Advanced	AE-26600B			
A) Firmware Test Procedure		•	23 June 1998	
•				
INPUT FROM: DATE	CDRL:	SPECIFICATION ENGINEER:		DATE
R. Schwantje	415			
CHECKED BY: // /	, DATE	JOB NUMBER:		DATE
RS Scheint	7/4/98			
APPROVED SIGNATURES			DEPT. NO.	DATE
Specifications Engineering (J. Kir	w RN9/en	unuas PMO	8631	
Specifications Engineering (5. Ki	N) W	1	0001	, ,
Product Toom Loader (A. Nieto)	611		8341	7/7/50
Product Team Leader (A. Nieto)	064/1		0041	l l
Systems Engineer (R. Platt)	X 91 Plut	!	8311	7/7/98
		7 0	0017	,
Safety (W. Neighbors)	1. Show h	his.	8331	7/7/98
Safety (vv. Neighbors)			0001	1 1 1
Design Assurance (E. Lorenz)	Cholo	e/ :	8331	7/7/98
Design Assurance (L. Lorenz)	4		0001] , ,]
Software Quality Assurance (M.	Santos)	Jah-	7831	7/10/98
Software Quality Assurance (M.		77	, , , , ,	
Technical Director/PMO (R. Hau	erwaas) RN9	Lauerwaas	4001	7/13/98
Toolinioal Bhooloin Wo (14. 11aa	(1		7/13/98
Configuration Management (J. C	avanaugh)	avenauch	8361	1112148
Comigaration Management (c. c	7			
	O			
This Revision incorporated ECN	CAMSU-1763			
			1	-
By my signature, I certify the above document hat requirements related to my area of responsibility.		nd concurs with the technical		
To your service is my drop of the particular in your				
RELEASE (Data Center) FINAL				
·				

A1-6] SCIENCE	1.EXE;35 DATA	EOS A1-C E1.EXE;35 FULL SCAN MODE [5] SCIENCE DATA ELEMENT 0000		P1 14-JUL-98:09:13 SCAN NUMBER		13
CONTROL	6] CONTROL/STATUS	ELEMENT	00			
[7] ENGINEERING	RING	ELEMENT	00			
		COMMANDS	NDS	PLLO POWER =	PLLO#1 [15	
9 SCANNER A1-	\leftarrow	POWER =	NO	COLD CAL POSITION 1 =	YES [16	_
10] SCANNER A1-	7	POWER =	NO	7	: NO [17	_
11] ANTENNA FULL SCAN MODE	A FULL SO	CAN MODE =	YES	H &	: NO [18	~
12]	WARM CAL	AL =	NO	COLD CAL POSITION 4 =	. NO [19	_
13]	COLD CAL	AL =	NO	RESET C&DH PROCESSOR	[20	_
[14]	NADIR	il	NO	GSE MODE	[21	
ENGR OK	POWER	ON CHI	ECKSUM I	ON CHECKSUM IN A71F CALC A71F SA28 SCREEN ONLY [2 PRINT [3 FULL	368 SA29	590
SELECT BUTTON 2	N 2					

AE-26600 PARAGRAPH 4.3.1

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			[]	[]	[17]	[]	[19	[20	[21	<u> </u>	ב היי
NUMBER			PLLO#1 [15]	YES [16]	NO	NO	NO			0 SA29	בא ר ד
SCAN			д	ij	11	11	11				
Pl 14-JUL-98:14:00 SCAN NUMBER			11	Н	7	ω	COLD CAL POSITION 4 :	RESET C&DH PROCESSOR		ON CHECKSUM IN B49D CALC B49D SA28	7704
JUL-98			PLLO POWER =	COLD CAL POSITION			CAL POS	T C&DH I	GSE MODE	"CALC, B	· · · · ·
1 14-			PLLO	COLD			COLD	RESE	GSE	B49D	FKIN
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AN N			COMMANDS			11	ŧi	11	£1	CHEC	
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5 FUI ELE		ELI		POWER	POWER	SCAN	CAL	CAL			ñ
EXE;3 DATA	TATUS	NG		A1-1	A1-2	FULL	WARM CAL	COLD CAL	NADIR	POWER	2
EOS A1-6 E1.EXE;35 FULL SCAN MODE [5] SCIENCE DATA ELEMENT 0000	6] CONTROL/STATUS	7] ENGINEERING		SCANNER A1-	SCANNER A1-2 POWER	ANTENNA FULL SCAN MODE					SELECT BUTTON 2
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AE-26600 PARAGRAPH 4,3,2,0

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MBER			NO	55 58 58					SA29	•
SCAN NUMBER			DATA	63 105	വ വ	७०	114 171		75	
			ON	40 00	25	55 74	22 22		SA28	
14:33			DATA	64 106	64 44	63 16	64 67		Л.Т.	
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EOS]		- -	NO	40	IW4	אטי	o Γα	$\begin{bmatrix} 21 \end{bmatrix}$	ENGR OK	SELE

AE 26600 PAR AGRAPH 4.3.2.a

SCAN NUMBER 73			PLLO#1 [15]	YES [16]	NO [17]	NO [18]	NO [19]	[20]	[21]	25 SA29 49	L J KETUKN
P1 14-JUL-98 . J:17:29 SC			PLLO POWER =	COLD CAL POSITION 1 =	2 =	11 E	COLD CAL POSITION 4 =	RESET C&DH PROCESSOR	GSE MODE	ON CHECKSUM IN BISE CALC BISE SA28	FRINI [3] FOLL
10DE	00	00	DS	OFF	OFF	NO	NO	YES	NO	CKSUM	. ז
COLD CAL PELEMENT 00	ELEMENT	ELEMENT	COMMANDS	POWER =	POWER =	SCAN MODE =	= 1	H	IJ	ON CHE	SCREEN ON
L.EXE;35 DATA	STATUS	SING					WARM CAL	COLD CAL	NADIR	POWER	2
EOS A1-C E1.EXE;35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	[6] CONTROL/STATUS	[7] ENGINEERING		[9] SCANNER A1-1	[10] SCANNER A1-2	[11] ANTENNA FULL	[12]	[13]	[14]	ENGR OK	SELECT BUTTON 2

AE-26600 PARAGRAM 4,3,2,1

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AE-26,600 PARAGRAPH 4.3,2,15

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39 SA29 78	ON CHECKSUM IN 6CDD CALC 6CDD SA28	IECKSUM NILY (ENGR OK POWER
[21]	GSE MODE	NO	11	14] NADIR
[20]	RESET C&DH PROCESSOR	: YES	CAL =	13] COLD CAL
YES [19]	COLD CAL POSITION 4 =	NO	CAL =	12] WARM CAL
NO [18]	e e	NO	FULL SCAN MODE =	11] ANTENNA FULL
NO [17]	22	OFF	-2 POWER =	10] SCANNER A1-2
NO [16]	COLD CAL POSITION 1 =	OFF	-1 POWER =	9] SCANNER A1-1
PLLO#1 [15]	PLLO POWER =	NDS	COMMANDS	
		00	ELEMENT	7] ENGINEERING
		00	ELEMENT	6] CONTROL/STATUS
SCAN NUMBER 88	P1 14-JUL-98 . J:19:29 S	MODE 000	S COLD CAL ELEMENT C	EOS A1-C E1.EXE,35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000

AE-26600 - PARAGRAPH 4,3,2,C

7			DATA	97 97 94 9	128 65 65	20 40	თ	HIRN 86	
MBER			NO	7 2 3 8 4	001 001	63 63 63	64	SA29	
SCAN NUMBER			DATA	103 103				43	1
			NO	400 601				SA28	
/:20:01			DATA	103 103				1111	
			64 NO	444 124	.44 340	4 4 7	48	C GDDD	,)
14-JUL-98			TO DATA	22 48 48 7	23 63 65 65	53 44	227	D CALC	•
P1 14			NO 1	დო 6	390 190	დ ოო	40	IN 6DI	
			SAM	128 128 5		38 9	176 JWN	M. ∠	- 1
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S COLD ELEME	ELE	ELE	NO	118				O.C.))
A1-C E1.EXE;35 SCIENCE DATA	TATUS	NG	DATA	000	000	00	0	POWER	7
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A1-C SCIE		ENGI	DATA	ر وسور	1 74, 14,0	191	$17\frac{4}{1}$	OK	SELECT BUTTON
EOS [5]	[9]	[7]	NO	H20	04ሴ	101	$\begin{bmatrix} 21 \end{bmatrix}$	ENGR OK	SELE

AE-26600 PARAGRAPH 4.3,2,C

3R 13			PLLO#1 [15]	[16]	[17]	3 [18]	[19]	[20]	[21]	54 SA29 108	1770 177
SCAN NUMBER			PLLO#1	NO	NO	YES	NO			54 SA2	7 - 1
				II	11	11	11	ኧ		758	
P1 14-JUL-98 . J:21:29			II	COLD CAL POSITION 1	7	m	COLD CAL POSITION 4	RESET C&DH PROCESSOR		ON CHECKSUM IN 6E45 CALC 6E45 SA28	
UL-98			POWER	CAL P(CAL P	C&DH	ODE	CALC	<u> </u>
. 14-JI			PLLO POWER =	COLD			COLD	RESET	GSE MODE	6E45 OPTNT	TATAT
P1										NI	7
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EXE; DA1	STAT	ING		A1-1	A1-2	FULL	WARM	COLD	NADI	POWER	0
EOS A1-C E1.EXE;35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	[6] CONTROL/STATUS	[7] ENGINEERING		SCANNER A1-1	[10] SCANNER A1-2	ANTENNA					SELECT BITTON 2
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EOS [5	9]	[7		[9]	1([1.	[12	[13	[14	ENG	C.

AE-2660D PARAGRAPH 4.3,2,J

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EXE;35 COLD CAL MODE DATA ELEMENT 0000	TATUS ELEMENT	ELEMENT	DATA
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AE-26600 PARMSRAPH

SA29 112] RETURN

SA28

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72222 7822 7832 7832 7832

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2 187074979

38			[15]	[16]	[17]	[18]	[19]	[20]	[21]) 158	
SCAN NUMBER			PLLO#1 [15	NO	YES	NO	NO			79 SA29 1	74
				11	11	11	11				
J:24:49			l i	SITION 1	7	m	SITION 4	PROCESSOR		193 SA2	T 0777
Pl 14-JUL-98 J:24:49			PLLO POWER	COLD CAL POSITION 1			COLD CAL POSITION 4	RESET C&DH PROCESSOR	GSE MODE	ON CHECKSUM IN 7193 CALC 7193 SA28	
P1			[A	ŭ			ΰ	R	Ü	NI D	-
00E	00	00	SC	OFF	OFF	ON	ON	YES	NO	CKSUM	7 1
AL MC			COMMANDS			II	11	11	H	CHEC	
S COLD CZ ELEMENT	ELEMENT	ELEMENT	CO	POWER =	OWER =	ANTENNA FULL SCAN MODE	CAL	CAL		NO NO	SCNEE
EXE;35 DATA	STATUS	ING		A1-1	A1-2 I	FULL 8	WARM CAL	COLD CAL	NADIR	POWER	7
EOS A1-C E1.EXE;35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	6] CONTROL/STATUS	7] ENGINEERING		SCANNER A1-1 POWER	10] SCANNER A1-2 POWER =	ANTENNA				OK	SELECT BUTTON 2
EOS [5]	[9]	[7]		[6]	[10]	[11]	[12]	[13]	[14]	ENGR 0	SELEC

RE-ZUGOO PARAGRAPH 4,3,2,R

SCAN NU		
J:24:58		
P1 14-JUL-98 J:24:58 SCAN NU		
P1		
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EOS A1-(E1.EXE,35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	ELEMENT	ELEMENT
1.EXE;35 DATA	[6] CONTROL/STATUS ELEMENT	RING
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AG-26600 PARAGRAP 4,3,2,C

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			ij	16	17	18	19	20	[21	``;	
SCAN NUMBER			PLLO#1 [15	YES	NO	NO	ON	_	_	90 SA29 179	1 J KET
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P1 14-JUL-98 . J:26:10			ER =	COLD CAL POSITION 1 =	2 =	ς. Π	COLD CAL POSITION 4 =	RESET C&DH PROCESSOR		ON CHECKSUM IN 716F CALC 716F SA28	7 T L CTT
1 14-JUL-3			PLLO POWER	COLD CAL			COLD CAL	RESET CE	GSE MODE	716F CAL	FRINI
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00 00 00	00	00	SC	OFF	OFF	NO	NO	YES	NO	KSUM	ז ר
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COLD	ELEMENT	ELEMENT	ŭ	OWER =	OWER =	CAN MO	'AL	'AL		NO	SCRE A
EXE;35 DATA	STATUS	ING		A1-1	A1-2 E	FULL S	WARM CAL	COLD CAL	NADIR	POWER	73
EOS A1-C E1.EXE;35 COLD CAL MODE [5] SCIENCE DATA ELEMENT 0000	6] CONTROL/STATUS	ENGINEERING		SCANNER A1-1 POWER	10] SCANNER A1-2 POWER =	ANTENNA FULL SCAN MODE				U	SELECT BUTTON 2
7) (Ξ.		_	_	_	_	_	_	Ŏ	ECI
EOS [5	9]	7		[6]	[10	[11	[12	[13	[14	ENGR OK	SEL

AE-26600 PARAGRAPH 4,3,2,F

E S S S	Al-C El.EXE;35 SCIENCE DATA	EI.EXE;35 COLD CAL MODE 3E DATA ELEMENT 0000	FI 14-00L-98 J:26:26	SCAN NUMB
[9]	CONTROL/STATUS	ELEMENT 00		
[7]	ENGINEERING	ELEMENT 00		

DATA	1 0200000 040803420	182 URN
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DATA	1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	[6]
NO	400000000 00400400	SA28
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TO DATA	0 0 0 00040000 40000040	F CAL NT [
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NO	1111444 78001484	SCI
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NO	444444 60428484	TTON
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NO	1284327 12 12	ENGR

AC-2660 PARAGRAPH 4.3,2,F

NETOKAN				SELECT BUTTON 2
103 SA29 206	ON CHECKSUM IN 725B CALC 725B SA28 SCREEN ONLY () DEINT () THILL	CHECKSUM	POWER OF	ENGR OK
[21]	GSE MODE	= YES	NADIR	[14]
[20]	RESET C&DH PROCESSOR	= NO	COLD CAL	[13]
[61] ON	COLD CAL POSITION 4 =	NO	WARM CAL	[12]
NO [18]	п В	MODE = NO] ANTENNA FULL SCAN MODE	[11] ANTENNA
NO [17]	2 = 1	\ = OFF	A1-2 POWER	[10] SCANNER A1-2 POWER
YES [16]	COLD CAL POSITION 1 =	t = OFF	A1-1 POWER	[9] SCANNER A1-
PLLO#1 [15]	PLLO POWER =	COMMANDS		
		ELEMENT 00		[7] ENGINEERING
		ELEMENT 00		[6] CONTROL/STATUS
SCAN NUMBER 62	P1 14-JUL-98 3:28:02 SC	OIR MODE EMENT 0000	.EXE;35 NAI DATA ELE	EOS A1-, E1.EXE;35 NADIR MODE [5] SCIENCE DATA ELEMENT 0000

AE-26600 PARAGRAPH 4.3.2.9

NUMBER
SCAN
1:28:10
14-JUL-98
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63			DATA	0400000 0400000 040000400	Z U 9 URN
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·			64 NO	44444444 1064060	3 J.
14-JUL-98			TO DATA	2 2 2 62646762 4676264 46762644	ZINT [
P1 14			NO NO	шшшшшшш ш4гипри 2000 г	N /3/ PRI
			EAM DATA	128 128 245 624 0172 0WN	[2]
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A1-C SCIE	CONTROL/STATU	ENGINEERING	DATA	1 1 9 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	NGR OK SELECT BUTTON
EOS [5]	[6]	[7]	NO	128470C81	ENGR

AE 26600 PARACRAPH 4.3,2.9

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82			15	16	17	18	19	20	21	124 SA29 247	3
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SCAN NUMBER			PLLO#1 [15	YES	NO	NO	NO			A25	Ā
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SCA											
				11	2	11)R		128	
: 47				Z	.,	.,	COLD CAL POSITION 4	RESET C&DH PROCESSOR		S	7
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J			11	SI			SI	PR		722	_
98			PLLO POWER	COLD CAL POSITION			Ϋ́	ъDН	.,	ູ່	າ .
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MOD	00	00	COMMANDS							ON CHECKSUM IN 722B CALC 722B SA28	X Z
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EXE	TAT	NG		A1-	A1-	FUL	WARM	COLD	NADIR	POWER	2
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NC.	ROJ	NE		SCANNER A1-1	SCANNER A1-2	ANTENNA FULL					TT.
CIE	LNO	NGI		SCA	SCA	ANT					E
EOS A1-L E1.EXE,35 WARM CAL MODE [5] SCIENCE DATA ELEMENT 0000	6] CONTROL/STATUS	7] ENGINEERING			_	_	$\overline{}$	_	_	ENGR OK	SELECT BUTTON 2
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EC L	_	_		_	_		<u></u>	_	_	EN	ഗ

AE-2660 MARAGRAPH 4,3,2,4

NUMBER
SCAN
3:30:57
14-JUL-98
P1

EOS		EOS A1-, E1.EXE;35 [5] SCIENCE DATA [6] CONTROL/STATUS	WARM CAL MODE ELEMENT 0000 ELEMENT 00	P1 14-JUL-98):30:57
	. ,_	ENGINEERING	ELEMENT 00		

SA29 251 RETURN DATA 78809887 78801784 0 N DATA 00040000 60040000 60040000 400000000 60126400 SA28 7329 S DATA 44444444 10645978 64 NO 2 2 2 02040202 46020240 7329 C PRINT Z Z 128 128 128 625 176 176 DOWN DATA STREAM DATA NO DATA CHECKSUM ONLY [2 2200010 MUMMAMMM-130 130 128 128 128 14 122 ON SCREEN 22220987 43210987 POWER DATA 0000000 $^{\circ}$ 401111109 6014830109 SELECT BUTTON 8 112933 1242 170 170 170 DATA S 220145027 8

AG-2660 PARAGRAPH 4,3,2,h

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		11	SITION 1	7	m	SITION 4	PROCESSO		SES SA	7704
		POWER :	CAL POS			CAL POS	T C&DH 1	MODE	"CALC A	
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00	00	DS	OFF	OFF	YES	NO	NO	NO	CKSUM	7] [7
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	ELEMEN	S	POWER =	POWER =	SCAN MOD	CAL	CAL		NO	SCREE BER
STATUS	ING		A1-1	A1-2	FULL	WARM	COLD	NADIR	POWER	7
CONTROL/S	ENGINEERI			SCANNER	ANTENNA				X	SELECT BUTTON 2
[6]	[7]		[9]	[10]	[11]	[12]	[13]	[14]	ENGR O	SELEC
	[6] CONTROL/STATUS ELEMENT 00	ELEMENT ELEMENT	rus element Element Comman	FUS ELEMENT 00 ELEMENT 00 COMMANDS PLLO POWER = COMMANDS COLD CAL POSITION 1 =	CONTROL/STATUS ELEMENT 00 ENGINEERING ELEMENT 00 COMMANDS SCANNER A1-1 POWER = OFF COLD CAL POSITION 1 = SCANNER A1-2 POWER = OFF COLD CAL POSITION 2 = 2 = 000000000000000000000000000000	CONTROL/STATUS ELEMENT 00 PLLO POWER PLLO POWER PLLO PLLO <td>CONTROL/STATUS ELEMENT 00 PLLO POWER PLLO POWER PLLO PLLO POWER PLLO PLLO</td> <td>CONTROL/STATUS ELEMENT 00 PLLO POMER PLLO PLLO</td> <td>SONTROL/STATUS ELEMENT 00 PLLO POWER PLLO POWER PLLOH PLLOH POWER PLLOH PLLOH POWER PLLOH PLLOH</td> <td>CONTROL/STATUS ELEMENT 00 ENGINEERING ELEMENT 00 COMMANDS PLLO POWER = PLLO POWER = SCANNER A1-1 POWER = OFF COLD CAL POSITION 1 = YES ANTENNA FULL SCAN MODE = YES 3 = NO MARM CAL = NO COLD CAL POSITION 4 = NO COLD CAL POSITION 4 = NO COLD CAL POSITION 4 = NO MADIR = NO GSE MODE NO MADIR = NO GSE MODE 140 \$A25</td>	CONTROL/STATUS ELEMENT 00 PLLO POWER PLLO POWER PLLO PLLO POWER PLLO PLLO	CONTROL/STATUS ELEMENT 00 PLLO POMER PLLO PLLO	SONTROL/STATUS ELEMENT 00 PLLO POWER PLLO POWER PLLOH PLLOH POWER PLLOH PLLOH POWER PLLOH PLLOH	CONTROL/STATUS ELEMENT 00 ENGINEERING ELEMENT 00 COMMANDS PLLO POWER = PLLO POWER = SCANNER A1-1 POWER = OFF COLD CAL POSITION 1 = YES ANTENNA FULL SCAN MODE = YES 3 = NO MARM CAL = NO COLD CAL POSITION 4 = NO COLD CAL POSITION 4 = NO COLD CAL POSITION 4 = NO MADIR = NO GSE MODE NO MADIR = NO GSE MODE 140 \$A25

AE-26600 PARAGRAPH 4,3,2,i

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SCAN NUMBER			DATA	1 1- 08000000 4 6000000 4-1
			NO	SA28 SA288 SA28
:33:23			DATA	404080408 JIL
£ 86			64 NO	4444444
14-JUL-98			TO DATA	10 A759 CALC
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AE 26600 PARAGRAPH 4,4,3

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- Block 15. Supplementary Notes. Information not included

- elsewhere: affiliation of authors if additional space is required for Block 9, notice of work sponsored by another agency, monitor of contract, information about supplements (file, data tapes, etc.) meeting site and date for presented papers, journal to which an article has been submitted, note of a report made from a thesis, appendix by author other than shown in Block 7.
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EOS/AMSU-A1, SYSTEM COMPREHENSIVE AND LIMITED PERFORMANCE TESTS TEST PROCEDURE

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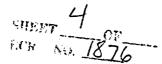
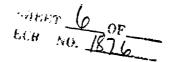


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37°		



1. SCOPE

- 1.1 Scope. This process specification establishes the requirements for the Comprehensive Performance Test (CPT) and Limited Performance Test (LPT) of the Earth Observing System Advanced Microwave Sounding Unit A1 (EOS/AMSU-A1), referred to as the unit. The unit is defined on Drawing 1356008.
- 1.2 Procedure sequence. The sequence of CPT/LPT testing is shown in Figure 1. At the discretion of the test engineer the order of tests may be changed.

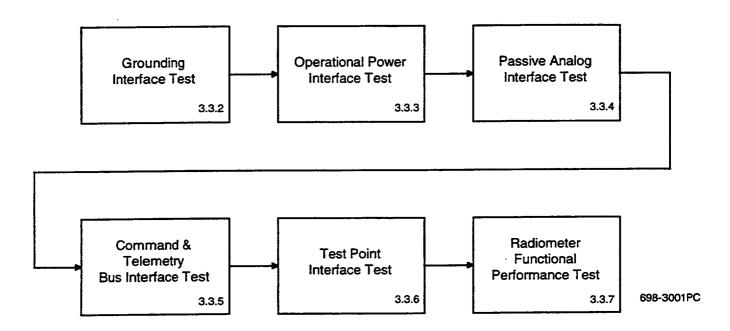


Figure 1. Sequence of EOS/AMSU-A1 CPT/LPT Testing

AE-26156/9 18 June 1998

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2. APPLICABLE DOCUMENTS

2.1 Government documents. The following documents form a part of this specification to the extent specified herein. The latest issue is applicable.

SPECIFICATIONS

NASA (Goddard Space Flight Center (GSFC))

S-480-80	Performance and Operation Specification for the EOS/ METSAT Integrated Programs AMSU-A Instrument (POS)
S-480-79	Performance Assurance Requirements for the EOS/METSAT Integrated Programs AMSU-A Instrument (PAR)
422-11-12-01	General Interface Requirements Document for EOS Common Spacecraft /Instruments EOS PM Project (GIRD)
422-12-12-02	Unique Instrument Interface Document for the Advanced Microwave Sounding Unit (AMSU-A) EOS PM Project (UIID)

STANDARDS

MIL-STD-45662 Calibration Systems Requirements

(Copies of government documents should be obtained as indicated in the Department of Defense Index of Specifications and Standards).

2.2 Non government documents. The following documents form a part of this specification to the extent specified herein. The latest issue is applicable.

2.2.1 TRW documents

SPECIFICATIONS

D24844	Interface Control Document for Advanced Microwave Sounding Unit - A1 (ICD)

D25092 Instrument Interface Database for the AMSU-A1

(Copies of TRW documents may be obtained from TRW Inc.)

2.2.2 Aerojet documents

STANDARDS

STD-2454	Requirements for Electrostatic Discharge Control
SPECIFICATIONS	
AE-26002/1	AMSU-A1 Antenna Drive Subsystem Test Procedure
AE-26156/7	EOS/AMSU-A1 Subsystem Integration Procedure

AE-26357 AMSU-A Transportation and Handling Procedure

AE-26600

EOS/AMSU-A Firmware Test Procedures

SHEET 9 OF STE

REPORTS

10353 EOS/AMSU-A Contamination Control Plan

10443 EOS/AMSU-A Software User's Guide (STE Software)

10458 EOS/AMSU-A Firmware Requirements

10974 EOS/AMSU-A Firmware Test Report

DRAWINGS

1338427 Cover, ESD Shielded Bag

1356008 EOS/AMSU-A1 Assembly

1356648 Cable Assembly, EOS Lab Test

1356655 Console Assembly, METSAT and EOS STE

SK1358702 9 Pin Breakout Box

SK1358704 25 Pin Breakout Box

SK1358705 37 Pin Breakout Box

SK1360106 ON/OFF Switch

(Copies of Aerojet documents may be obtained from Gencorp Aerojet, Azusa Operations, CAGE 70143, P.O. Box 296, Azusa, California, 91702-0296).

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3. REQUIREMENTS

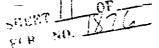
3.1 Equipment. All measurements shall be made using the test equipment or its equivalent as specified in Table I. Equivalent test equipment shall be approved by Systems Engineering and Quality Assurance. Test equipment and gauges required to perform examinations and tests shall be controlled by a calibration system as specified in MIL-STD-45662.

All inspection, measurement and test equipment used shall be currently calibrated to certified standards. The date of last calibration and calibration due date shall be displayed on each item of equipment subject to calibration and recorded at the time of test performance as specified in detailed procedures.

- 3.2 Materials. Not applicable.
- 3.3 Required procedures and operations. The unit shall be subjected to the tests shown in Figure 1 and Table II.

Table I. Required Test Equipment

				
Item	Qty	Equipment	Manufacturer	Model No,
1	1	9-Pin Breakout Box	Aerojet	SK1358702-1/ 2536-3743
2	1	25-Pin Breakout Box	Aerojet	SK1358704-1/ 2536-3746
3	1	37-Pin Breakout Box	Aerojet	SK1358705-1/ 2536-3745
4	1	AMSU-A Special Test Equipment (STE)	Aerojet	1356655-1
5	1	STE Interface Cable J1 (W31)	Aerojet	1356648-1
6	1	STE Interface Cable J2 (W32)	Aerojet	1356648-2
7	1	STE Interface Cable J3 (W33)	Aerojet	1356648-3
8	1	STE Interface Cable J4 (W34)	Aerojet	1356648-4
9	2	Liquid Nitrogen Container	Cole Parmer	N03726-20
10	1	Digital Multimeter	Fluke/Tektronix	77/DMM916
11	1	Spectrum Analyzer	Hewlett-Packard	8566B/8590L
12	1	Plotter	Hewlett-Packard	7475A
13	1	Digital Multimeter	Hewlett-Packard	34401A
14	1	Digital Oscilloscope	Tektronix	TDS386/2221A
15	1	Dynamic Signal Analyzer	Hewlett-Packard	3562A/3563
16	1	WR19 Harmonic Mixer (40-60 GHz)	Hewlett-Packard	HP11970V
17	1	WR19 Feedhorn	TRG	V861
18	1	Current Probe	Tektronix	AM503
19	1	Frequency Counter	Hewlett-Packard	5316A
20	1	Function Generator	Hewlett-Packard	3325A/B
21	1	Power Supply	Power Designs	3650-S
22	1	FREQUENCY SYNTHESIZER	HEWLETT-PACKARD	83623A
23	'	SOURCE MODULE/MULTIPLIER	HEWLETT-PACKAR	0 83557A
24	_1_	SOURCE MUDDLE / MULTIPLIER	HEWLETT-PACKARD	83558A



Item	Qty	Equipment	Manufacturer	Model No,
22	1	Oxygen Monitor	Bio Systems 3100	
23	2	CRYO Protective Gloves	Lab Safety · 5932L Supply	
24	1	Protective Face Mask	SELLSTROM	124-390/380
25	1	Cold Target Support	Aerojet	T-1291001-2
26	1	Cold Target Support	Aerojet	T-1291001-3
27	2	Cold Target	Aerojet	T-1291000-1
28	1	ON/OFF Switch	Aerojet	SK1260106
29	1	Power Supply	Hewlett-Packard	HP 6205 B
30	1	Protective Apron	Lab Safety Supply	8A-7549-3

3.3.1 Integration and test preliminary conditions

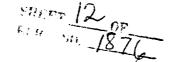
- 3.3.1.1 Limited performance test (LPT). The Limited Performance Test shall consist of the test procedures in the LPT column of Table II.
- 3.3.1.2 Comprehensive performance test (CPT). Three types of Comprehensive Performance Testing are shown in Table II. The first and final CPTs are the same except for paragraph 3.3.5.1 which is performed during the first protoflight unit CPT. The first CPT is performed prior to the start of environmental testing. Sub CPTs are intermediate comprehensive performance tests performed during environmental testing. The final CPT is performed after the completion of environmental testing. Table II shows the required tests for each CPT.
- 3.3.1.3 Integration and test facilities. Unless otherwise specified, all testing and inspection of the EOS/AMSU-A1 shall be conducted at Aerojet, Azusa Operations, Azusa, California.
- 3.3.1.4 Environment. Unless otherwise specified all testing and inspection operations shall be performed under the following laboratory ambient conditions:
 - a. Handling in accordance with AE-26357
 - b. Contamination control in accordance with Report 10353

c. Temperature: $+23 \pm 10$ degrees Celsius

d. Pressure: 610 to 810 torr

e. Humidity: $50 \pm 20\%$ (no condensation)

- f. The instrument shall be placed in its protective bag (1338427) when not in use.
- 3.3.1.5 Integration testing/inspection. Prior to the start of CPT/LPT testing, the unit should be in the final system configuration as determined by the successful completion of the subsystem integration procedure, AE-26156/7.
- 3.3.1.6 Electrostatic discharge (ESD) certification. Certification for handling ESD sensitive equipment in accordance with STD-2454 is required for all personnel working on the EOS/AMSU-A1 instrument.



3.3.1.7 CPT/LPT preparation checklist. Prior to starting the integration, perform the following procedures.

- Visually inspect the instrument. Check for physical damage and cleanliness.
- Verify proper installation of the ESD protective mat and wriststraps. Refer to STD-2454 for ESD protection instructions.
- 3. Verify that each connector of the spacecraft interface has a connector saver installed.
- 4. Obtain the required test equipment listed in Table I. Verify that the test equipment requiring calibration is currently calibrated.
- Verify operation of the Special Test Equipment (STE) shown in Figure 2 by itself. Ensure that the current limits on the two power supplies that interface to the instrument are set correctly. The Q supply should be set to 3 amps and the N/S supply should be set to 1.5 amps. Refer to Figure 3 for the STE power supply panel layout. Figures 4 through 6 show other panels on the STE that will be referenced later in this procedure.
- 6. Verify that all of the required procedures and drawings listed in 2.2.2 are available for reference.

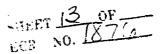


Table II. AMSU-A1 Performance Tests

Para.	Description	1st CPT	LPT	Sub CPT	Final CPT
3.3.2	Grounding Interface Test	х	х	Х	Х
3.3.3	Operational Power Interface Test				
3.3.3.1	Quiet Power Bus	WEST OF	1 1 1 1 1 1 1 1 1 1		
3.3.3.1.1	Quiet Power Bus Operational Power Test	х		X	X
3.3.3.1.2	Quiet Power Bus Operational Power Test (LPT Only)		X		
3.3.3.1.3	Quiet Power Bus Turn On Transient Test	х			Х
3.3.3.2	Noisy Power Bus				
3.3.3.2.1	Noisy Power Bus Operational Power Test	х		х	Х
3.3.3.2.2	Noisy Power Bus Turn On Transient Test	х			Х
3.3.3.3	Survival Heater Power Bus Interface Test				Х
3.3.4	Passive Analog Interface Test	х	х	Х	Х
3.3.5	Command & Telemetry Bus Interface Test				B. 216 A.
3.3.5.1	FQT of the EOS/AMSU-A1 Firmware (PFM Only)	х			
3.3.5.2	Instrument Commanding Verification	Х	х	х	Х
3.3.5.3	Science and Engineering Data Verification	Х	х	Х	Х
3.3.5.4	1553 Bus Interface Test	х			X
3.3.6	Test Point Interface Test				
3.3.6.1	Intentionally Left Blank				
3.3.6.2	8 Second Sync Pulse Verification	X		х	Х
3.3.6.3	Integrate/Hold & Dump Signal Verification	X		х	Х
3.3.6.4	Radiometer Channel Analog Output Verification	X		Х	X
3.3.6.5	PLO #1 and PLO #2 Lock Signal Verification	Х		Х	X
3.3.6.6	GSE-1 Mode Verification	х			X
3.3.6.7	GSE-2 Mode Verification	X			Х
3.3.6.8	GSE-3 Mode Verification	X			Х
3.3.6.9	GSE-4 Mode Verification	х			х
3.3.6.10	GSE-5 Mode Verification	х			Х
3.3.6.11	GSE-7 Mode Verification	Х			Х
3.3.7	Radiometer Functional Performance Test				
3.3.7.1	PLO Frequency Measurements	х			X
3.3.7.2	Relative Radiometer NEAT Measurements	Х	X	X	Х
33.8	CHANNEL I DENTIFICATION TEST	X			(

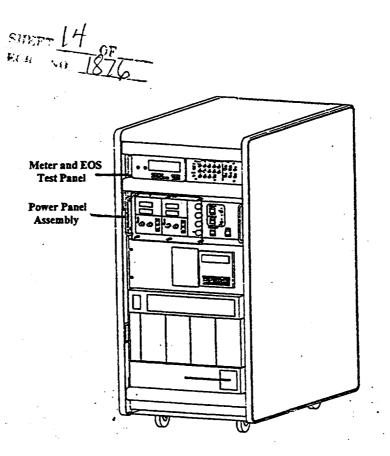


Figure 2. Special Test Equipment (STE) (1356655)

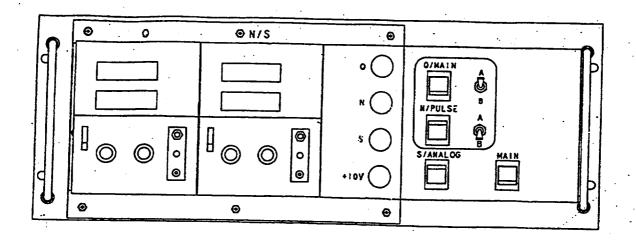
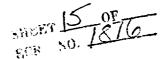


Figure 3. STE Front Power Supply Panel Layout



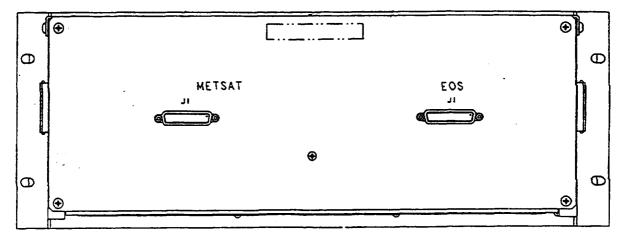


Figure 4. STE Rear Power Supply Panel Layout

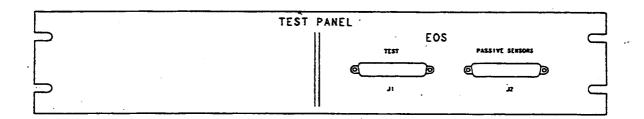


Figure 5. STE Rear Test Panel Layout

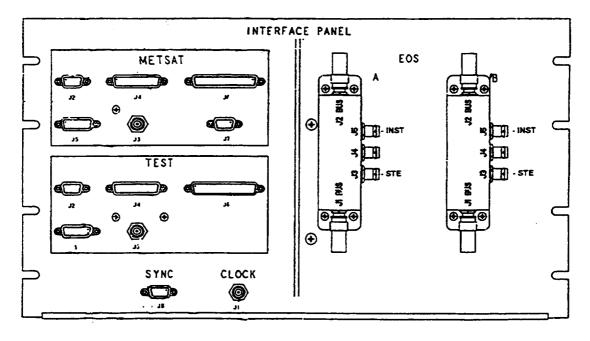
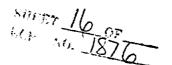


Figure 6. STE Rear Interface Panel Layout



3.3.2 Grounding interface test. This test provides the verification of the unit grounding requirements found in the following documents:

UIID Waiver 5 (12)

GIRD Sections 5.3 and 6.2.2 (except section 5.3.5.2)

POS Section 4.4.1

ICD Section 5.3

To verify these requirements, perform the following procedures.

- 1. Configure the unit as shown in Figure 7. Verify that connectors J1, J2, J3 and J4 have connector savers installed. Connect a 25 Pin breakout box at J1. Connect a 37 Pin breakout box at J2. Connect a 9 pin breakout box at J3. Connect a 37 pin breakout box at J4.
- 2. Measure and record continuity or isolation between the points as specified on Test Data Sheet (TDS) 1.
- 3. Remove the breakout boxes from J2 and J3 ensuring that the connector savers remain in place.

3.3.3 Operational power interface test. This test provides the verification of the operational power interface requirements found in the following documents:

UIID - Section 3.3 and waivers 5(3), 5(7), 5(9), and 5(11)

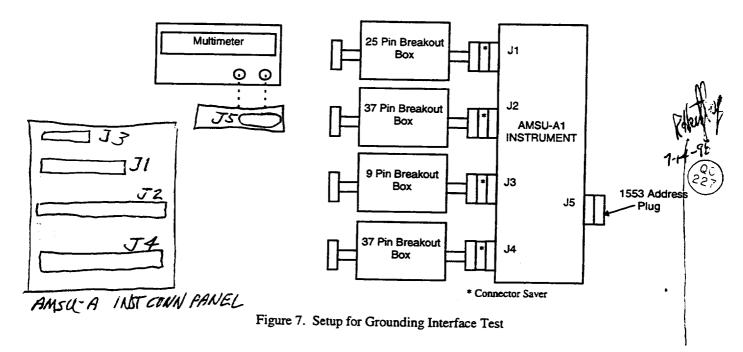
GIRD - Sections 5.1.2 and 5.2

POS - None

ICD - Sections 5.1.2 and 5.2

Operational power is delivered to the unit through spacecraft interface connector J1 as follows:

- 1. Quiet power bus (3.3.3.1)
- 2. Noisy power bus (3.3.3.2)



EHERT 1 OF BOY NO. 18 TIO

- 3. Survival heater power bus (3.3.3.3)
- 3.3.3.1 Quiet power bus interface tests. The quiet bus is active immediately upon the introduction of spacecraft power to the bus. There is no internal control within the unit. The quiet power bus shall be verified by performing the following tests:
 - 1. Quiet power bus operational power test (3.3.3.1.1)
 - 2. Quiet power bus operational power test (LPT only) (3.3.3.1.2)
 - 3. Ouiet power bus turn on transient test (3.3.3.1.3).
- 3.3.3.1.1 Quiet power bus operational power test. The Quiet Power Bus operational power shall be verified at combinations of three voltages (+27, +29, and +31 volts) and two PLO conditions (PLO #1 active and PLO #2 active). The operational power test will be conducted for the unit in full scan mode as follows:
 - 1. With the STE main power off and the STE power panel turned off (MAIN POWER, Q/MAIN, N/PULSE, and S/ANALOG switches as shown in Figure 3 in the OFF position), connect the instrument as shown in Figure 8. This setup assumes a dc impedance from the spacecraft supplied power through fuse and cabling to the unit on the order of 0.3 ohms.
 - 2. Ensure breakout boxes at J1 and J4 are connected to the unit as indicated in 3.3.2, testing.
 - 3. Connect the STE to the instrument using the following STE interface cables:
 - a. STE interface cable J1 (1356648-1)
 - b. STE interface cable J2 (1356648-2)
 - c. STE interface cable J3 (1356648-3)
 - 4. Connect STE interface cable J1 from EOS J1 found on the STE power panel shown in Figure 4 to the 25 pin breakout box. Connect the remaining end to the 25 pin breakout box to J1 of the instrument.
 - 5. Connect STE interface cable J2 from EOS J2 found on the STE test panel shown in Figure 5 to J2 on the unit.
 - 6. Connect STE interface cable J3 from EOS A&B J5 found on the STE interface panel shown in Figure 6 to J3 on the unit.
 - 7. Before turning on the power to the unit, verify that switches 1, 2, 14, and 15 of the 25 pin breakout box are in the open position.
 - 8. Disconnect the external power supply PS1 from the 25 pin breakout box. Turn on the external supply and using a multimeter, adjust its output to 27 ± 0.10 volts. Turn off the external supply and reconnect the supply as shown in Figure 8.
 - 9. Turn the STE MAIN POWER switch ON {refer to Figures 2 and 3 (computer should be on, STE power panel should be off)}. From the A1 directory and at the "\$" prompt, enter the command to the STE "RUN E1". The EOS/AMSU-A1 software program should be running as evidenced by the STE screen shown in Figure 9.
 - Turn the STE power supply panel MAIN POWER switch ON (refer to Figure 3).

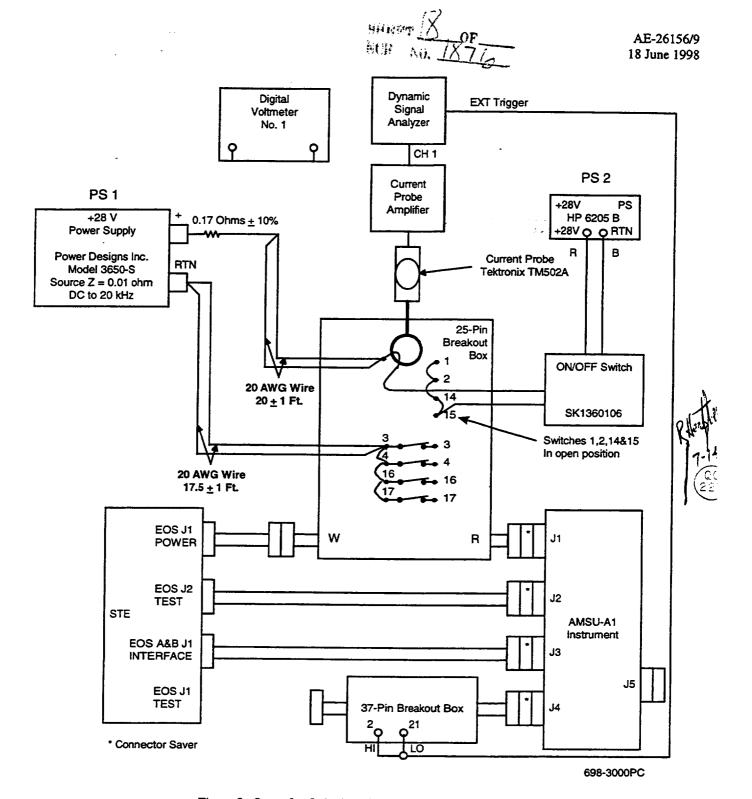


Figure 8. Setup for Quiet Bus Operational Power Tests

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Figure 9. EOS/AMSU-A1 STE Main Screen

- Turn the external power supply on. Place ON/OFF switch in the 'ON' position. With a multimeter, adjust the Quiet Bus voltage at the breakout box to 27 ± 0.10 volts (between J1-1 and J1-3).
- 12. Turn the STE power supply panel N/Pulse switch on (refer to Figure 3). With a multimeter, adjust the Noisy Bus voltage at the breakout box to 29 ± 0.10 volts (between J1-5 and J1-7).
- 13. Go to the Commands screen on the STE. From the main screen shown in Figure 9, enter the STE command "[2] MONITOR ONLY". The screen should now be as shown in Figure 10. Enter the STE command "[14] COMMANDS". The screen should now be as shown in Figure 11.
- 14. Enter the STE command "[11] ANTENNA FULL SCAN MODE". Wait 18 seconds before issuing the next command.
- 15. Enter the STE command "[9] SCANNER A1-1 POWER". Wait 18 seconds before issuing the next command.
- 16. Enter the STE command "[10] SCANNER A1-2 POWER". Wait 18 seconds before issuing the next command. The unit should now be scanning in full scan mode with PLO #1 active.
- 17. Look at the Quiet Bus voltage. If necessary, using the multimeter adjust the external supply PS1 to 27 ± 0.10 volts. Record the voltage on TDS 2.
- Observe the Quiet Bus current waveform on the dynamic signal analyzer. Configure the dynamic signal analyzer as follows:

Select MEAS MODE

Select Time Capture

Select Capture Select

Select Capture Length; Enter 1.0, Select Record

Select FREQ

Select Freq Span; Enter 100.0; Select Hz

Select E SMPL Off

Select Time Length; Enter 8.0; Select Sec

Select SELECT MEAS

Select Power Spec

Select CH1 Active

Select WINDOW

Select Hann

Select SOURCE

Select Source Off

Select AVG

Select Avg Off

Select Tim Av Off

Select RANGE

Select Aut 1 up&dwn

Select INPUT COUPLE

Select CH1 DC

Select CH 1 Ground

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Select SELECT TRIG

Select Trig Level; Enter 1.5; Select V

Select Arm AU

Select Ext

Select Slope +

Select TRIG DELAY

Enter 0.0; Select Sec

Select COORD

Select Real

Select VIEW INPUT

Select Time Buff

Select SCALE

Select X Fixd Scale; Enter 0.0, 8.0; Select Sec

Select Y Fixd Scale; Enter -10.0, 70.0; Select my

Select UNITS

Select Hz (sec)

-NOTE-

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- a) Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
- b) Depress "Start Capture" on the DSA.
- c) With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
- d) Position the current probe to its original location in accordance with Figure 8.

The instrument is now ready to capture and plot 8.0 seconds of data

- 19. Start the DSA signal capture by depressing "Start Capture"; Insure Relay Board is 'ON'.
- 20. Obtain a record of the Quiet Bus current waveform. On the Relay Board, turn the switch OFF.
- 21. Determine average power by the following:

Observe the current waveform on the DSA. Using the Y markers, place the lower horizontal bar on the 0.0 ma line and the upper bar on the current trace, adjusting the bar to the middle of the signal. This measures the average current over the 8.0 second span. Multiply this value by the current scale factor (20 ma/mv, which yields Average Quiet Bus Current. Record on TDS-2, Record the PS-1 measured Quiet Bus Voltage on TDS-2. Multiply the voltage times the current for the calculated average power. Record on TDS-2.

22. Determine peak power by the following:

Observe the current wave form taken above. Sweep the X marker across the current wave form stopping on each narrow spike to see which has the highest amplitude. Upon finding the largest one, leave the X marker indicating the Peak Current Amplitude. Record this on TDS-2. Make a plot of this screen and attach it to TDS-2. Record the PS-1 measured Quiet Bus Voltage on TDS-2. Multiply the voltage times the peak current to obtain the calculated Peak Power. Record this on TDS-2.

With the multimeter, adjust the external power supply PS1 to 29±0.10vdc as measured between J1-1 (high and J1-2 (low). Record this voltage on TDS 2.

- 24. Repeat steps 19 through 22.
- 25. With the multimeter, adjust the external power supply PS1 to 31±0.10vdc as measured between J1-1 (high) and J1-2 (low). Record this voltage on TDS-2.
- 26. Repeat steps 19 through 22.
- 27. Repeat the above steps after changing to PLZO#Z.

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	194#\$	* No. 10 11			
	XX OB.A1] E1. CE DATA ELEMI	7.~	29 -SEP-97 14:44:25 SCA	N NUMBER	
[6] CONTI	ROL/STATUS ELEM	ENT 00		Kg11	
[7] ENGIN	EERING ELEMI	ENT 00		8	
	COMMAN	IDS .	PLLO POWER	PLLO#2 [15	5]
[9] SCAN	NER A1-1 POWER	= OFF	COLD CAL POSITION 1	YES [16	6]
[10] SCAN	NER A1-2 POWER	= OFF	2	NO [17	7]
[11]ANTEN	NNA FULL SCAN MO	DE = NO	3	NO [18	8]
[12]	WARM CAL	= NO	COLD CAL POSITION 4	NO [19	9]
[13]	COLD CAL	= NO	RESET C&DH PROCESS	OR [20	0]
[14]	NADIR	= NO	GSE MODE	[2]	1]
ENGR OK I	SCREEN ONLY [CKSUM IN 2] PRINT [3	CALC SA28 B] FULL [1] RET	SA29 FURN	

Figure 10. EOS/AMSU-A1 STE Monitor Only Screen

EOS/AMSU-A1 WHAT TYPE	E OF TEST?
[2] MONITOR ONLY	[13] FUNCTIONAL TEST
[3] WARM PATH CALIBRATION	[14] S/C TARGET TEST
[4] CYCLE 1 CALIBRATION	[15] ARCHIVE
[5] CYCLE 2 CALIBRATION	[16] INIT AZONIX
[6] CYCLE 3 CALIBRATION	
[7] SPECIAL CYCLE CALIBRATION	[10] SELF TEST
[8] DISK/TAPE PLAYBACK	[11] ID NUMBER XX
[9] ERROR MESSAGES	OFF[] POWER
SELECT BUTTON	[1] RETURN

Figure 11. EOS/AMSU-A1 STE Commands Screen

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3.3.3.1.2 Quiet power bus operational power test (LPT only).

- 1. Configure the unit as shown in Figure 12.
- 2. Breakout box at J1 should still be connected to the unit from the grounding interface testing of paragraph 3.3.2.
- 3. Connect the STE to the instrument using the following STE interface cables:
 - a. STE interface cable J1 (1356648-1)
 - b. STE interface cable J2 (1356648-2)
 - c. STE interface cable J3 (1356648-3)
- 4. Connect STE interface cable J1 from EOS J1 found on the STE power panel shown in Figure 4 to the remaining end of the 25 pin breakout box connected to J1 on the unit.
- 5. Connect STE interface cable J2 from EOS J2 found on the STE test panel shown in Figure 5 to J2 on the unit.
- 6. Connect STE interface cable J3 from EOS A&B J5 found on the STE interface panel shown in Figure 6 to J3 on the unit.
- 7. Turn the STE main power switch on {refer to Figures 2 and 3 (computer should be on, STE power panel should be off)}. From the A1 directory and at the "\$" prompt, enter the command to the STE "RUN E1". The EOS/AMSU-A1 software program should be running as evidenced by the STE screen shown in Figure 9.
- 8. Turn the STE power supply panel main power switch on (refer to Figure 3).
- 9. Turn the STE power supply panel Q/Main switch on (refer to Figure 3). With a multimeter adjust the Quiet Bus voltage at the breakout box to 29 ± 0.10 volts (between J1-1 and J1-3).

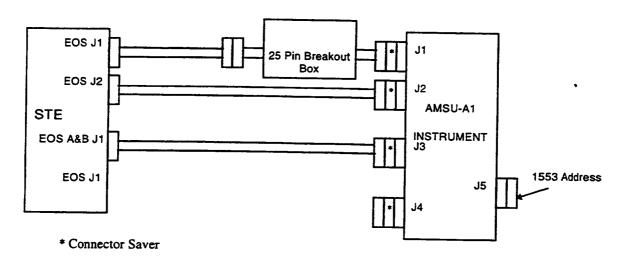
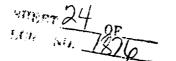


Figure 12. Test Setup of Unit Connected to STE



- 10. Turn the STE power supply panel N/Pulse switch on (refer to Figure 3). With a multimeter adjust the Noisy Bus voltage at the breakout box to 29 ± 0.10 volts (between J1-5 and J1-7).
- Go to the Commands screen on the STE. From the main screen shown in Figure 9, enter the STE command "[2] MONITOR ONLY". The screen should now be as shown in Figure 10. Enter the STE command "[14] COMMANDS". The screen should now be as shown in Figure 11.
- 12. Enter the STE command "[11] ANTENNA FULL SCAN MODE". Wait 18 seconds before issuing the next command.
- 13. Enter the STE command "[9] SCANNER A1-1 POWER". Wait 18 seconds before issuing the next command.
- 14. Enter the STE command "[10] SCANNER A1-2 POWER". Wait 18 seconds before issuing the next command.
- 15. Look at the Quiet Bus voltage. If necessary, using the multimeter adjust the external supply to 29 ± 0.05 volts. Record the voltage and current on TDS 3. The current is read directly from the Q/Main power supply panel meter.
- 16. Compute the operating power in watts on TDS 3 using the equation provided on TDS 3.
- 17. Turn the STE power supply panel N/Pulse switch off (refer to Figure 3).
- 18. Turn the STE power supply panel Q/Main switch off (refer to Figure 3).
- 19. Turn the STE power supply panel main power switch off (refer to Figure 3).
- 20. Leave the setup intact for paragraph 3.3.4 testing.
- 3.3.3.1.3 Quiet power bus turn on transient test. The Quiet Power Bus turn on transient shall be verified at +31 volts as follows:
 - 1. The setup should be intact from paragraph 3.3.3.1.1 testing
 - 2. Verify the external power supply (PS1) is adjusted to 31±.1vdc, make appropriate adjustments.
 - 3. Configure the Dynamic Signal Analyzer (DSA) as follows:

Select MEAS MODE

Select Time Capture

Select Capture Select

Select Capture Length; Enter 400.0, Select msec

Select FREQ

Select Freq Span; Enter 100.0; Select KHz

Select E SMPL Off

Select Time Length; Enter 400.0; Select msec

Select SELECT MEAS

Select Power Spec

Select CH1 Active

Select WINDOW

Select Hann

Select SOURCE

Select Source Off

Select AVG

SHEET SO. LOTTE

Select Avg Off

Select Tim Av Off

Select RANGE

Select Chan 1 Range; Enter 1; Select V

Select INPUT COUPLE

Select CH1 DC

Select CH 1 Ground

Select INPUT TRIG

Select Trig Level; Enter 100; Select mv

Select Arm AU

Select Chan 1 Input

-Select 'Ext' Select Slope +

Select TRIG DELAY

Enter 0.0: Select Sec

Select COORD

Select Real

Select VIEW INPUT

Select Time Buff

Select SCALE

Select X Fixd Scale; Enter 0.0, 400.0; Select msec

Select Y Fixd Scale; Enter -10.0, 320.0; Select mv

Select UNITS

Select Hz (sec)

-NOTE-

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- Remove the current probe from the circuit and close the probe. Place the probe in a magnetic a) benign location.
- b) Depress "Start Capture" on the DSA.
- With the "capture in process", adjust the "output DC level" control on the current amplifier to c) indicate zero current on the DSA.
- Position the current probe to its original location in accordance with Figure 8. d)
- 4. Adjust PS2 for +28vdc.
- Start the DSA signal capture by depressing "Start Capture"; wait for the DSA message "waiting for trigger" 5. before proceeding.
- On the Relay Board, turn the switch ON and obtain a record of the Quiet Bus Turn on current waveform. 6. On the Relay Board, turn the switch OFF. Adjust the display time base and voltage sensitivity to allow for adequate current and pulse duration measurements. Plot the obtained waveform and attach a hard copy of the scan to TDS 4. See Figure 13-A & Figure 13-B.
- 7. Measure the Turn On pulse width; record this value in TDS 4.
- 8. Compute the peak current as follows: Multiply the maximum Ya value by the current/ div as selected on the current amplifier. As an example: if the current amplifier is set up to display 200 ma/ 10 mv per division, and the maximum Ya value = 276mv:

RCh = NO $\sqrt{50}$ 276mv x (200ma/ 10mv) = 5520ma = 5.52 amps

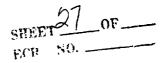
Record this value on TDS 4.

9. The 1st derivative of the current waveform must be calculated. Compute the dI/dT as follows:

The most probable location of the greatest current demand is during the first positive transition after voltage application. If this is the case, expand that segment of the display and measure the greatest voltage transition in the smallest time transition. The change in voltage times the current div as selected on the current amplifier produces the change in current. Next divide this change in current by the change in time (in microseconds). This value is dI/dT. Example:

144mv x (200ma/ 10mv)/ 19.5 us = 147.7ma per us

- 10. Record the computed value on TDS 4.
- 11. With the multimeter, adjust the external power supply PS1 to 29±0.10vdc as measured between J1-1 (high) and J1-3 (low).
- 12. Repeat steps 3 through 10.
- 13. With the multimeter, adjust the external power supply PS1 to 27±0.10vdc as measured between J1-1 (high) and J1-3 (low).
- 14. Repeat steps 3 through 10.
- 15. Turn the STE power supply panel N/ pulse switch OFF (refer to Figure 3).
- 16. Turn the STE power supply panel main power switch OFF (refer to Figure 3).



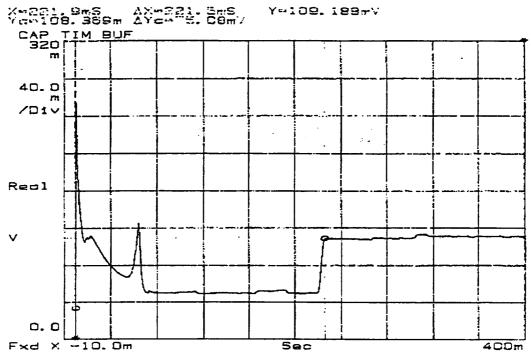


Figure 13-A. Typical Quiet Bus Turn On Transient

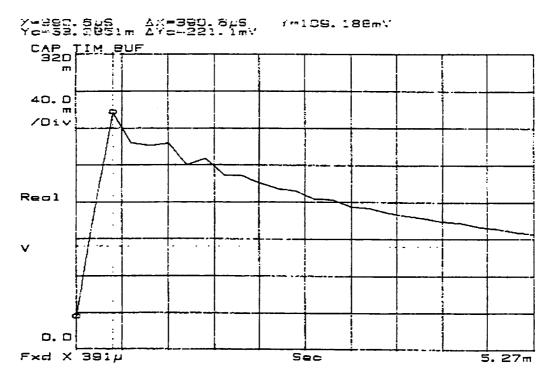
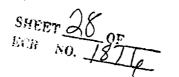
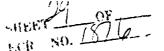


Figure 13-B. Typical Quiet Bus Turn On Expanded



- 3.3.2 Noisy power bus interface tests. The noisy bus is not active upon the introduction of spacecraft power to the bus. Two relays, one for the A1-1 scan drive and one for the A1-2 scan drive, must be turned on before the noisy bus is active within the unit. During normal turn on operation, each scan drive relay is enabled separately a minimum of eighteen seconds apart. Similarly, during normal shut down each drive system is disabled separately. The noisy bus shall be verified by performing the following tests:
 - 1. Noisy power bus operational power test (3.3.3.2.1)
 - 2. Noisy power bus turn-on transient test (3.3.3.2.2)
- 3.3.3.2.1 Noisy power bus operational power test. The Noisy Power Bus operational power shall be verified at combinations of three voltages (+27, +29, and +31 volts). The operational power test will be conducted for the unit in full scan mode as follows:
 - 1. With the STE main power off and the STE power panel turned off (MAIN POWER, Q/MAIN, N/PULSE, and S/ANALOG switches as shown in Figure 3 in the OFF position), connect the instrument as shown in Figure 14. This setup assumes a dc impedance from the spacecraft supplied power through fuse and cabling to the unit on the order of 0.3 ohms.



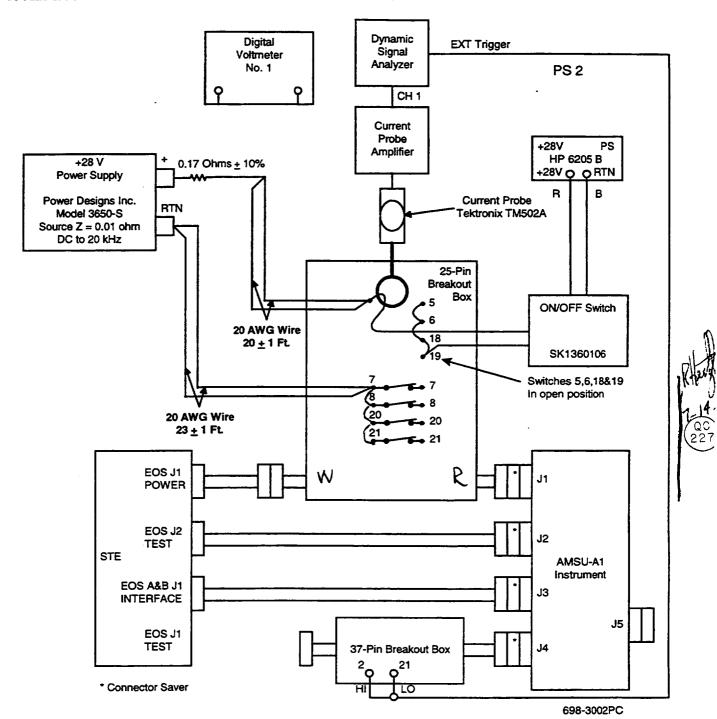


Figure 14. Setup for Noisy Bus Operational Power Tests

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- 2. Before turning on the power to the unit, verify that switches 5, 6, 18, and 19 of the 25 pin breakout box are in the open position.
- 3. Disconnect the external power supply from the 25 pin breakout box. Turn on the external supply PS1 and using a multimeter, adjust its output to 27 ± 0.05 volts. Turn off the external supply and reconnect the supply as shown in Figure 14.
- 4. Turn the STE main power switch on {refer to Figures 2 and 3 (computer should be on, STE power panel should be off)}. From the A1 directory and at the "\$" prompt, enter the command to the STE "RUN E1". The EOS/AMSU-A1 software program should be running as evidenced by the STE screen shown in Figure 9.
- 5. Turn the STE power supply panel main power switch on (refer to Figure 3).
- 6. Turn the STE power supply panel Q/MAIN switch on (refer to Figure 3). With a multimeter, adjust the Ouiet Bus voltage at the breakout box to 29 ± 0.05 volts (between J1-1 and J1-3).
- 7. Turn the external power supply PS1 on. Place ON/OFF switch in the 'ON' position. With a multimeter, adjust the Noisy Bus voltage at the breakout box to 27 ± 0.05 volts (between J1-5 and J1-7).
- 8. Go to the Commands screen on the STE. From the main screen shown in Figure 9, enter the STE command "[2] MONITOR ONLY". The screen should now be as shown in Figure 10. Enter the STE command "[14] COMMANDS". The screen should now be as shown in Figure 11.
- 9. Enter the STE command "[11] ANTENNA FULL SCAN MODE". Wait 18 seconds before issuing the next command.
- 10. Enter the STE command "[9] SCANNER A1-1 POWER". Wait 18 seconds before issuing the next
- 11. Enter the STE command "[10] SCANNER A1-2 POWER". Wait 18 seconds before issuing the next command. The unit should now be scanning in full scan mode with PLO #1 active.
- 12. Look at the Noisy Bus voltage. If necessary, using the multimeter, adjust the external supply to 27 ± 0.10 volts. Record the voltage on TDS 5.
- 13. Observe the Noisy Bus current waveform on the dynamic signal analyzer. Configure the dynamic signal analyzer as follows:

Select MEAS MODE

Select Time Capture

Select Capture Select

Select Capture Length; Enter 1.0, Select Record

Select FREQ

Select Freq Span; Enter 100.0; Select Hz

Select E SMPL Off

Select Time Length; Enter 8.0; Select Sec

Select SELECT MEAS

Select Power Spec

Select CH1 Active

Select WINDOW

Select Hann

Select SOURCE

Select Source Off

Select AVG

SHEET 3/ OF ECH NO. 1874

Select Avg Off

Select Tim Av Off

Select RANGE

Select Aut 1 up&dwn

Select INPUT COUPLE

Select CH1 DC

Select CH 1 Ground

Select SELECT TRIG

Select Trig Level; Enter 1.5; Select V

Select Arm AU

Select Ext

Select Slope +

Select TRIG DELAY

Enter 0.0; Select Sec

Select COORD

Select Real

Select VIEW INPUT

Select Time Buff

Select SCALE

Select X Fixd Scale; Enter 0.0, 8.0; Select Sec

Select Y Fixd Scale; Enter -10.0, 70.0; Select mv

Select UNITS

Select Hz (sec)

-NOTE-

Prior to collecting any current data, the current meter and DSA have to be "zeroed out"; zero current reference has to be established on the DSA. Follow this interim procedure to zero reference the current meter and DSA.

- a) Remove the current probe from the circuit and close the probe. Place the probe in a magnetic benign location.
- b) Depress "Start Capture" on the DSA.
- c) With the "capture in process", adjust the "output DC level" control on the current amplifier to indicate zero current on the DSA.
- d) Position the current probe to its original location in accordance with Figure 8.

The Instrument is now ready to capture and plot 80 seconds of data.

- 14. Start the DSA signal capture by depressing "Start Capture."
- 15. Obtain a record of the Noisy Bus current waveform. On the Relay Board, turn the switch OFF. Using the Y markers, mark the maximum current amplitude as indicated in Figure 15. Plot the obtained waveform and attach a hard copy of the scan to <u>TDS 5</u>.
- 16. Examine the expanded waveform to find the peak current over the entire 80 second scan. Record the peak current on TDS 5. A representative Noisy Bus Current is shown in Figs 15-A & 15-B.
- 17. Calculate the Average Noisy Bus Current as follows:

Select VIEW INPUT

Select Time Record: Note - the display shows the first 8 seconds of data and the heading changes to read "Cap Tim Rec"